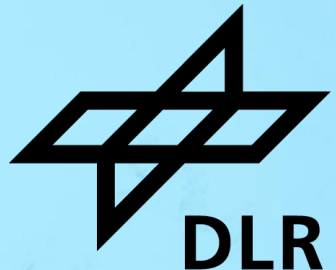


# PRECONDITIONED FEM-BASED NEURAL NETWORKS FOR SOLVING INCOMPRESSIBLE FLUID FLOWS AND RELATED INVERSE PROBLEMS

Fabrice von der Lehr, Franziska Griesse, Katharina Rauthmann, Philipp Knechtges  
(SC-HPC, Cologne)

WAW Machine Learning 11



## Mechanistic modelling:

- Numerical simulations of partial differential equations (PDEs)
- Expensive, especially when PDEs have to be solved for several parameters
- Often infeasible in real-time applications / with limited computing capacity

 **Performant surrogate models needed**

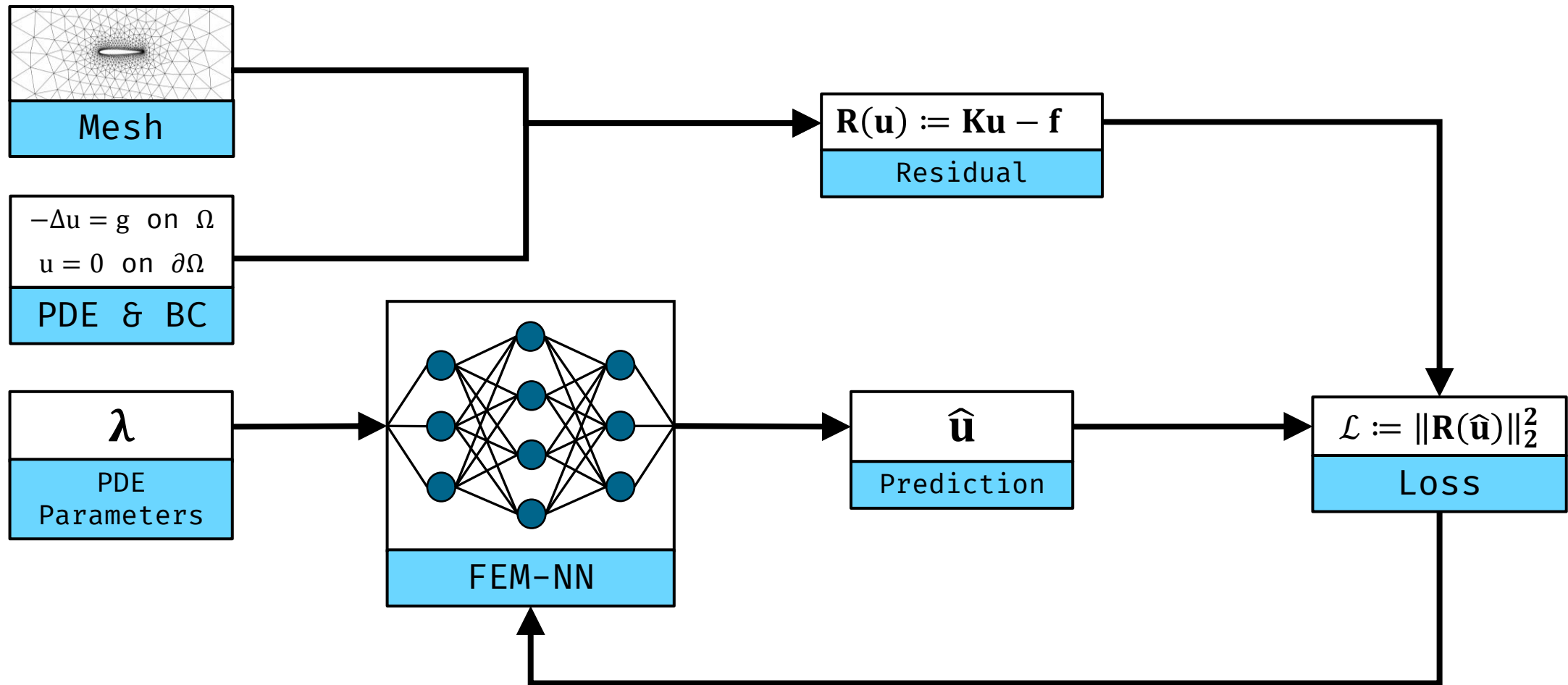
## Data-driven ML approaches:

- Trained with data from simulations or real-world measurements
- Physical laws (e.g., conservation of energy) are not / only poorly considered

- Combining finite element method (**FEM**) with neural networks (**NNs**)
- Ideal outcome: **Uniting strengths** of both methods while **compensating their weaknesses**

| FEM                                       | NN                               |
|-------------------------------------------|----------------------------------|
| ☹ No real-time capacity                   | 😊 Fast prediction after training |
| ☹ No cost amortization over multiple runs | 😊 Parameterizable                |
| 😊 Sound mathematical foundation           | ☹ Black box model                |
| 😊 Numerical theory of errors              | ☹ Rudimentary convergence theory |

# FEM-based Neural Networks



# Incompressible 2D Navier-Stokes Flow around an Airfoil

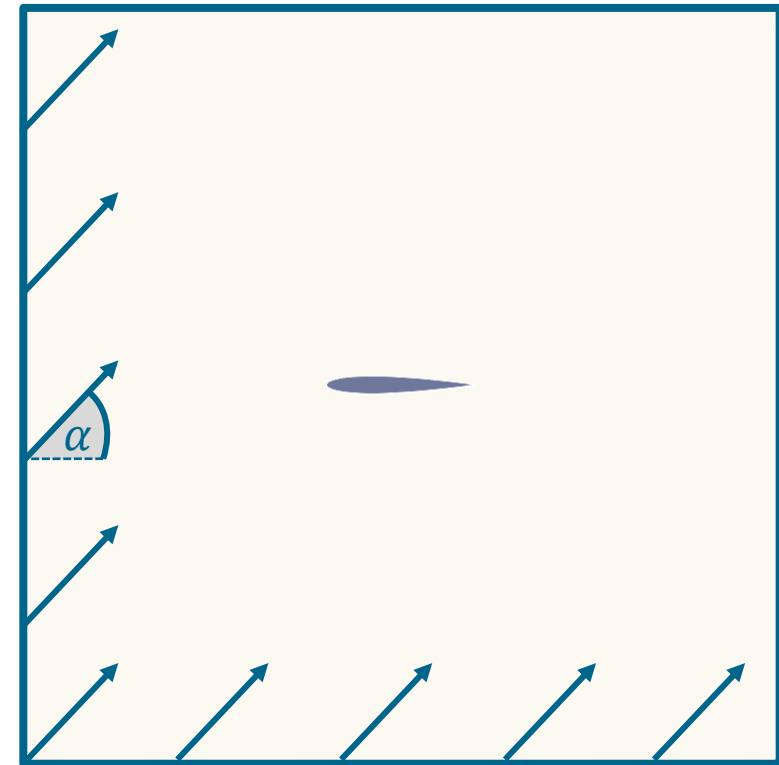
## PDE:

$$\begin{aligned} -\eta \Delta u + u \cdot \nabla u + \nabla p &= 0, \\ \nabla \cdot u &= 0 \end{aligned}$$

with viscosity  $\eta$ , velocity  $u$ , and pressure  $p$ .

