

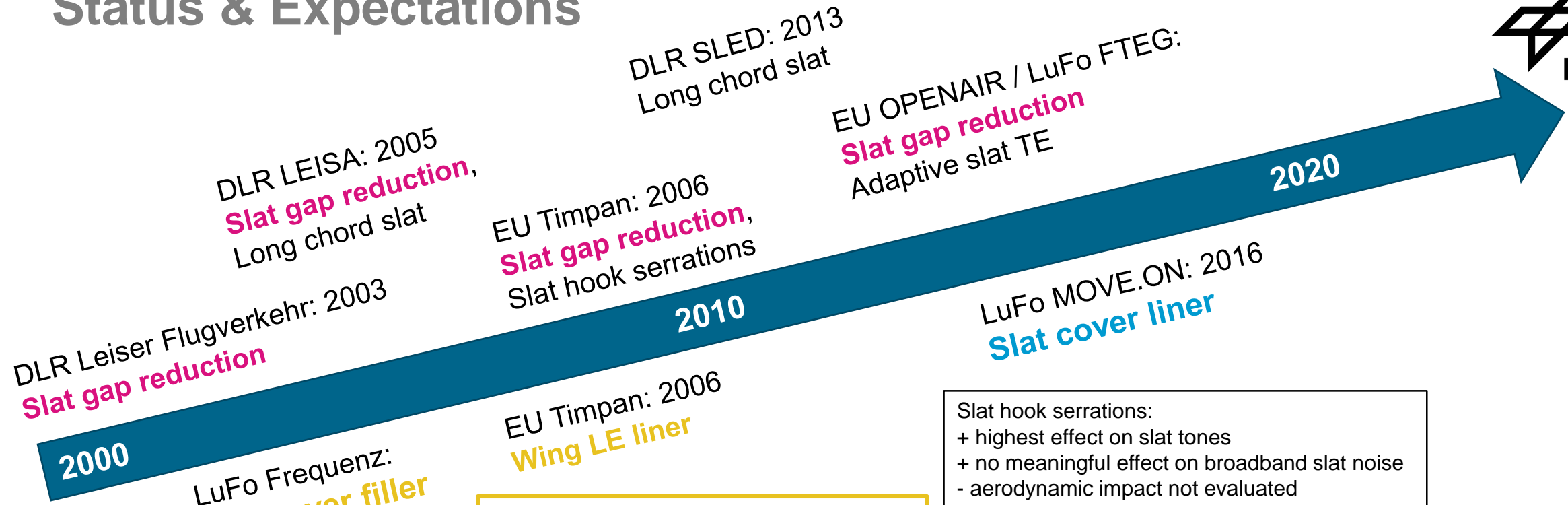
THE CHALLENGE OF SLAT NOISE REDUCTION BY RETRO-FIT TECHNOLOGY

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Status & Expectations



Slat gap reduction & Adaptive slat TE:
+ > 6 dB
 + up to noise cancellation for closed gap
 - high aerodynamic penalty
 - high system complexity
 - assumed to include additional weight

Slat cover filler:
+ up to 6 dB
 + low aerodynamic impact
 - complex design
 - high dependence on angle of attack

Slat hook serrations:
 + highest effect on slat tones
 + no meaningful effect on broadband slat noise
 - aerodynamic impact not evaluated

Long chord slat:
+ up to 7 dB
 + no aerodynamic penalty
 - larger slat tracks
 - additional weight

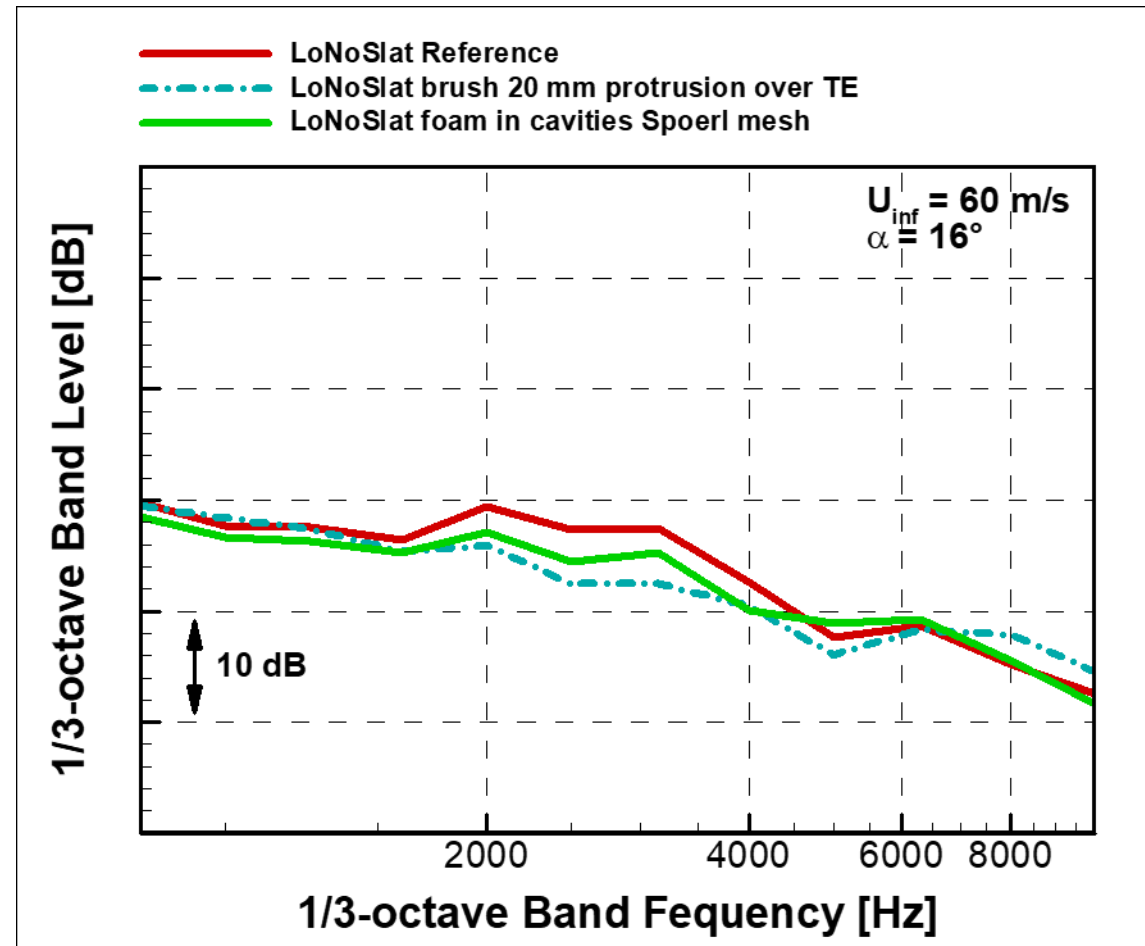
Slat cover liner:
+ up to 3 dB (improve by InVentor)
 + no aerodynamic impact
 + good options for integration (AI)
 - modest noise reduction



