

Prototypical Control for the Digital Twin of Aircraft Environmental Control System

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Imprint

Topic: Prototypical Control for the Digital Twin of Aircraft

Environmental Control System

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Institute: Institute of System Architectures in Aeronautics (SL) – Air-Vehicle

Thermal and Energy Systems

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Hardware test rig





- Test on a single system / a few systems together
- Real conditions but difficult to test all cases
- Real behaviour













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Digital Twin: Virtual test rig





- Possibility to carry out impact assessment at aircraft level
- Tests in multiples conditions and configurations
- Tests of all systems together
- Test possible at an early stage of development
- Results depend on model validity











Hardware test rig



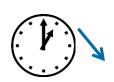


- Test on a single system / a few systems together
- Real conditions but difficult to test all cases
- Real behaviour









- Possibility to carry out impact assessment at aircraft level
- Tests in multiples conditions and configurations
- Tests of all systems together
- Test possible at an early stage of development
- Results depend on model validity

Complementarity

- Hardware tests for subsystem model validation
- Digital Twin for overall system demonstration















- Interface definition
- Simple and robust models of all systems
- Surrounding models (mission, controllers, environment)

1. Initial Digital
Twin Model

2. Digital Twin Improvement

- Development of highfidelity models shared as FMU by the partners
- Integration inside the Digital Twin Model



- Validation of the highfidelity models by hardware tests
- Validation of the FMU integrations inside the Digital Twin

3. Models validation

4. Virtual Demonstration

- Simulation for various environmental and mission conditions
- Outputs for the different performance test cases









Initial Digital Twin Model





- Interface definition
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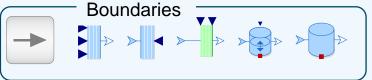
DLR ThermoFluid Stream Library

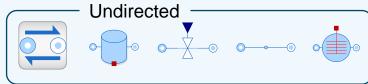
Highlights

Free open-source library

https://github.com/DLR-SR/ThermofluidStream

Modeling complex thermodynamic systems using gases and/or fluids



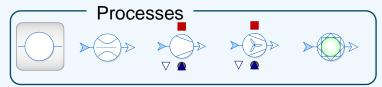


Robust

No larger non-linear equation systems

User friendly

- Simple parameterization, easy to initialize
- Many base components easy to adapt
- Many examples to start with





