

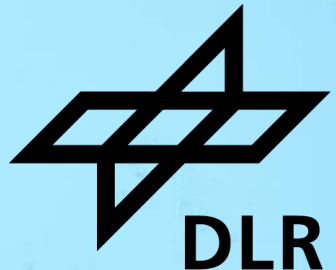
A COMPACT LRI FOR CUBESAT PLATFORMS

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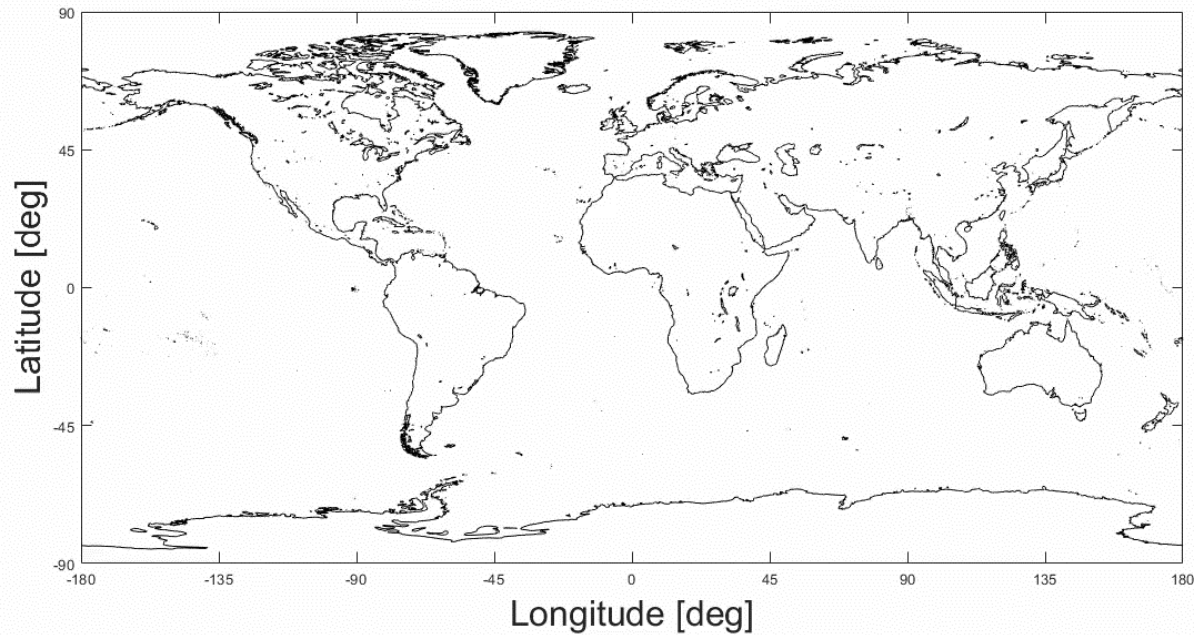
³ Institute of Aerospace Technology, City University of Applied Sciences Bremen



Ground-Track Coverage

GRACE-FO

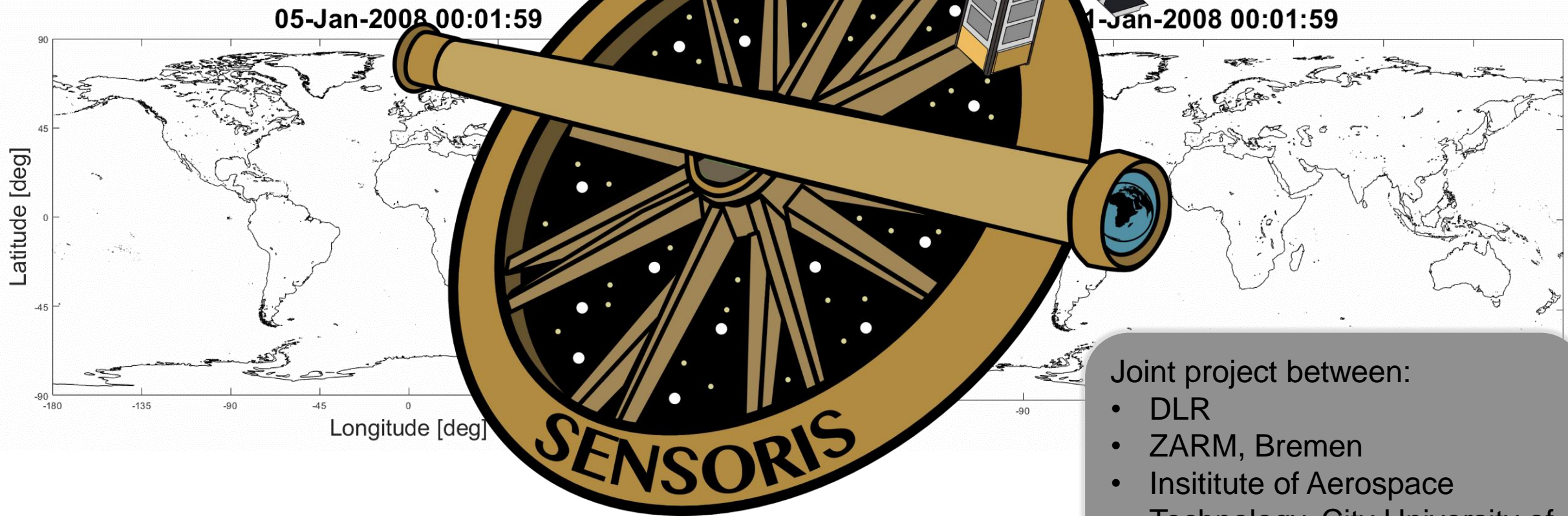
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Ground-Track Coverage

