

RESULTS OF THE DLR PROJECT GROUP “MACHINE LEARNING AND QUANTUM COMPUTING – DIGITALIZATION OF AIRCRAFT DEVELOPMENT 2.0”

Deutscher Luft- und Raumfahrtkongress 2024

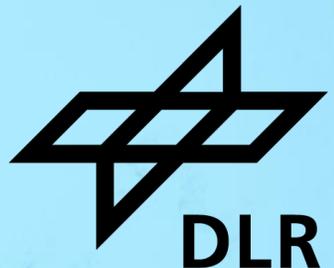
30. September – 02. Oktober 2024 – Hamburg

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Institute of Aerodynamics and Flow Technology

Braunschweig

Department C²A²S²E



Outline



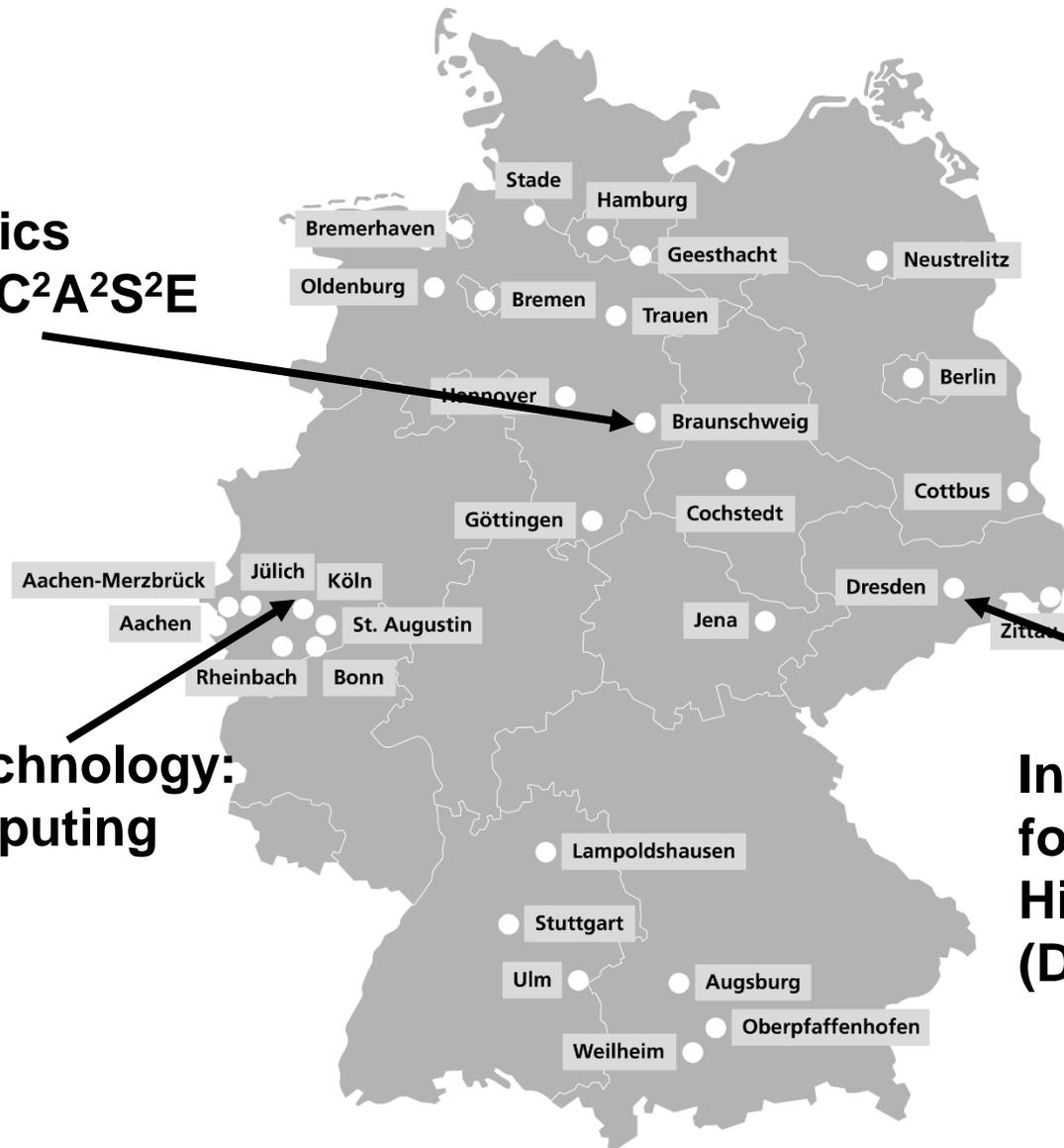
- Overview of the project
- Presentation of selected topics and results
- Summary and outlook

Participating DLR Institutes and Departments



**Institute of Aerodynamics
and Flow Technology: C²A²S²E
(Braunschweig)**

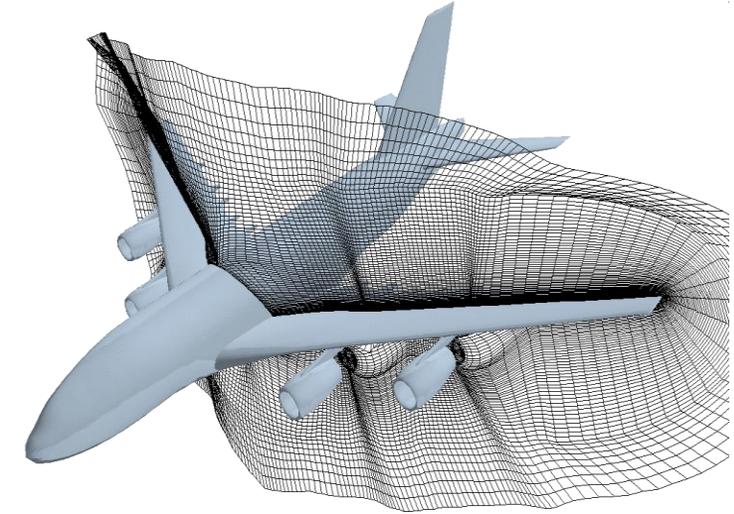
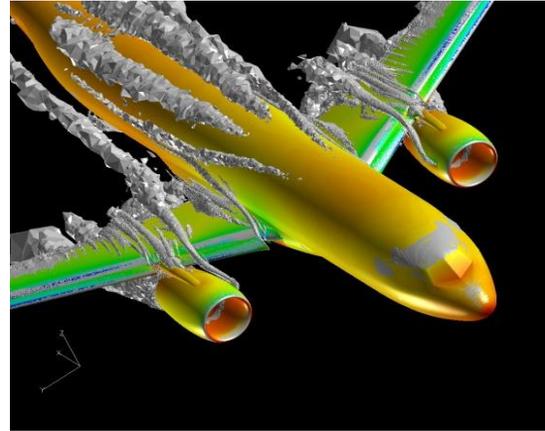
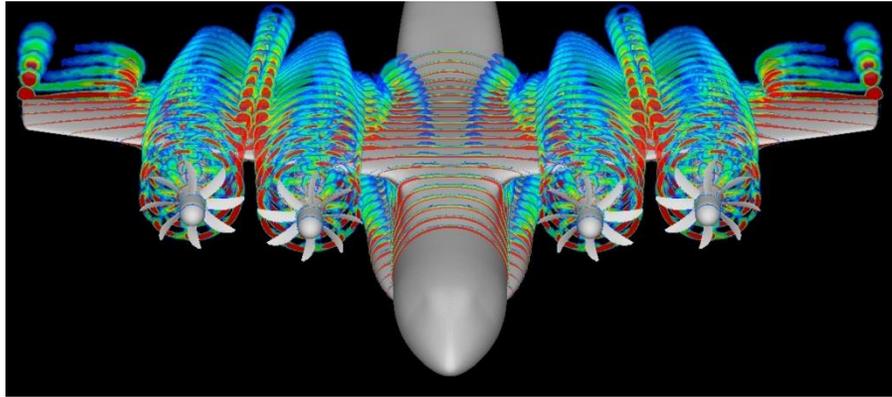
**Duration of the project:
01.04.2021 – 31.03.2024
(Final meeting 06.03.2024)**



**Institute of Software Technology:
High-Performance Computing
(Köln)**

**Institute of Software Methods
for Product Virtualization:
High Performance Computing
(Dresden)**

Computational Fluid Dynamics



- Computational Fluid Dynamics (CFD) is an important, established tool in the aircraft industry