Starting Point

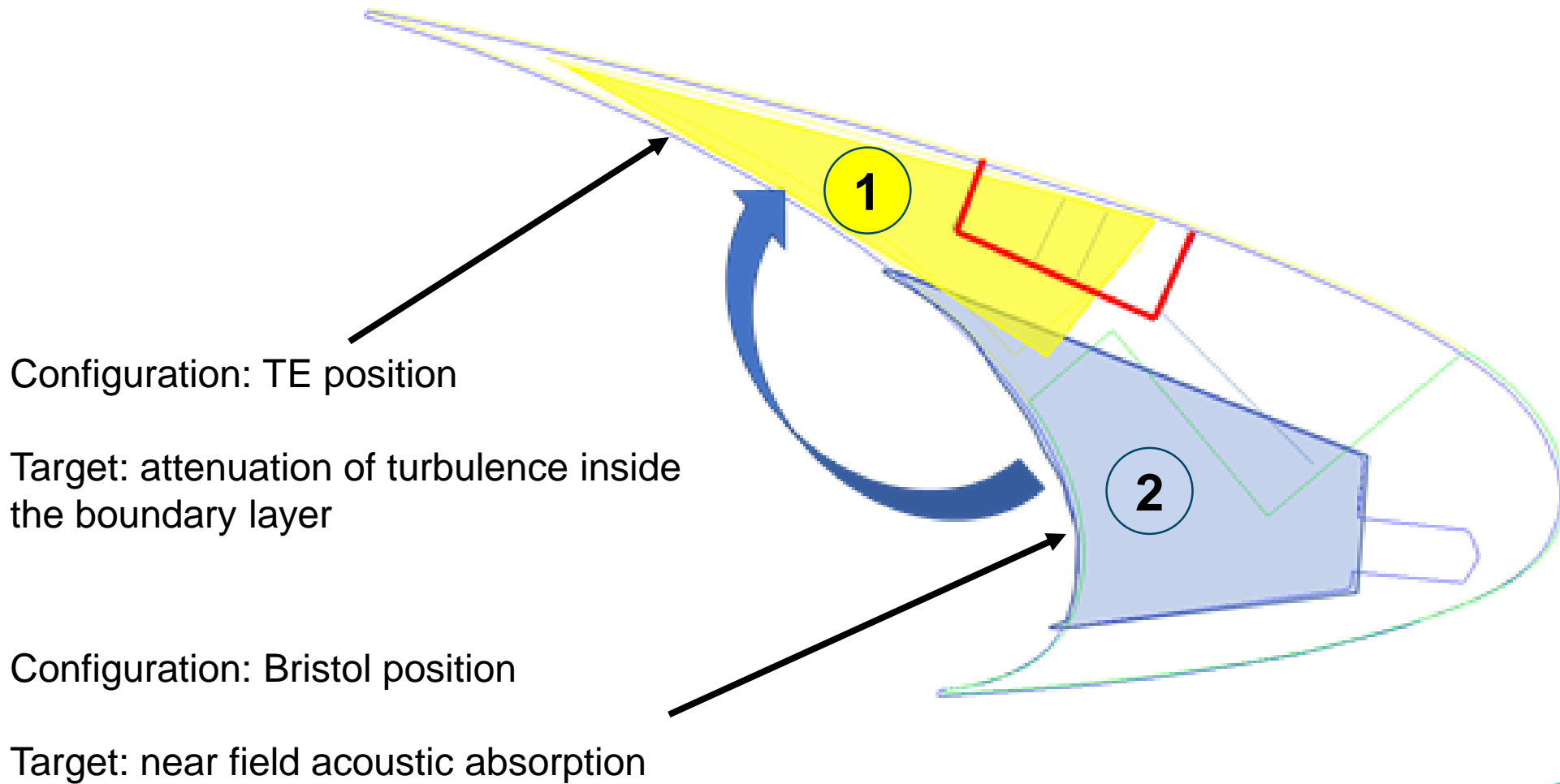
- German LuFo **MOVE.ON**
 - TE position
 - F16 model
 - Open cell PU foam covered with wire mesh
 - 3 – 4 dB noise reduction in frequency range 2 – 4 kHz
 - Cross check with TE brushes, know as „best solution“
 - Brushes ~1 dB quieter

