

Numerical Simulations in Ultrasonic Guided Waves Analysis for the Design of SHM Systems – Benchmark Study based on the Open Guided Waves Online Platform Dataset

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Knowledge for Tomorrow



Outline

- Guided Waves based Structural Health Monitoring (SHM)
- Guided Waves in thin structures
- Simulation methods for ultrasonic wave propagation analysis
- Open Guided Waves Project
- Simulation results
- Conclusions and further steps



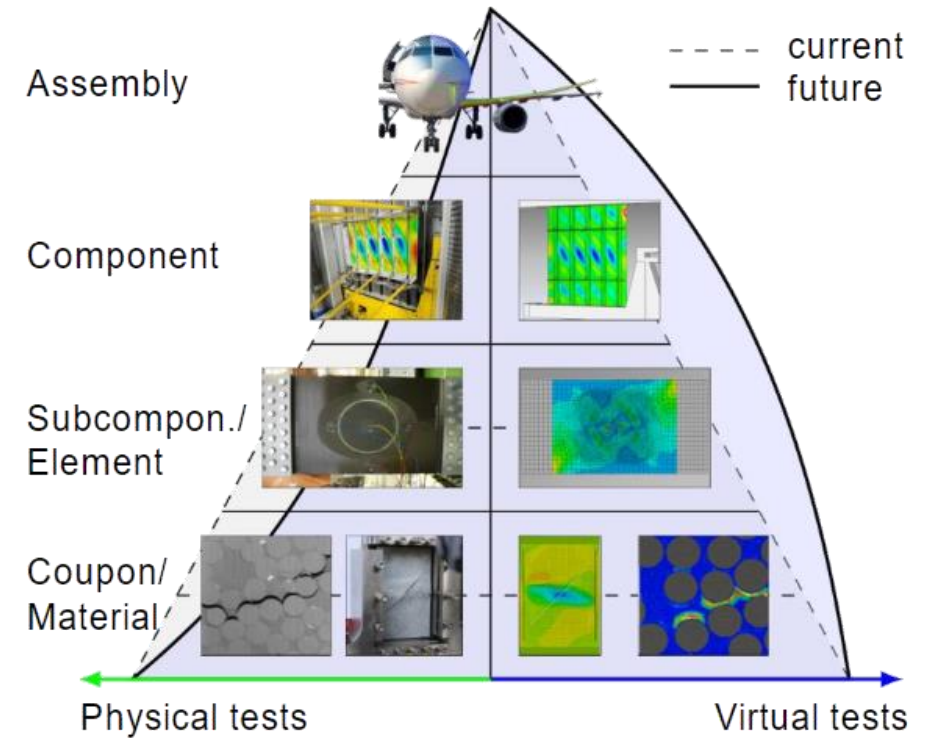
Guided Waves based Structural Health Monitoring (SHM)

- GW has become a major topic in research on SHM since ~2000
 - ✓ Knowledge about underlying physics in isotropic and anisotropic materials
 - ✓ Simulations
 - ✓ Measurement equipment
 - ✓ Demonstrators
- First technical rules and regulations exist
 - ✓ SAE-ARP6461: Guidelines for Implementation of Structural Health Monitoring on Fixed Wing Aircraft (September 2013)
 - ✓ SHM 01 E: Structural Testing with Guided Waves (December 2014, 56 p.)
 - ✓ SAE-AIR6245: Perspectives on Integrating Structural Health Monitoring Systems into Fixed-Wing Military Aircraft (September 2019)



Need for efficient simulation methods in SHM

- For the approval of SHM systems, proof of damage detection is required
- Enormous effort in development and adaption cannot be covered by laboratory testing
- Use of model assisted methods is an alternative to laboratory testing
- Development of numerical methods to analyze ultrasonic wave propagation is still a challenge
- Detection of typical defects and damages occurring in fiber reinforced composites by ultrasonic guided waves (UGW)



→ Aim: **Simulation strategies for UGW propagation in complex composite structures**