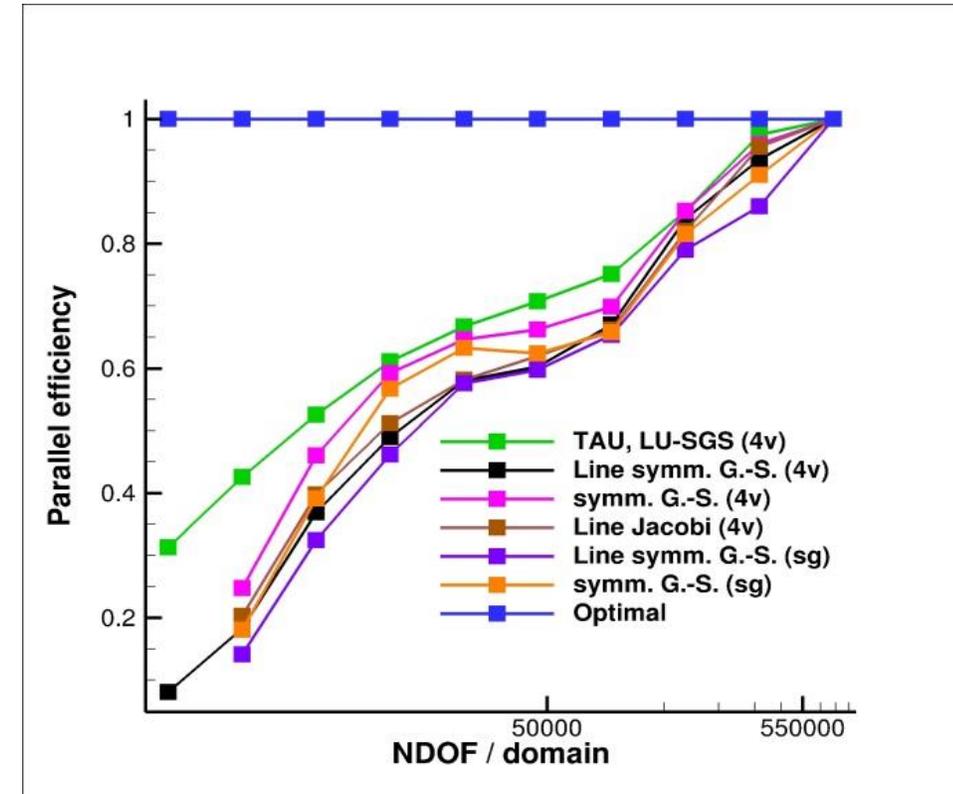
In order to significantly increase further capabilities and prediction accuracy, more and more complex models and larger computational meshes are required.

In particular **scale resolving (e.g. large eddy simulation, direct numerical simulation) and **unsteady computations** are **unfeasible** for industrial test cases on classical computer hardware, **even under very optimistic assumptions.****

How much acceleration is still possible with classic methods?

Increasing complexity of the algorithms

- Multigrid (linear and nonlinear)
- Multistage implicit Runge-Kutta schemes
- Combination of linear solvers:
 - GmRes
 - Jacobi and Gauss-Seidel Iterations
 - LU-Solvers
- Truncation criteria for all the components
-



The developed algorithms show a severe loss of parallel efficiency

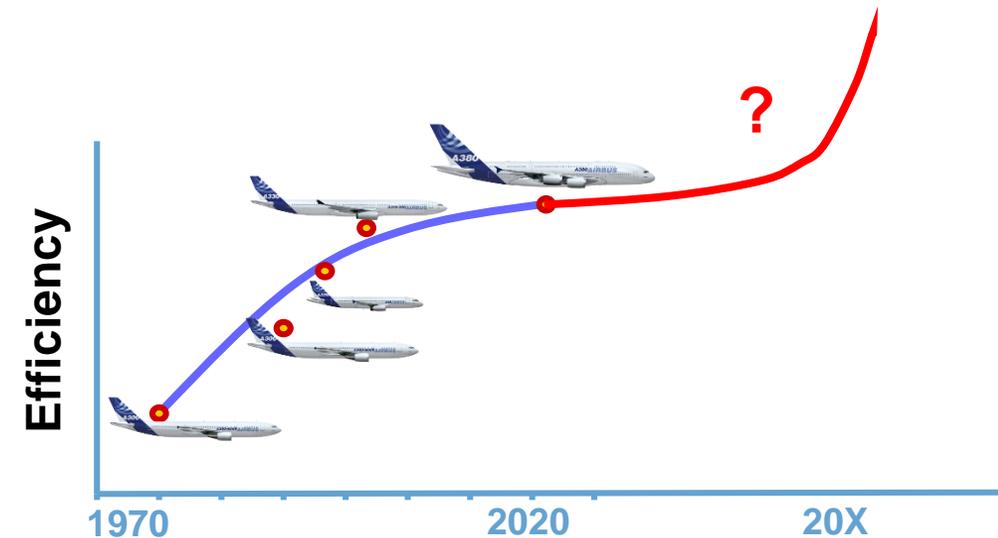
→ It is becoming more and more difficult to achieve a significant acceleration with the classic approaches

Motivation

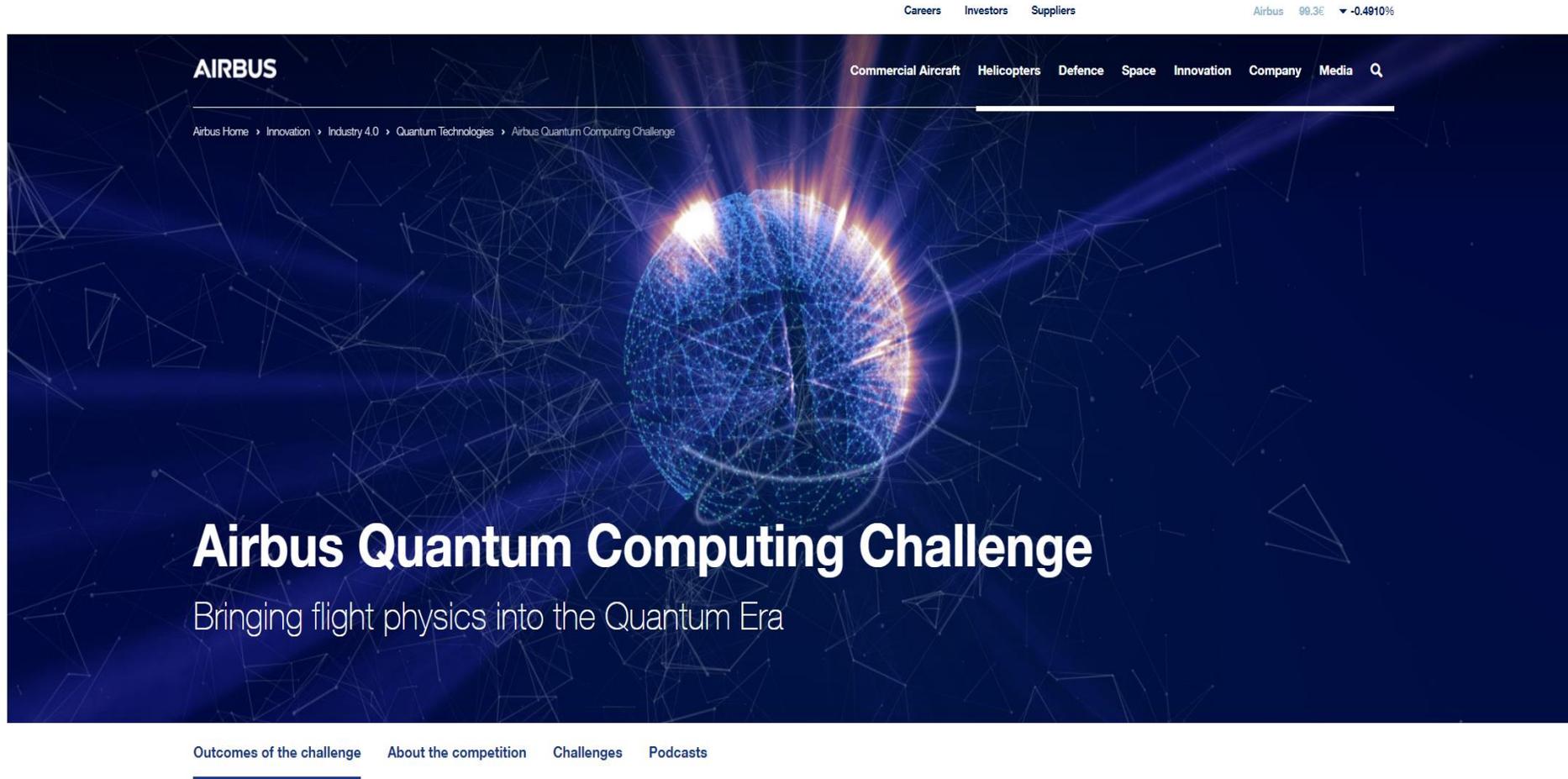
Industry requirements:

- Real-time simulation of full aircraft
- Optimization and certification before first flight
- Scale resolving simulations

Significantly more efficient aircraft are required to achieve climate goals



**Can quantum computers be used to design future aircraft?
Is quantum computing a game changer for the aerospace industry?**



Toulouse, 10 December 2020

The Airbus Quantum Computing challenge has ended and we are proud to announce the 2020 winner : the team Machine Learning Reply.