Topology









Tool compatible

- Dymola
- **OpenModelica**















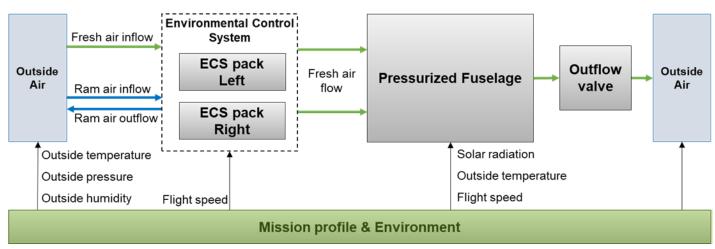


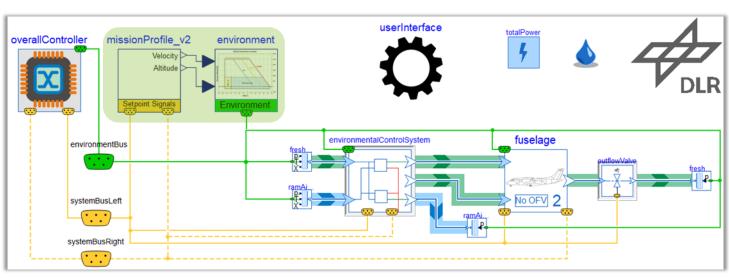


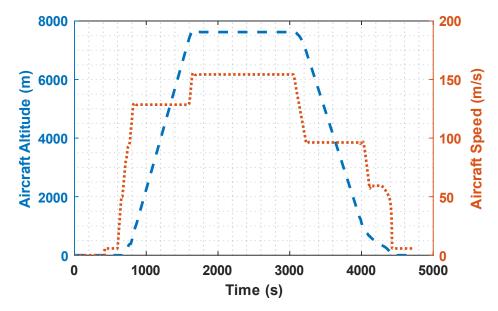


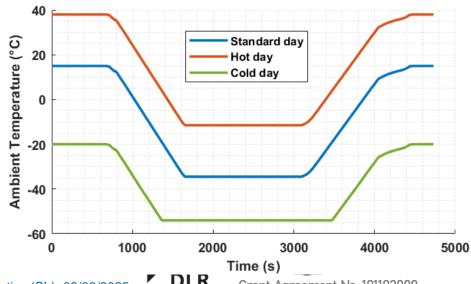






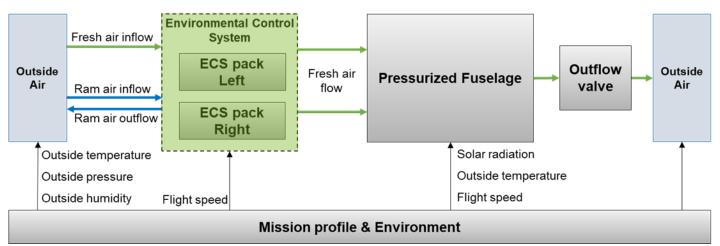


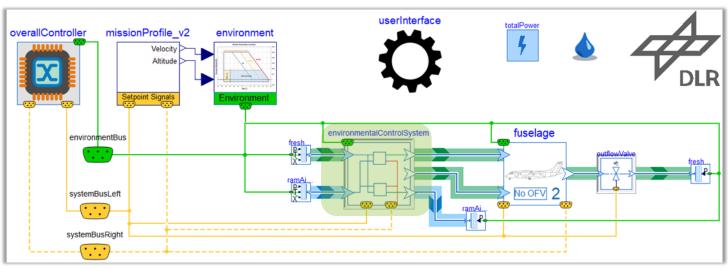


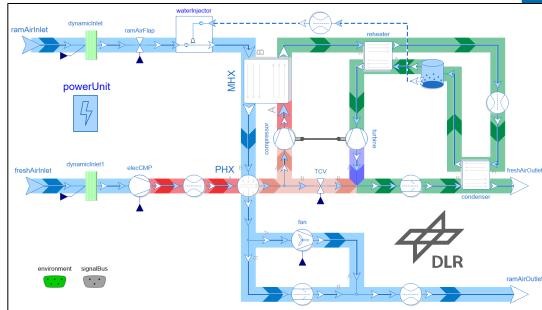










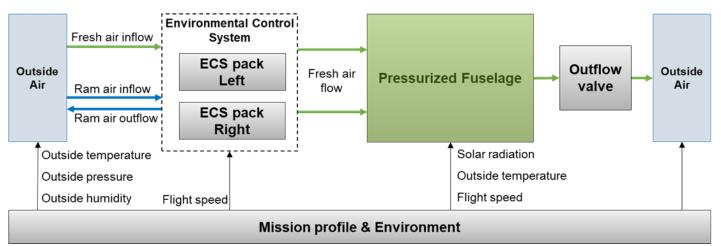


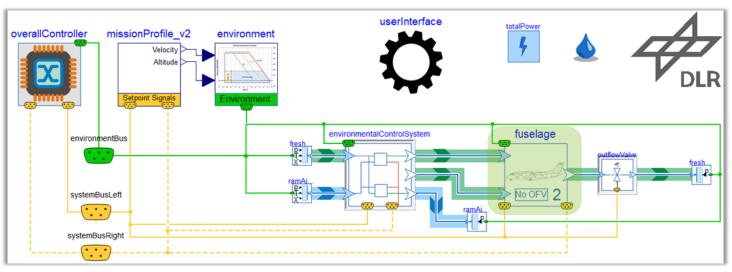
Electrified conventional ECS pack. Bleed air from the engines has been replaced by outside air compressed by an electrical compressor