Guided waves in thin structures [WILLBERG2013, WILLBERG2015]

- Appearance of Lamb waves in thin elastic shells and plates
- Dispersive properties of Lamb wave modes

$$\frac{\tan(\tilde{p}h)}{\tan(\tilde{q}h)} + \frac{4\tilde{p}\tilde{q}k^2}{(k^2 - \tilde{p}^2)^2} = 0$$

$$\frac{\tan(\tilde{q}h)}{\tan(\tilde{p}h)} + \frac{(k^2 - \tilde{p}^2)^2}{4\tilde{p}\tilde{q}k^2} = 0$$

$$\tilde{p}^2 = \frac{\omega^2}{c_L^2} - k^2, \quad \tilde{q}^2 = \frac{\omega^2}{c_T^2} - k^2$$

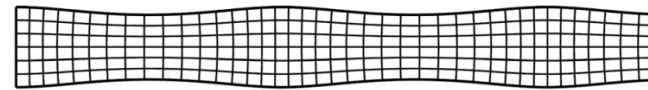
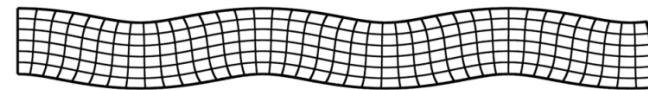
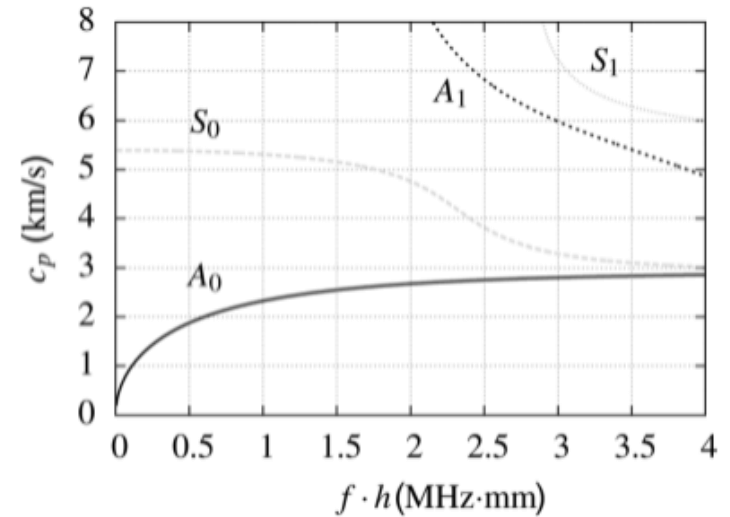
(a) S_0 -mode(b) A_0 -mode

Fig. 1 Lamb wave mode shapes



(a) Phase velocity dispersion curves

Fig. 2 Dispersion curves for the first two symmetric and anti-symmetric Lamb modes in an aluminum plate ($E = 7 \cdot 10^{10} \text{ N/m}^2$, $\nu = 0.33$)

c_L, c_T ...longitudinal, transversal wave velocity

$$c_p = \frac{\omega}{k}, \quad k = \frac{2\pi}{\lambda}$$



Elastodynamic Finite Integration Technique (EFIT) [SCHUBERT2004, TSCHÖKE2018]

- Discretization scheme to model elastic waves (Finite Volume Method)
- EFIT uses a velocity-stress formalism on a staggered grid
- The time integration scheme is an explicit Leapfrog scheme

Cauchy's equation of motion

1. Calculate \dot{v} in space
2. Update v in time using \dot{v}

$$\rho \dot{v}_h^{k-\frac{1}{2}} = F \left(\sigma_h^{k-\frac{1}{2}}, f_h^{k-\frac{1}{2}} \right)$$

$$v_h^k = v_h^{k-1} + \Delta t \cdot \dot{v}_h^{k-\frac{1}{2}}$$

Hooke's law

3. Calculate $\dot{\sigma}$ in space
4. Update σ in time using $\dot{\sigma}$

$$\dot{\sigma}_h^k = \tilde{F}(v_h^k)$$

$$\sigma_h^{k+\frac{1}{2}} = \sigma_h^{k-\frac{1}{2}} + \Delta t \cdot \dot{\sigma}_h^k$$