Challenges:

- 1) Aircraft Climb Optimization
- 2) Computational Fluid Dynamics on Quantum Computers
- 3) Quantum Neural Networks for Solving Partial Differential Equations
- 4) Wingbox Design Optimisation
- 5) Aircraft Loading Optimisation

Direct reference
to the project

→ A new Quantum Computing challenge is raised in 2024 by BMW and Airbus

Goals and Perspective



Dealing with scientific and technical issues in order to **determine the potential of quantum computing and machine learning in aviation**

Possibility to permanently establish participating institutions as a **partner, supporter and pioneer** of the German and European aviation industry for questions regarding the use of quantum computing in the context of digitalization

Establish close relationships between aviation industry and DLR in questions regarding machine learning and quantum computing

Short overview of some of the topics and a selection of results

- Correcting the discretization error of coarse grid CFD simulations with Machine Learning
- Physics-Informed neural networks for solving compressible flow equations
- Quantum machine learning for partial differential equations

A composite image showing a white commercial airplane flying through a blue sky with white clouds. The airplane is overlaid with a colorful computational fluid dynamics (CFD) simulation. The simulation uses a color gradient from blue to red to represent flow velocity or pressure. The wings and fuselage are primarily blue and green, while the wake behind the aircraft is shown as a complex, swirling structure of red and orange lines, indicating high-velocity flow regions. The text "CORRECTING THE DISCRETIZATION ERROR OF COARSE GRID CFD SIMULATIONS WITH MACHINE LEARNING" is overlaid in white on a dark blue semi-transparent banner at the bottom of the image.

CORRECTING THE DISCRETIZATION ERROR OF COARSE GRID CFD SIMULATIONS WITH MACHINE LEARNING

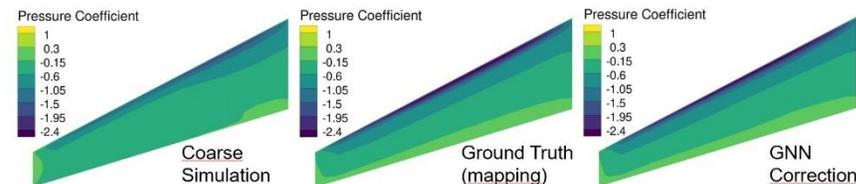
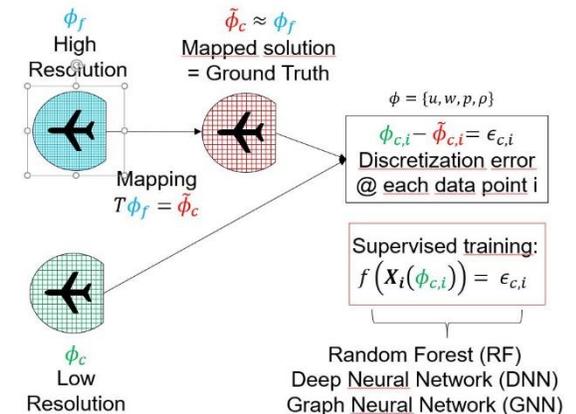
- Take **fast but comparatively inaccurate** CFD simulations, such as:
 - Coarse grid finite-volume method (FVM) solution
 - Low-order discontinuous Galerkin (DG) solution

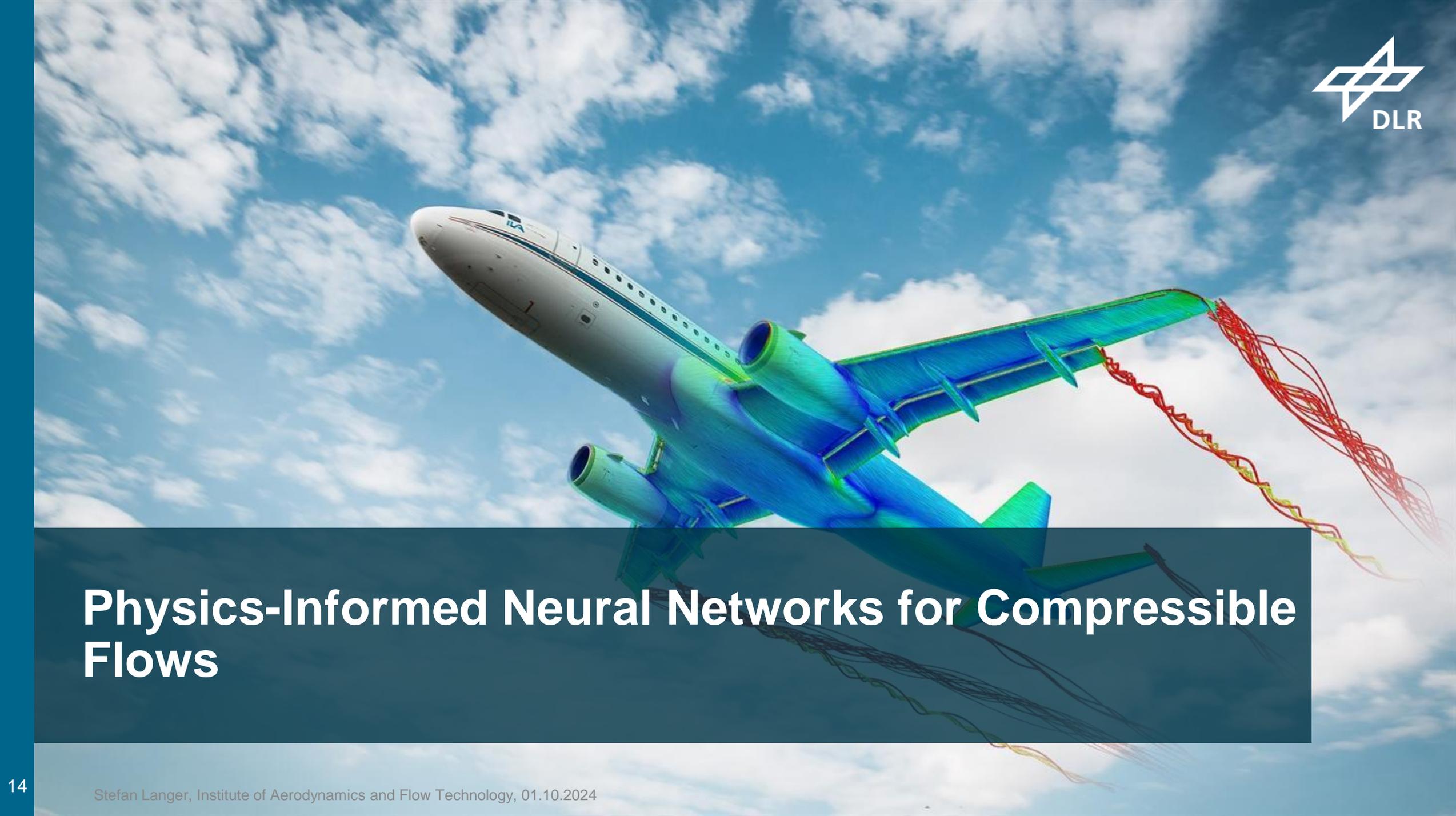
→ And then, try to improve accuracy with data-driven methods

Investigated topics

1. Post-Processing Correction of Coarse Grid **FVM** Solutions
2. Post-Processing Correction of Lower-Order **DG** Solutions
3. Correcting **Unsteady** DG Simulations
4. **External Source Term** Correction and **Reinforcement Learning**

→ **Talk of Anna Kiener**



A photograph of a white commercial airplane flying through a blue sky with scattered white clouds. The aircraft is overlaid with a 3D flow visualization. The flow field is color-coded, with blue and green representing smoother flow regions and red and yellow representing areas of high turbulence or shock waves. The flow lines are visible around the wings, engines, and tail, showing the complex aerodynamic environment.

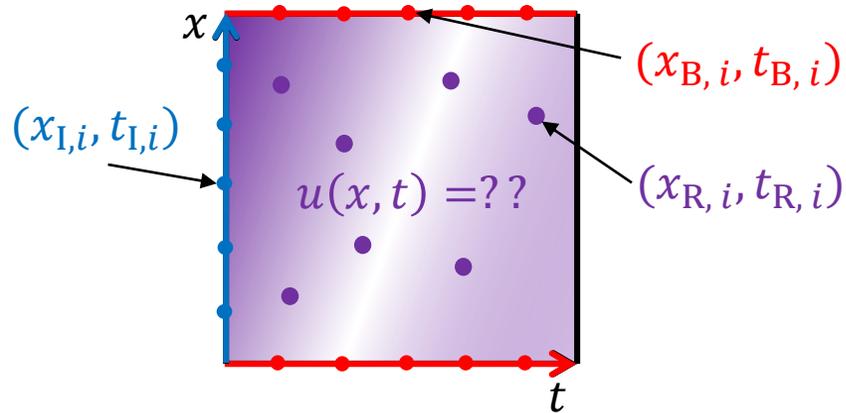
Physics-Informed Neural Networks for Compressible Flows

