Ionospheric impact on space-borne GNSS reflectometry: studying satellite and sounding rocket scenarios

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MAPHEUS-14 rocket launched from Esrange Northern Sweden Feb 28, 2024

Outline



Background and Motivation

Prelim. Results from Satellites

Preparation and Simulation of Rocket Obs.

Summary & Outlook

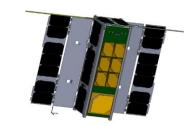


Background and Motivation

Motivation GNSS Reflectometry

A: Low Earth Orbiter

Wickert et al. 2016 Semmling et al. 2016 Moreno et al. 2023



B: Aircraft

Semmling et al. 2014 Moreno et al. 2022



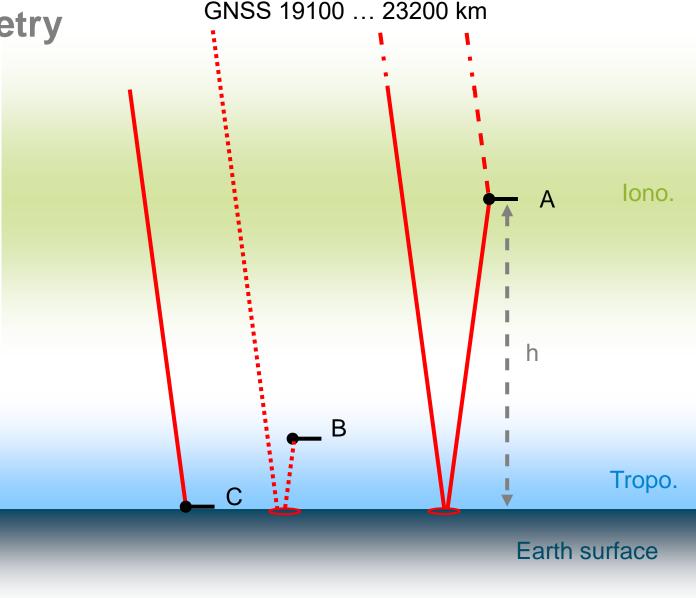
C: Research Vessels

Wang et al. 2019 Semmling et al. 2019, 2022 Semmling et al. 2023



Application

sea surface altimetry sea state estimation sea-ice detection water vapor estimation ionosphere monitoring



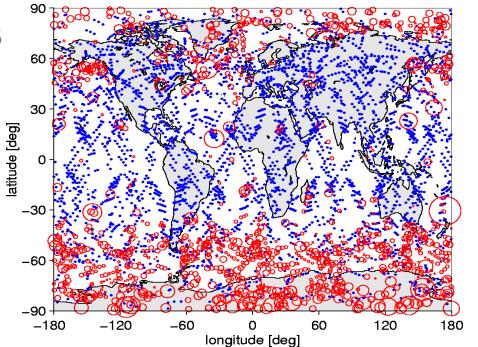
A: e.g. PRETTY, h ~ 540 km

C: e.g. Polarstern, h ~ 25 m

B: e.g. HALO, h ~ 3500 m

Opportunities and Challenges

- GNSS signal are freely available with global coverage
- Coherent signature have been observed in various scenarios also from space
- Main goal: understand & correct ionospheric effects, exploit them for earth observation
- Disturbances to be considered
 - Irregularities on Earth surface (land, ocean roughness)
 - Irregularities in Earth's atmosphere (ionosphere, troposphere)
- Best Opportunities for coherent reflectometry
 - Over sea ice, calm ocean and in coastal areas
 - At grazing elevation angles
- New points in this study
 - Sat. obs. down to grazing elevation -> increased atmo. effect
 - Rocket obs. focusing on E-layer detection

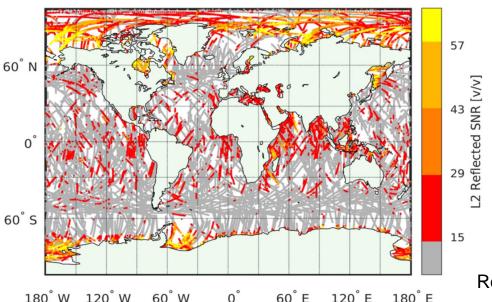




Radio Occultation events recorded with CHAMP mission (one month)

red with reflection blue w/o reflection

Beyerle et al. 2002



Reflectometry events recorded by Spire constell. (four months)

coherent obs. coincide with higher SNR

Roesler et al. 2021

Considerable Factors

Sea Surface

- Roughness (Sea State)
- Penetration (e.g. Sea Ice)
- ...

