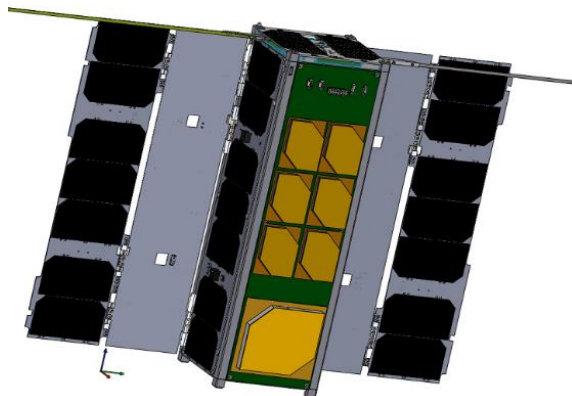


GNSS REMOTE SENSING OF THE PRETTY CUBESAT MISSION: A DEMONSTRATION TO VALIDATE MODELS OF IONOSPHERIC ELECTRON DENSITY

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15th IAA Symposium on
Small Satellites for Earth System Observation,
Berlin May 2025



beyond gravity

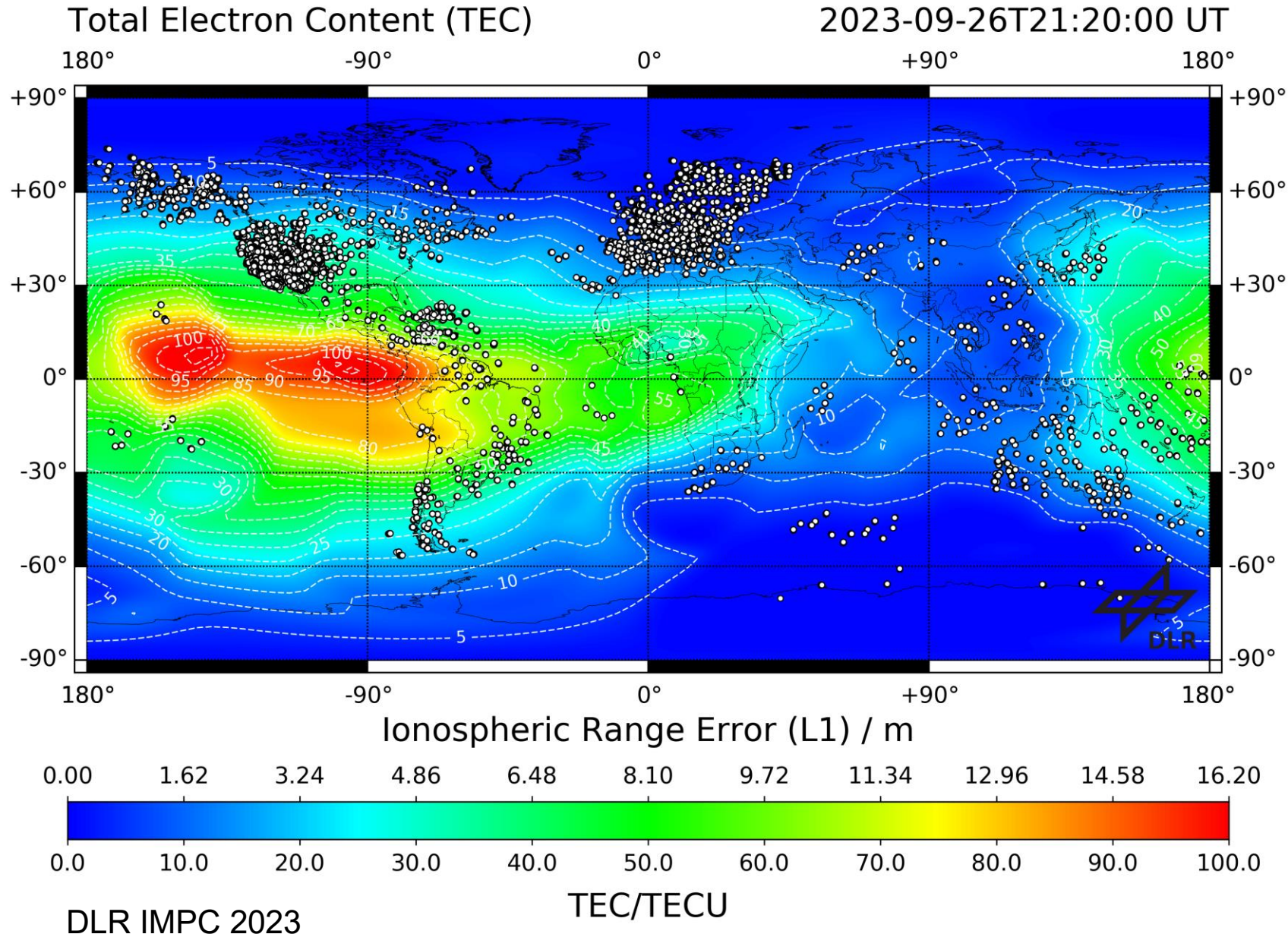
Outline



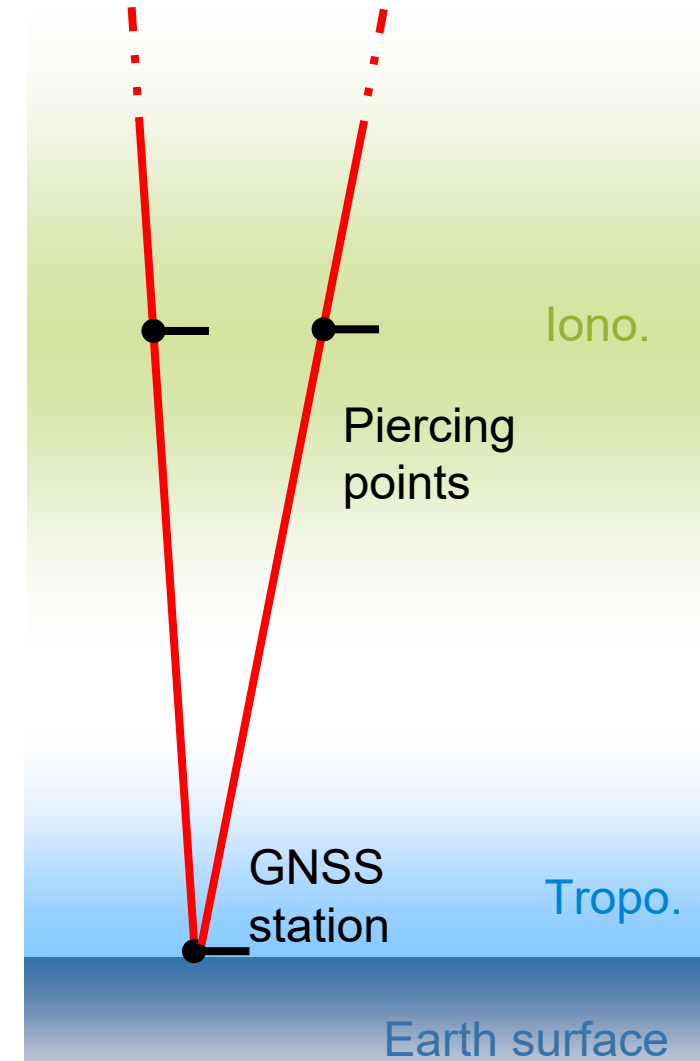
- Background of GNSS Remote Sensing
- Grazing Angle Reflectometry with PRETTY Mission
- Preliminary Results on Ionosphere Validation
- Conclusions & Outlook

Background of GNSS Remote Sensing

Ionospheric TEC Monitoring with GNSS



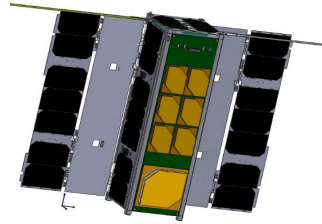