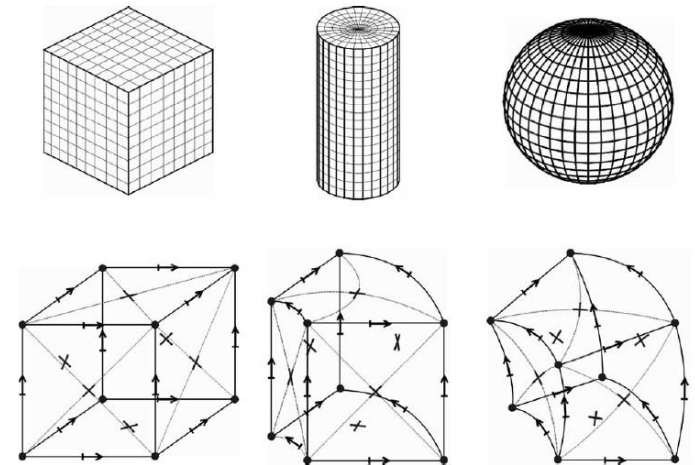
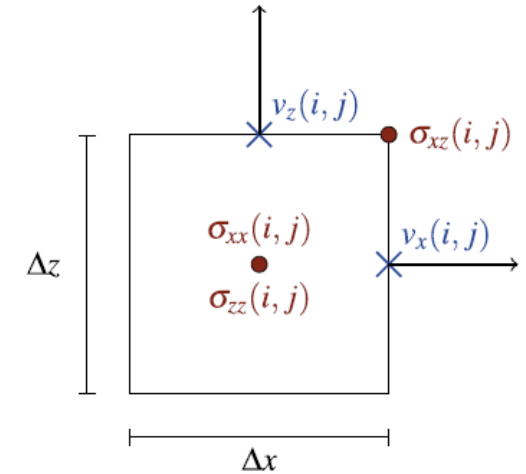
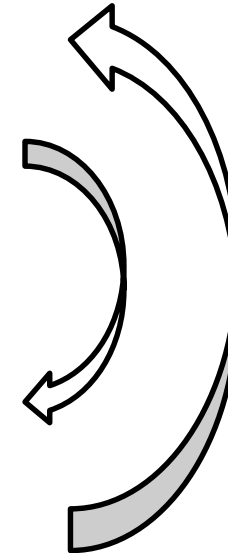
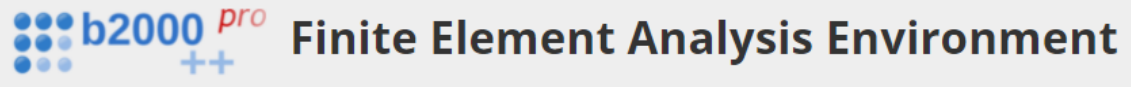


Figure: Staggered grid in 3D.

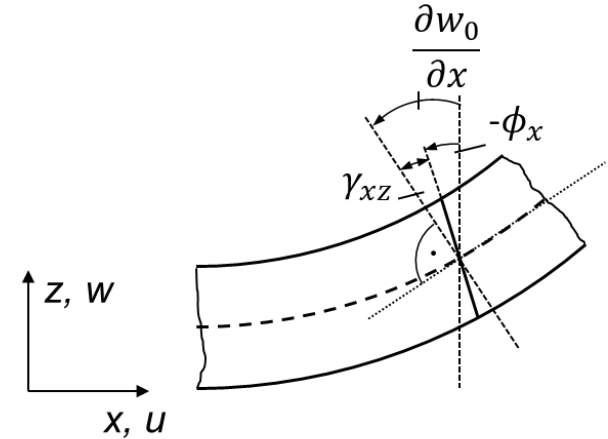


Finite shell elements in b2000++pro



<https://www.smr.ch>

- 4-node and 9-node shell elements following First Order Shear Deformation Theory (FSDT) [REDDY1999]
- MITC – Mixed Interpolation of Tensorial Components [BATHE2002]