Atmosphere

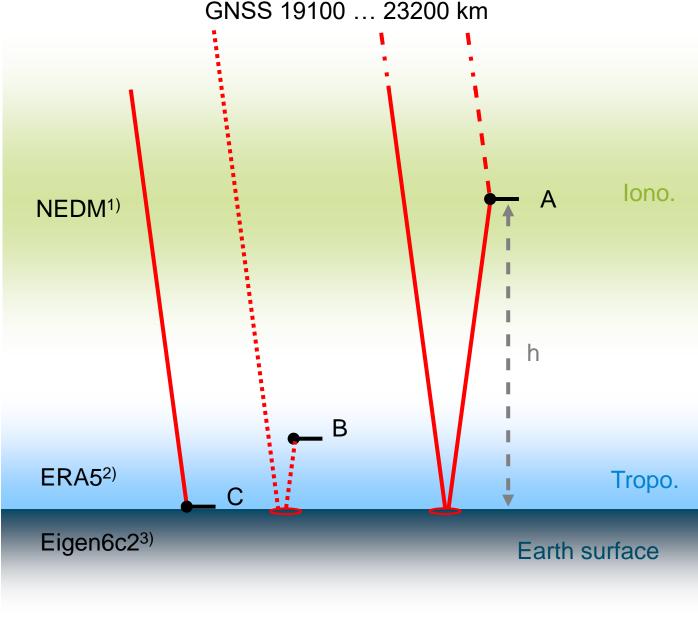
- Refraction (neutral gas and ionosphere)
- Scintillation (Plasma Depletion, Space Weather)
- ...

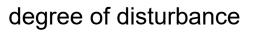
Receiver & Transmitter

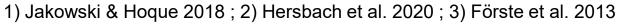
- Position & Attitude uncertainty (of vessel, aircraft or satellite)
- Antenna & Instrumental parameter (e.g. gain pattern)
- **.** . . .







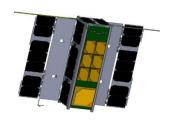




Data for space-borne reflectometry









Mission:

TDS-1

PRETTY

MAPHEUS-15

Platform type:

small sat

cube sat

sounding rocket

Observation alt.:

~ 650 km

~ 560 km 80 ... 240 km

Major field of view:

near-nadir

grazing

grazing

Supported signals:

GPS L1 C/A

GPS L5C & GAL E5

GNSS L1 & L5

Observation area:

Hudson Bay, Canada

Arctic Ocean

Northern Europe

Time period:

Jan 2015

May – July 2024

Nov 2024 (planned)