GRACE-FO

SENSORIS



Joint project between:

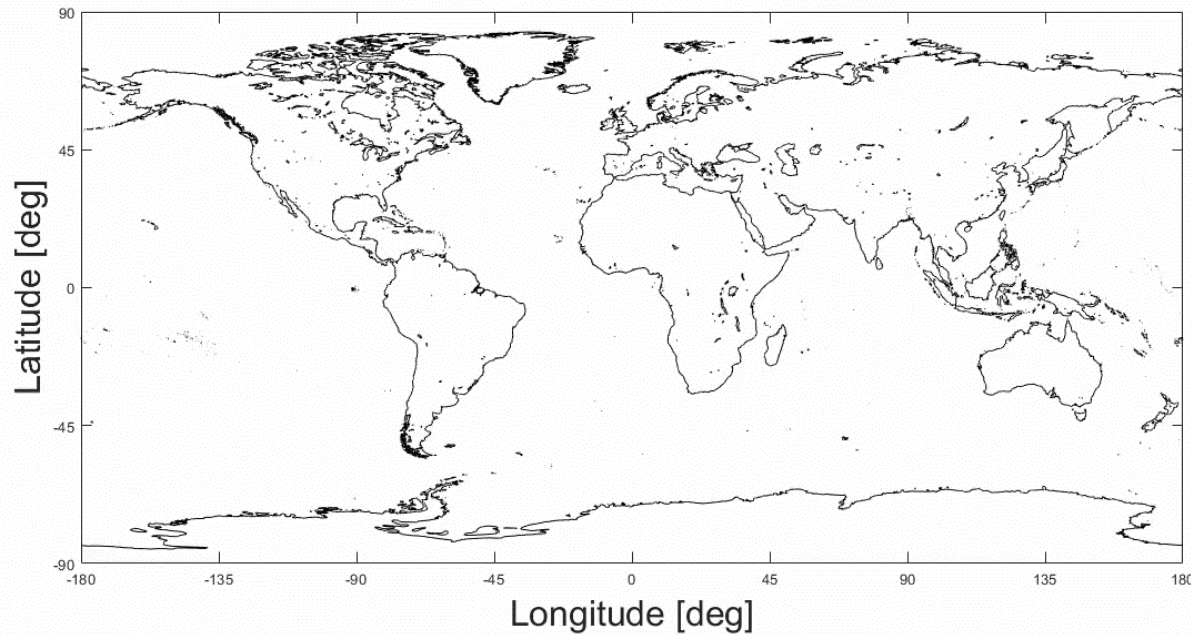
- DLR
- ZARM, Bremen
- Institute of Aerospace Technology, City University of Applied Sciences Bremen