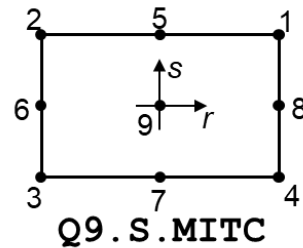
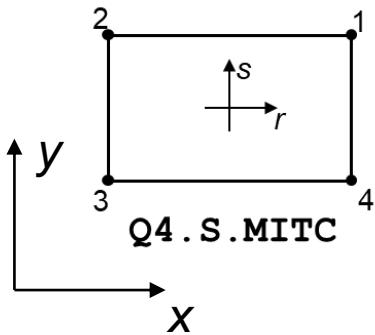


$$u(x, y, t) = u_0(x, y, t) + z\phi_x(x, y, t)$$

$$v(x, y, t) = v_0(x, y, t) + z\phi_y(x, y, t)$$

$$w(x, y, t) = w_0(x, y, t)$$

- Lamb wave A_0 -mode → shell bending mode
- Lamb wave S_0 -mode → shell membrane mode



Open Guided Waves Project [MOLL2020]

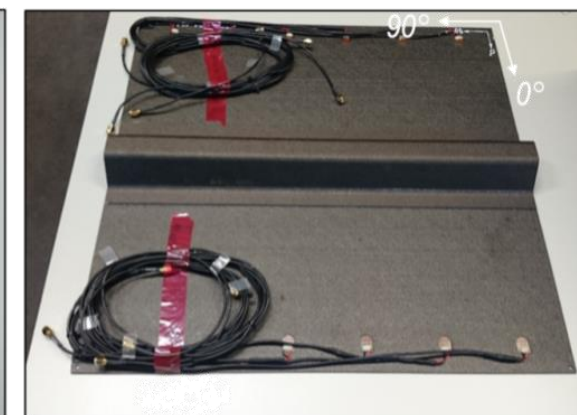
“Rely on Collective Data, Emphasize Your Work!”

- Provides a transparent data set of wide-range measurements that is freely available
- Focused on guided wave inspection techniques for carbon fiber reinforced polymers
- <http://openguidedwaves.de>
- <https://doi.org/10.5281/zenodo.5105861>

(a) Wave field plate (with stringer)



(b) SHM plate (with stringer)



- 500 mm x 500 mm
- Piezoelectric transducers
- 3D Laser Doppler vibrometer



Plate with stringer – material and composite lay-up [MOLL2020]

- Material stringer:

Hexply ® M21/34%/UD194/IMA-12K

```
material 1 type orthotropic
  e1 171500.
  e2 8659.
  nu12 0.324
  g12 5882.
  g13 5882.
  g23 3331.
  density 1.58e-09
end
```

- Material plate:

Hexply ® M21/34%/UD134/T700/300

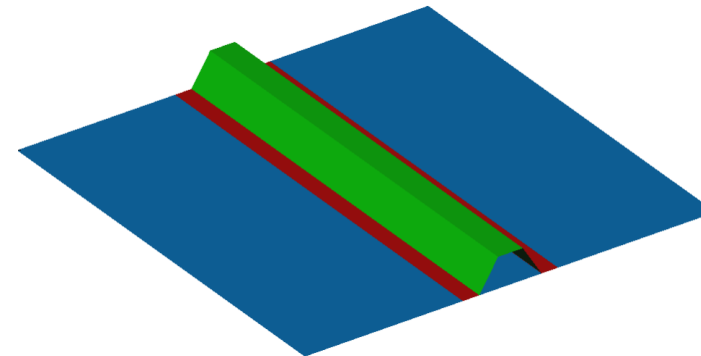
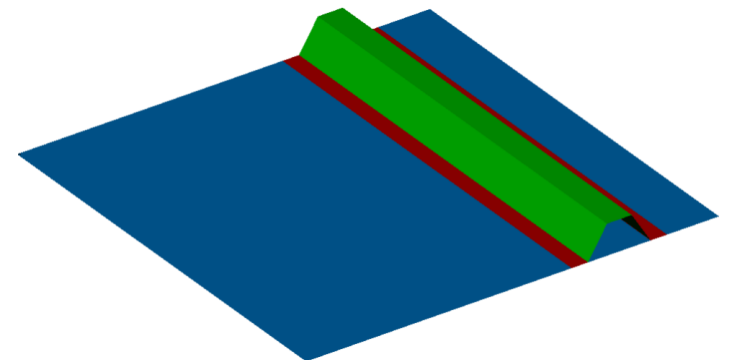
```
material 2 type orthotropic
  e1 125462.
  e2 8701.
  nu12 0.372
  g12 4200.
  g13 4200.
  g23 3000.
  density 1.571e-09
end
```

- Lay-up plate

```
material 3 type laminate
  0.125 +45 2
  0.125 +0 2
  0.125 -45 2
  0.125 +90 2
  0.125 -45 2
  0.125 +0 2
  0.125 +45 2
  0.125 +90 2
  0.125 +90 2
  0.125 +45 2
  0.125 +0 2
  0.125 -45 2
  0.125 +90 2
  0.125 -45 2
  0.125 +0 2
  0.125 +45 2
end
```

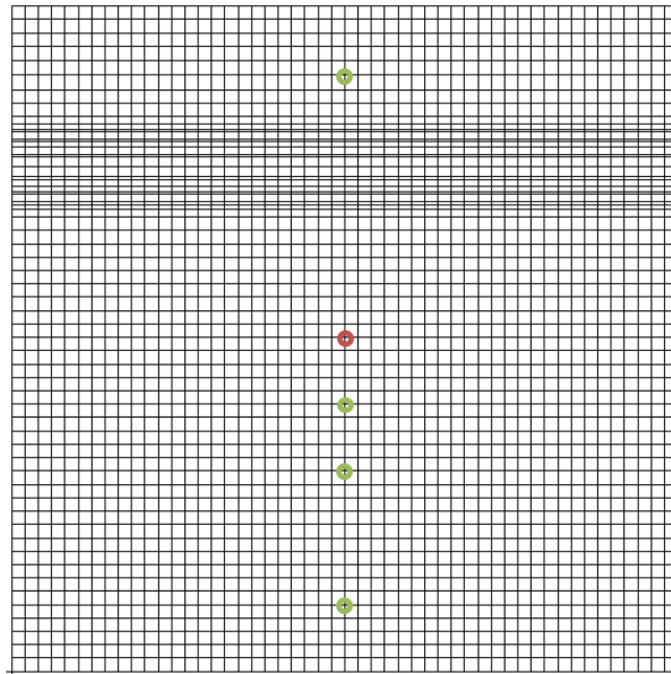
- Lay-up stringer

```
material 5 type laminate
  0.125 -45 1
  0.125 +0 1
  0.125 +90 1
  0.125 +45 1
  0.125 +90 1
  0.125 -45 1
  0.125 -45 1
  0.125 +90 1
  0.125 +45 1
  0.125 +90 1
  0.125 +0 1
  0.125 -45 1
end
```