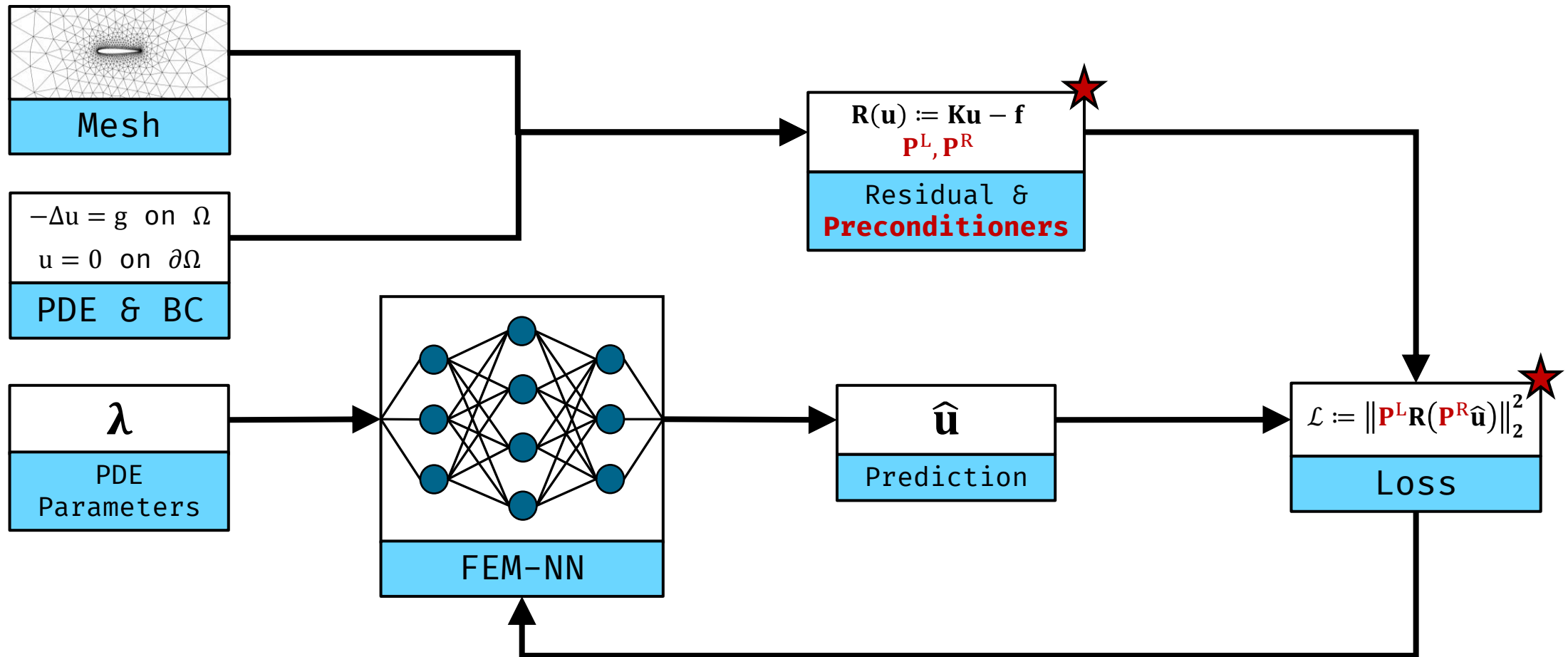
**BC:** Dirichlet at inflow (dependent on  $\alpha$ ), Neumann at outflow, no-slip at airfoil.

- Nonlinear saddle point problem
- Even for FEM hard, depending on  $\eta$



Domain around NACA 0012  
airfoil with an angle of attack  $\alpha$ .

# Preconditioned FEM-based Neural Networks

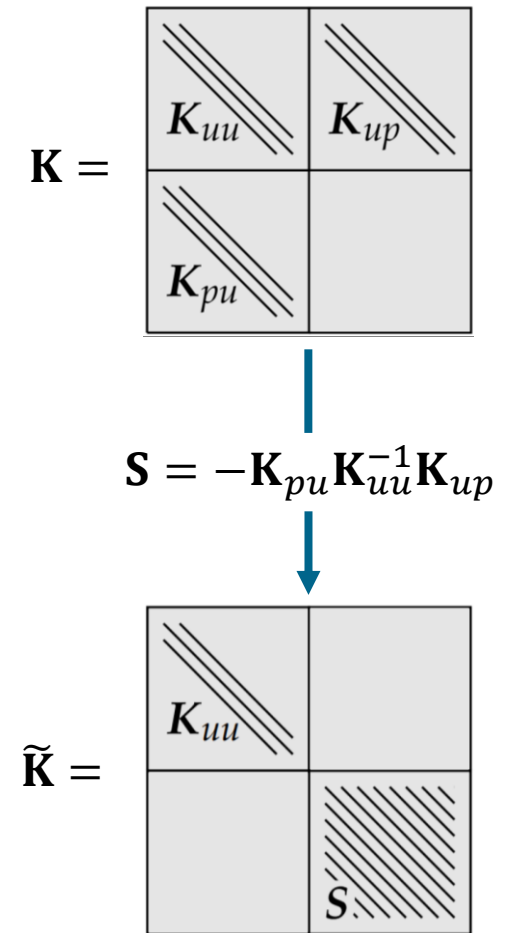


- Saddle point structure of the stiffness matrix  $\mathbf{K}$  allows for **block-diagonal preconditioners** featuring the **Schur complement**

$$\mathbf{K} = \begin{bmatrix} \mathbf{K}_{uu} & \mathbf{K}_{up} \\ \mathbf{K}_{pu} & \mathbf{0} \end{bmatrix} \rightarrow \mathbf{S} = -\mathbf{K}_{pu} \mathbf{K}_{uu}^{-1} \mathbf{K}_{up}$$

