

Investigation of magnetic heat shielding using high-temperature superconductor

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1. Introduction

- Spacecraft undergoing atmospheric entry dissipates a vast amount of heat, which must be shielded.
- A proposed concept is to employ a strong magnetic field using high-temperature superconductor (HTS) to reduce the direct interaction between the spacecraft and the plasma, thus lowering the heat load on the shields.
- In collaboration with German Aerospace Centre (DLR), the proof of concept will be investigated in the HEG shock tunnel, which can recreate flow conditions for hypersonic flight configurations
- Shock stand-off distance will be measured by the Schlieren imaging.
- This poster introduces the magnet design and the mechanical structures required for this study in the shock tunnel.

2. DLR's HEG high-enthalpy shock tunnel

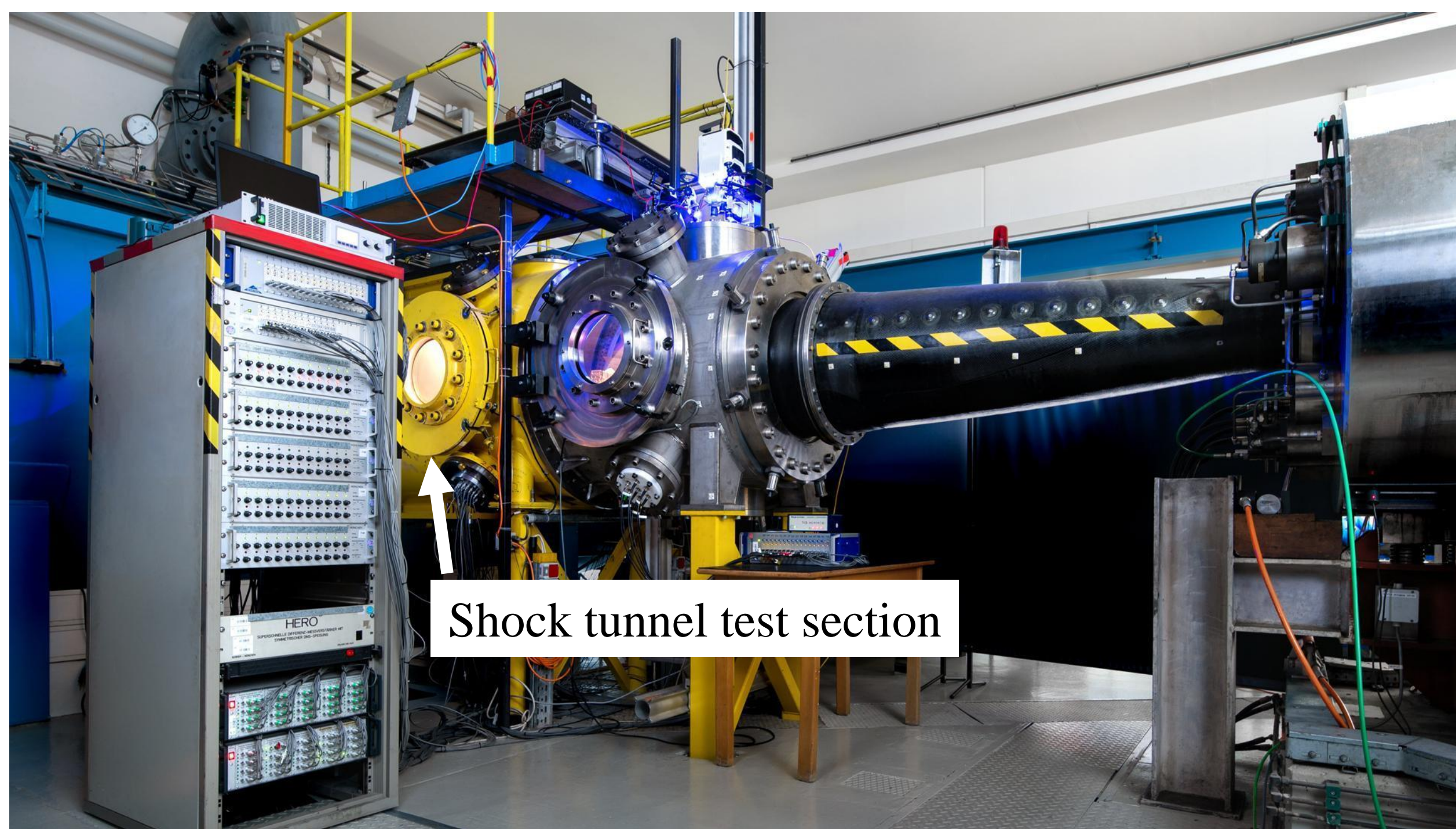
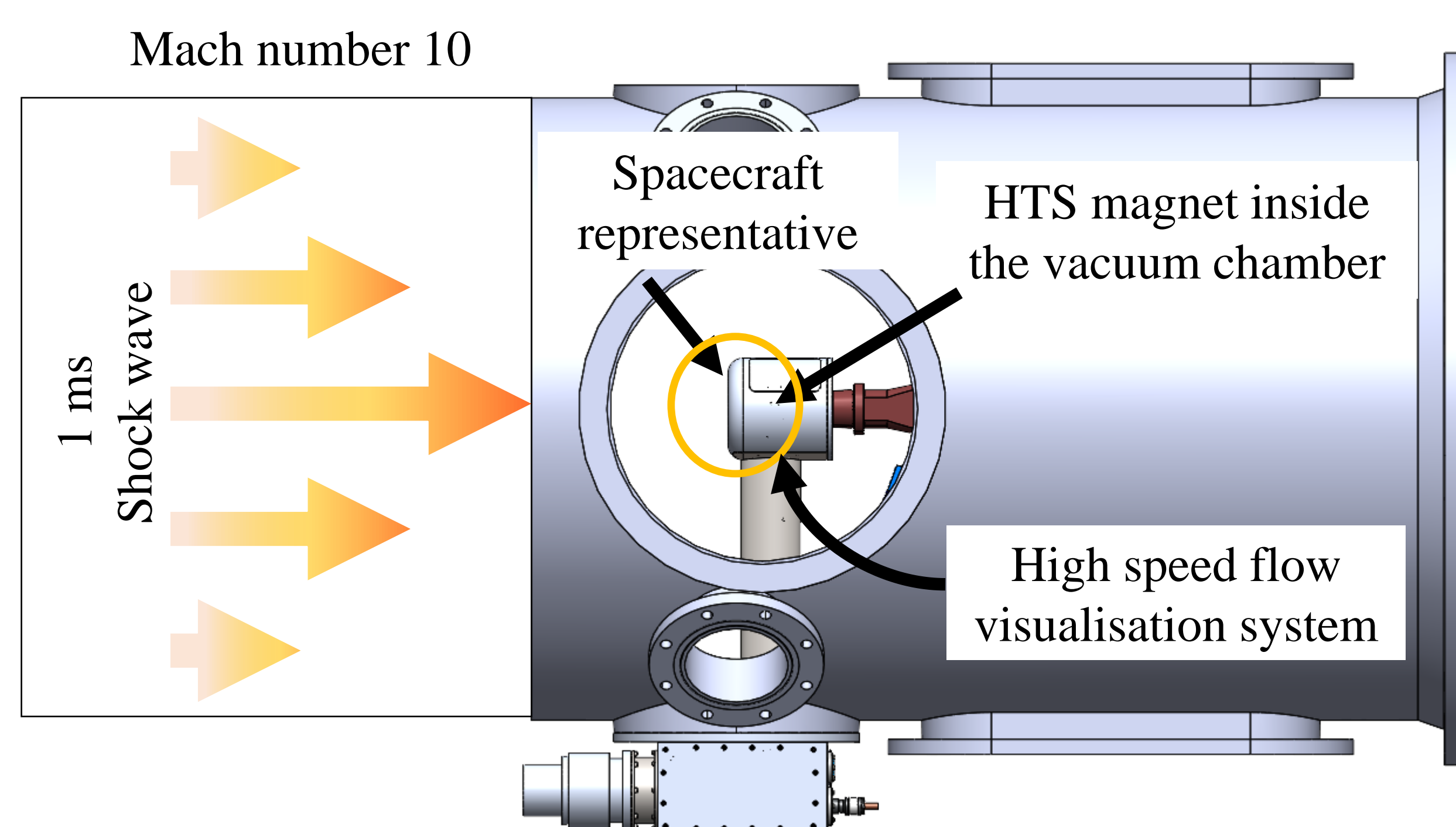