GNSS 19100 ... 23200 km



Motivation GNSS Remote Sensing

- A: Low Earth Orbiter

Wickert et al. 2016
Semmling et al. 2016
Cardellach et al. 2020
Moreno et al. 2023



- B: Aircraft

Semmling et al. 2014
Moreno et al. 2022



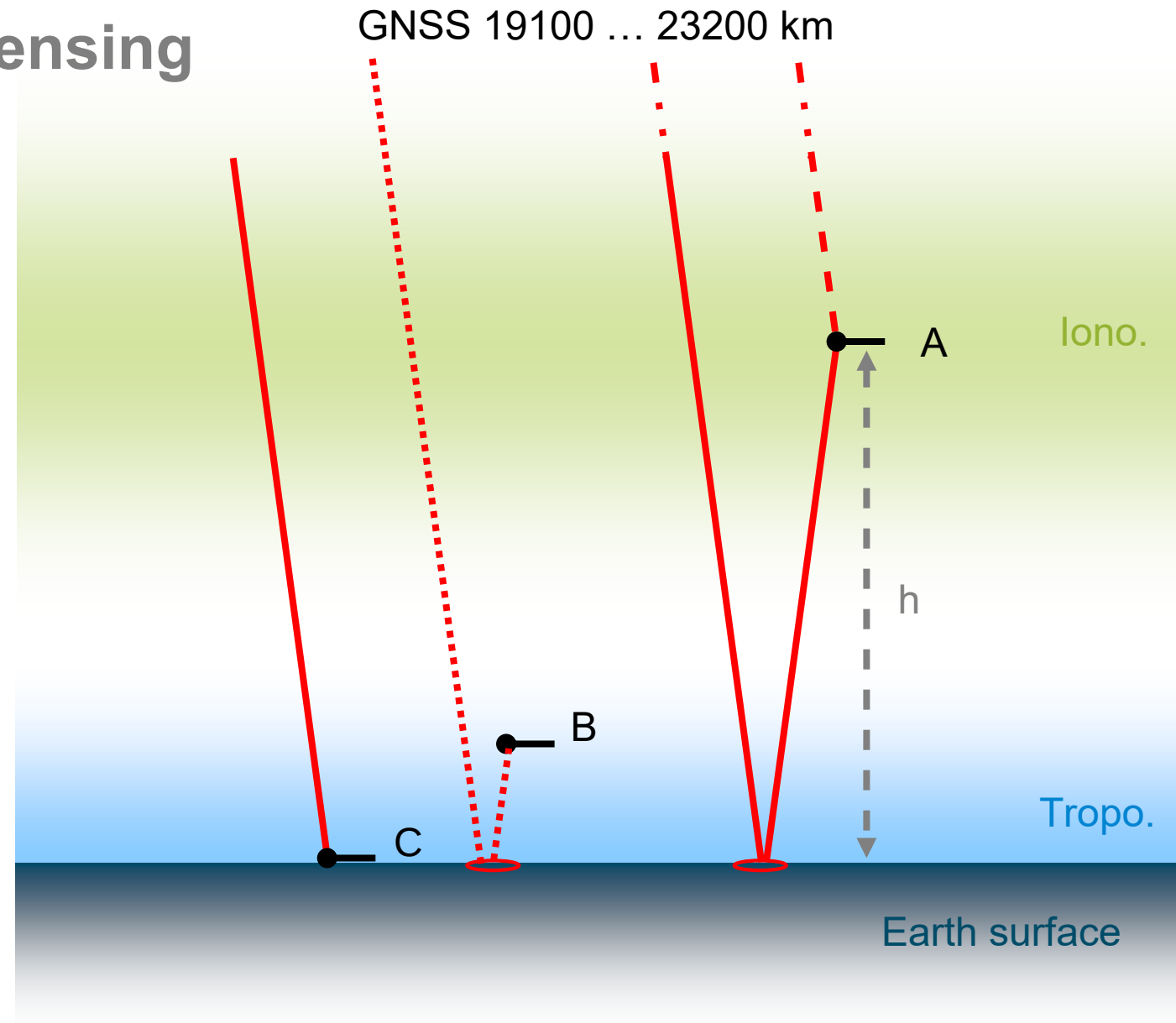
- C: Research Vessels

Wang et al. 2019
Semmling et al. 2019, 2022
Semmling et al. 2023



- Application

sea surface altimetry water vapor estimation
sea state estimation ionosphere monitoring
sea-ice detection



A: e.g. PRETTY, $h \sim 560$ km
B: e.g. HALO, $h \sim 3500$ m

C: e.g. Polarstern, $h \sim 25$ m

Considerable Factors Reflectometry

Sea Surface

- Roughness (Sea State)
- Penetration (e.g. Sea Ice)
- ...