➔ Cove filler meets expectations



Slat Cove Filler

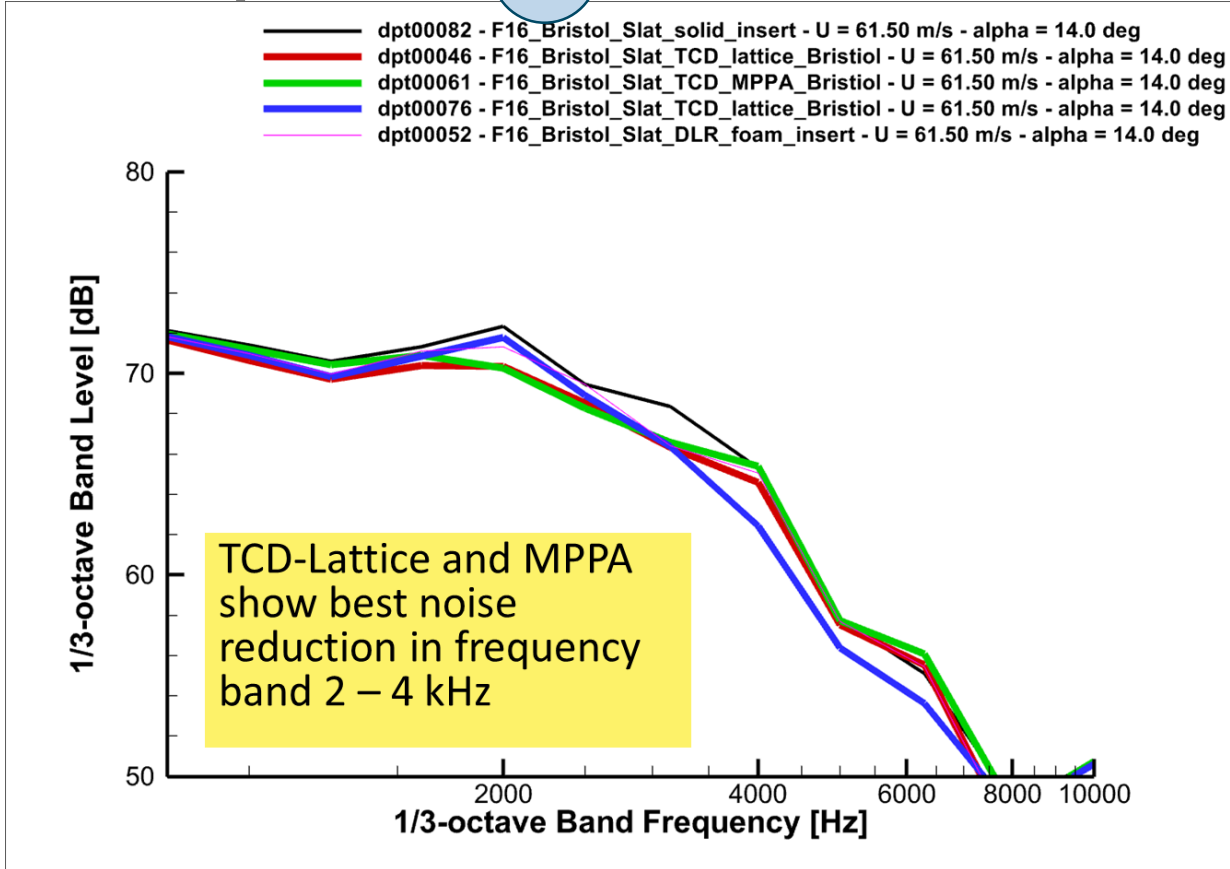
InVentor: Tested Configurations



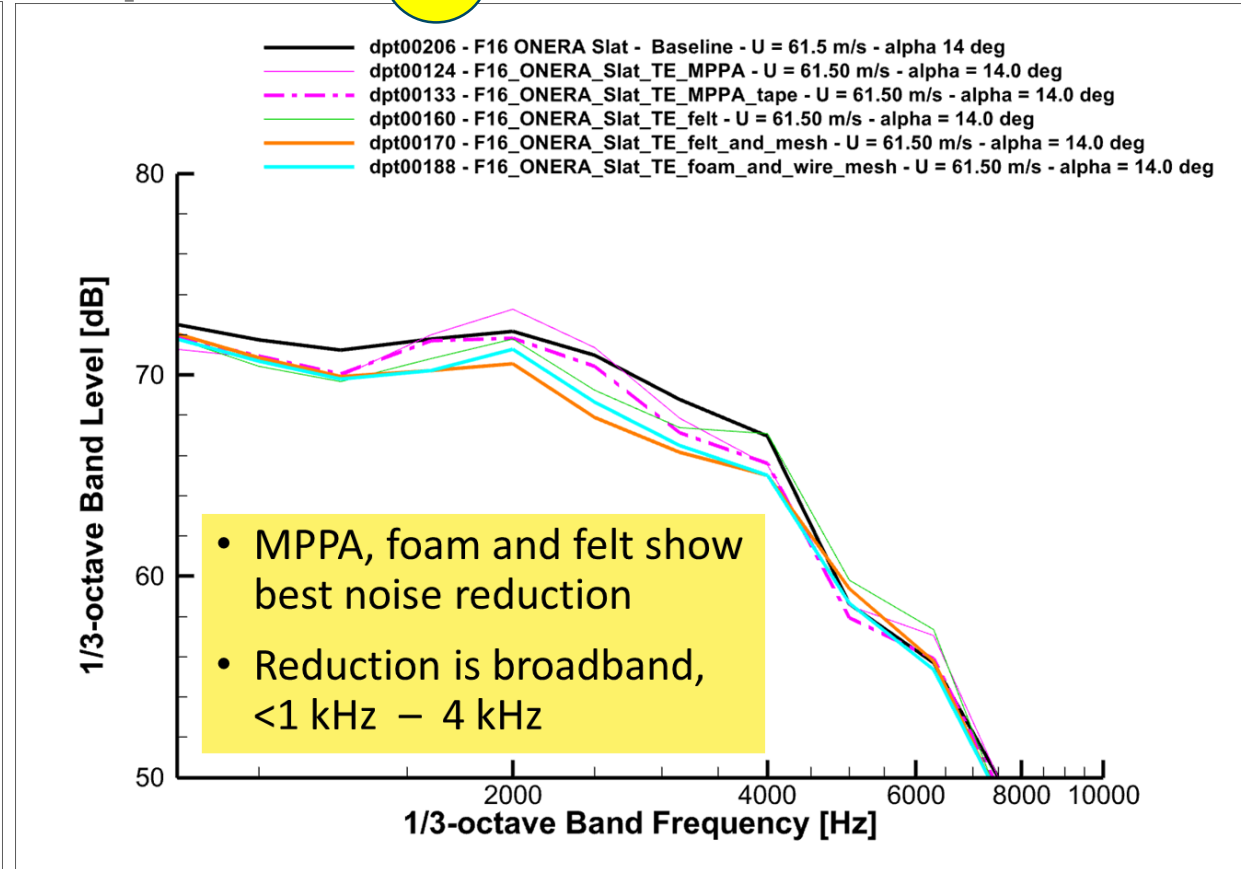
Large Number of Configurations Selected Results



Bristol position 2



TE position 1

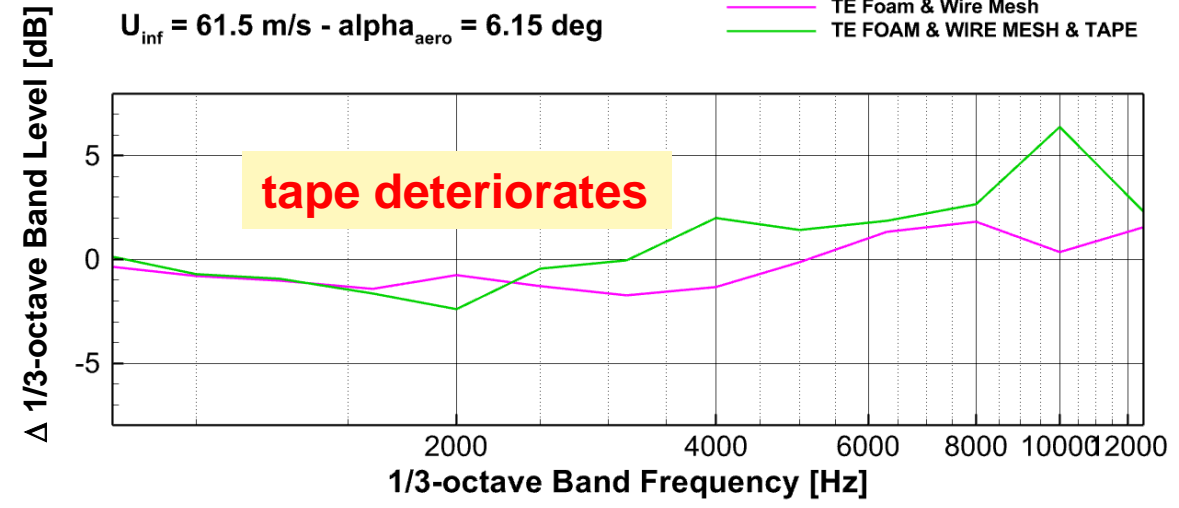
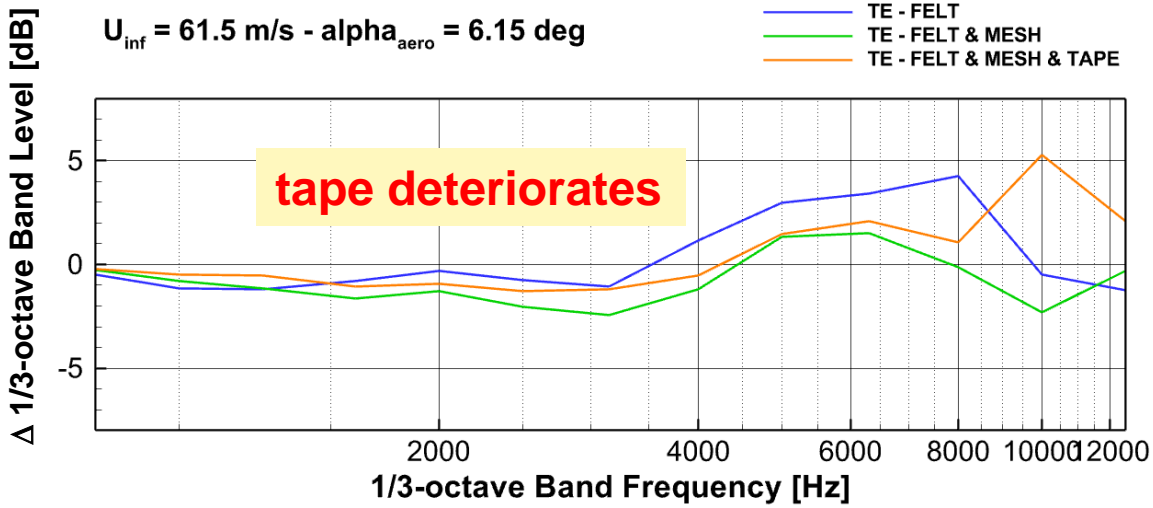
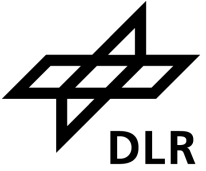


- Noise reductions are achieved – well done!?