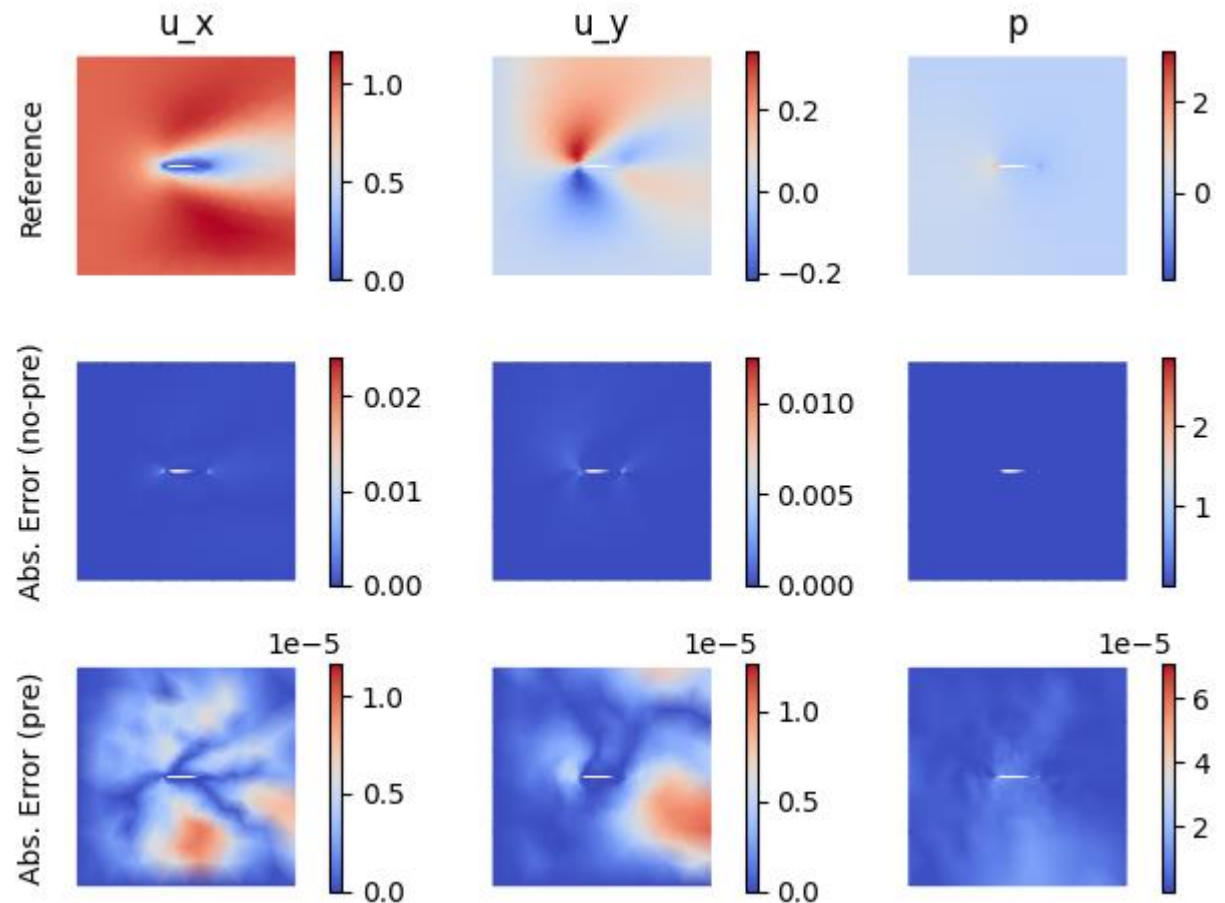
$$\rightarrow \mathbf{P}^L := \begin{bmatrix} \mathbf{L}^{-1} & 0 \\ 0 & \mathbf{M}^{-1} \end{bmatrix}, \quad \mathbf{P}^R := \begin{bmatrix} \mathbf{L}^T & 0 \\ 0 & \mathbf{M}^T \end{bmatrix}$$