Ground-Track Coverage

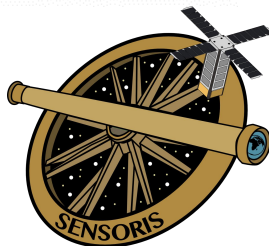
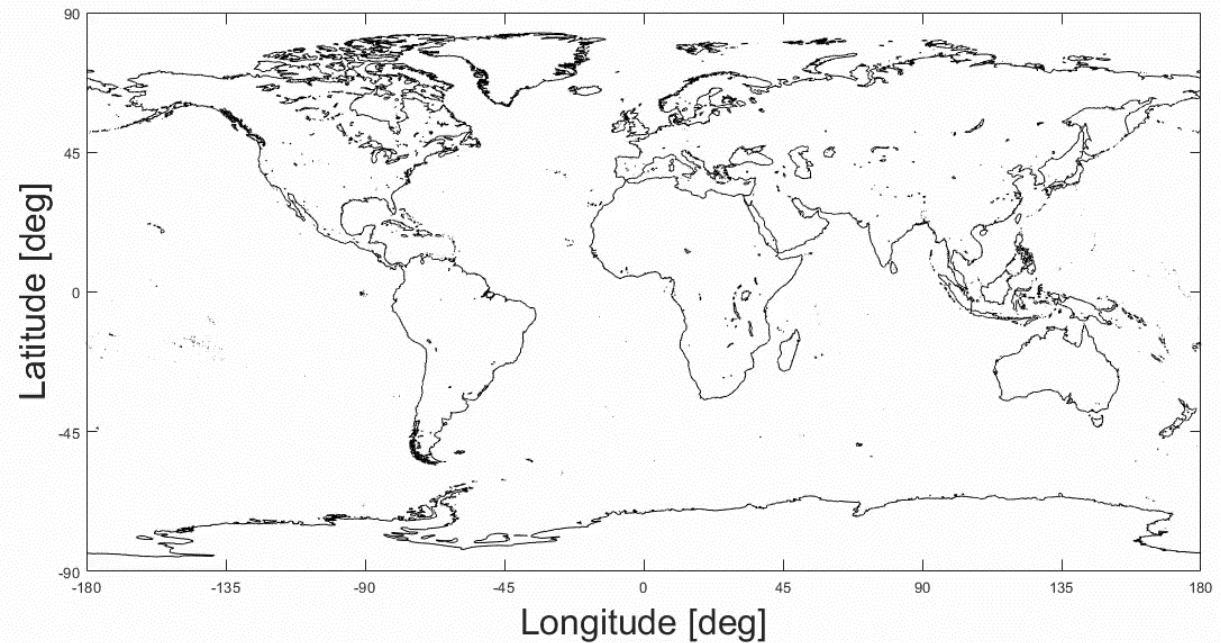
GRACE-FO

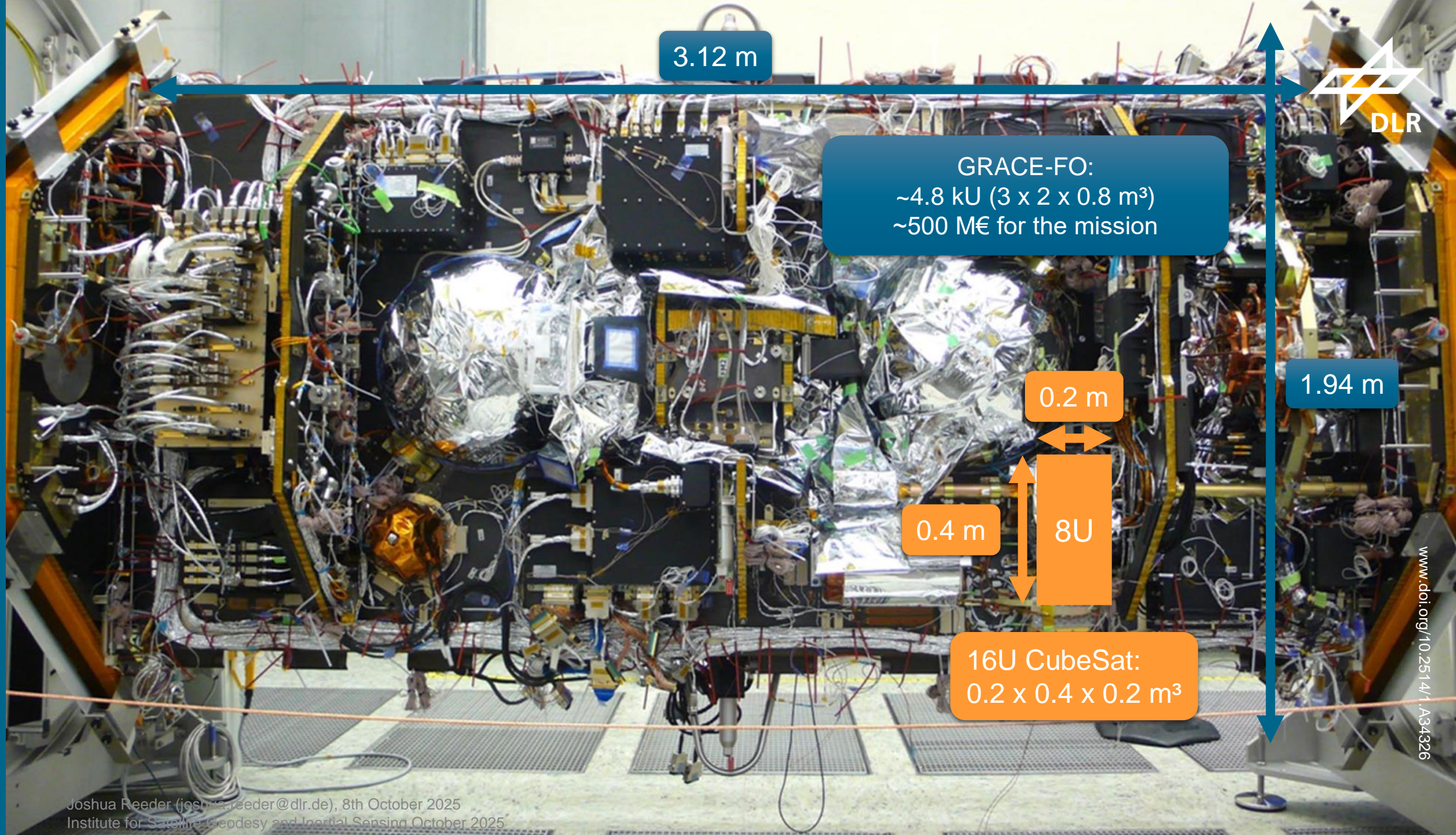
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3.12 m