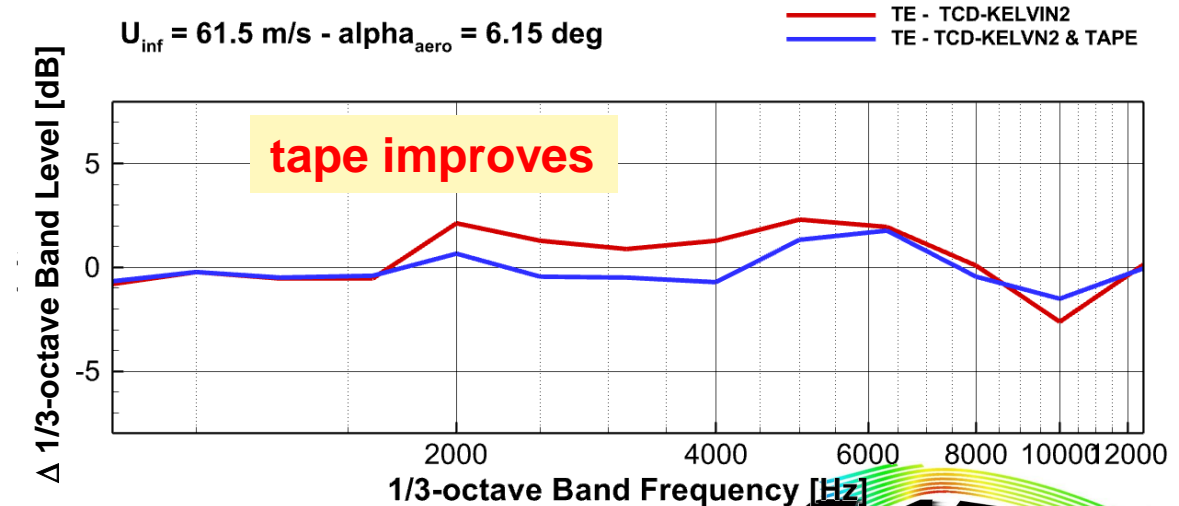
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How-to: Dimensioning – Installation – Noise Reduction



Open questions:

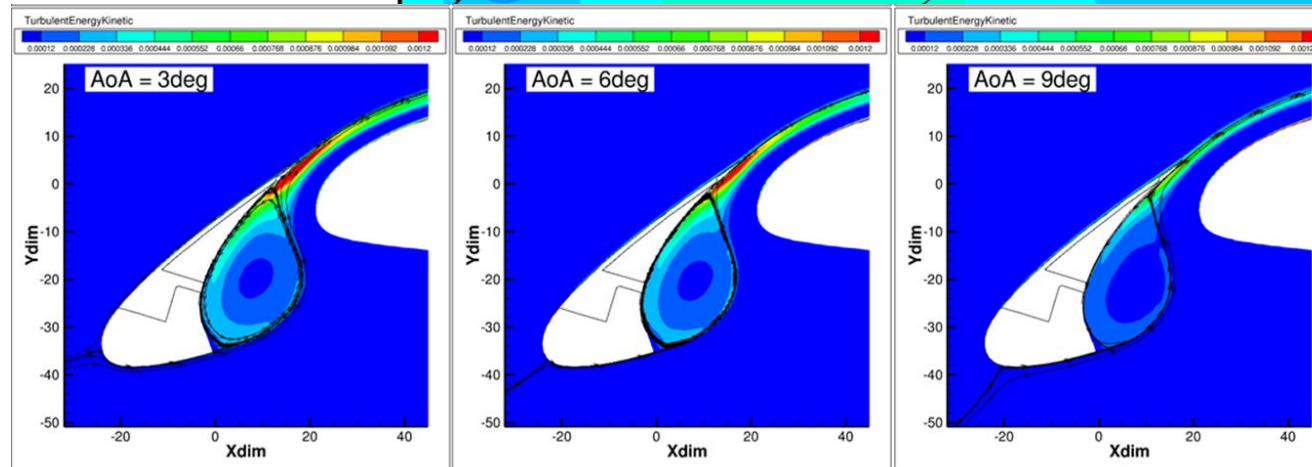
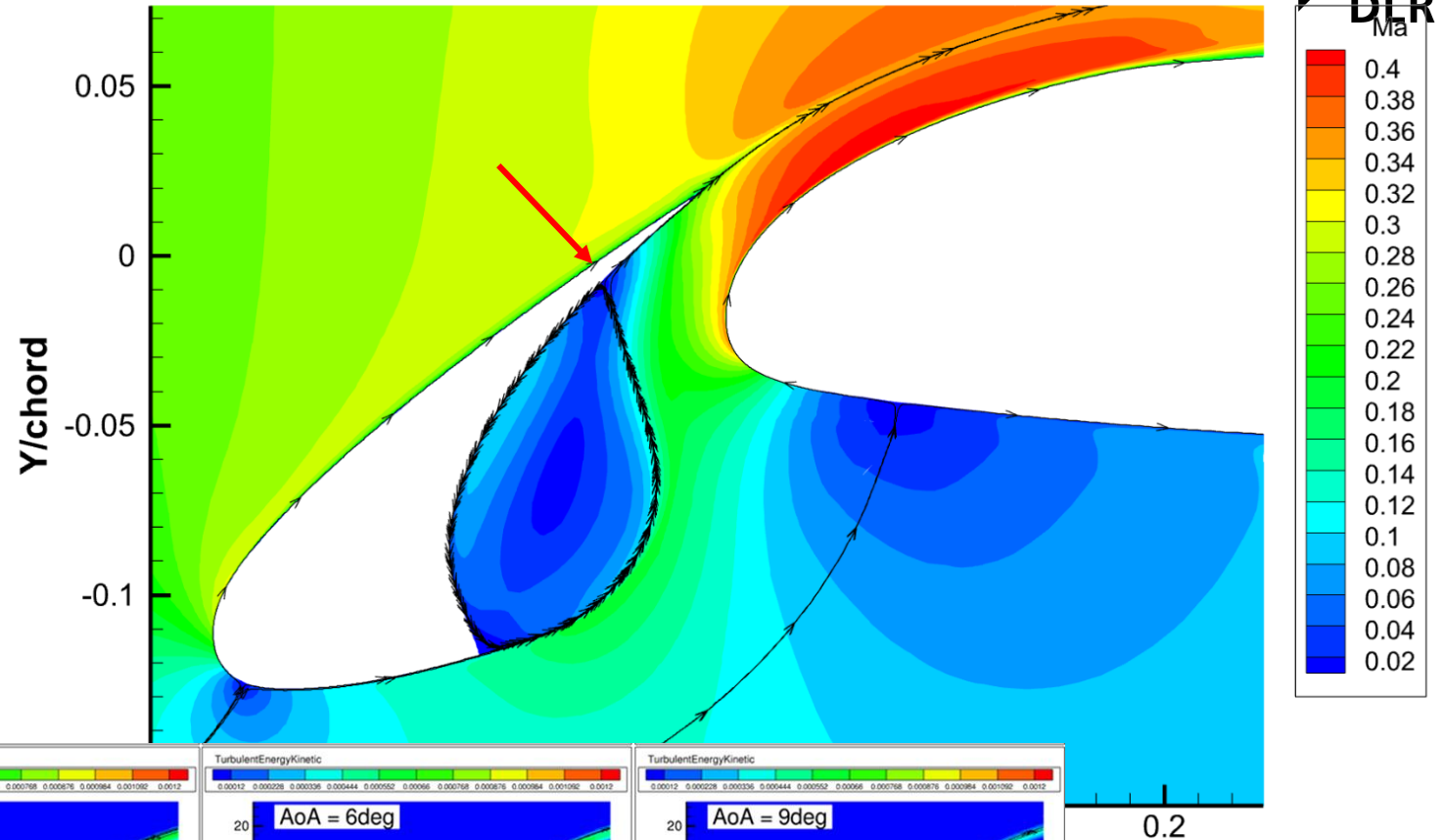
- Extend of porous surface
- Effect of contour discontinuities and tape