- Here,  $\mathbf{K}_{uu} = \mathbf{L}\mathbf{L}^T$  and  $-\mathbf{S} = \mathbf{M}\mathbf{M}^T$  denote Cholesky decompositions, respectively



# Results for 2D Navier-Stokes Equations

- Training with  $\alpha = 1^\circ$ ,  $\eta = 0.1$ , and 10k iterations (no-pre) / 250 iterations (pre) of L-BFGS
  - Training time: 4 min (pre) / 9 min (no-pre)
- **Error improvement by >3 orders of magnitude**
- **Speedup of x2**

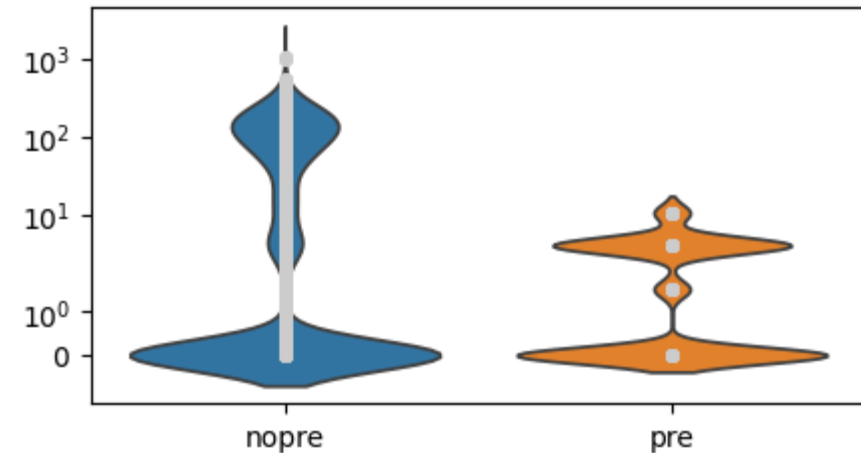


# Eigenspectra of the Loss Hessian

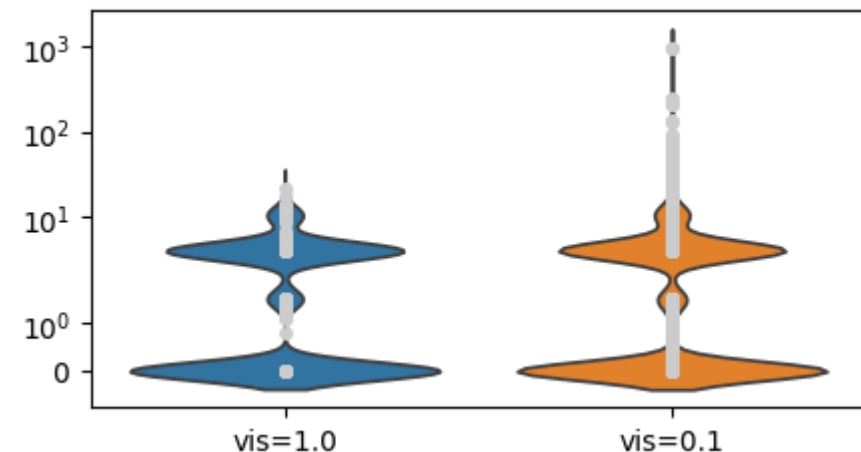
- **Without preconditioning:**  
unclustered eigenspectrum,  
extending over several orders  
of magnitude
- **With preconditioning:**  
only few clusters at rather  
small eigenvalues

→ **Reduction of the condition number is transferred through the nonlinearity of the NN to its parameter space**

Stokes



Navier-Stokes



# Conjugate Gradients (CG) on the Normal Equations



- Solving the least-squares problem

$$\min_{\mathbf{u}} \|\mathbf{K}\mathbf{u} - \mathbf{f}\|_2^2$$

using BFGS is equivalent to applying CG to the normal equations

$$\underbrace{\mathbf{K}^T \mathbf{K}}_{\tilde{\mathbf{K}}} \mathbf{u} = \underbrace{\mathbf{K}^T \mathbf{f}}_{\tilde{\mathbf{f}}}$$

- It is well known that **CG converges more quickly** if the **eigenspectrum** of the system matrix (here:  $\tilde{\mathbf{K}} = \mathbf{K}^T \mathbf{K}$ ) **has few clusters**
- This is the case if the **singular values** of the matrix  $\mathbf{K}$  are well clustered