GRACE-FO:
~4.8 kU (3 x 2 x 0.8 m³)
~500 M€ for the mission

1.94 m

0.2 m

0.4 m

8U

16U CubeSat:
0.2 x 0.4 x 0.2 m³

www.doi.org/10.2514/1.A34326

Compact LRI for CubeSat Platforms

Envisioned Requirements

- 16U Cubesat
- Performance: $< 1 \frac{\mu\text{m}}{\sqrt{\text{Hz}}}$ (better than GNSS)
 - (GRACE-FO: $< 80 \frac{\text{nm}}{\sqrt{\text{Hz}}}$)

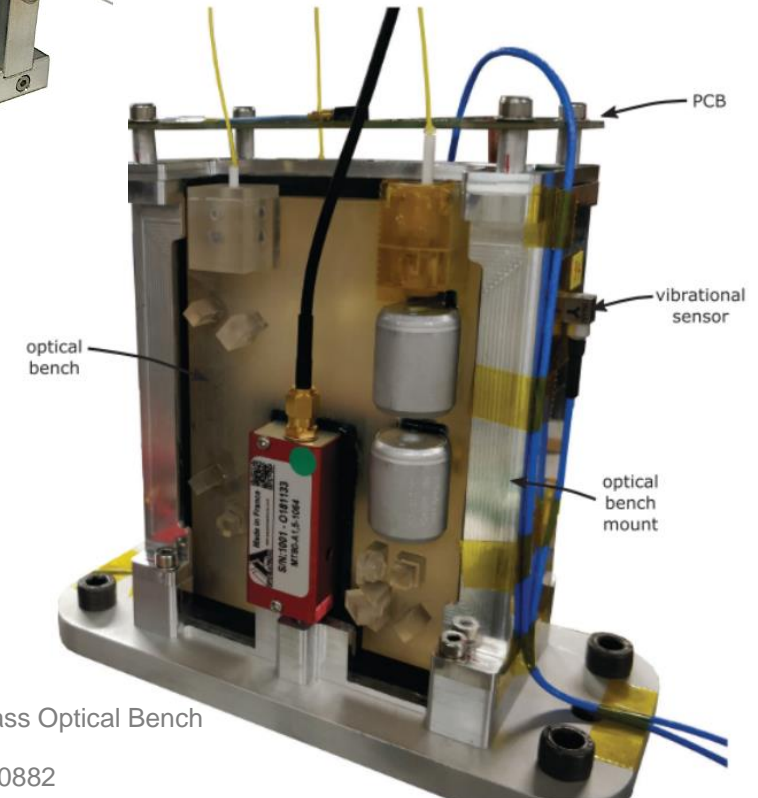
Trade-Offs and Technology

- Wavelength
- Laser Frequency Stabilisation Method
- Thermal Environment → Low CTE Glass Breadboard



