p-anisotropic h-isotropic adaptive Discontinuous Galerkin methods for turbulent flows

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- Introduction to modal anisotropic DG
- **EXERGINAL EXERGINAL EXERGINAL EXERGINAL EXERGINAL EXERGINAL EXERGINAL EXECUTABLE IN A USE**
- **Example 1** Introduction of refinement indicators
- Description of adaptation strategies
- Numerical experiments
- Conclusion and outlook

■ This work has been conducted in the CFD software by ONERA, DLR and Airbus (CODA)

Isotropic adaptive modal Discontinuous Galerkin in CODA

$$
\langle \phi_i, \phi_j \rangle_{\kappa} = \delta_{i,j}
$$

- Full basis is constructed up to a certain maximum degree
- This maximum degree can be chosen individually for each element

Bassi, F., Botti, L., Colombo, A., Di Pietro, D. A., & Tesini, P. (2012). On the flexibility of agglomeration based physical space discontinuous Galerkin discretizations. *Journal of Computational Physics*, *231*(1), 45–65. https://doi.org/10.1016/j.jcp.2011.08.018

Anisotropic adaptive modal Discontinuous Galerkin in CODA

■ Orthonormal hierarchical modal basis in physical space

$$
\langle \phi_i, \phi_j \rangle_{\kappa} = \delta_{i,j}
$$

- Full basis is constructed up to a certain maximum degree
- Any subset of basis functions is allowed to be chosen in each element
	- The constant basis function is always included
	- The linear basis functions are also included as we start adapting from second order

Bassi, F., Botti, L., Colombo, A., Di Pietro, D. A., & Tesini, P. (2012). On the flexibility of agglomeration based physical space discontinuous Galerkin discretizations. *Journal of Computational Physics*, *231*(1), 45–65. https://doi.org/10.1016/j.jcp.2011.08.018

Adaptive mesh refinement using subdivision

- When compared to other techniques (remeshing, ...)
	- + Possible for every element type (→hybrid meshes) except for polyhedrons
	- + Fewer elements are affected
	- + Pre-existing expert knowledge for meshing is reused
	- − Spatial orientation of elements is fixed
	- − There are non-conforming element interfaces
	- − Pyramids are problematic

Residual-based error Indicator

- Introduced by Hartmann, Houston and Leicht in [1, 2]
- **Measure for the discretization error in each element** 1

$$
\eta_{\kappa} = h_{\kappa} \left\| \mathbf{R}_{\kappa}(\mathbf{u}_{h}) \right\|_{L^{2}} + h_{\kappa}^{\overline{2}} \left\| \mathbf{r}_{\partial \kappa}(\mathbf{u}_{h}) \right\|_{L^{2}}
$$

- $\mathbf{R}_{k}(\mathbf{u}_{h})$ is the strong element residual
- $\mathbf{r}_{\partial K}(\mathbf{u}_h)$ is the difference between the numerical and exact fluxes
- Implementation details for the indicator with lifting operators can be found in [Wegener, Hartmann, JCP, submitted]
- Not oriented at a specific goal (e.g. force coefficient) but aims at resolving **all** flow features

[1] Hartmann, R., & Houston, P. (2006). Symmetric Interior Penalty DG Methods for the Compressible Navier–Stokes Equations II: Goal–Oriented A Posteriori Error Estimation. *International Journal of Numerical Analysis & Modeling*, *3*(2), 141–162.

Malte Wegener, Institute of Aerodynamics and Flow Technology, 8.11.2023 [2] Leicht, T., & Hartmann, R. (2010). Error Estimation and Anisotropic Mesh Refinement for 3d Laminar Aerodynamic Flow Simulations. *J. Comput. Phys.*, *229*(19), 7344–7360.