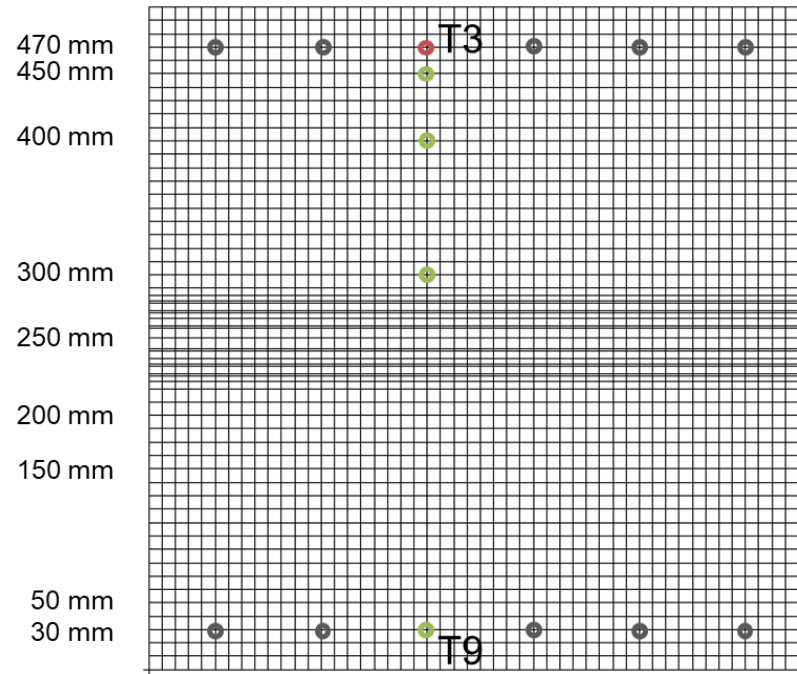


Input/output setup of test plates

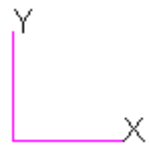
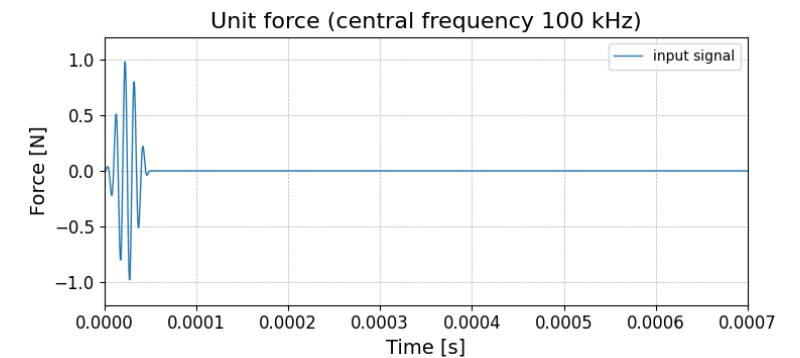
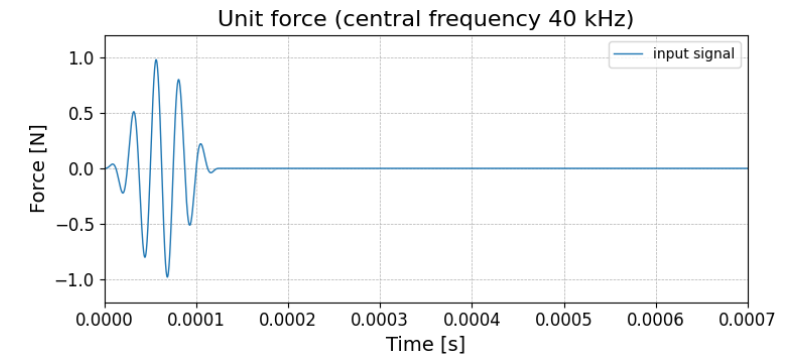
Wave field plate



SHM plate



Excitations (modulated unit force)



- Transducer/excitation
- Evaluation
- Other transducer positions

$$F(t) = \hat{F} \sin^2\left(\frac{2\pi f_c t}{2n}\right) \sin(2\pi f_c t), \quad n = 5$$