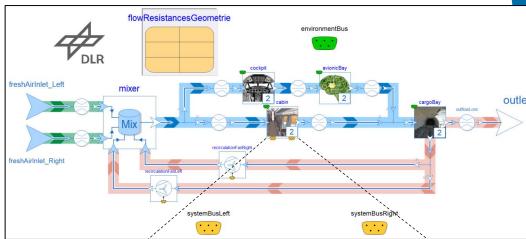


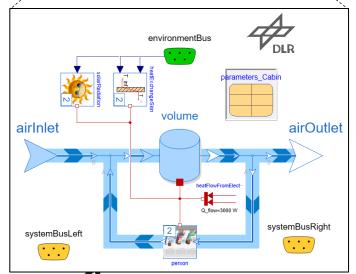








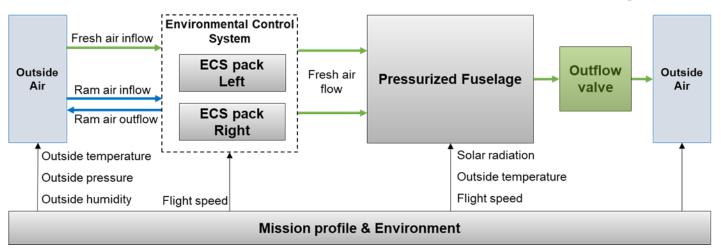


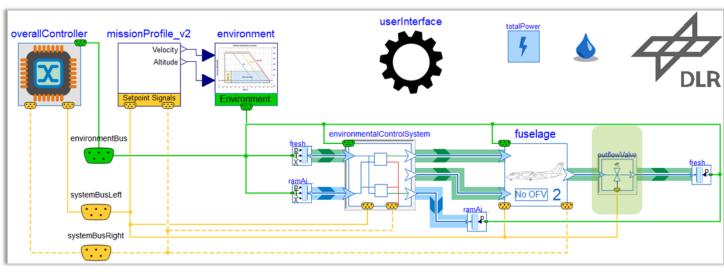


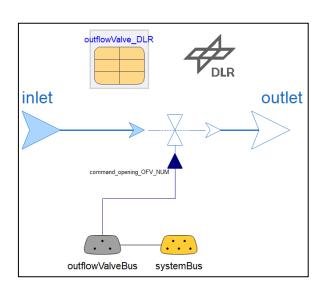




















Digital Twin Improvement





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- Simple and robust models of all systems
- Surrounding models (mission, controllers, environment)

1. Initial Digital
Twin Model

2. Digital Twin Improvement

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- Integration inside the Digital Twin Model



- Validation of the highfidelity models by hardware tests
- Validation of the FMU integrations inside the Digital Twin

3. Models validation

4. Virtual Demonstration

- Simulation for various environmental and mission conditions
- Outputs for the different performance test cases