Fig 1. Photograph of the test shock tunnel

3. Experimental setup



Measurements

- Surface pressure;
- Heat flux;
- Shock stand-off distance

Fig 2. Overall view of the experimental setup in HEG shock tunnel

7. Magnetic field simulation

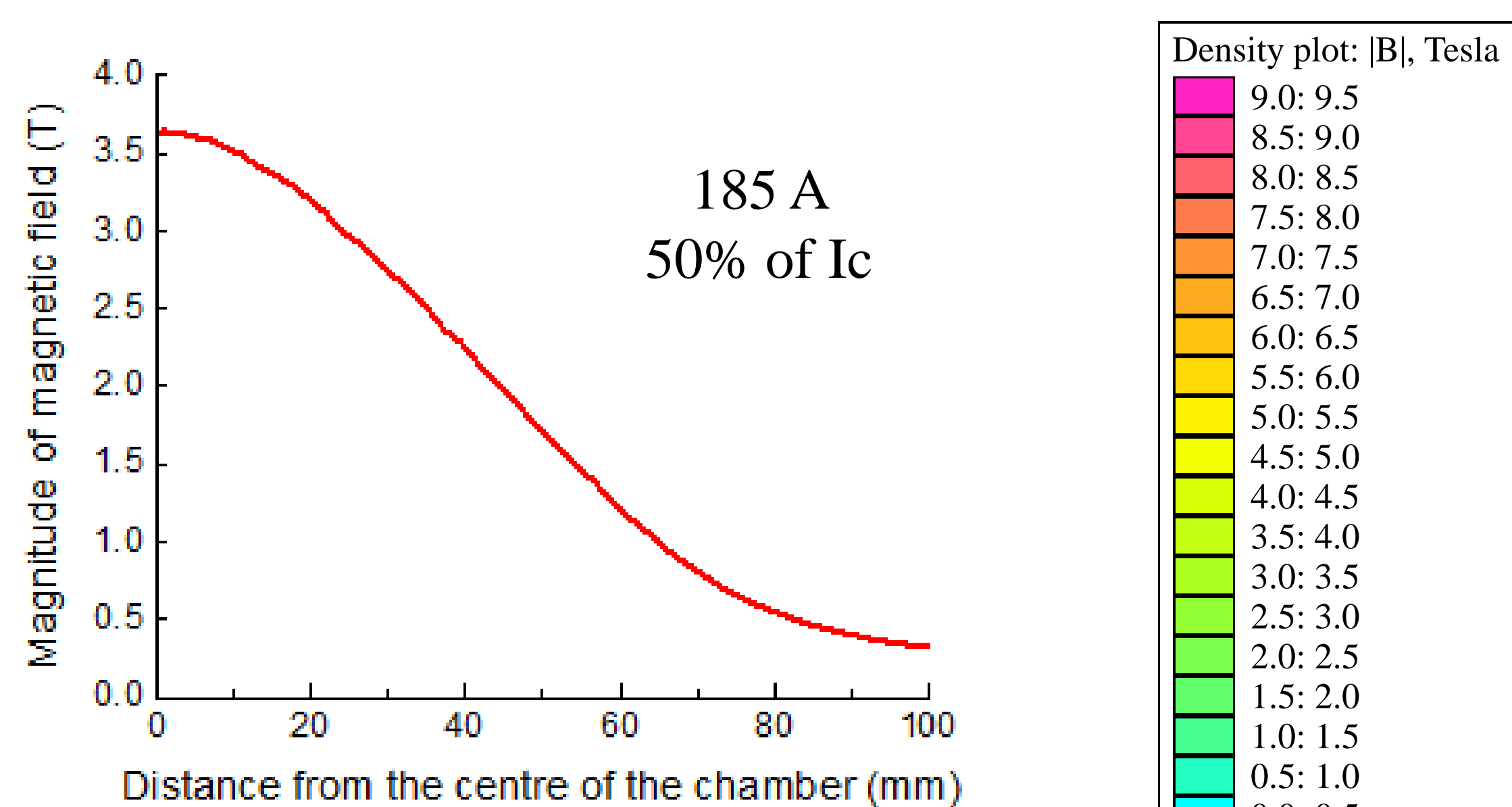
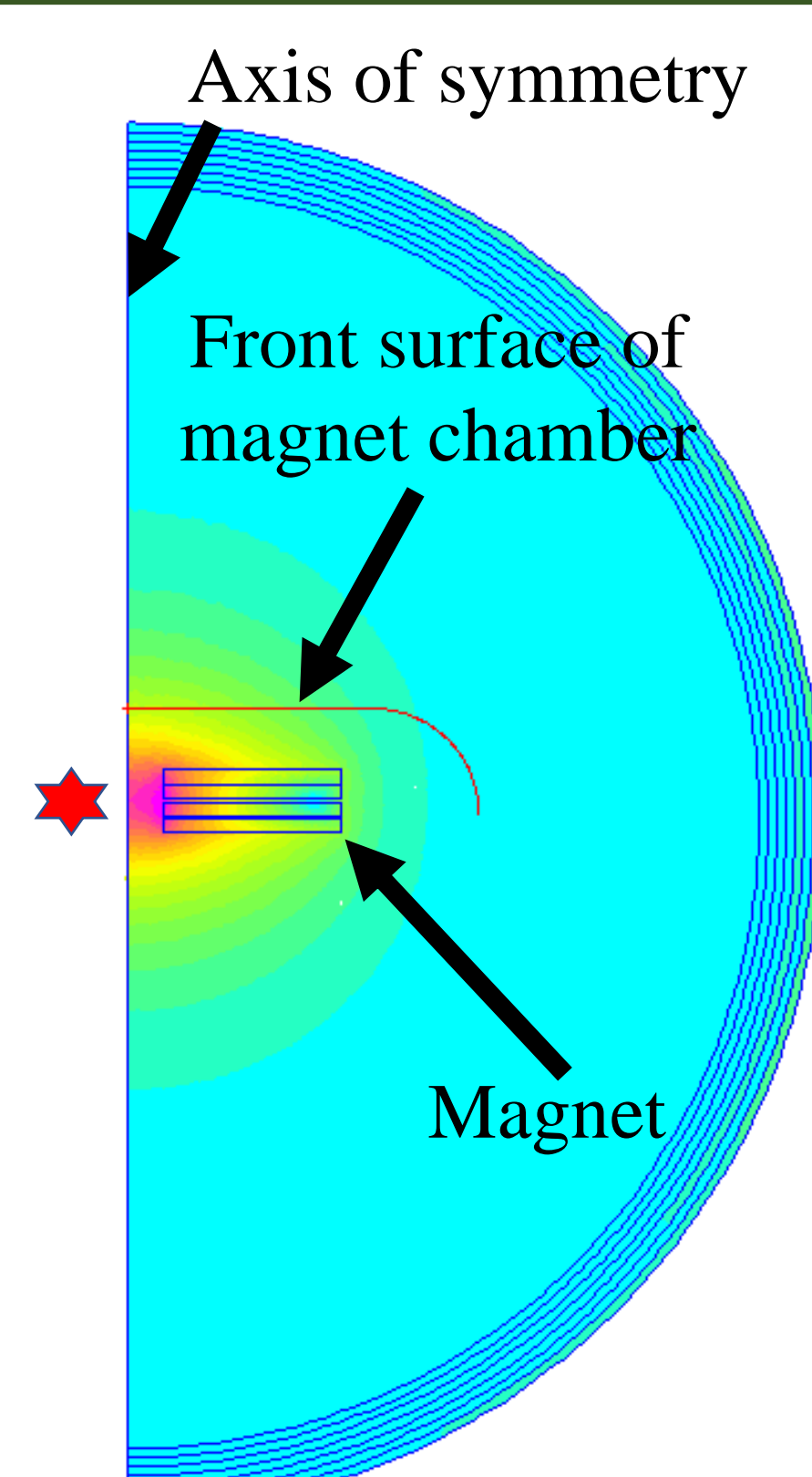
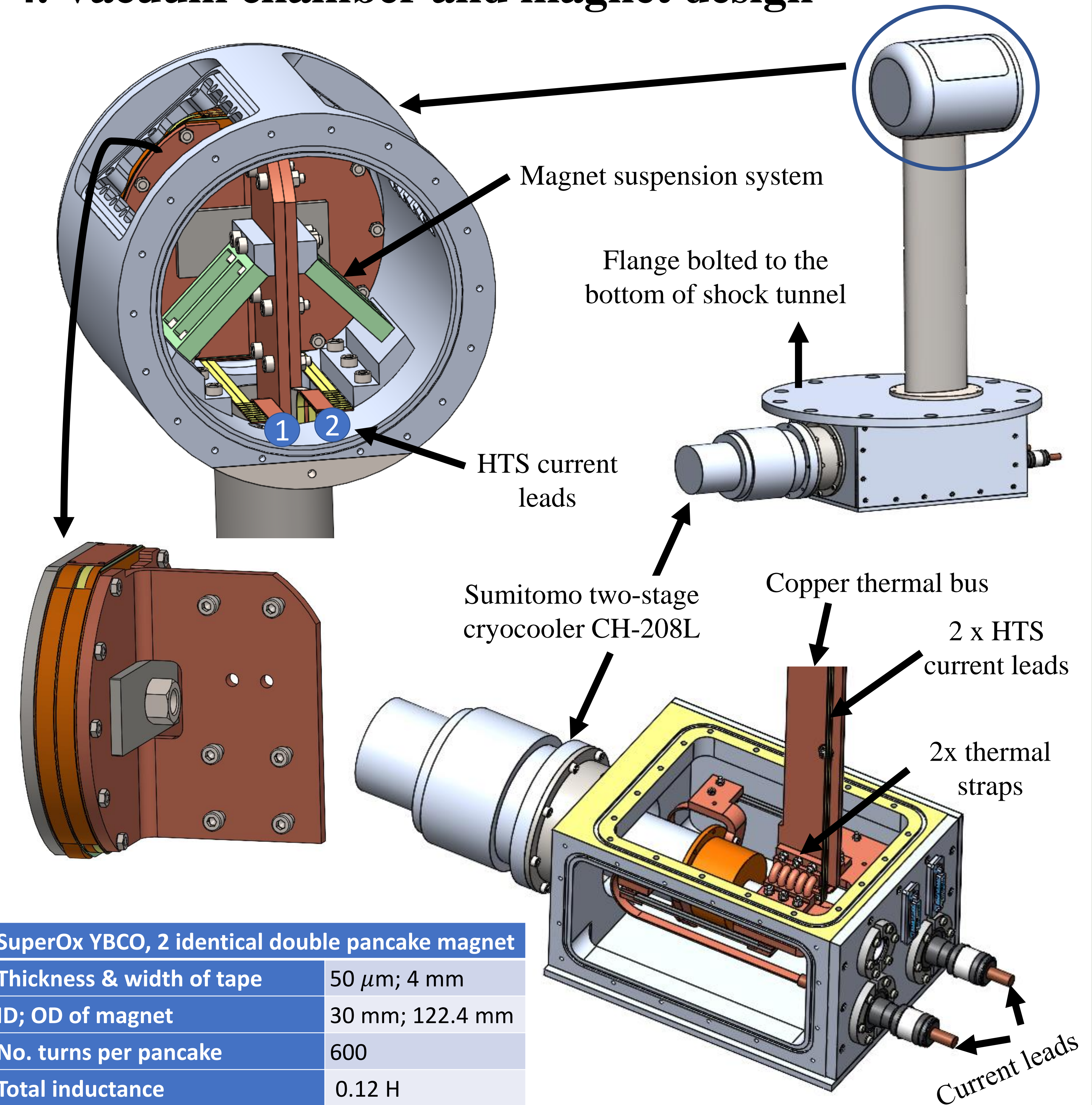