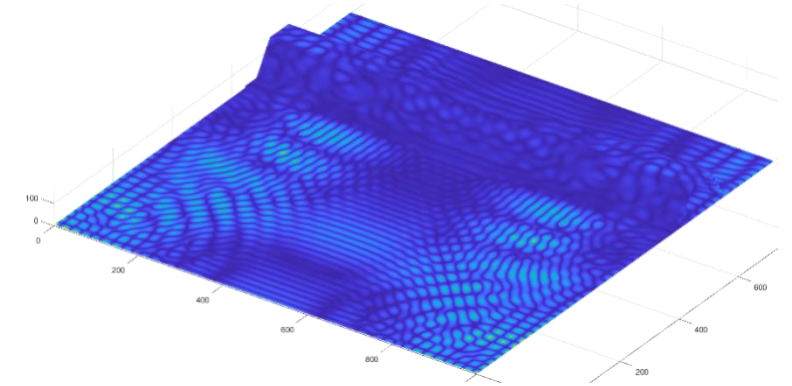
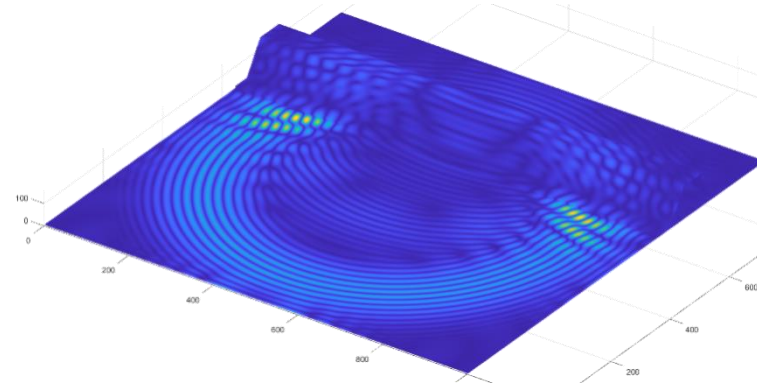
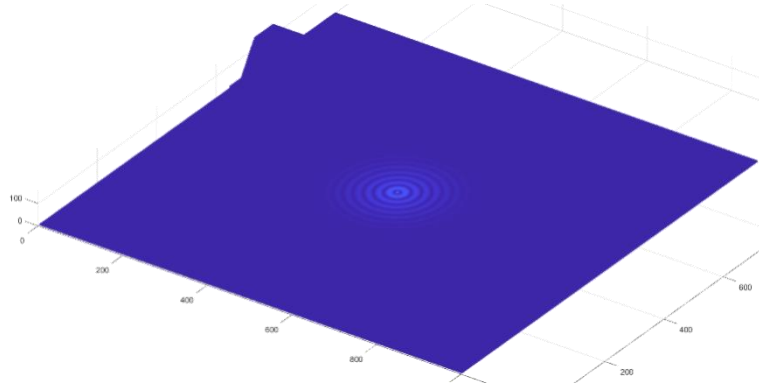
Wave field plate – $f_c = 40$ kHz, Displacement amplitudes (normalized)