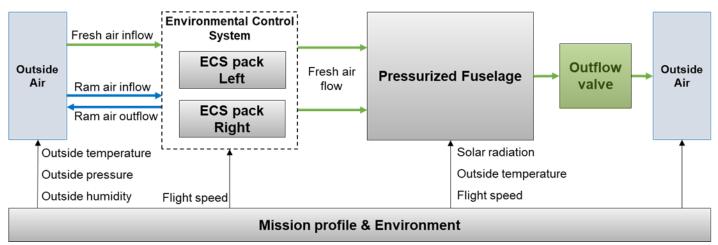


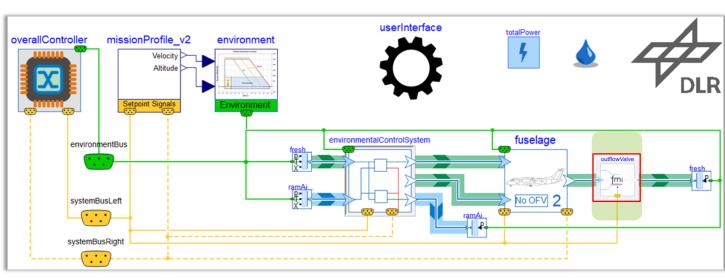


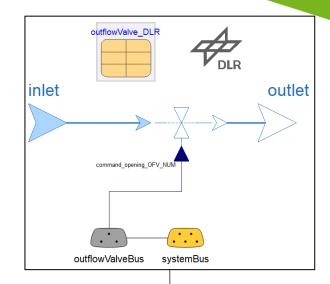




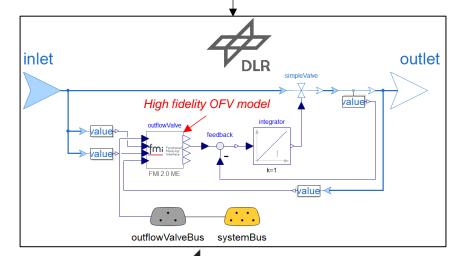
Digital Twin Improvement







Is replaced by high fidelity model shared as FMU



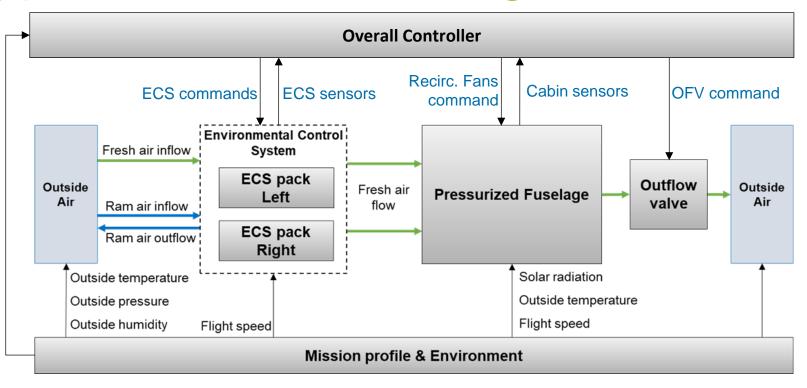






Mission Setpoints

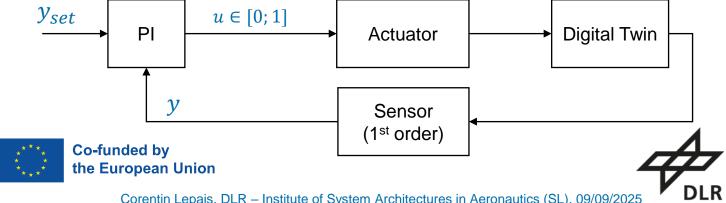
- Cabin Temperature
- Cabin pressure
- Cabin ventilation
- Fresh air supply