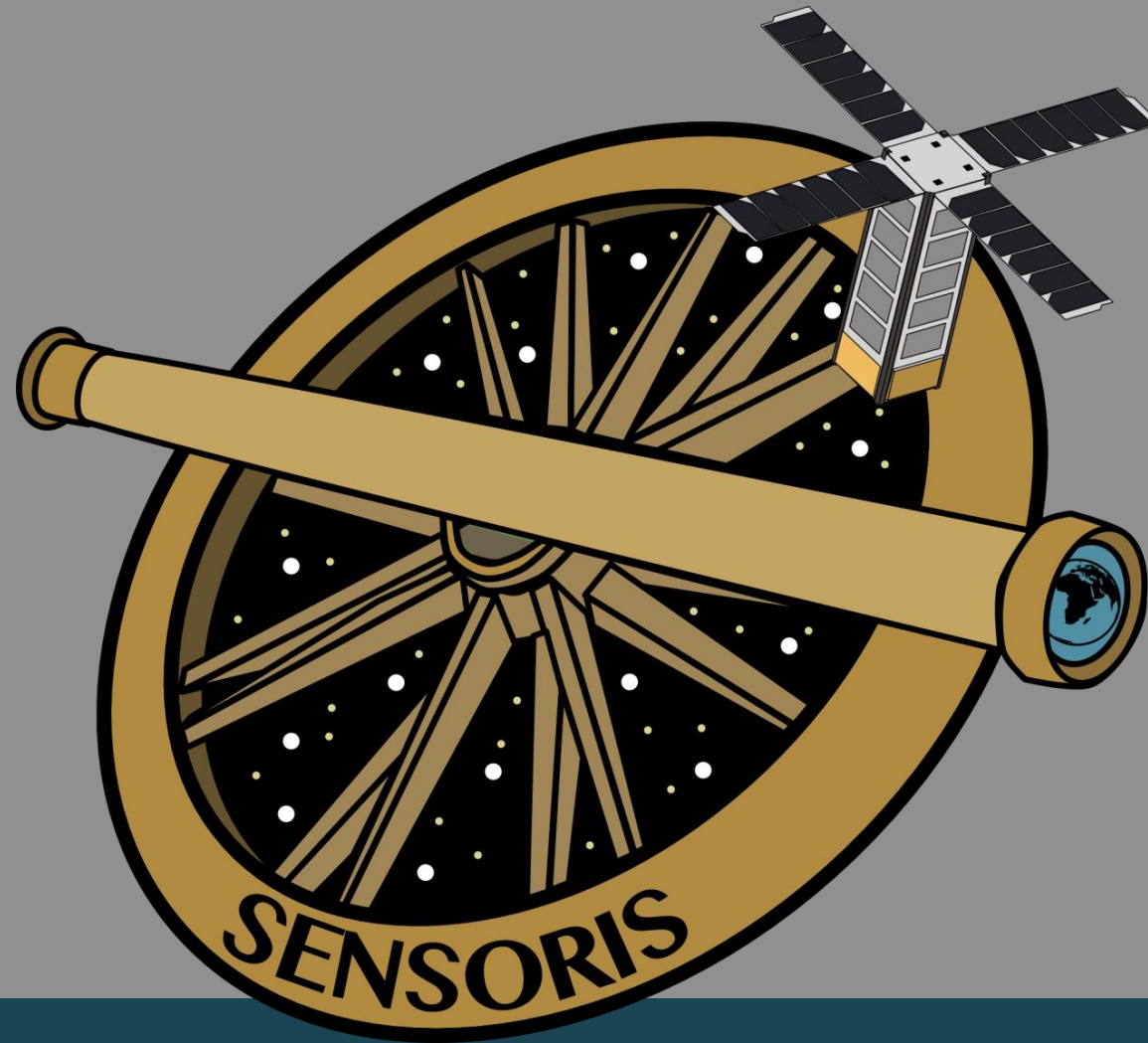
<1U Rubidium Frequency Reference

<https://doi.org/10.1364/JOSAB.420875>



Micro-Integrated Low CTE Glass Optical Bench

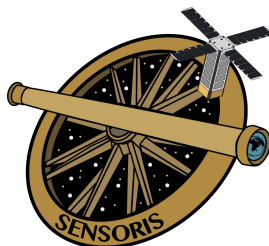
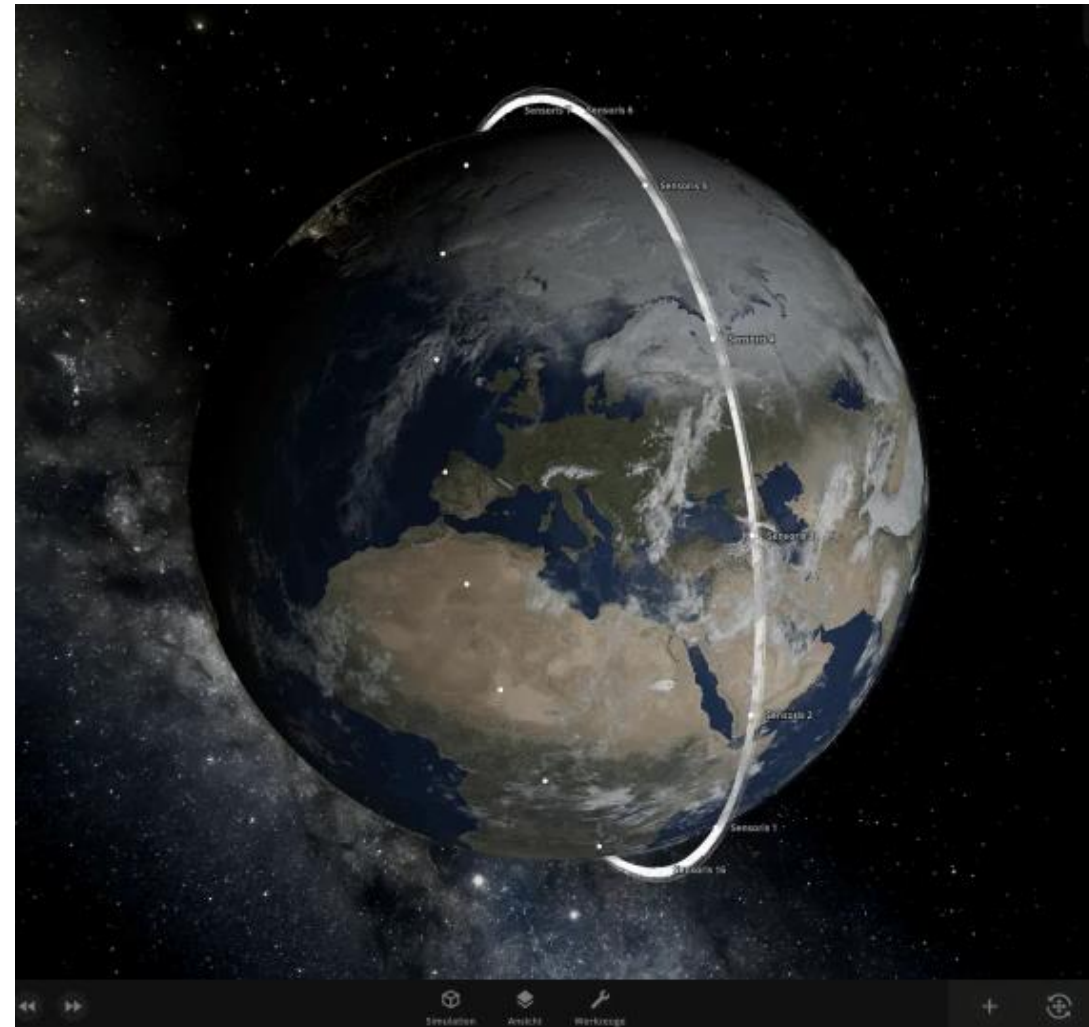
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THANK YOU!

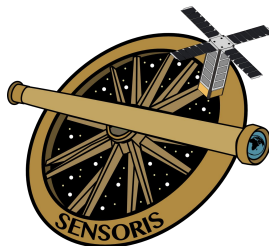
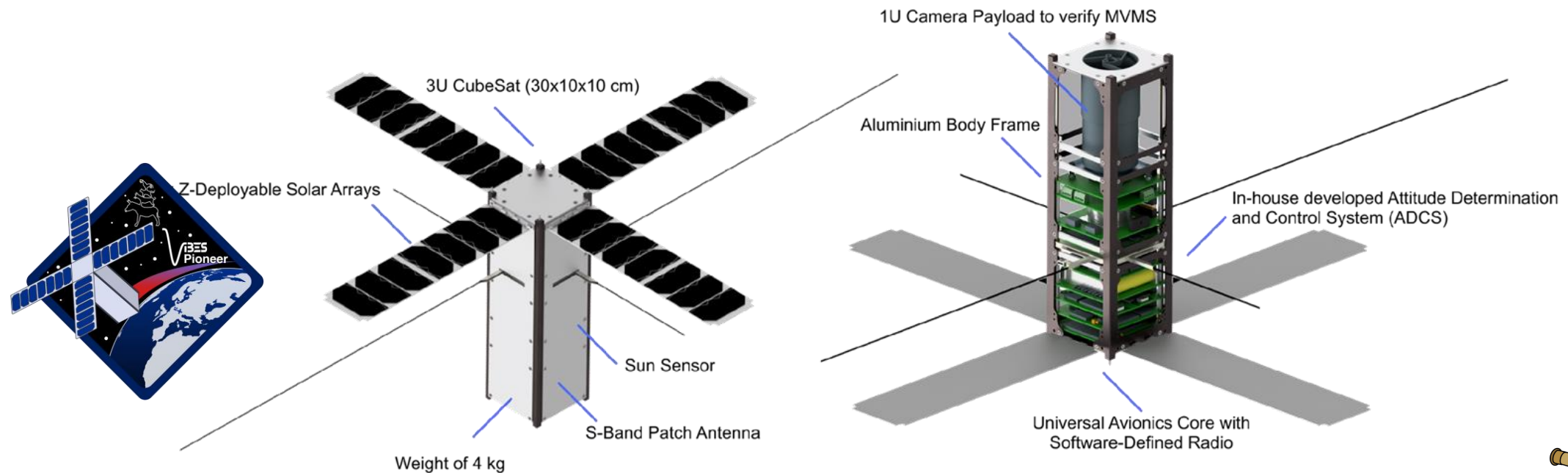
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- Polar Low-Earth Orbit
- Initially 16 Satellites
 - Evenly distributed over the same orbit (TBD)
- High-low Satellite-to-Satellite Tracking
- GNSS as primary observation
 - Accelerometry Optional
- Technology Basis:
 - VIBES Pioneer 3U cubesat
 - New Space approach
 - Commercial off the shelf components