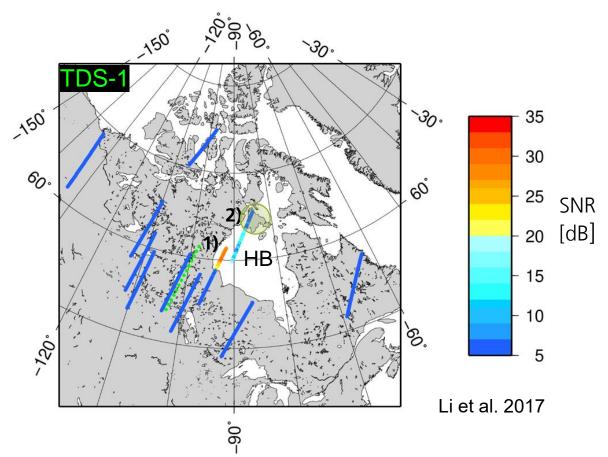


Prelim. Results from Satellites

Altimetric Scenario with TDS-1

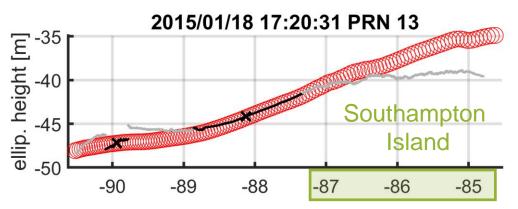


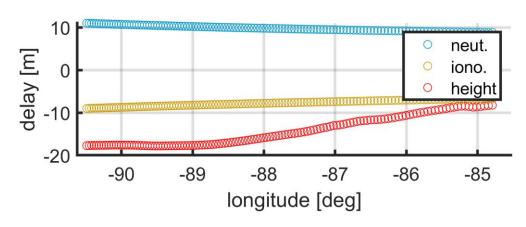
Coherent Phase Tracks of TDS-1 Mission



- Two tracks over Hudson Bay (HB) with rather high SNR selected for analysis, they run over sea ice
- Reflection at spec. Point with high and mid elevations:
 - 1) western HB track ~ 58°
 - 2) eastern HB track ~ 30°

Retrievals for Eastern HB Track 2)



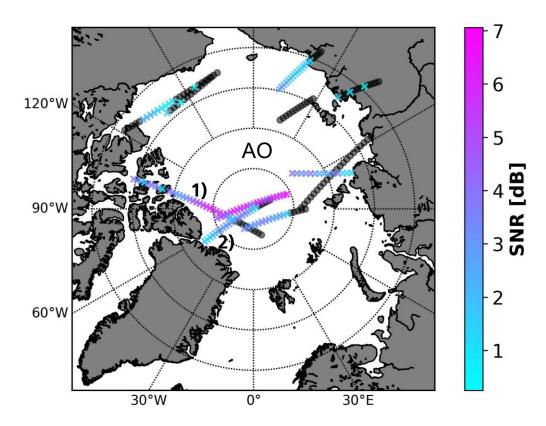


- Surface height retrievals (black marker) follow the sea surface height model (red marker)
- Altimetric scenario because expected delay (based on model) is dominated by sea surface height effect

Atmospheric Scenario with PRETTY

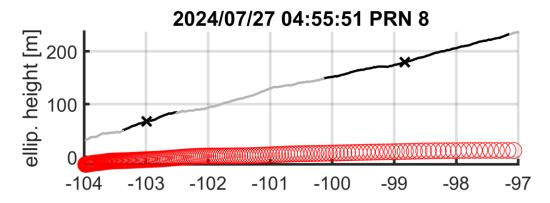


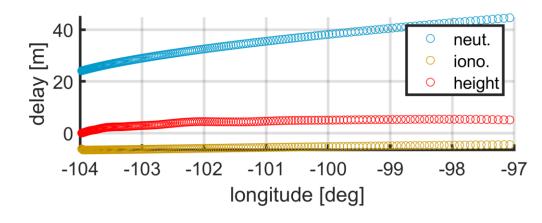
Coherent Phase Tracks of PRETTY Mission



- Nine sea-ice tracks over Arctic Ocean (AO) with reflection signature (rather low SNR), segments without reflection (grey)
- Reflection at spec. Point with grazing elevation:
 - 1) western AO track 0 ... 11°
 - 2) Greenland track 0 ... 10°

Retrievals for Western AO Track 1)

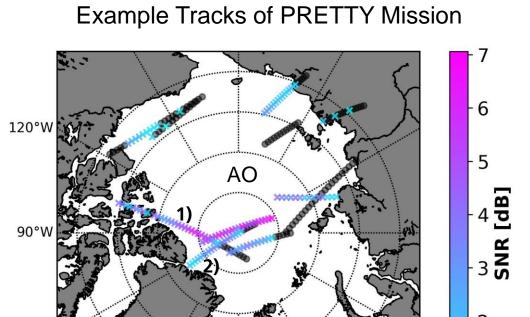


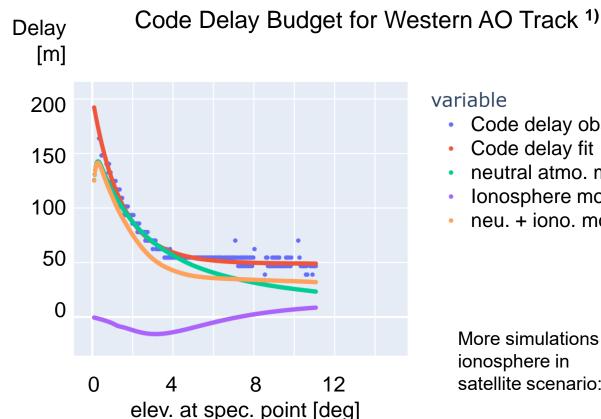


- Surface height retrievals (black marker) follow not the sea surface height model (red marker)
- Atmospheric scenario because expected delay (based on model) is dominated by neutral atmo. effect