Least-Squares reconstruction (LSQ)

- Find a higher order reconstructed solution, which is indistinguishable from the numerical solution on the nearest neighbor stencil S_{κ} in a weak sense
- As the reconstruction is overconstrained, it is solved in a leastsquares sense
- $\textbf{u}_k + \textbf{u}_R$, $\boldsymbol{\phi}_i$ $\rangle_{\kappa'} = \langle \textbf{u}_h, \boldsymbol{\phi}_i \rangle_{\kappa'}$ $\forall \boldsymbol{\phi}_i \in \boldsymbol{U}^{\rho}_{\kappa'}$ $\sum_{\kappa'}^p \kappa' \in S_{\kappa}$
- As the basis is orthonormal, the coefficients of u_R are the energy contained in the **candidate functions**

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Adaptation strategies

p-isotropic h-isotropic

- Each element is either flagged for prefinement or h-refinement according to the **exponential decay indicator**
- **Elements are sorted by their indicator** value
- Elements are refined until the requested increase in DoF is reached

Adaptation strategies

p-anisotropic h-isotropic split

- Each element is either flagged for prefinement or h-refinement according to the **exponential decay indicator**
- p-refinement is chosen according to the LSQ indicator
- **Elements are refined until the** requested increase in DoF is reached

NUMERICAL RESULTS

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- **E** Laminar flow at
	- Ma: 0.3
	- Re: 4000
	- \blacksquare α: 12.5°
- **Examediate Structured hexahedral grid with** degenerate hexahedrons subdivided into regular element types
- Has already been part of the European ADIGMA project, thus reliable reference results are available

- **hp-adaptation outperforms pure** h-adaptation
- **In early stages p-anisotropic** outperforms p-isotropic
- **p-anisotropic adaptation stalls**

Adaptation strategies

p-anisotropic h-isotropic combined

- Each element is either flagged for prefinement or h-refinement according to the **exponential decay indicator**
- Indicator is split between different candidate functions according to their energy
- **Elements are refined until the** requested increase in DoF is reached

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- New refinement strategy keeps benefits from early stages
- Can keep the advantage and converges to the correct value

Juskowski airfoil

- Part of HiFiCFD 2024
- RANS-SA negative model with the QCR2000 modification
	- \blacksquare Ma: 0.15
	- Re: 6,000,000
	- \blacksquare α : 0°
- Comparison to the reference drag from the SANS solver
- hp-adaptive simulations outperform h-adaptive simulations in terms of accuracy per DoF
- **•** p-anisotropic adaptation outperforms p-isotropic adaptation

2D Multi-Element airfoil

- HLPW4, Case 1
	- 3 element high-lift airfoil
- RANS-SA negative model
	- Ma: 0.2
	- Re: 5,000,000
	- \blacksquare α: 16°
- **. Initial mesh has 6645 elements**
- Each iteration increases DoF by a factor of 1.2 each iteration

2D Multi-Element airfoil

- **hp-adaptive simulations** outperform h-adaptive simulations in terms of accuracy per DoF
- p-anisotropic adaptation outperforms p-isotropic adaptation after the 3rd adaptation
- All methods converge to the reference solution

2D Multi-Element airfoil

- **hp-adaptive simulations** outperform h-adaptive simulations in terms of accuracy per DoF
- **p-anisotropic adaptation** outperforms p-isotropic adaptation after the 3rd adaptation
- All methods converge to the reference solution

Conclusion and Outlook

Conclusion

- The combined formulation for refinement is necessary to take advantage of p-anisotropic refinement
- The p-anisotropic refinement strategy outperforms the p-isotropic refinement

Outlook

- Application to a more complex 3D flow
- Compare the effectiveness of a LSQ driven anisotropic adaptation to an adjoint based anisotropic adaptation

- This work has been conducted and implemented in the CFD software by **ONERA**, **DLR** and **Airbus** (CODA). **CODA** is the computational fluid dynamics (CFD) software being developed as part of a collaboration between the French Aerospace Lab ONERA, the German Aerospace Center (DLR), Airbus, and their European research partners. CODA is jointly owned by ONERA, DLR and Airbus.
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