Controller Requirements

- Simplicity
- Adaptability
- Robustness

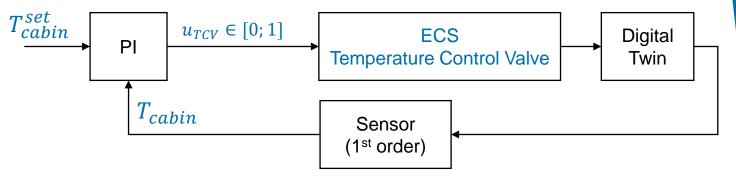


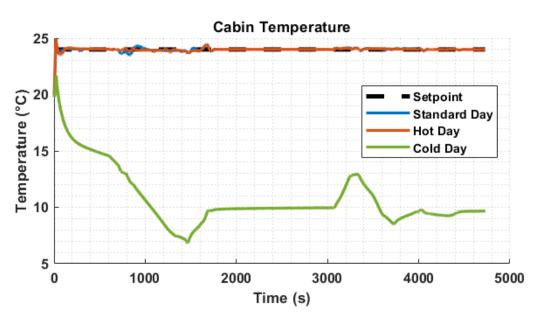


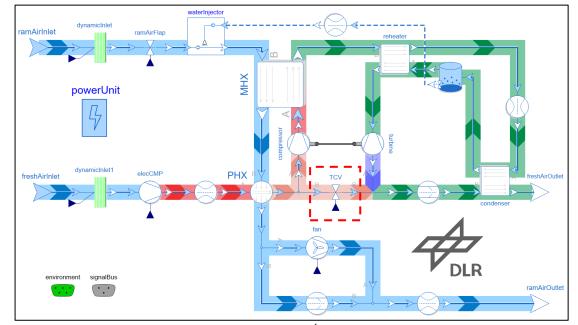


Setpoint Actuator Command Signal

Cabin
Temperature ECS – TCV TCV opening







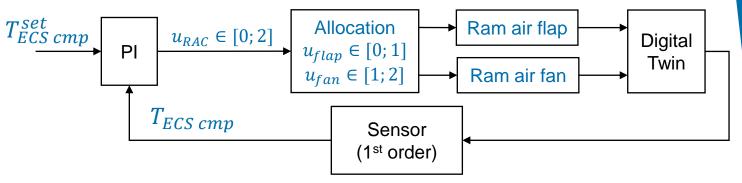


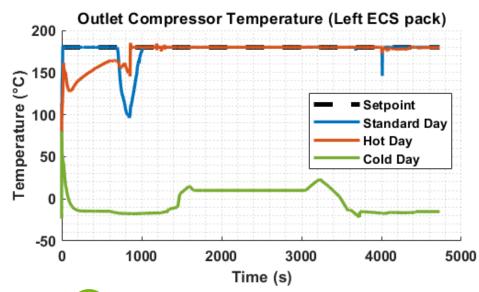


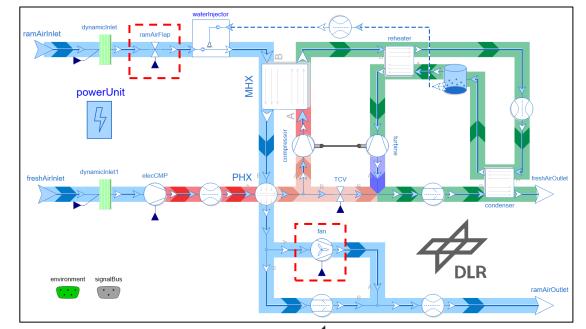




Setpoint	Actuator	Command Signal
Cabin Temperature	ECS – TCV	TCV opening
ECS compressor temperature	Ram Air channel	Ram air flap Ram air fan





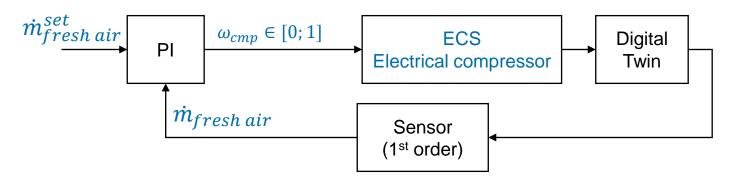


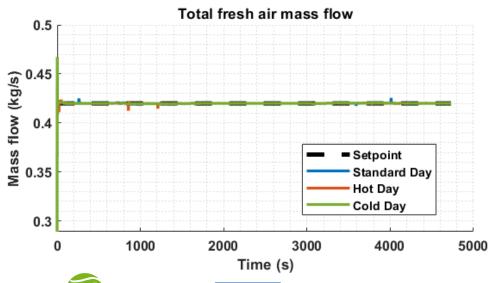




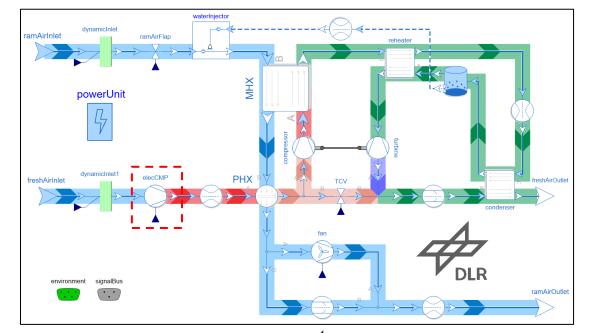


Setpoint	Actuator	Command Signal
Cabin Temperature	ECS – TCV	TCV opening
ECS compressor temperature	Ram Air channel	Ram air flap Ram air fan
Fresh air mass flow	ECS – Elec. Compressor	Compressor speed