- Classical preconditioners **P** are designed to **cheaply approximate the inverse  $K^{-1}$**  and often guarantee a certain **clustering of eigenvalues**
- In our case, however, since **K is not necessarily symmetric** (e.g., Navier-Stokes), **its eigenvalues do generally not relate to its singular values**; in the worst case, such a near-inverse preconditioner could even **spread out the singular values further**
- By contrast, an **ideal left-preconditioner  $P^L$  for the normal equations** is a **near-orthogonalizer**, i.e.,

$$(P^L K)^T (P^L K) \approx I,$$

as the singular values of an orthogonal matrix are all equal to 1

- **LQ decomposition**

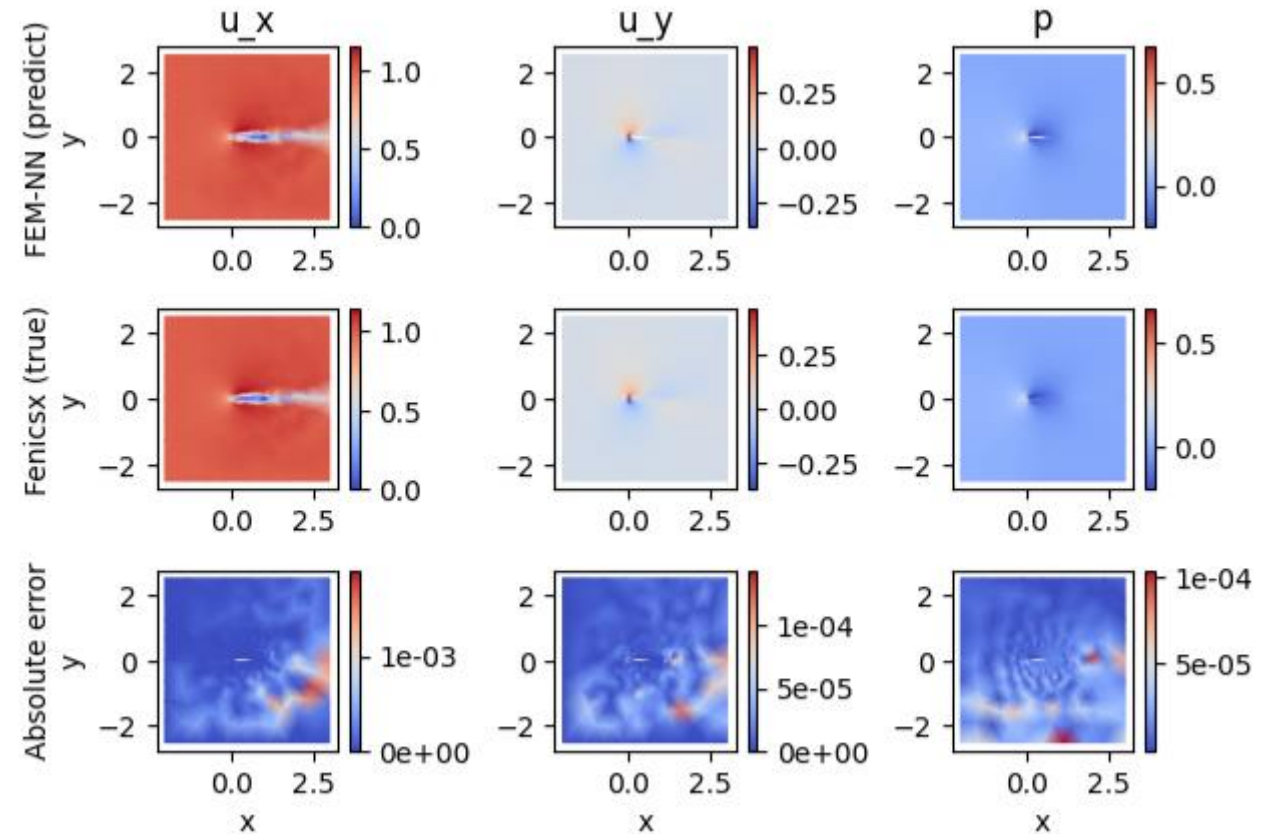
$$\mathbf{LQ} = \mathbf{K}$$

provides an **ideal orthogonalizing left-preconditioner**  $\mathbf{P}^L = \mathbf{L}^{-1}$ , but is expensive and has to be recomputed in the case of Navier-Stokes due to the nonlinear convection term