1 Assumptions

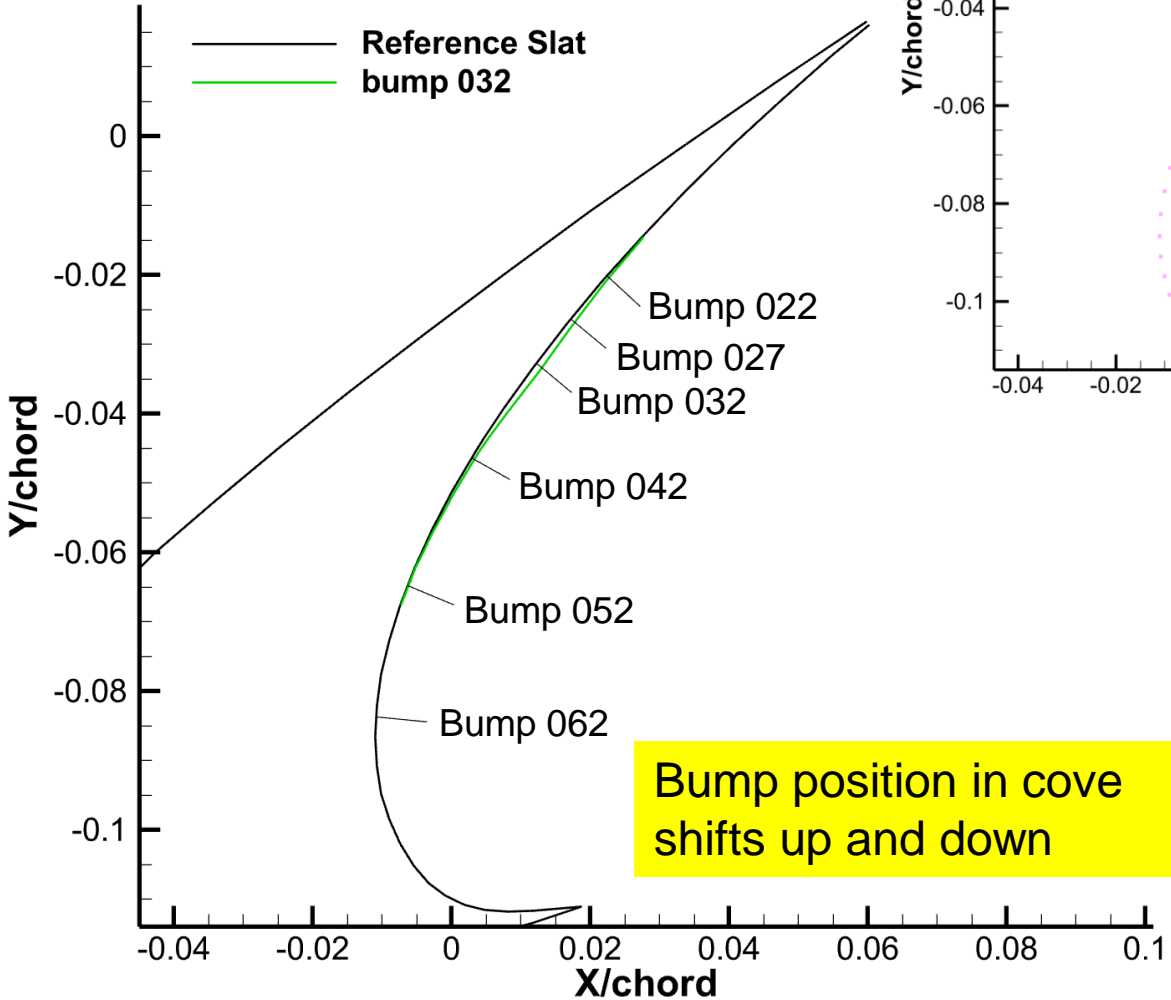
- Place the absorbing treatment at the **stagnation point**
- Account for movement with angle of attack
- Try to simulate different extents by taping regions
- Mixed effects (as shown)