Solving Differential Equations with Physics-Informed Neural Networks

boundary value problem

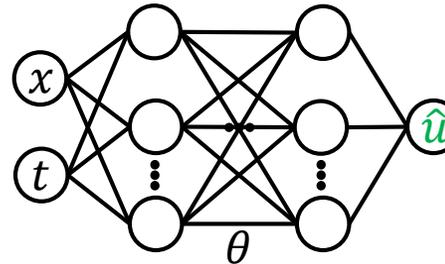
$$\mathcal{D}(u(x, t), x, t) = 0, \text{ in } \Omega \times [0, T]$$

$$\mathcal{B}(u(x, t), x, t) = 0, \text{ on } \partial\Omega \times [0, T]$$

$$\mathcal{J}(u(x, 0), x) = 0, \text{ in } \Omega$$


neural network as global ansatz function for solution

$$u(x, t) \approx \hat{u}_\theta(x, t)$$



calculate partial derivatives with automatic differentiation

$$\partial_x \hat{u}(x_{R,i}, t_{R,i}),$$

$$\partial_t \hat{u}(x_{R,i}, t_{R,i}),$$

$$\partial_x \hat{u}(x_{B,i}, t_{B,i}),$$

$$\dots$$

find optimal parameters for network
 $\hat{=}$
 find solution

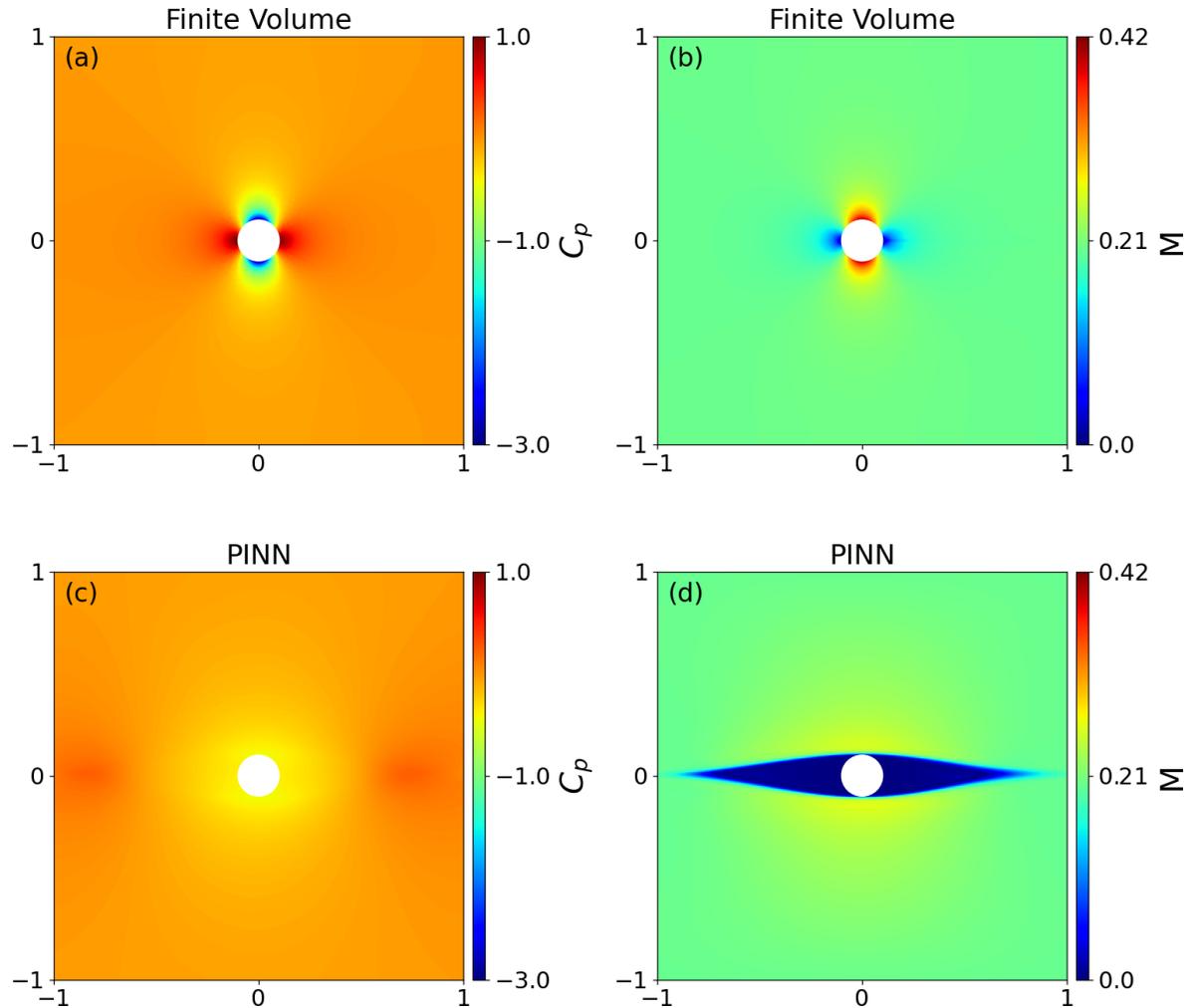
$$\arg_\theta (\min(\mathcal{L}(\hat{u}_\theta)))$$

training

$$\mathcal{L} = \iint_{\Omega \times (0, T)} \mathcal{D}(\hat{u})^2 dx dt + \iint_{\partial\Omega \times (0, T)} \mathcal{B}(\hat{u})^2 ds dt + \int_{\Omega} \mathcal{J}(\hat{u})^2 dx$$

$$\approx \sum_{i=0}^n \mathcal{D}(\hat{u}(x_{R,i}, t_{R,i}))^2 + \sum_{i=0}^n \mathcal{B}(\hat{u}(x_{B,i}, t_{B,i}))^2 + \sum_{i=0}^n \mathcal{J}(\hat{u}(x_{L,i}, t_{L,i}))^2$$

Unphysical Results with Naive PINN Implementation



**Naive/direct PINN
implementation yields
unphysical results!**

(a)-(b) reference finite volume solution
(c)-(d) naive PINN results

How to avoid unphysical results?

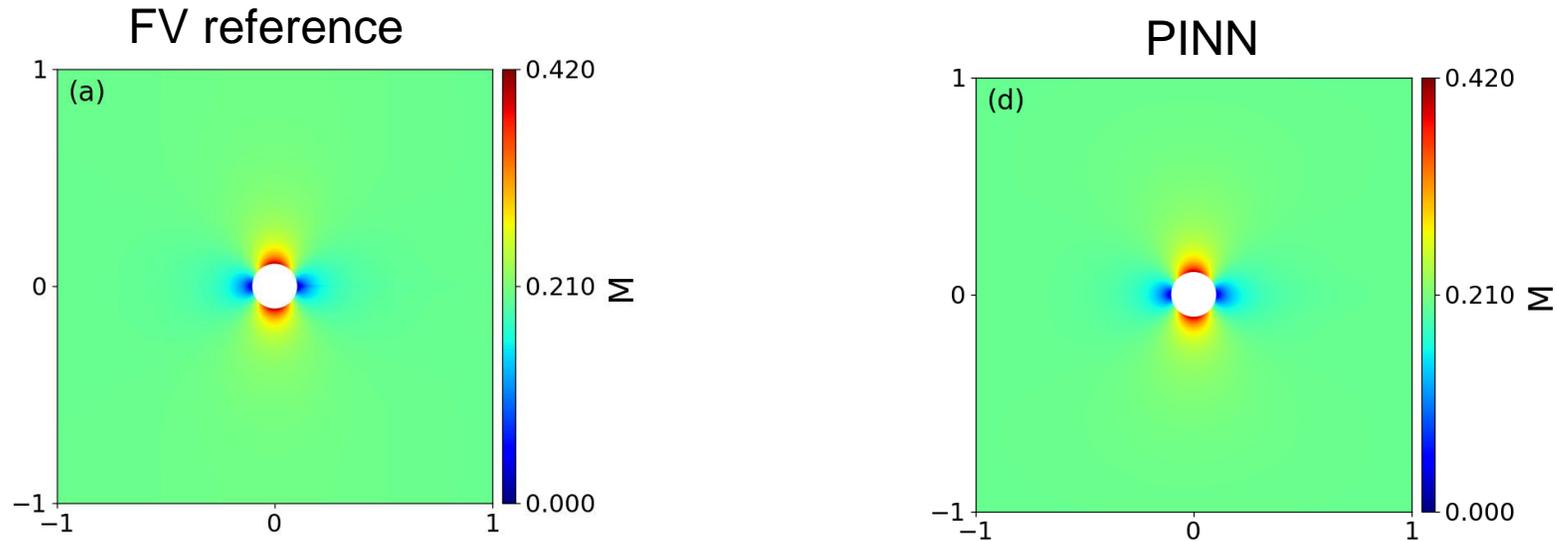
- From the theory of finite volume methods we know that **artificial dissipation** is necessary to stabilize compressible flow simulations
- For PINNs we can add a **dissipative/viscous regularization term** to the PDE
- Similarly to a simplistic scalar valued artificial viscosity in finite volume methods the dissipation is locally scaled based on the spectral radius of the Euler equations
$$\eta = \nu \cdot (a + |\mathbf{q}|)$$
- First, we determine initial ν empirically and reduce during training