Atmospheric Scenario with PRETTY







- variable
 - Code delay obs.
 - Code delay fit
 - neutral atmo, model
 - Ionosphere model
 - neu. + iono. model

More simulations on ionosphere in satellite scenario:

Moreno et al. 2023

Nine sea-ice tracks over Arctic Ocean (AO) with reflection signature (rather low SNR), segments without reflection (grey)

30°E

0°

- Reflection at **spec. Point with grazing elevation**:
 - 1) western AO track 0 ... 11°

30°W

2) Greenland track 0 ... 10°

- In grazing reflection geometry (elev. < 15°) atmo. delay increases exponentially
- retrieved delay and model agree in this trend, bias remains (10m range underestimation)

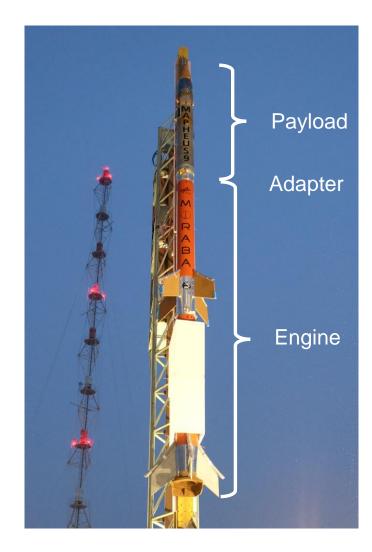
60°W



Preparation and Simulation of Rocket Obs.

GNSS setup on MAPHEUS rocket





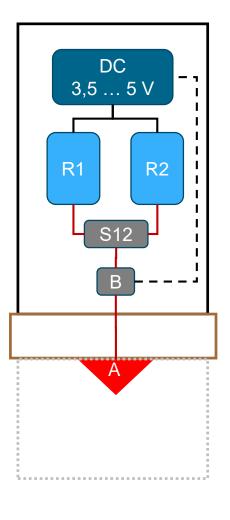
Receiver unit in payload module

GNSS Bitgrabber (redundant)

Payload Adapter

GNSS Antenna

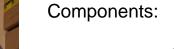
Clear view to Earth once engine is thrown off



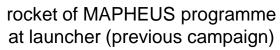
Design & layout for GNSS remote sensing



Payload adapte



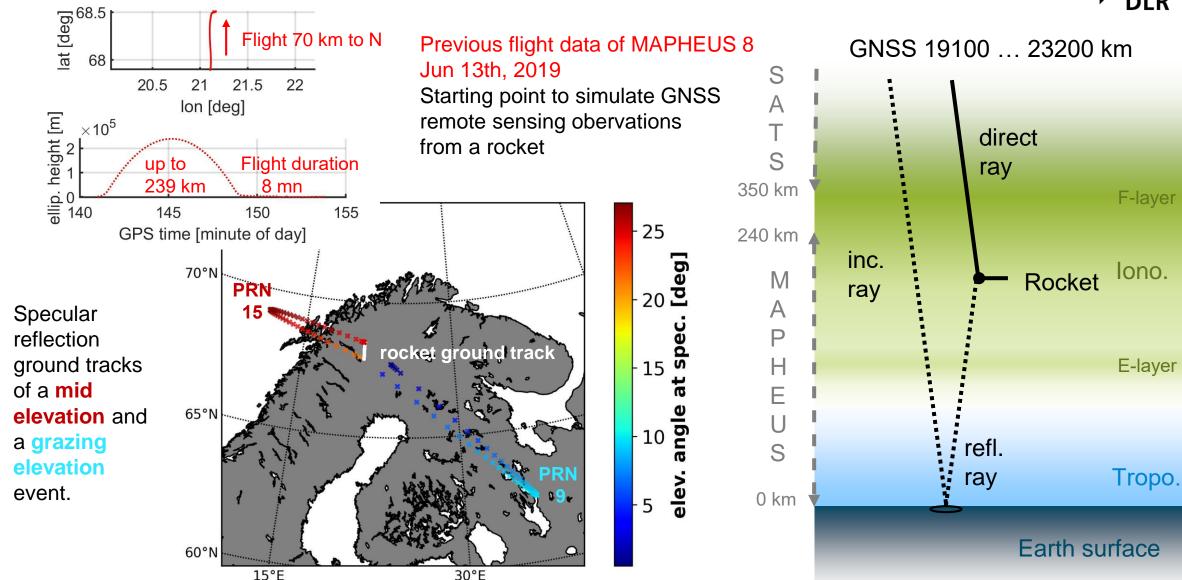
- Bitgrabber (R1,2)Syntony GNSS
- Antenna (A)
 matterwaves
- Bias-tee (B)
- Splitter (S12)
- Powercontrol (DC)