Atmosphere

- Refraction (neutral gas and ionosphere)
- Scintillation (Plasma Depletion, Space Weather)
- ...

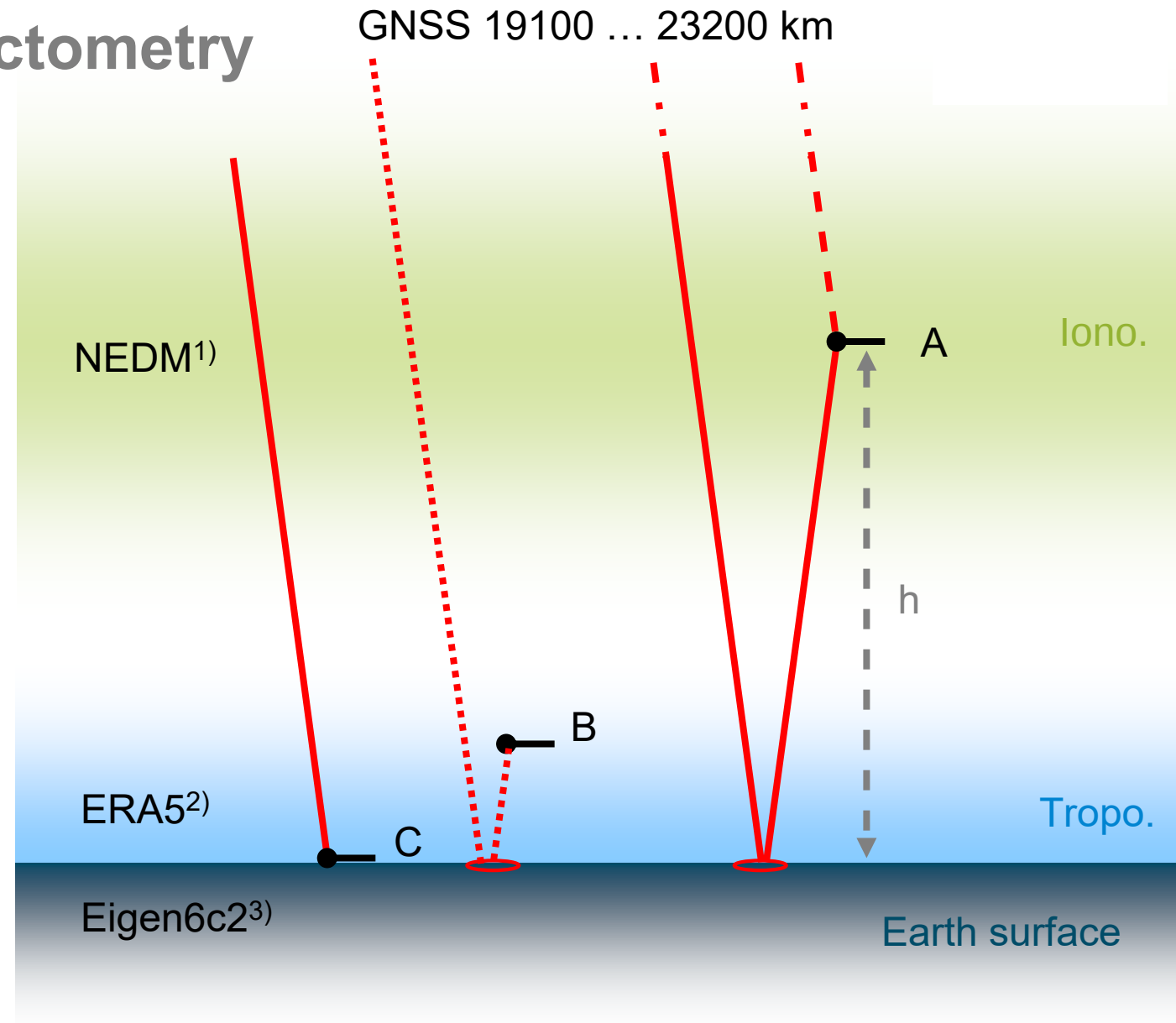


Receiver & Transmitter

- Position & Attitude uncertainty (of vessel, aircraft or satellite)
- Antenna & Instrumental parameter (e.g. gain pattern)
- ...