Reference $\alpha_{aero} = 6.15 \text{ deg}$

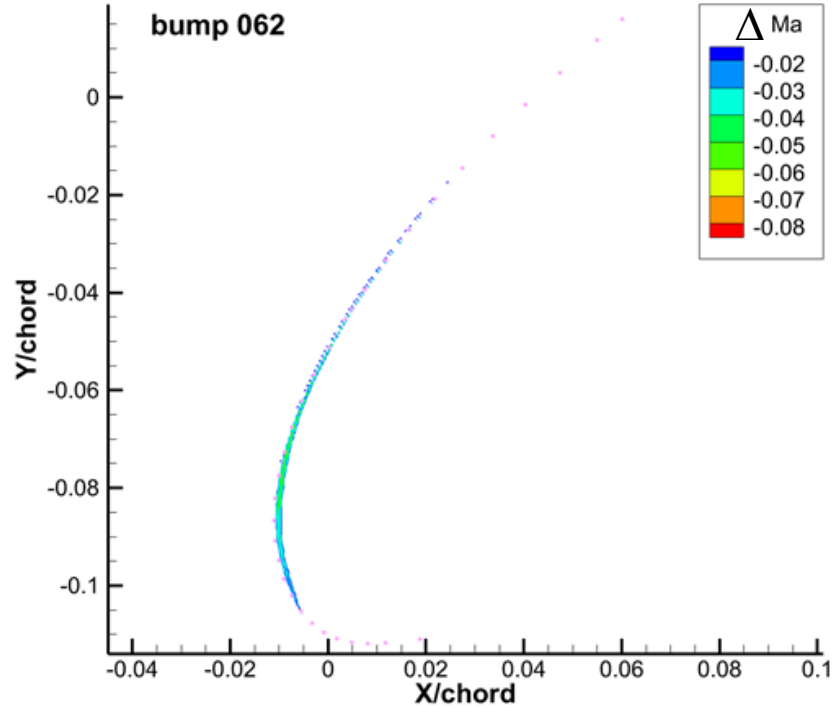
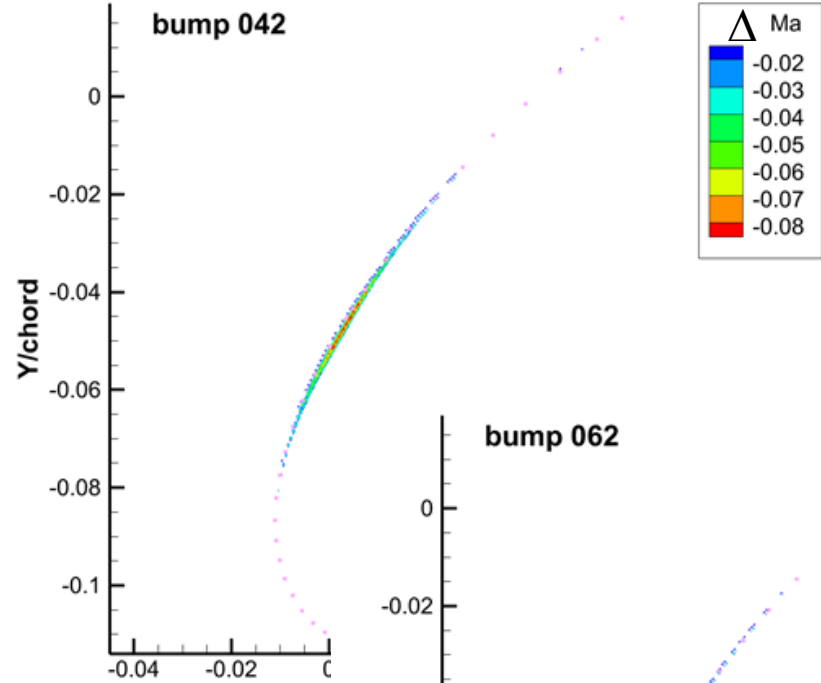
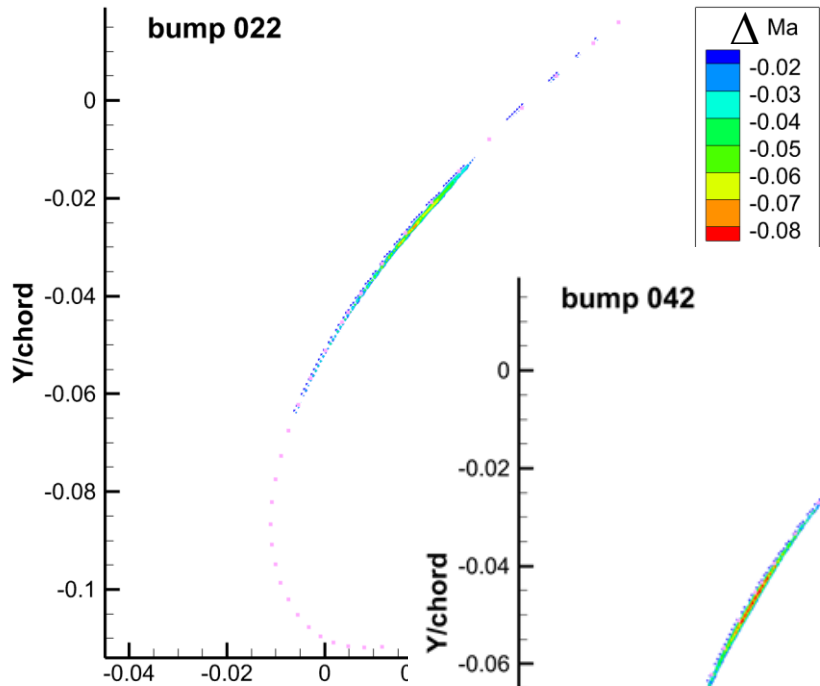


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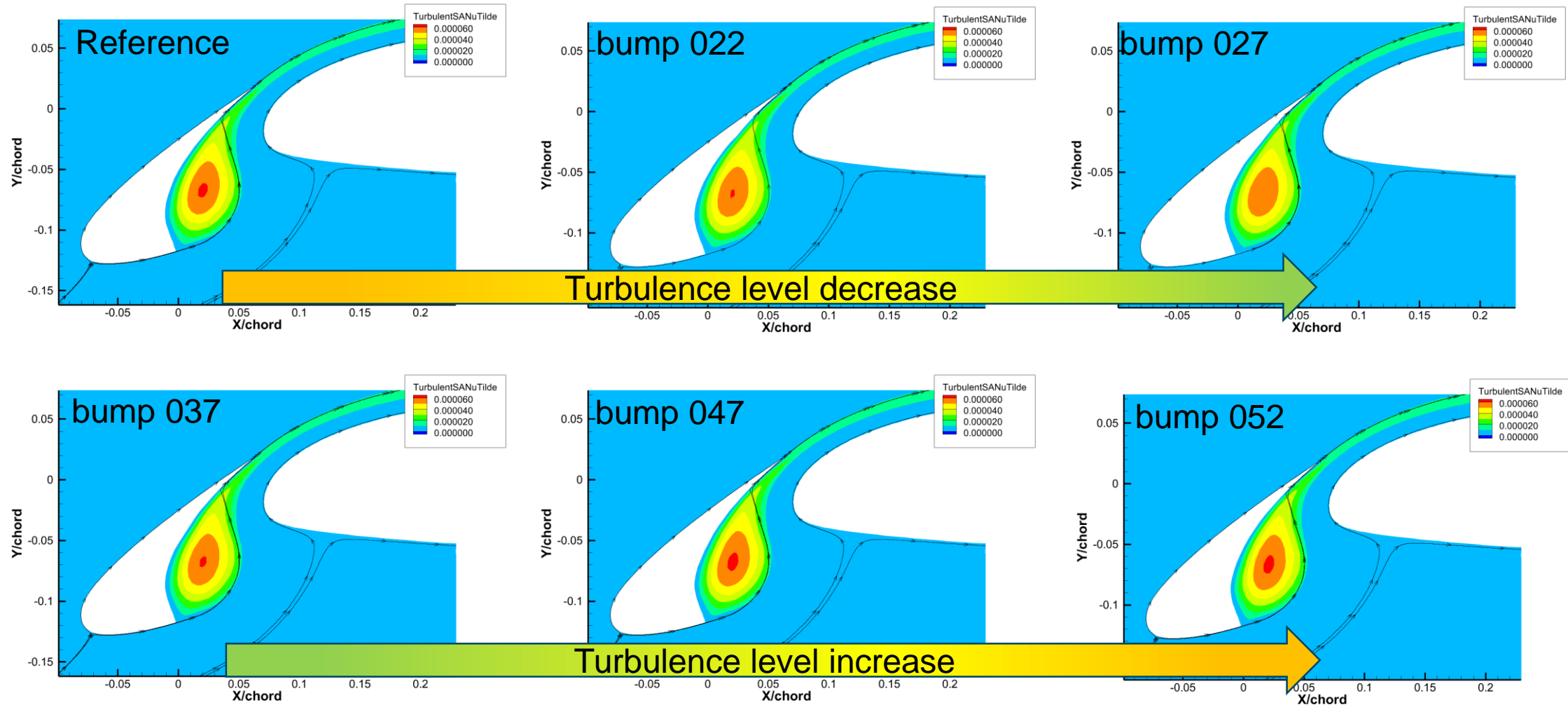
Slat Geometry Modifications



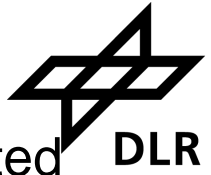
Bump position in cove shifts up and down



1 Turbulence generation - eddy viscosity $\tilde{\nu}$



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- Based on the 1-equation Spalart-Almaras turbulence model the eddy viscosity $\tilde{\nu}$ is selected to represent the turbulent kinetic energy
- The proportionality $\tilde{\nu} \sim \frac{l_t^2}{t} \sim \frac{k^2}{\varepsilon} \sim \frac{k}{\omega}$ links the eddy viscosity with the turbulent kinetic energy k and the dissipation rate ω (*Boussinesq hypothesis*)