Fig 3. Magnitude of magnetic field on the front surface of the chamber

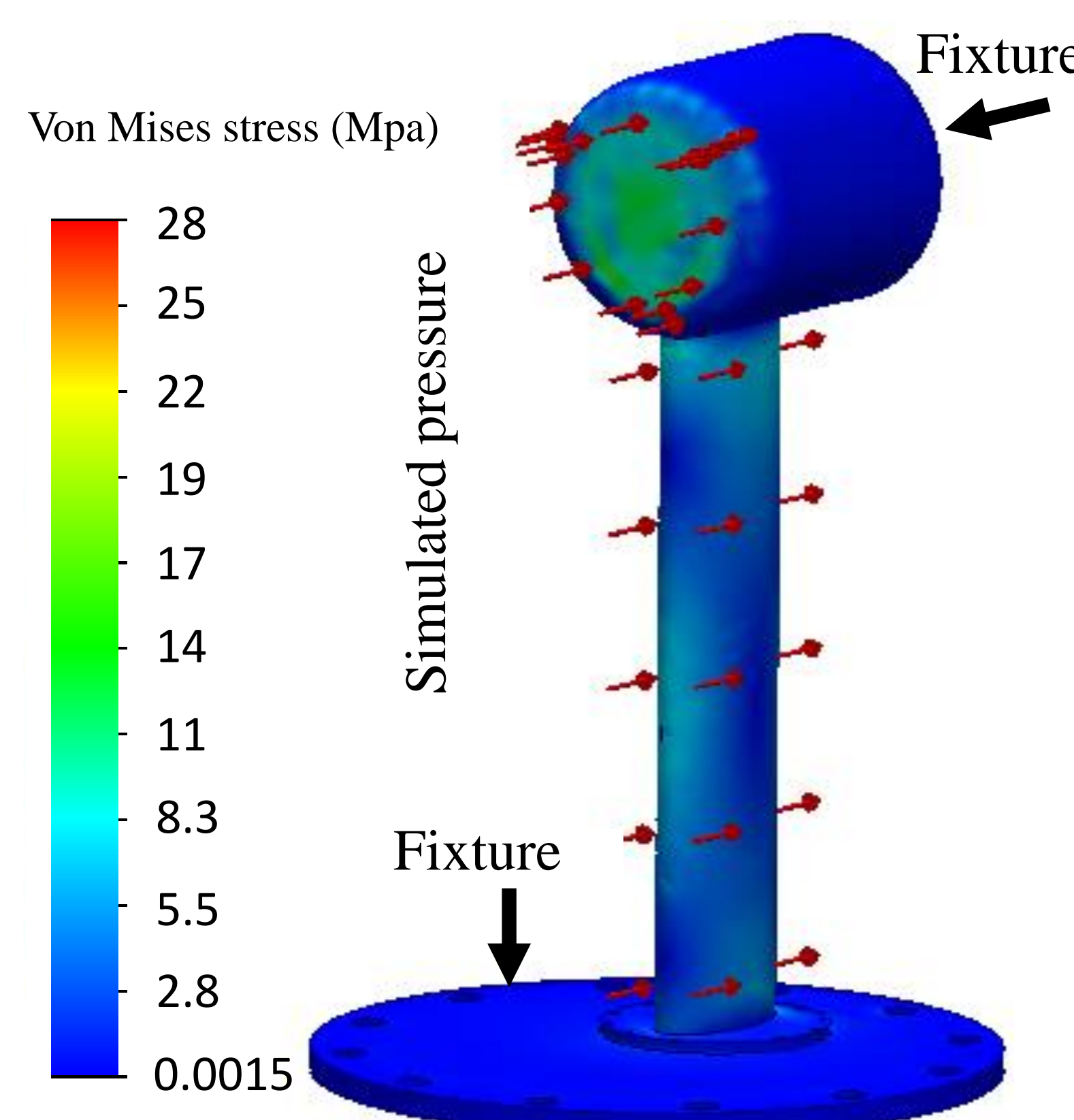
★ Central field 9.0 T



4. Vacuum chamber and magnet design

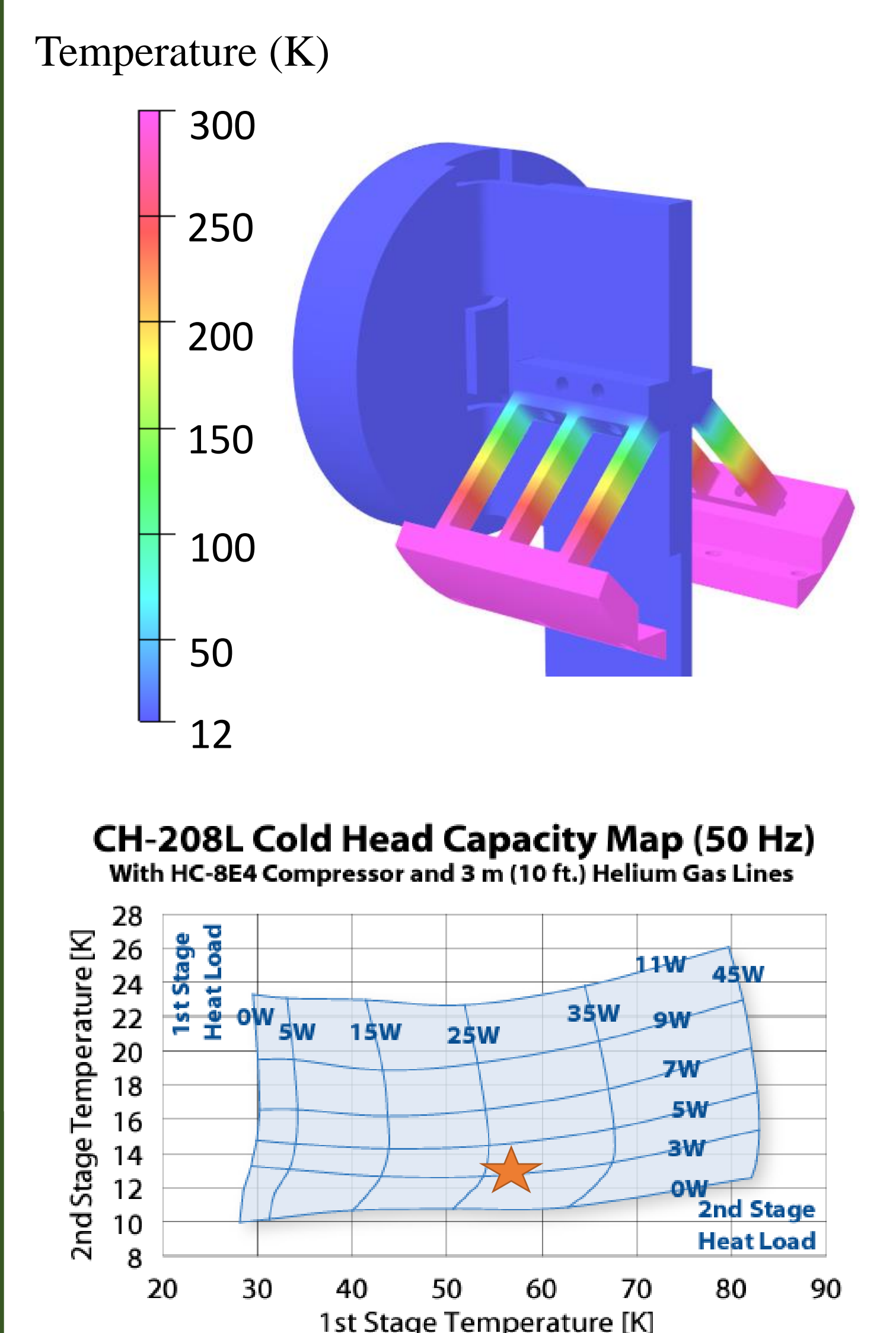


5. Structure analysis



Factor of safety above 8 for Aluminium alloy 6061 and stainless steel 304 under the simulated shock wave

6. Thermal simulation



Magnet is expected to be operated at 12 K

8. Conclusion

- A HTS magnet which can generate a **central field of 9.0 T** is currently being built at Robinson Research Institute.
- The magnet will generate **3.6 T** on the front surface of the magnet chamber. The shock stand-off distance under this magnetic field will be investigated in HEG shock tunnel in **mid 2024**.
- The vacuum chambers experiencing the shock in the tunnel have a factor of safety of 8.

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