degree of disturbance



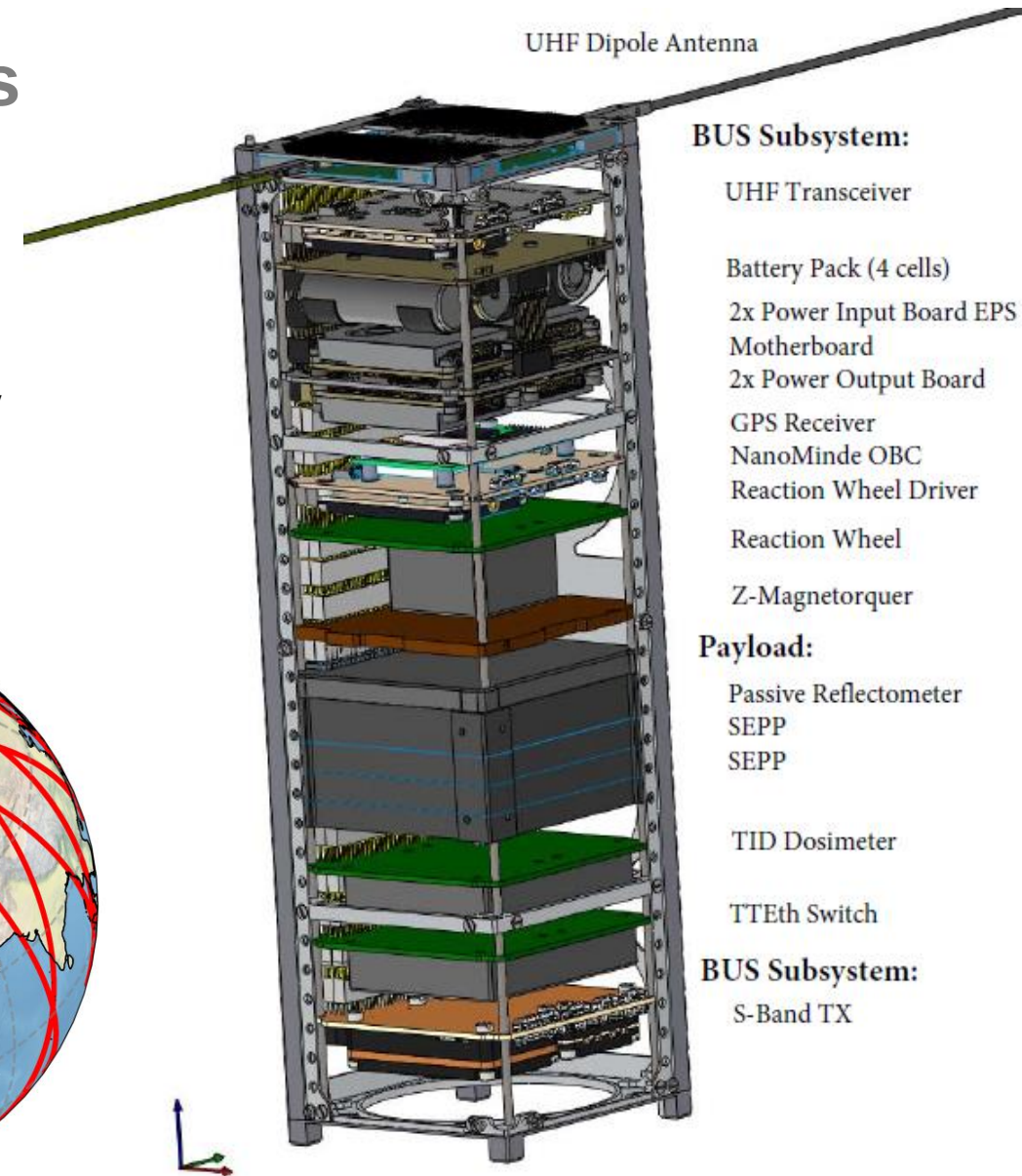
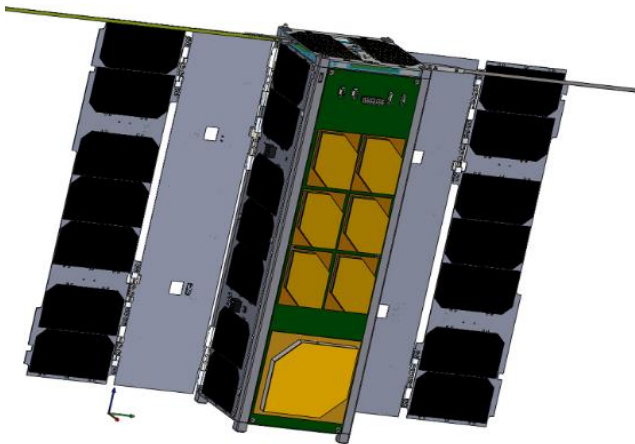
1) Hoque et al. 2022 ; 2) Hersbach et al. 2020 ; 3) Förste et al. 2013

Grazing Angle Reflectometry with PRETTY Mission

PRETTY Satellite Mission Parameters

Passive **RE**flecTometry and dosime**Tr**Y

- ESA CubeSat mission, developed by an Austrian consortium led by Beyond Gravity Austria
- Size: 30 x 10 x 10 cm³
- Orbit: polar SSO, altitude 560 km
- GNSS-R antenna: RHCP, limb pointing
- GNSS-R grazing elevations: 0° to 15°
- GNSS-R signal carrier: L5



BUS Subsystem:

UHF Transceiver

Battery Pack (4 cells)

2x Power Input Board EPS

Motherboard

2x Power Output Board

GPS Receiver

NanoMinde OBC

Reaction Wheel Driver

Reaction Wheel

Z-Magnetorquer

Payload:

Passive Reflectometer

SEPP

SEPP

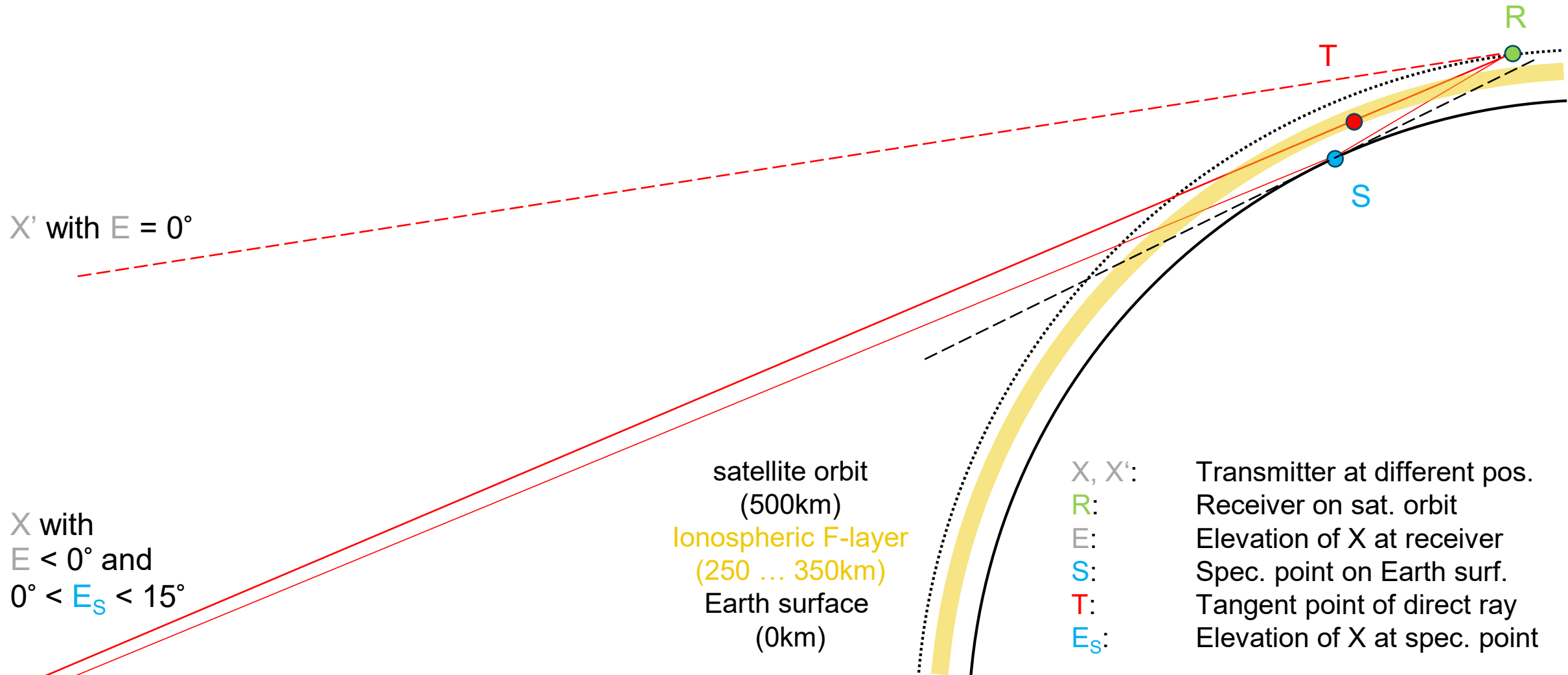
TID Dosimeter

TTeth Switch

BUS Subsystem:

S-Band TX

Ray Geometry at Grazing Angles



Scenarios of Ionospheric Delay



(1) Tangent point above F-layer:
iono. delay on reflected rays (XS, SR)
exceed the one on direct ray (XR)

X' with $E = 0^\circ$

(2) Tangent point in F-layer:
iono. delay on direct ray (XR) can exceed
the ones on reflected rays (XS, SR)

X with
 $E < 0^\circ$ and
 $0^\circ < E_s < 15^\circ$

satellite orbit
(500km)
ionospheric F-layer
(250 ... 350km)
Earth surface
(0km)

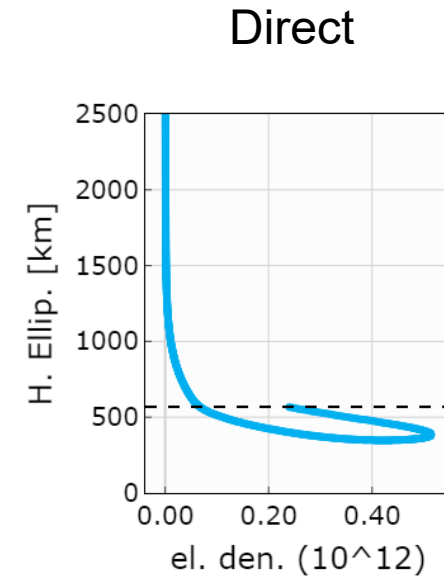
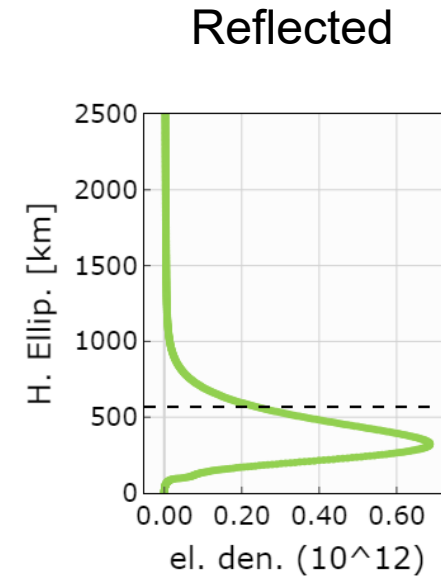
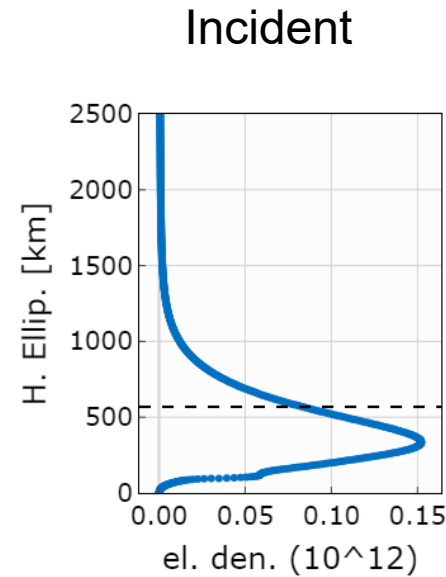
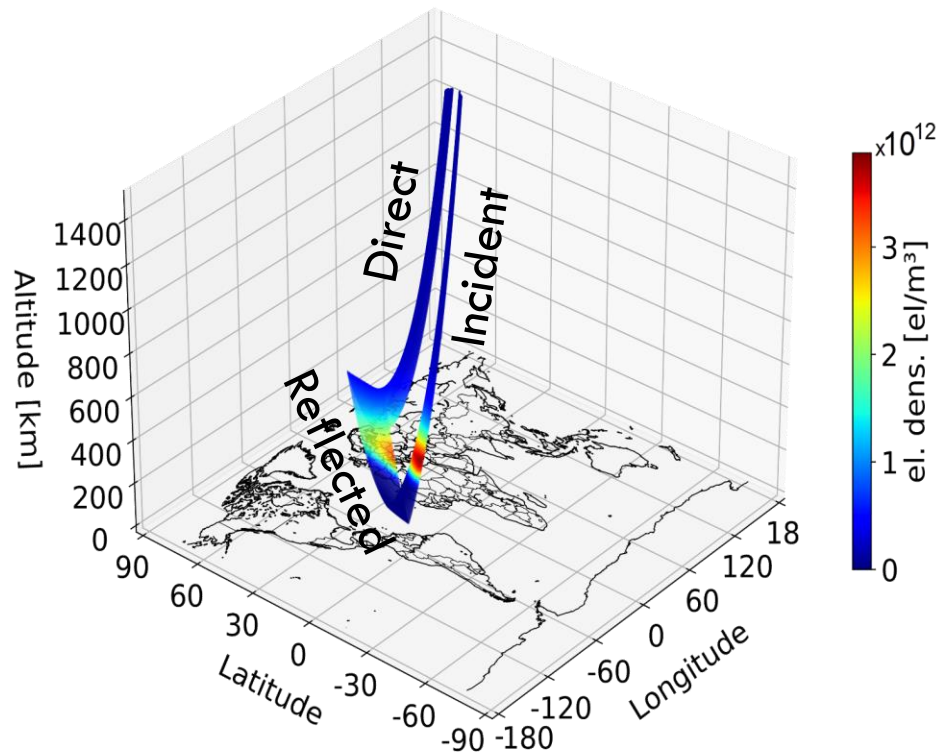
X, X': Transmitter at different pos.
R: Receiver on sat. orbit
E: Elevation of X at receiver
S: Spec. point on Earth surf.
T: Tangent point of direct ray
E_s: Elevation of X at spec. point