- **Cheaper incomplete LQ algorithms and data-driven preconditioners** have shown to cluster singular values **only insufficiently**
- Although the **set of all orthogonal matrices grows quadratically** with the matrix dimensionality, the design of cheaper orthogonalizing preconditioners has proven to be **difficult**

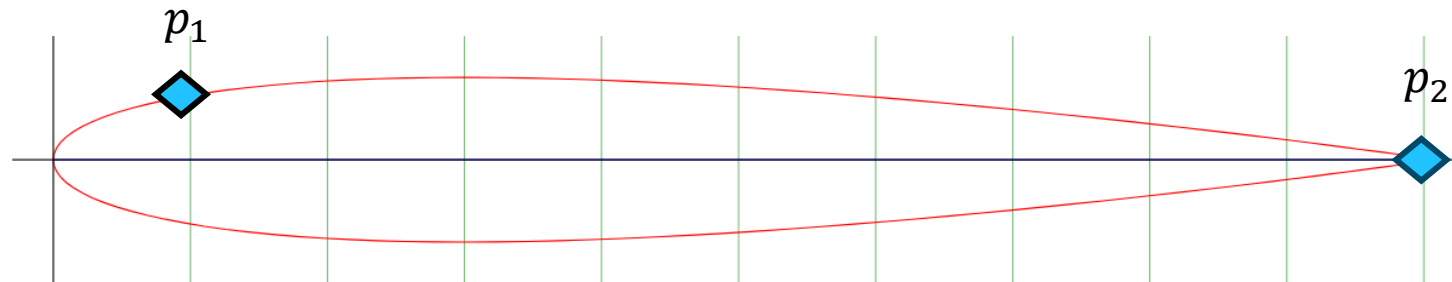
# Results for 2D Navier-Stokes Equations (Low Viscosity)

- **Continuation Method:** In order to compute the preconditioner  $\mathbf{L}$  to  $\mathbf{K}(\mathbf{u})$  for a viscosity  $\eta_k$ , we predict  $\mathbf{u}$  using a FEM-NN previously trained for viscosity  $\eta_{k-1} > \eta_k$
- This **iterative training process** is started by training w.r.t. the corresponding Stokes problem
- Accurate predictions are possible up to a **Reynolds number of 500** ( $\eta = 0.02$ , right figure)



## Inverse problem:

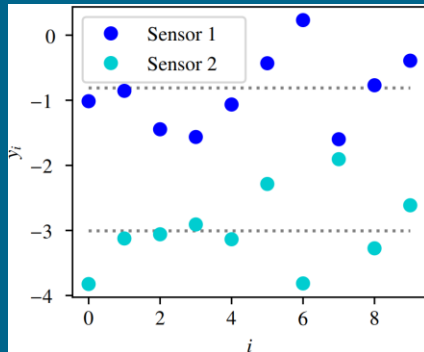
Given noisy pressure measurements  $p_i$  at specific locations of the airfoil, infer the angle of attack  $\alpha$ .



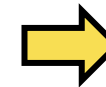
## Inverse problem:

Given noisy pressure measurements  $p_i$  at specific locations of the airfoil, infer the angle of attack  $\alpha$ .

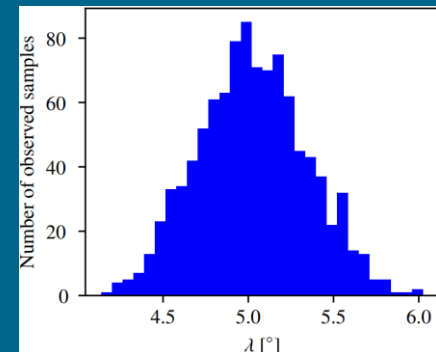
**Input:** Trained FEM-based NN,  
noisy measurements  $p_i$



Hamiltonian  
Monte Carlo  
(HMC)



**Output:** Distribution for  
angle of attack  $\alpha$



- This talk presented **FEM-NNs** as performant **surrogate models** to solve **incompressible fluid flow problems**
- Since **efficient preconditioners** are **absolutely necessary**, but **difficult to obtain**, the method currently only performs well for **low to moderate Reynolds numbers** ( $10^2$  to  $10^3$ )
- The presented results have been created as part of the **DLR projects PISA (finished)** and **TIARA (ongoing)**