Technological Basis: VIBES Pioneer

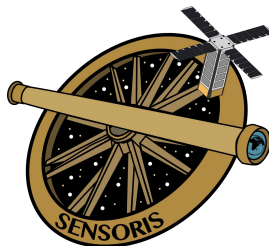
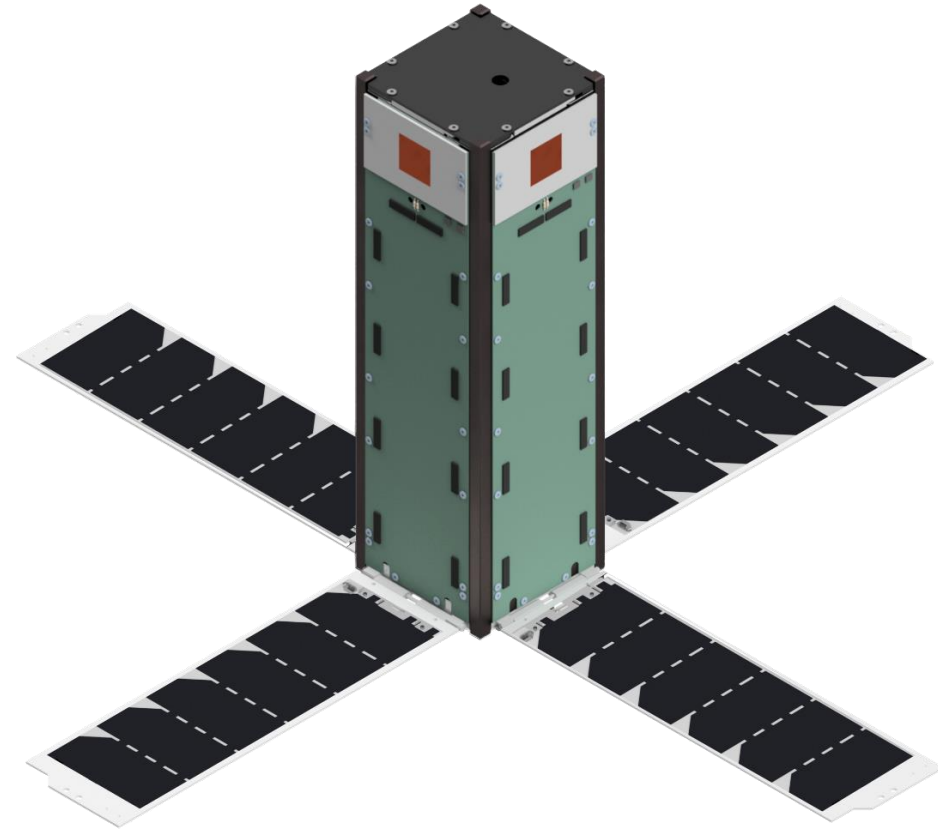
- Bremen's 1st Student-Built Satellite to improve the optical performance of spacecrafts using consumer electronics
- Launch scheduled for 2025 to a 500km polar SSO