2D Compressible Euler Equations with artificial dissipation:

$$\frac{\partial \mathbf{W}}{\partial t} + \frac{\partial \mathbf{F}_x}{\partial x} + \frac{\partial \mathbf{F}_y}{\partial y} = \eta \nabla^2 \mathbf{W}$$

$$\mathbf{W} = \begin{pmatrix} \rho \\ \rho u \\ \rho v \\ \rho E \end{pmatrix}, \mathbf{F}_x = \begin{pmatrix} \rho u \\ \rho u^2 + p \\ \rho uv \\ \rho Eu \end{pmatrix}, \mathbf{F}_y = \begin{pmatrix} \rho v \\ \rho uv \\ \rho v^2 + p \\ \rho Ev \end{pmatrix},$$

Initial Results with Global Viscosity

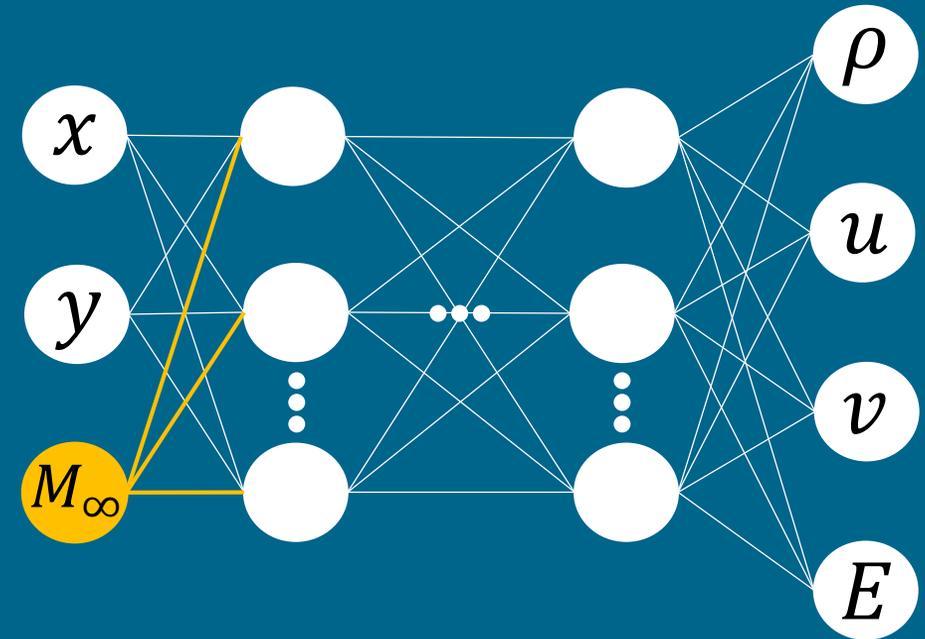


Unphysical results can be avoided!

- Accuracy is still relatively low (e.g. slight asymmetries in solution were observed)
 - Training time is of the order of multiple hours
 - Once trained, evaluation is possible in $< 1s$
- What are possible advantages compared to classical solvers besides QC ?

Extension with Parametric Boundary Conditions

- For parametric problems the parameter can be added to the input space of the NN
- A single PINN is trained in a continuous parameter space



→ Talk of Simon Wassing

A composite image showing a white commercial airplane in flight against a blue sky with white clouds. The airplane is overlaid with a colorful aerodynamic simulation, with the fuselage and wings in shades of blue and green, and the engine nacelles in purple. A complex, tangled red and orange structure extends from the rear of the aircraft, representing a wake or a specific flow field visualization.

Solving Transport Equations on Quantum Computers

Solving transport equations on quantum computers



Goal: Compare performance of **Physics-Informed Neural Networks (PINN)** and its quantum computing analogon, the **Physics-Informed Quantum Circuit (PIQC)**.

Challenge: Very **limited availability and access** to quantum hardware.

- Computations are performed on a **simulator**
- Simplistic 1D problems

Outline:

- PINNs and PIQCs
- Experimental Setup
- Results

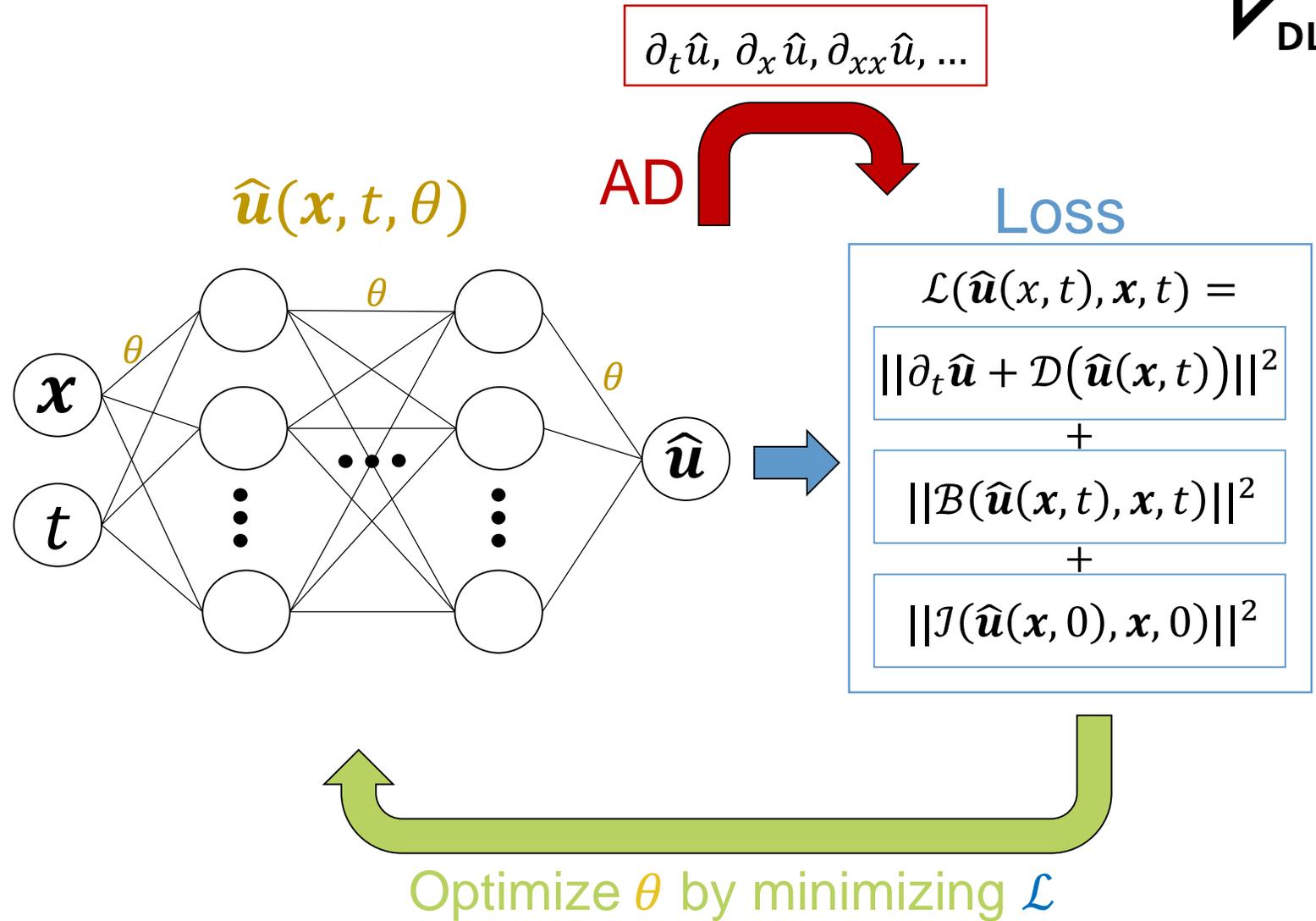
Physics-Informed Neural Network (PINN)



Boundary Value Problem:

$$\begin{aligned} \partial_t u + \mathcal{D}(u) &= 0, \text{ in } \Omega \times [0, T) \\ \mathcal{B}(u(x, t), x, t) &= 0, \text{ in } \partial\Omega \times [0, T) \\ \mathcal{I}(u(x, t), x, t) &= 0, \text{ in } \Omega \end{aligned}$$

- $\hat{u}(x, t, \theta)$ approximates solution $u(x, t)$
- **Loss** terms measure the agreement with PDE, initial cond. \mathcal{I} and boundary cond. \mathcal{B}
- **Automatic Differentiation (AD)** is used to calculate derivatives in **loss**
- Network parameters θ are **optimized** based on the calculated loss