Preliminary Results on Ionosphere Validation

Modelling Ionospheric Electron Content

- Ray points every 10 km along ray paths (Direct, Incident, Reflected)
- Retrieve electron density from NEDM 2020 Model

PRN 303 - 2024.02.23 @ 13:30:26

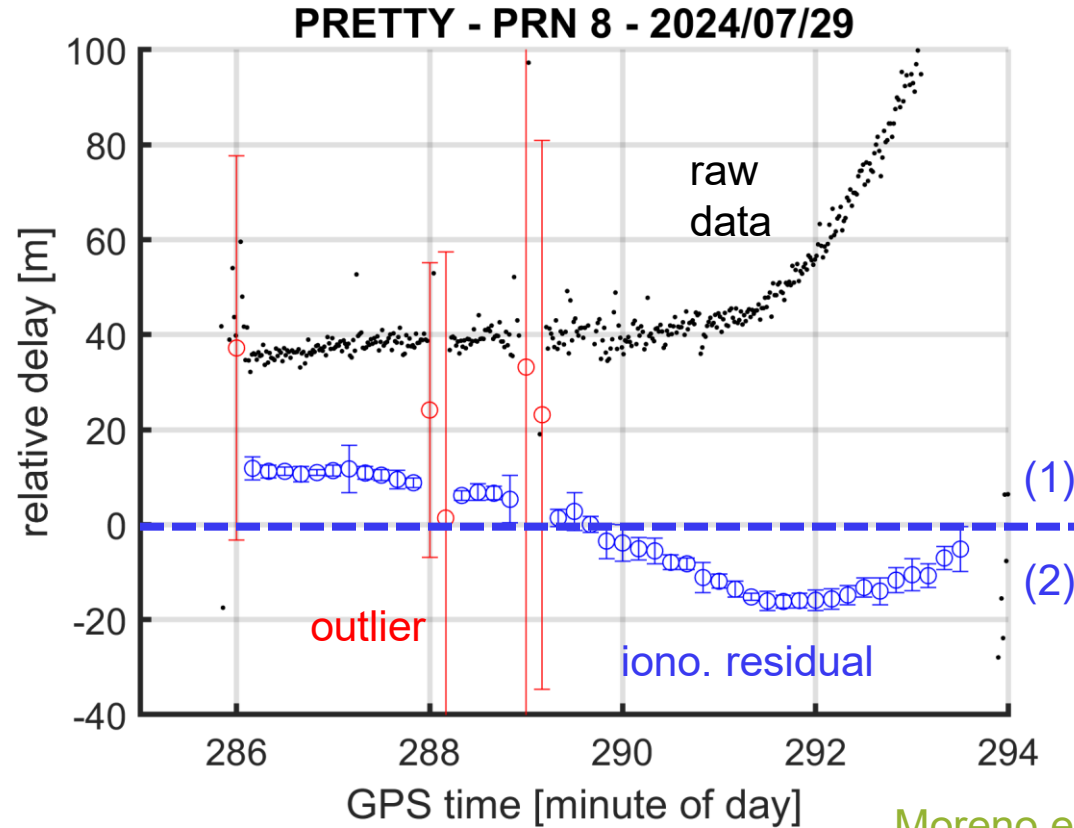


Moreno et al. 2023

elec. dens. $\rightarrow$ integrated to $\rightarrow$ elec. content

Observation Example GPS PRN 8 – 2024/07/29