SENSORIS Next Steps



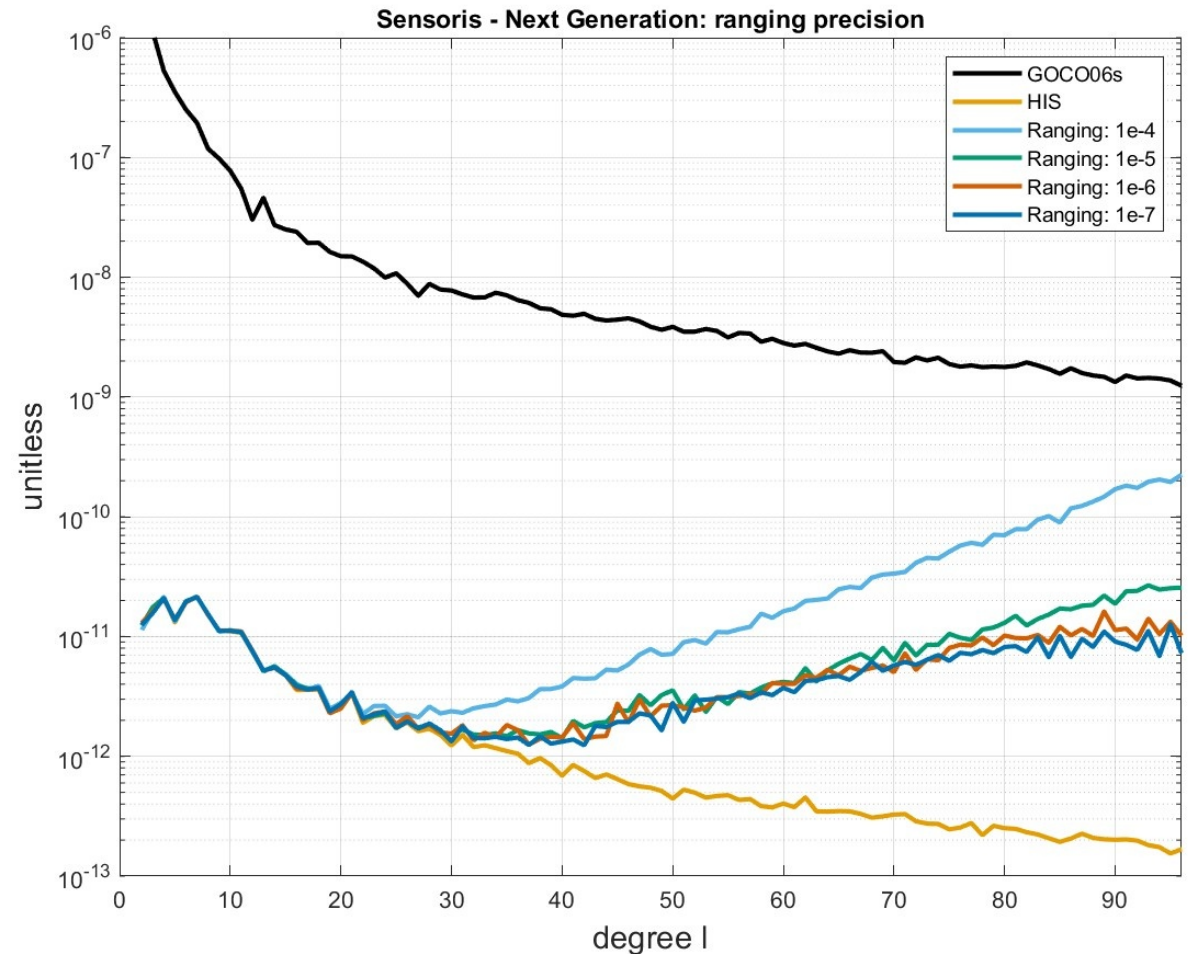
- 9 month feasibility study (starting November):
 - Usability and TRL of consumer electronics
 - Adaption of VIBES Pioneer
 - Orbital configuration including number of orbital planes
 - Payload requirements
 - Accelerometry
 - Performance
- Implementation phase of 3 years
- Long-term Goals:
 - Scalability: Use 64+ satellites to increase to half- or quarter-daily gravity solutions
→ directly measure tidal effects
 - Implement accelerometry using quantum technology
 - Implement low-low satellite-to-satellite tracking (LL-SST)



SENSORIS – Ranging Precision



- Comparison of spherical harmonic degree RMS signal
- Black = static field
- Yellow = HIS signal we want to measure
- Others: Solutions for different levels of precision of ranging measurement



Topic: **A Compact LRI for Cubesat Platforms**

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Institute: Institute for Satellite Geodesy and Inertial Sensing

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