GNSS setup components

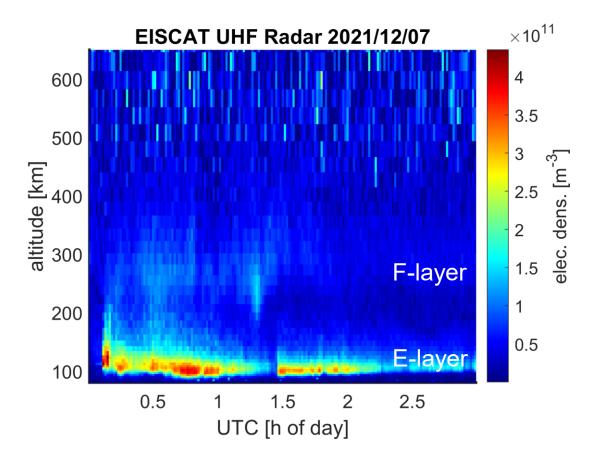
GNSS remote sensing simulation

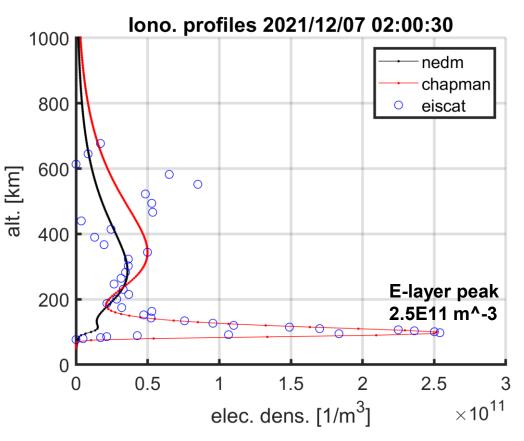




Ionospheric test scenario





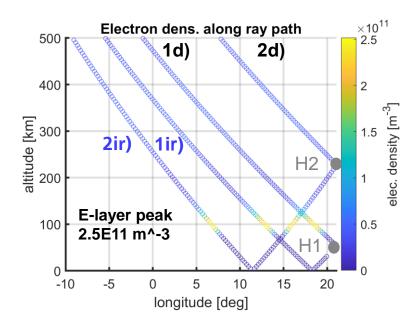


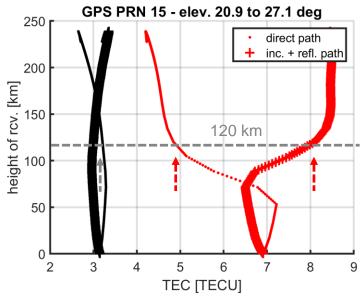
- 3h of elec. density data from EISCAT (European Incoherent Scatter) radar site near Tromsø, Norway
- Polar night period with E-layer dominated ionosphere