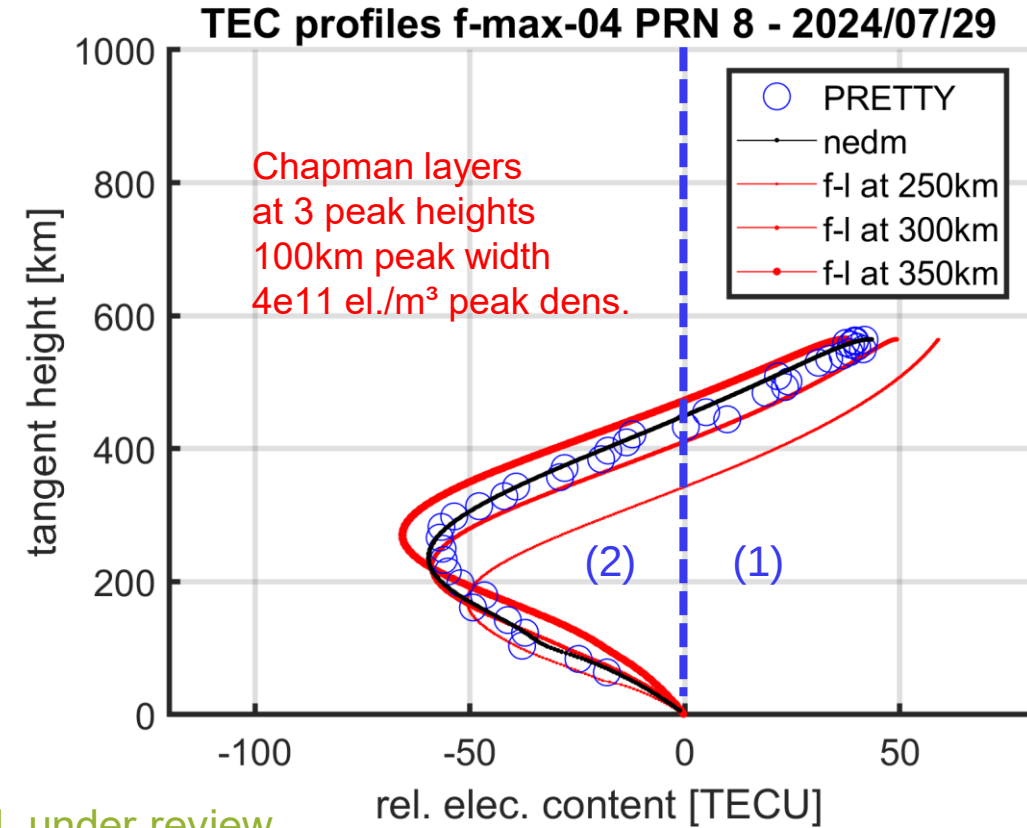
Satellite Obs. Data



Moreno et al. under review

Ionospheric residual is retrieved after correction of tropospheric delay. Residual below zero belong to scenario (2), tangent point in F-layer

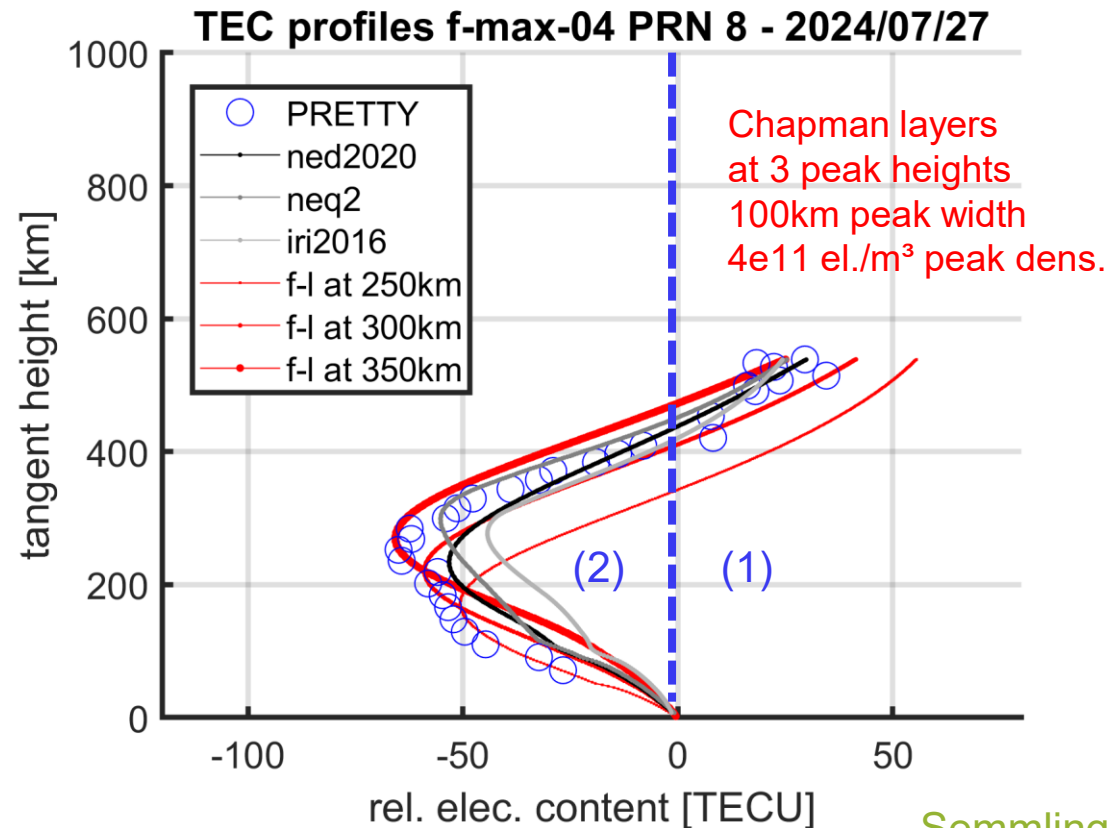
Comparison with model profiles



Iono. Residual is converted to relative electron content and compared to predictions with NEDM and chapman layers