- Moving the bump leads to
 - a turbulence level **decrease** between positions 22% and approx. 30%
 - and a turbulence level **increase** for positions greater approx. 40%

inside the cove vortex

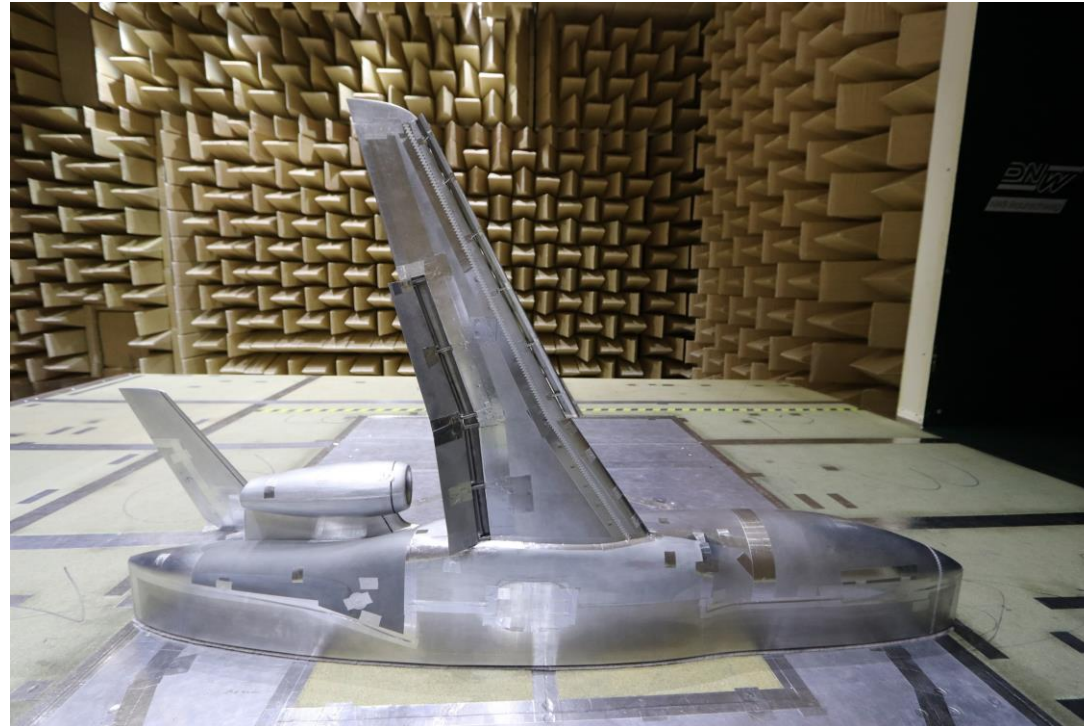
- This turbulence is swallowed out of the vortex, accelerated and passes the slat trailing edge finally leading to higher noise levels

- Numerical methods are required to assess these problems and support the liner design
- Similar problems will occur for impedance jumps from solid to porous surfaces



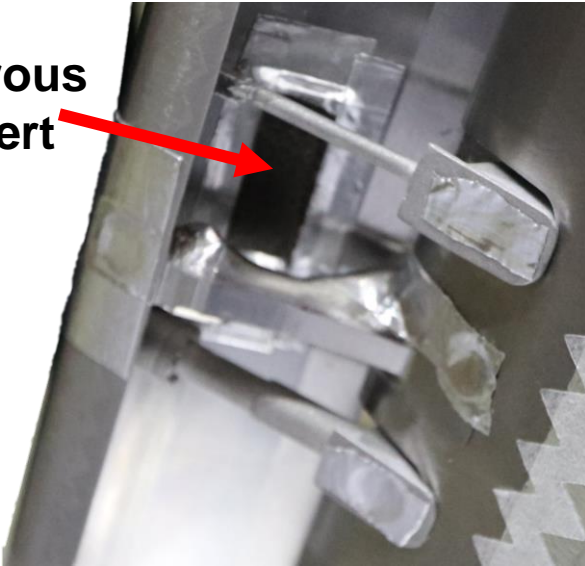
2 Application

- InVentor: Generic Business Jet Aircraft Platform in WP5
- Use the near field absorber to mitigate slat track noise
- DAv wind tunnel model in DNW's low speed acoustic wind tunnel in Braunschweig (DNW-NWB)
- Proof of concept on 3D semi span model (~1.2 m span)

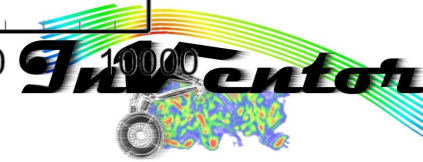
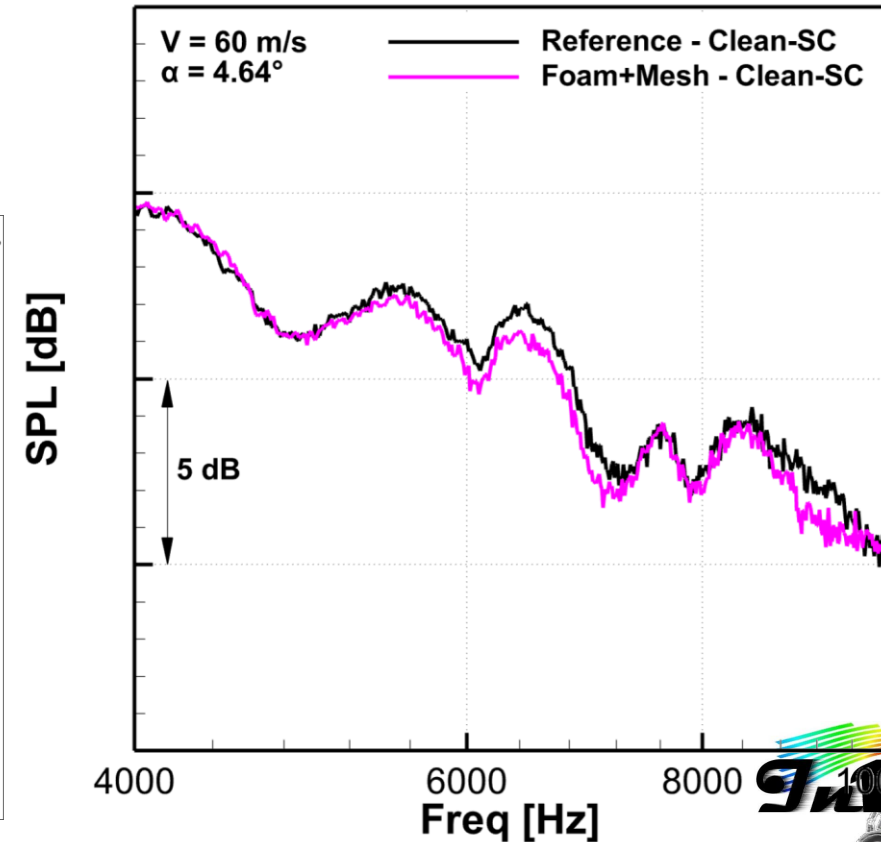
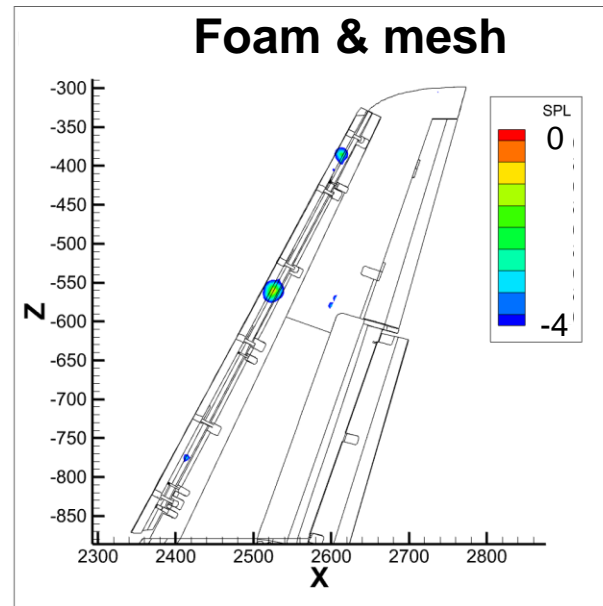
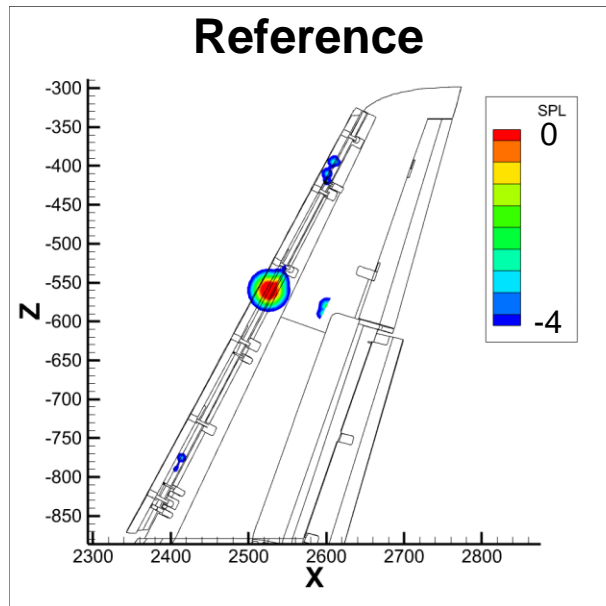