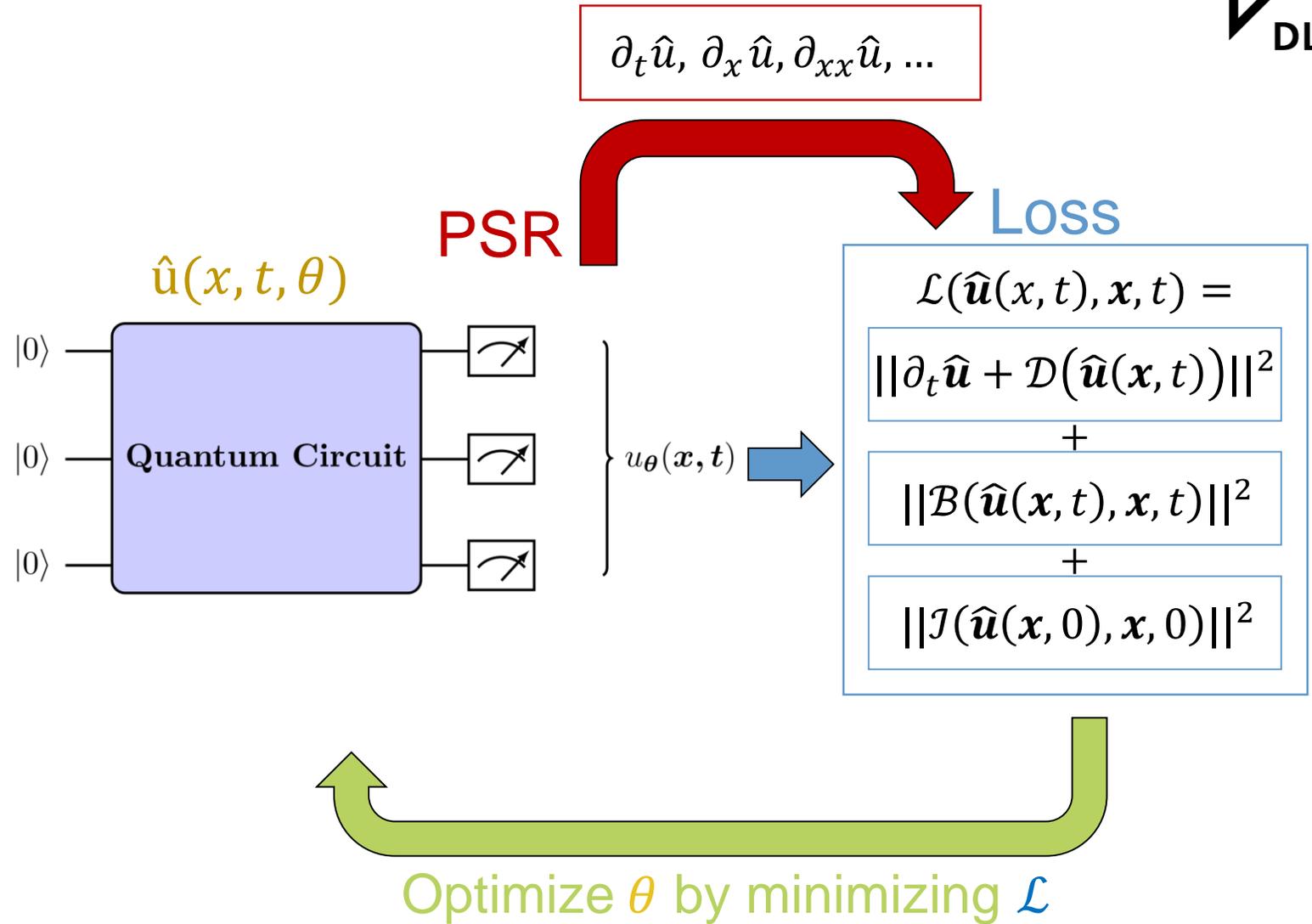
Physics-Informed Quantum Circuit (PIQC)

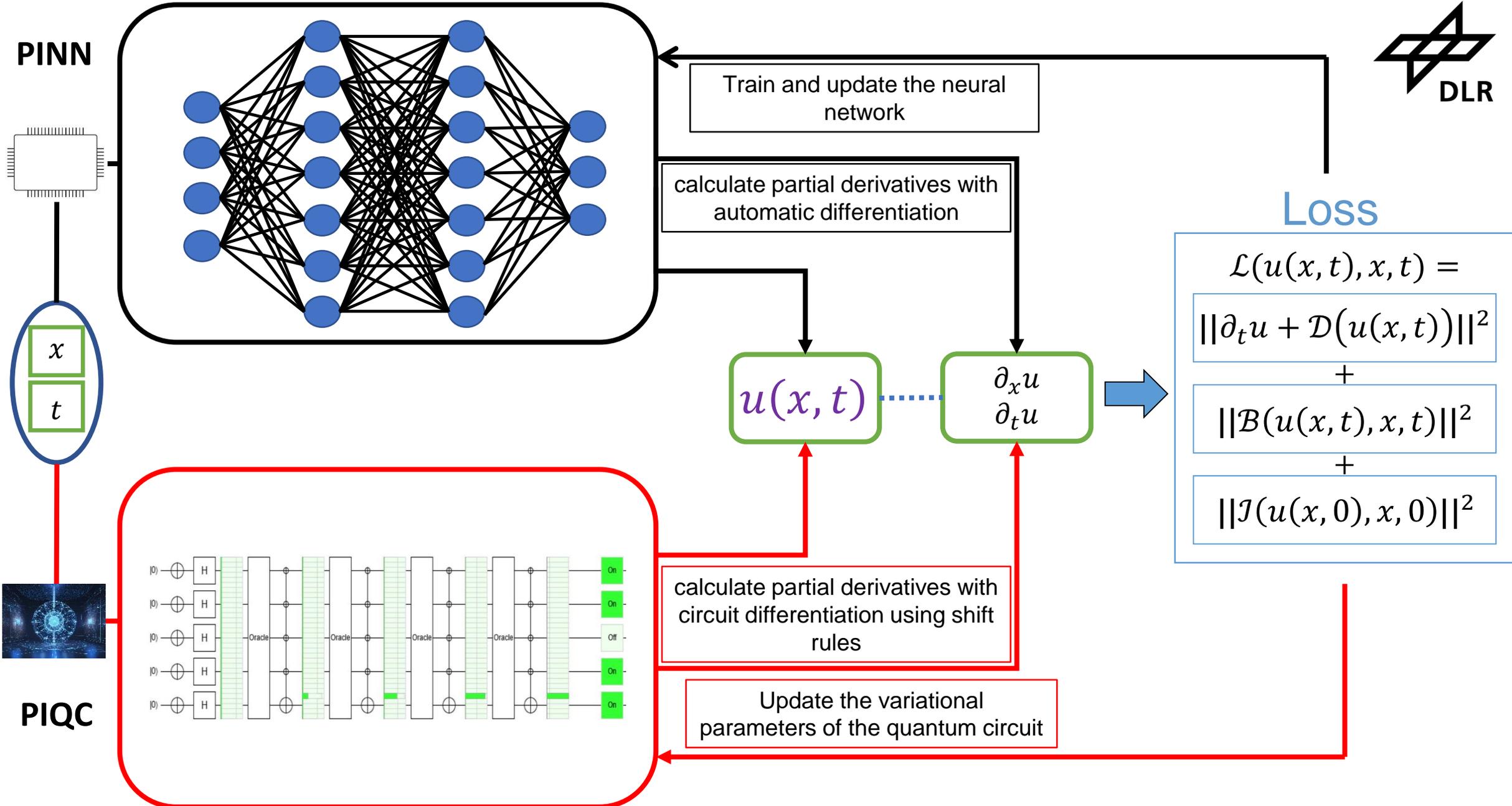


Boundary Value Problem:

$$\begin{aligned} \partial_t u + \mathcal{D}(u) &= 0, \text{ in } \Omega \times [0, T) \\ \mathcal{B}(u(x, t), x, t) &= 0, \text{ in } \partial\Omega \times [0, T) \\ \mathcal{I}(u(x, t), x, t) &= 0, \text{ in } \Omega \end{aligned}$$

- $\hat{u}(x, t, \theta)$ approximates solution $u(x, t)$
- **Loss** terms measure the agreement with PDE, initial cond. \mathcal{I} and boundary cond. \mathcal{B}
- **Parameter Shift Rule (PSR)** is used to calculate derivatives in **loss**
- Network parameters θ are **optimized** based on the calculated loss



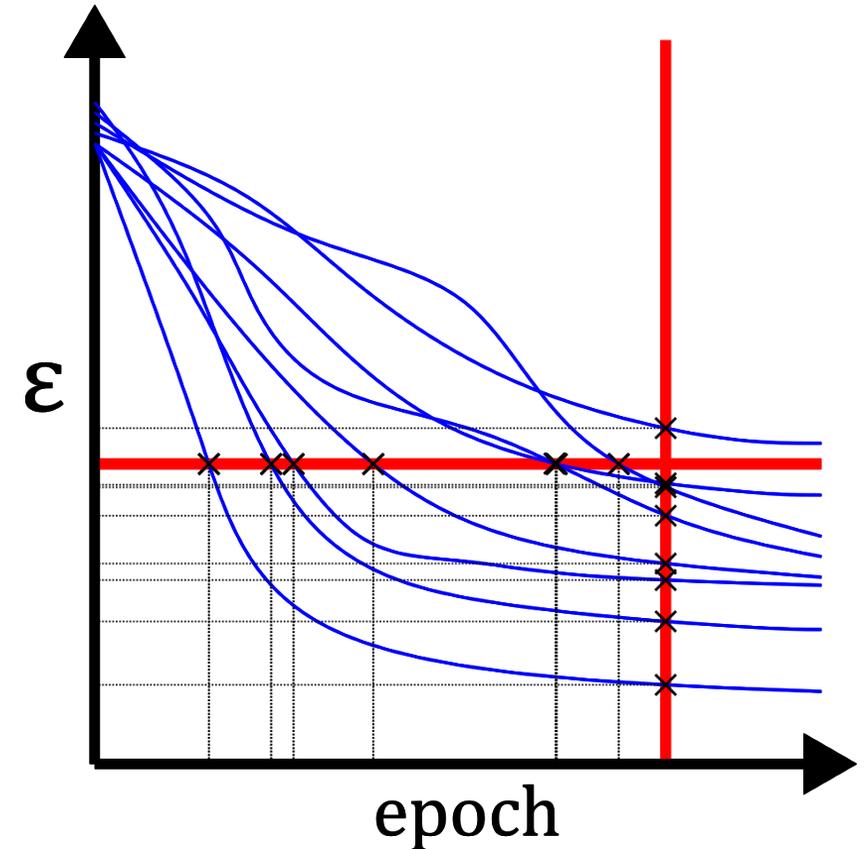


How we compare Accuracy and Convergence

- Repeating runs for varying quantum circuit size and neural network shape
- Consider the **mean absolute error** ε :

$$\varepsilon = \frac{1}{N_{val}} \sum_{i=1}^{N_{val}} |\hat{u}_{\theta}(x_i, t_i) - u(x_i, t_i)|$$