More Examples for GPS and Galileo

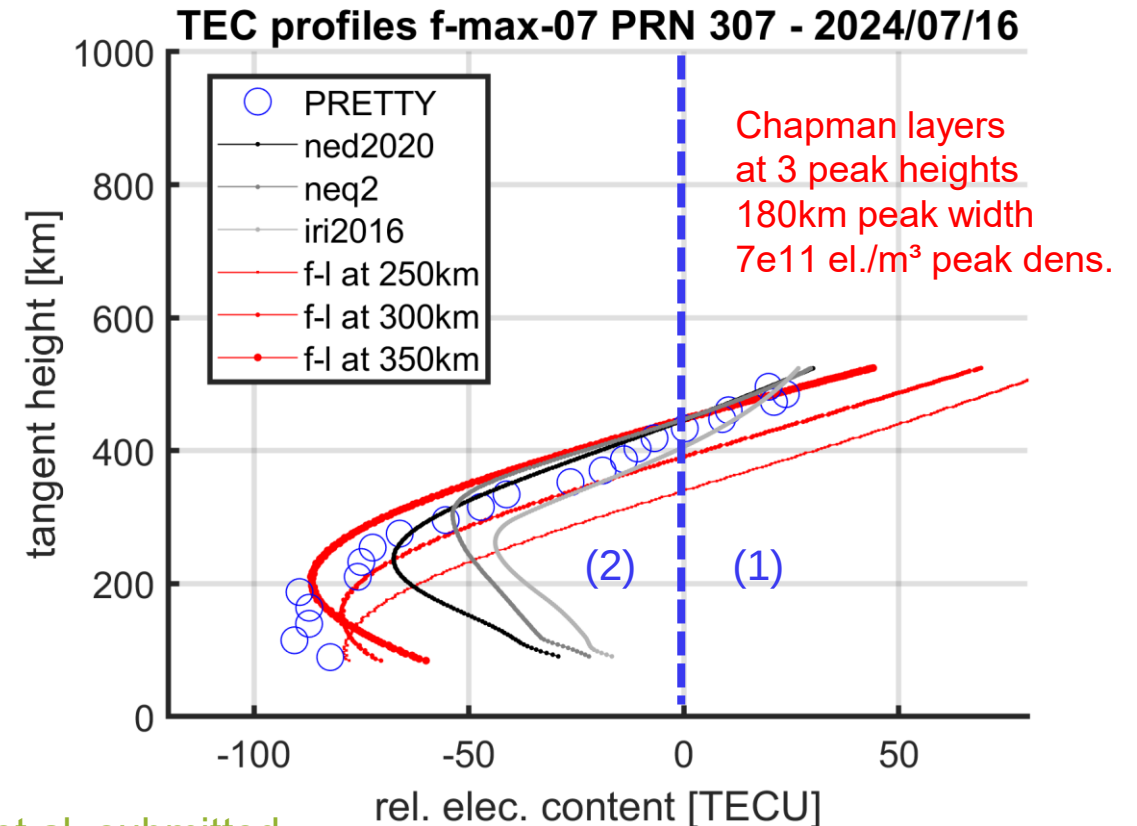
Comparison with model profiles



Semmling et al. submitted

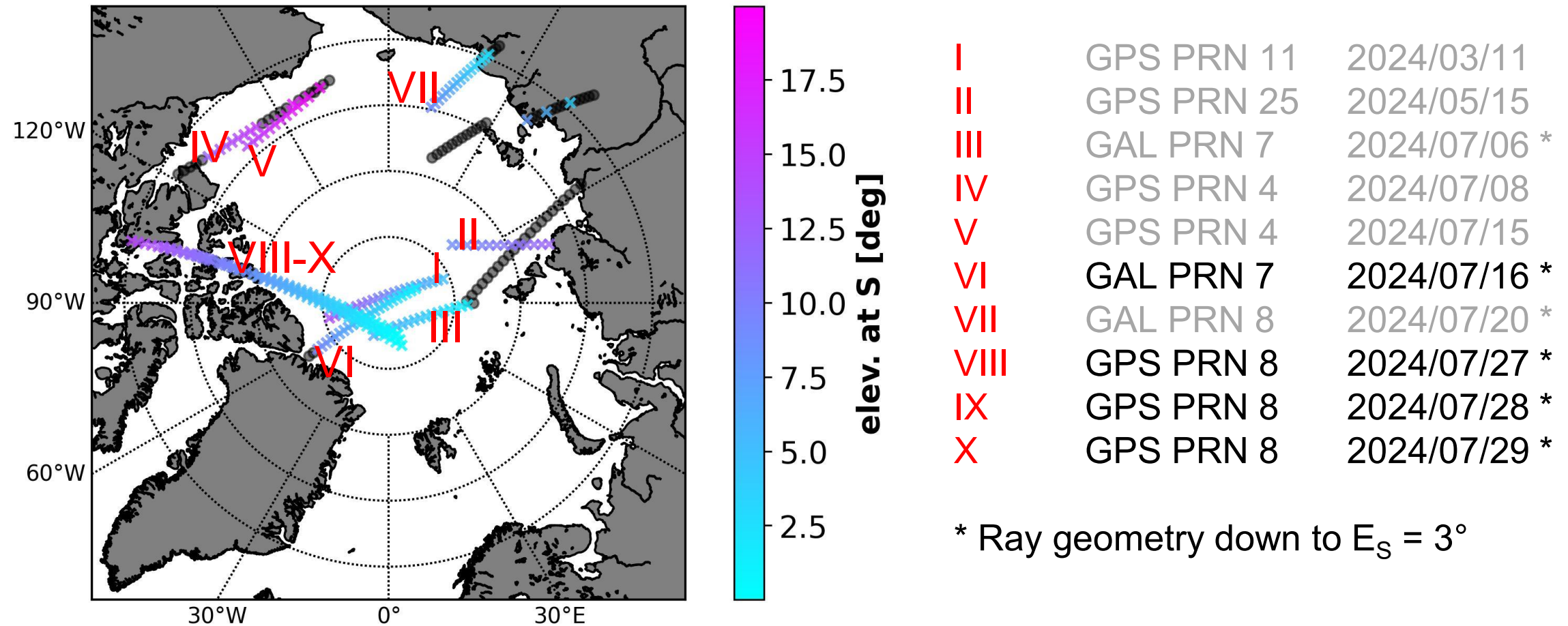
Very similar to previous example,
two days earlier same ground track

Comparison with model profiles



Significantly different example,
about 2 weeks earlier, different track
larger peak width and peak density

Geo-Reference and Uncertainties



* Ray geometry down to $E_s = 3^\circ$

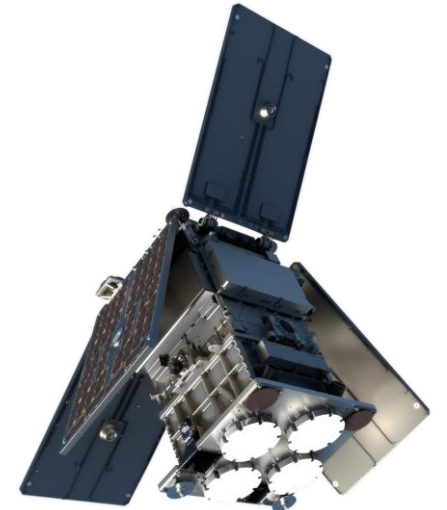
Tropo. Unc.: ~60cm (2 TECU) *
Surface Unc.: ~10cm (< 1 TECU) *

Bending Effect: ~10% of tang. Height *
Receiver Orbit Unc.: t.b.d.

Conclusions & Outlook



- **PRETTY** cubesat mission provides **grazing-angle reflectometry data**
- Reflection with **sufficient SNR over smooth Earth surface** e.g. Arctic sea ice
- **Elec. content profiles** modelled and retrieved from PRETTY observation
- Some retrievals give **excellent validation of model computation** (NEDM)
- Local minimum of profiles when **sensitive to F-layer structure**
- Fitted layer allow **to estimate ionospheric peak- and scale height**
- **Open uncertainties** (especially orbit and bending) to be determined



Acknowledgements

<https://www.hydrognss.org/>

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Thank you for your attention

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