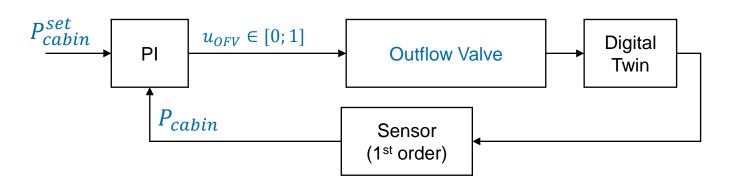
CLEAN AVIATION

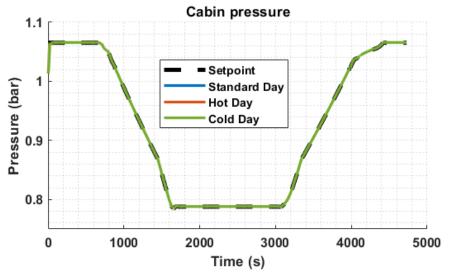


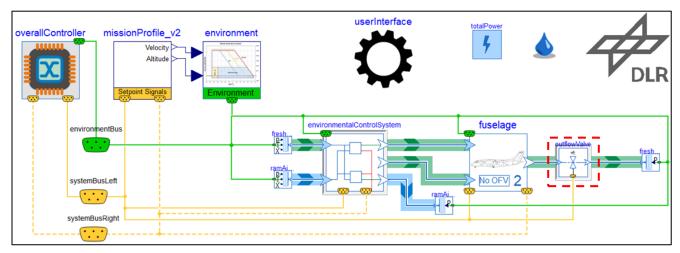




Setpoint	Actuator	Command Signal
Cabin Temperature	ECS – TCV	TCV opening
ECS compressor temperature	Ram Air channel	Ram air flap Ram air fan
Fresh air mass flow	ECS – Elec. Compressor	Compressor speed
Cabin Pressure	Outflow valve	Outflow valve opening