2 Comparison: Reference to Foam & Mesh – Clean-SC

porous insert



- Near field absorber is efficient
- Noise source maps show clear reduction, $f_m = 6.3$ kHz
- Spectral representation confirms source map result over larger frequency domain



Summary & Conclusion



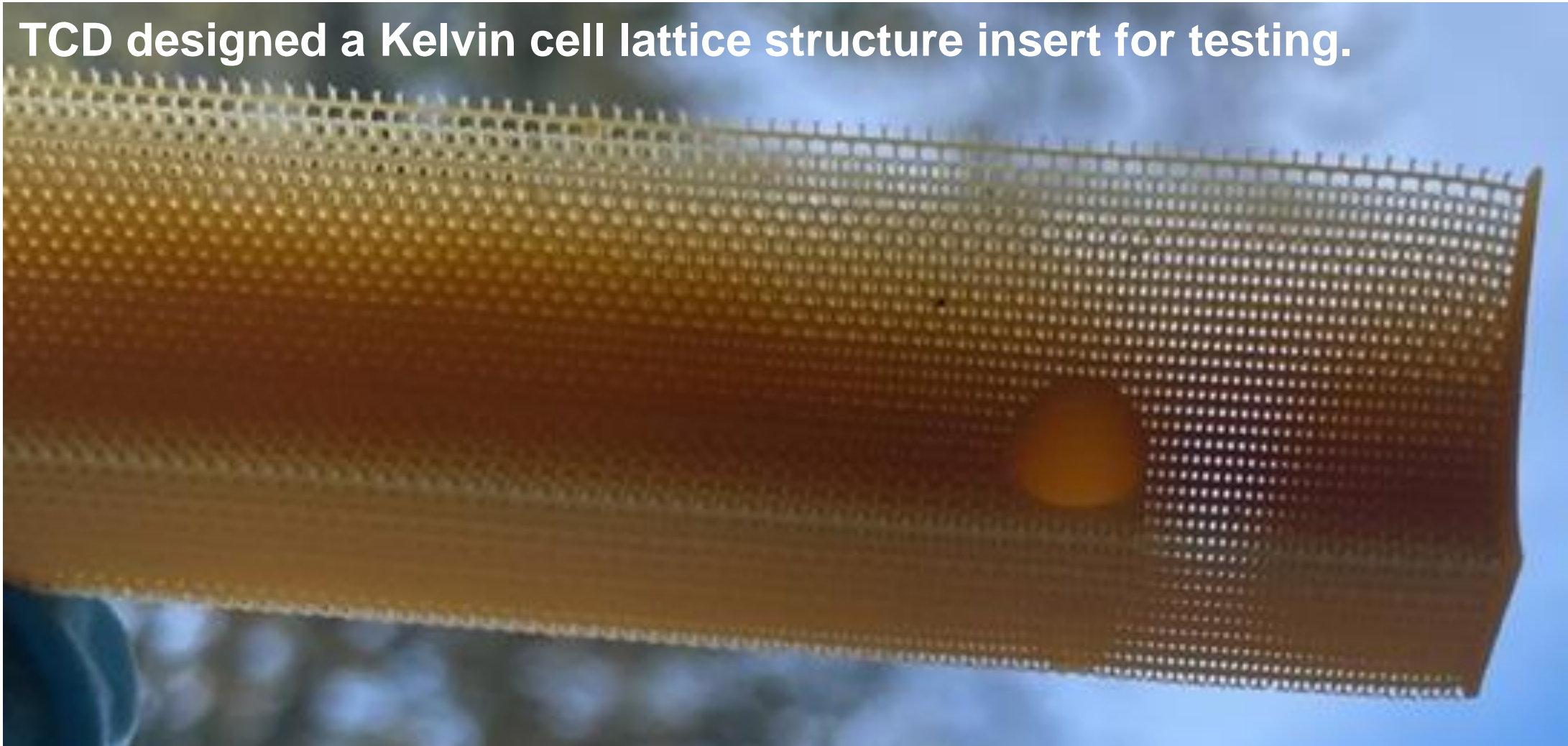
- 2 different types of slat cove liners were investigated in the mainframe of the EU co-funded research project InVentor
- Both liner types show a noise reduction of up to 3 dB when tested on the 2-D F16 high lift system model
 - the near field absorber acts in a limited frequency band
 - the TE insert works on a larger frequency range
 - both liner types suffer from installation issues, demonstrated for the TE insert
- Numerical methods (CFD & CAA) should be used within a combined design process to assess flow properties and noise generation
- Based on the InVentor work also porous media and their installation issues might be assessed in the near future



Thank you for your attention!



TCD designed a Kelvin cell lattice structure insert for testing.



The EU INVENTOR (Innovative Design of Installed Airframe Components For Aircraft Noise Reduction) project receives funding from the European Union's Horizon 2020 Research and Innovation Programme under Grant Agreement No 860538. INVENTOR is a collaborative effort between ONERA (coordinator), Airbus Operations, Dassault Aviation, Safran Landing Systems, DLR, CERFACS, University of Southampton, TC Dublin, NLR, RWTH Aachen, Chalmers University, Von Karman Institute, University of Bristol, TU Delft, Upstream CFD and ERDYN Consultants.