- Compare PINNs and PIQC by measuring:
 - 1) No. of epochs required to reach a threshold value of ε
 - 2) Error ε reached after set number of epochs
- Use same optimization parameters (e.g. learning rate) for fair comparison



Problem I: Ordinary Differential Equation

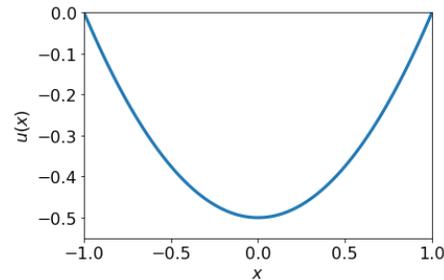
$$\frac{d^2 u}{dx^2} = 1 \quad x \in [-1, 1]$$

Boundary Conditions:

$$u(x = -1) = u(x = 1) = 0$$

Analytical Solution:

$$u(x) = \frac{1}{2}(x^2 - 1)$$



Problem II: Linear Transport of a Gauss Pulse

$$\frac{\partial u}{\partial t} + c \frac{\partial u}{\partial x} = 0 \quad x \times t \in [-1, 1] \times [0, 0.5]$$

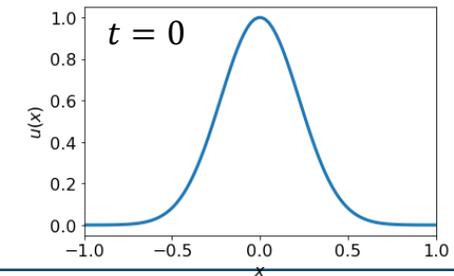
$$c = 0.5$$

Initial Conditions:

$$I(x) = e^{-x^2/0.1}$$

Analytical Solution:

$$u(x, t) = I(x - ct)$$



Problem III: Burgers Equation (Non-Linear)

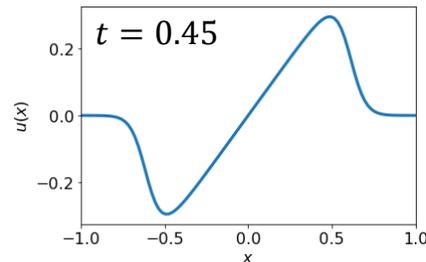
$$\frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} = \vartheta \frac{\partial^2 u}{\partial x^2} \quad x \times t \in [-1, 1] \times [0, 1]$$

$$\vartheta = 0.01$$

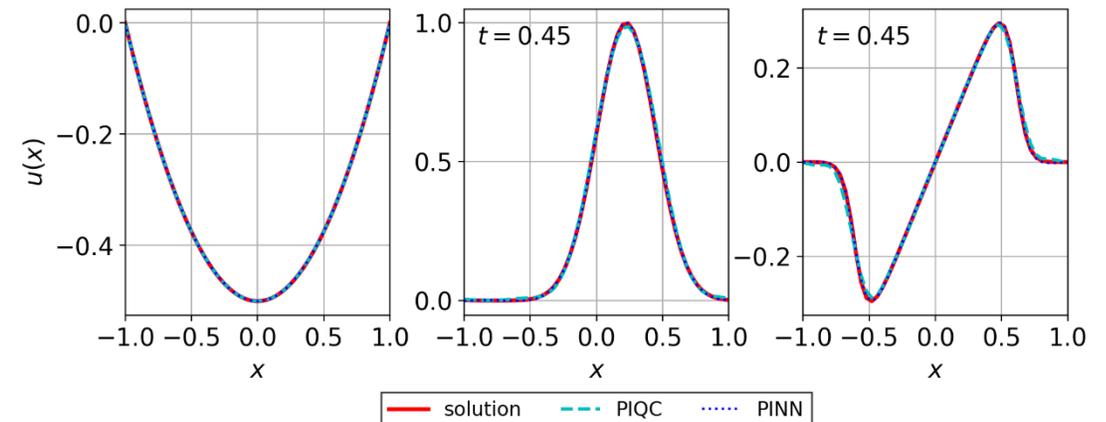
Analytical Solution:

$$u(x, t) = \frac{\frac{x}{t+1}}{1 + \sqrt{\frac{t+1}{t_0} e^{4\vartheta(t+1)} \frac{x^2}{t_0}}}$$

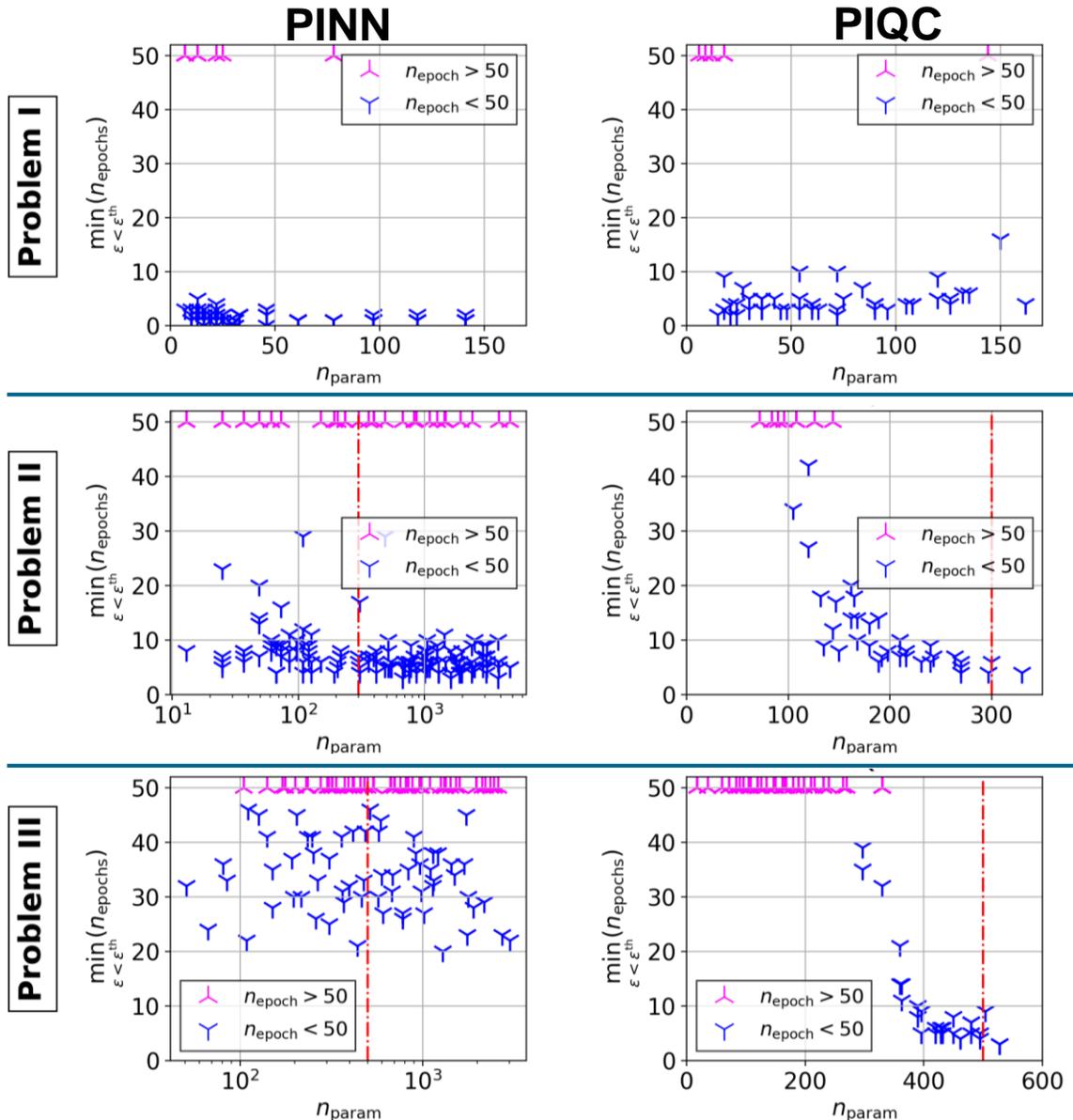
$$t_0 = e^{\frac{1}{8\vartheta}}$$



Problems I-III can be well represented by PINNs and PIQCs



Results: Number of Epochs



- Both approaches need **< 10 epochs** for most runs
- PINNs converge slightly faster

- PIQCs show **clear trend** with n_{param}
- PINN has **training inconsistencies** over whole range of n_{param}

- PIQC shows **faster convergence** as soon as $n_{\text{param}} > 350$.