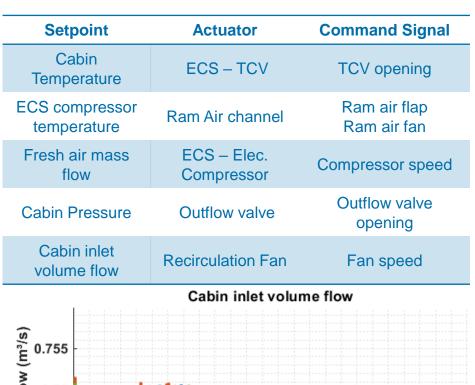


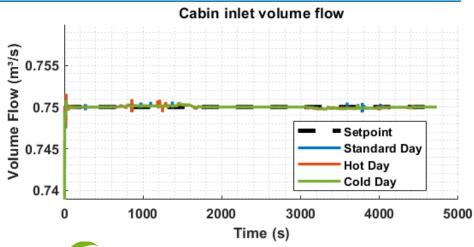


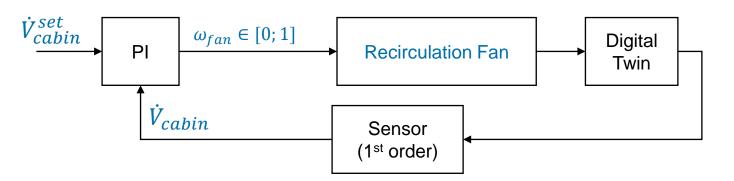


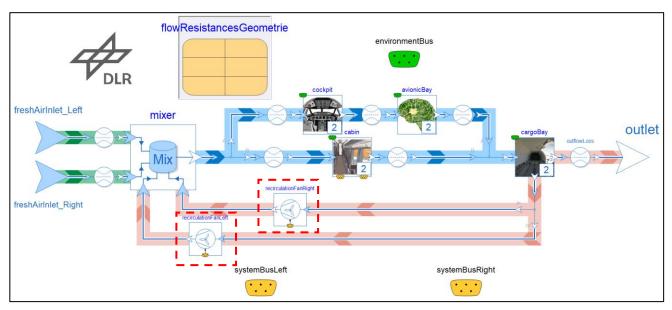












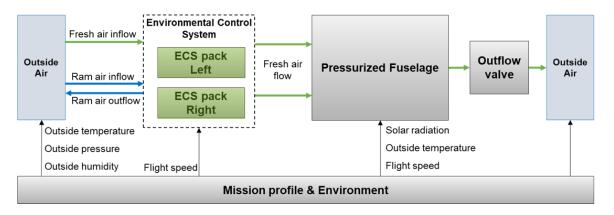




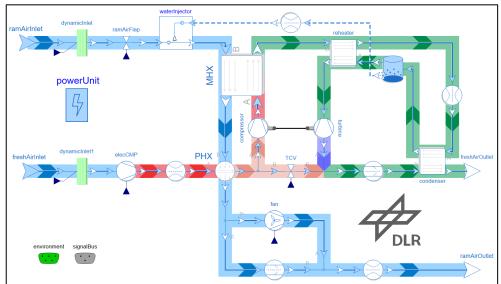




Prototypical Controller Robustness



Initial ECS pack architecture



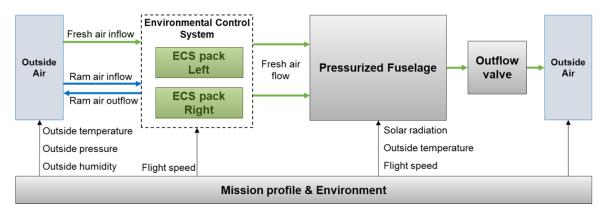




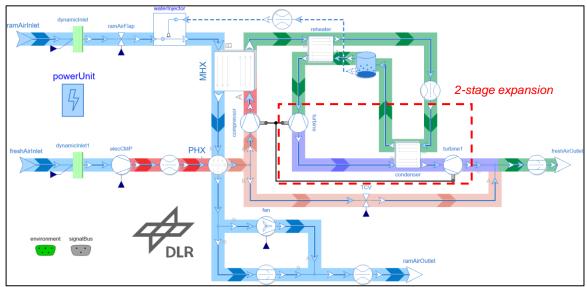


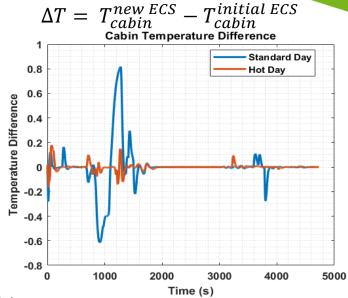


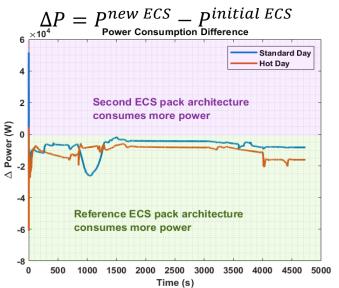
Prototypical Controller Robustness



New ECS pack architecture













Outlook

- Complete model of an aircraft environment control system based on simple models
- Working prototypical control for complete gate-to-gate mission and various ambient conditions
 - Interface definition
 - Simple and robust models of all systems
 - Surrounding models (mission, controllers, environment)
 - 1. Initial Digital
 Twin Model

2. Digital Twin Improvement

- Development of highfidelity models shared as FMU by the partners
- Integration inside the Digital Twin Model
 - ✓ Prototypical control still working when used with a different ECS pack
 - → Test to be done now with an high-fidelity FMU models

→ Once all simple models are replaced by high-fidelity ones, validation of the model integrations can be performed and then the complete system can be virtually demonstrated

- Validation of the highfidelity models by hardware tests
- Validation of the FMU integrations inside the Digital Twin

3. Models validation

4. Virtual Demonstration

- Simulation for various environmental and mission conditions
- Outputs for the different performance test cases









Conclusion

- Complete initial Digital Twin of an Environmental Control System for hybrid electric aircraft based on simple models
- Prototypical controller design to allow simulation on gate-to-gate mission with desired temperature and pressure inside the cabin
- Prototypical controller robustly designed to make it compatible with high-fidelity models

Contact

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Clean Aviation is the EU's leading research and innovation program for transforming aviation towards a sustainable and climate neutral future. As a European public-private partnership, Clean Aviation pushes aeronautical science beyond the limits of imagination by creating new technologies that will significantly reduce aviation's impact on the planet, enabling future generations to enjoy the social and economic benefits of air travel far into the future. Visit the website to find out more about Clean Aviation; www.clean-aviation.eu