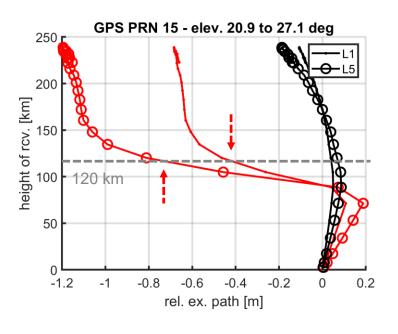
- Chapman layer profile fitted to EISCAT data, dominant E-layer peak and moderate F-layer peak
- Profile from empirical NEDM (Neustrelitz Elec. Density Model) for comparison, E-layer underestimated

Results for mid elevation event









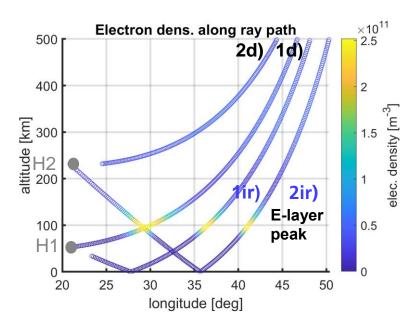
- Rays path for two receiver heights (H1 ~ 50km, H2 ~ 240km)
- For H1: incident-reflected (ir) and direct (d) signals hit E-layer
- For H2: only incident-reflected signal hits E-layer (even twice)

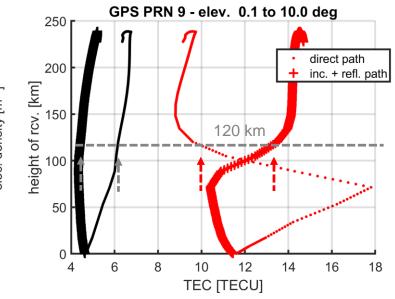
- Comparison of TEC along direct and incident-reflected paths dependent on height of receiver (rocket)
- NEDM scenario (black), E-layerdomin. scenario (red)
- TEC at 120km (above E-layer) differs significantly between scenarios:
 - ~ 5 TECU on ir path
 - ~ 2 TECU on d path

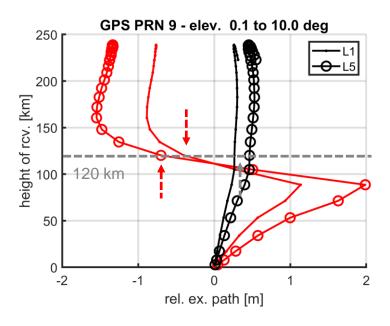
- Comparison of relative ionosphere excess path (between ir and d path)
- NEDM scenario (black) and E-layer domin. scenario (red) for L1 and L5
- ex. path at 120km (above E-layer) differs in dm range between scenarios:
 - ~ 4 dm for L1
 - ~ 8 dm for L5

Results for grazing elevation event









- Rays path for two receiver heights (H1 ~ 50km, H2 ~ 240km)
- For H1: incident-reflected (ir) and direct (d) signals hit E-layer
- For **H2**: **only incident-reflected** signal hits E-layer (even twice)

- Comparison of TEC along direct and incident-reflected paths dependent on height of receiver (rocket)
- NEDM scenario (black), E-layerdomin. scenario (red)
- TEC at 120km (above E-layer) differs significantly between scenarios:
 - ~ 8 TECU on ir path
 - ~ 4 TECU on d path

- Comparison of relative ionosphere excess path (between ir and d path)
- NEDM scenario (black) and E-layer domin. scenario (red) for L1 and L5
- ex. path at 120km (above E-layer) differs in dm to m range between scenarios:
 - ~ 7 dm for L1
 - ~ 10 dm for L5



Summary & Conclusion

Conclusion



- GNSS signals offer opportunities for atmospheric remote sensing incl. GNSS-R
- Coherent reflection tracks often occur over smooth surface (e.g. sea ice)
- Altimetry or atmosphere dominated satellite obs., elev. angle plays major role
- Can we detect E-layer dominated ionosphere with GNSS-R?
- Rocket experiment is currently prepared to answer
- **Delay** resolution in **dm range** (2-4 TECU) is required ...

Acknowledgements

The rocket experiment is conducted within DLR's RESITEK project (RESIliente TEchnologien für den Katastrophenschutz).

The PRETTY data study is partly funded by ESA.

Thank you for your attention

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 J. Space Weather Space Clim.

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