Summary



- PINNs and PIQCs reach similar accuracies
- PIQCs depend more on n_{param}
- PIQCs converge more consistently for problem II & III
- PIQCs converge faster for large enough n_{param} for problem III
- The PIQC simulation is very time-consuming, allowing only for simple problems



Summary and outlook



The **S**urrogate **M**odeling for **AeRo** data **T**oolbox Python Package [1]

NN Module

PyTorch
Deep learning
framework

PyTorch L-BFGS

**Gradients
&
NNs/circuits**

**Loss function
&
Optimization**

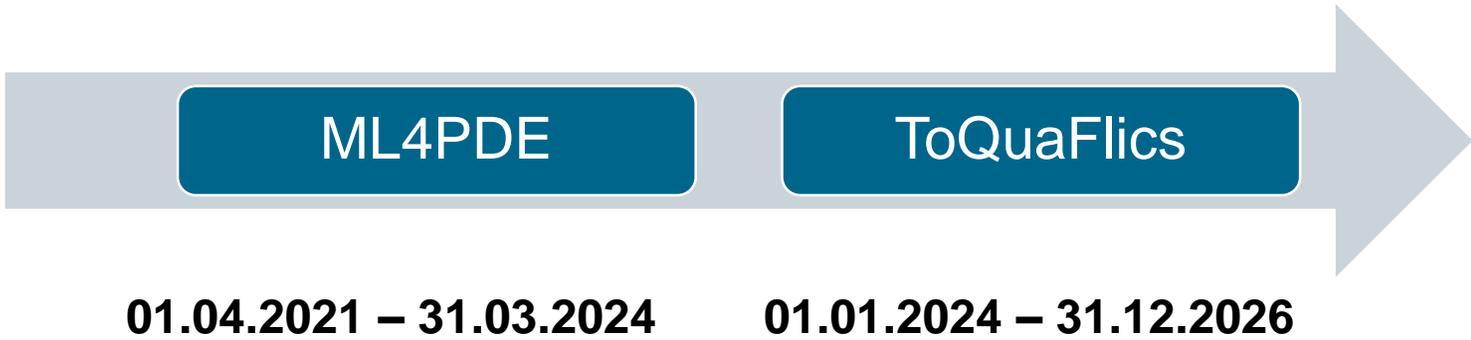
QC Module

PennyLane
Quantum machine
learning framework

PyTorch L-BFGS

- Capabilities developed in the project are available in SMARTy and can be used
 - Physics-informed Neural Networks
 - Correction of Coarse Grid Solutions
 - Quantum Circuits
- Interactive Workshops on some topics have been provided
- Reach out to Anna Kiener, Simon Wassing or smarty@dlr.de for further information

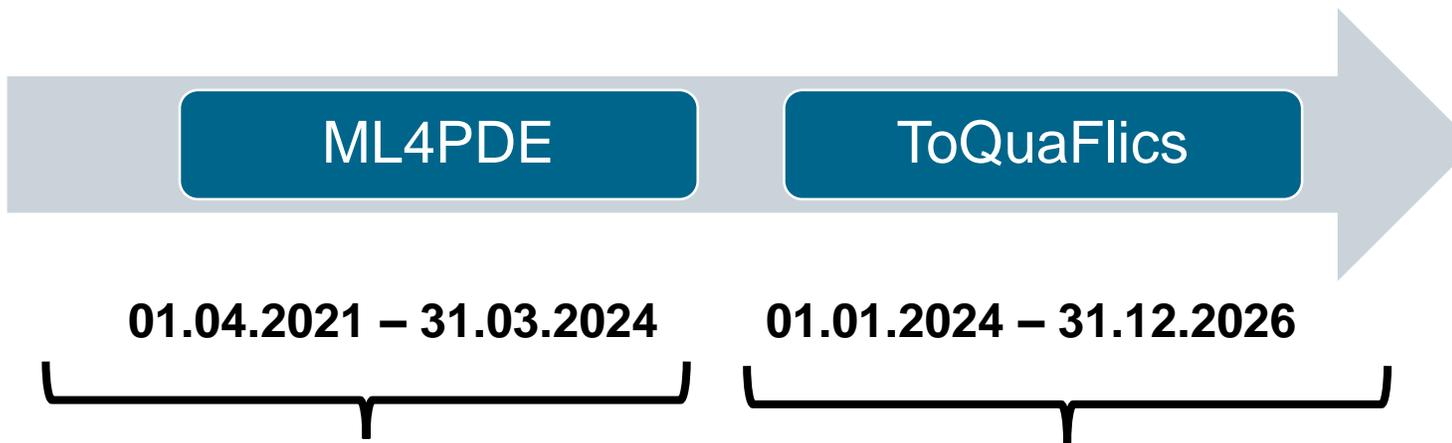
What's next?



XX.XX.2(0)XX

I'm sure it will be on a Monday, but please don't ask me about the year!

What's next?



Funded by
Vorstandsreserve VO-L

Funded by DLR QC-I



XX.XX.2(0)XX

**I'm sure it will be on a Monday, but
please don't ask me about the year!**



**Many challenges, but
potential is huge**

Availability of quantum computers - assumptions in BDLI Roadmap 2021

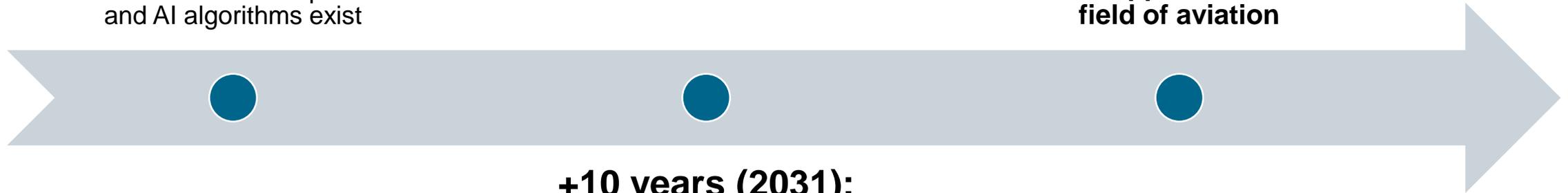


+5 years (2026):

First commercial quantum computers & hybrid quantum HPC methods in use;
middleware products for hybrid use of QC and established optimization and AI algorithms exist

+15-20 years (2036-2040):

Error-corrected, universally programmable quantum computers in commercial use
Various applications in the field of aviation



+10 years (2031):

NISQ computers (Noisy Intermediate-Scale Quantum) achieve **>100 Qbit**;
low-error QC systems

Reality check:

2021:
IBM quantum processor „Eagle“ 127-qubit
2022:
IBM "Osprey" 433-Qubit

2023:

- IBM Condor 1.121 qubit
- IBM Heron 133-qubit tunable-coupler QPU (eliminates crosstalk errors)
- IBM Quantum System Two (modular utility-scaled quantum computer system)

IBM roadmap:

2029-2033:
fully error-corrected systems, capable of running 100 million operations in a single quantum circuit in 2029, and a billion operations by 2033 (100,000 physical qubits form 2,000 logical qubits)

Goals of the project

- Perspective how to exploit potential of quantum computers for applications in aerodynamics
- Identification which parts of a solution algorithm can gain from bringing it on a quantum computer
- Estimate how many QuBits are required to solve relevant test cases representing industrial problems
- Investigation of error propagation in Quantum algorithms to formulate requirements on accuracy for hardware

- Pia Siegl, Simon Wassing, Markus Mieth, Stefan Langer, Philipp Bekemeyer: **Solving transport equations on Quantum Computers - Potential and limitations of Physics-Informed Quantum Circuits**, accepted in CEAS Journal (will be published soon)
- S. Langer: **Application of the iteratively regularized Gauss-Newton method to parameter identification problems in Computational Fluid Dynamics**, Journal of Computers & Fluids, 284: 104638, 2024, <https://doi.org/10.1016/j.compfluid.2024.106438>
- S. Wassing, S. Langer, P. Bekemeyer: **Physics-Informed Neural Networks for Parametric Compressible Euler Equations**, Journal of Computers & Fluids, 270:106164. 2024, <https://doi.org/10.1016/j.compfluid.2023.106164>
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- T. Bode: **The two-particle irreducible effective action, for classical stochastic processes**, Journal of Physics A: Mathematical and Theoretical, 55 (2022) 265401, <https://doi.org/10.1088/1751-8121/ac73c6>

Thank you for your attention!

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