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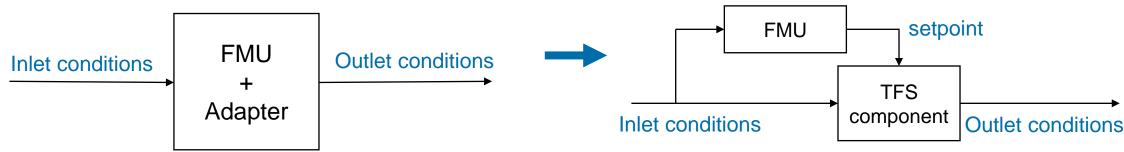




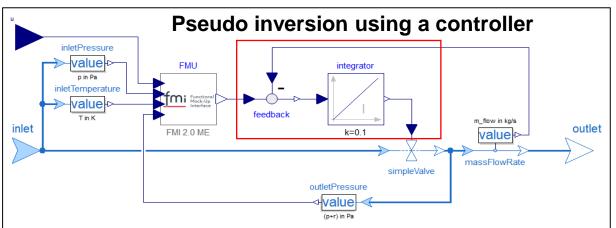


Digital Twin Improvement - Integration of FMU

Modelica Standard Library approach



Control of a TFS component to mimic the behaviour of the FMU model



Paper: Gebhart, Raphael and Lepais, Corentin (2025) *Novel Control-like Approach for the Robust Integration of Functional Mock-Up Units into Digital Twins.* Engineering Proceedings, 90 (1), pp. 1-12. Multidisciplinary Digital Publishing Institute (MDPI). doi: 10.3390/engproc2025090012. ISSN 2673-4591

Presentation: Gebhart, Raphael and Lepais, Corentin (2024) <u>Digital Twin for simulation of Aircraft Thermal Management System architectures in TheMa4HERA.</u> 14th EASN International Conference, 2024-10-08 - 2024-10-11, Thessaloniki, Greece. (Unpublished)



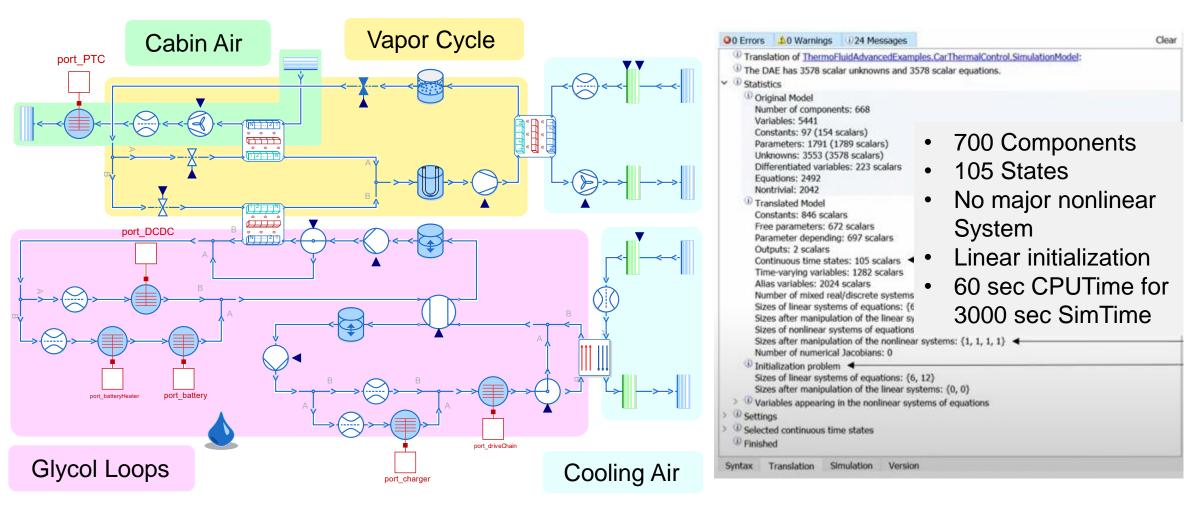




Control-like approach



Thermal Management System of an electric car











Prototypical Controller Robustness