$t = 0.7E-4$ s

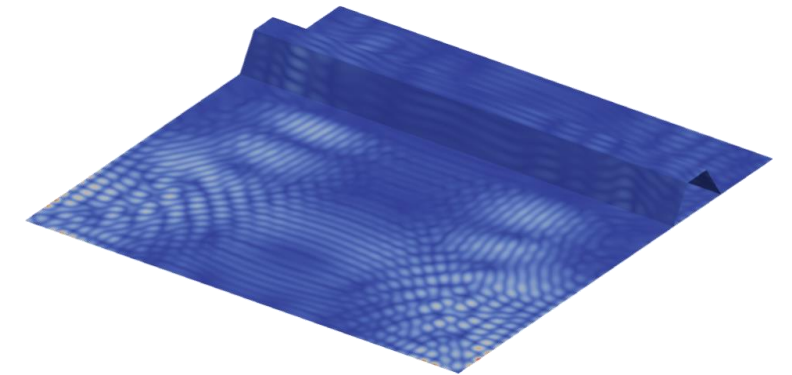
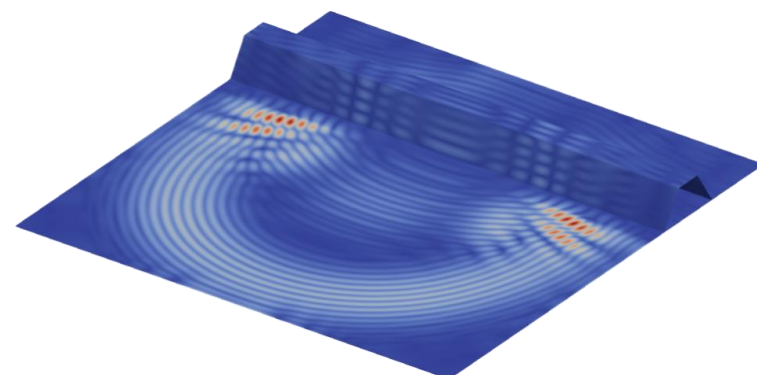
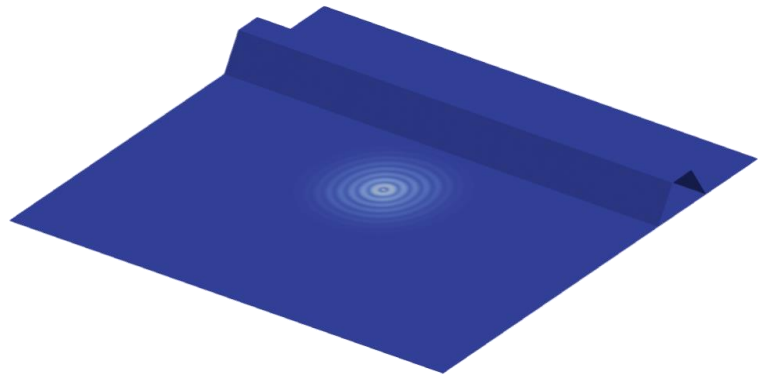
$t = 2.1E-4$ s

$t = 3.5E-4$ s

EFIT



FEM



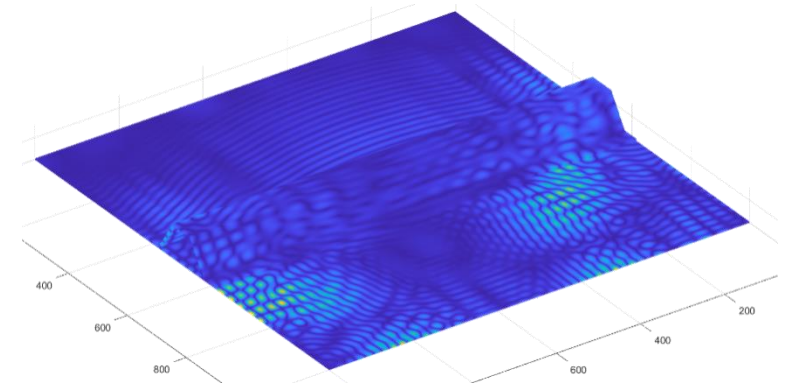
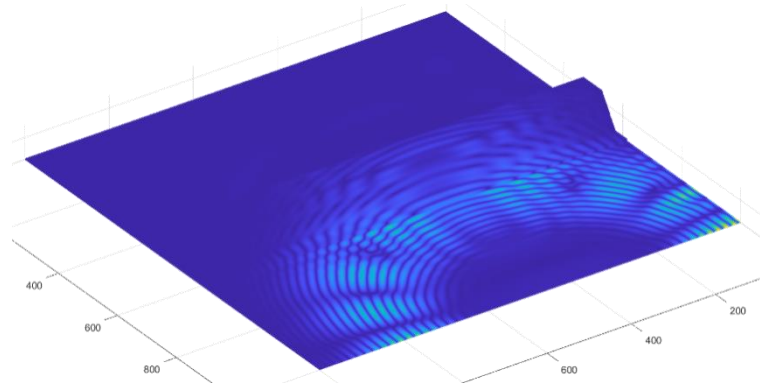
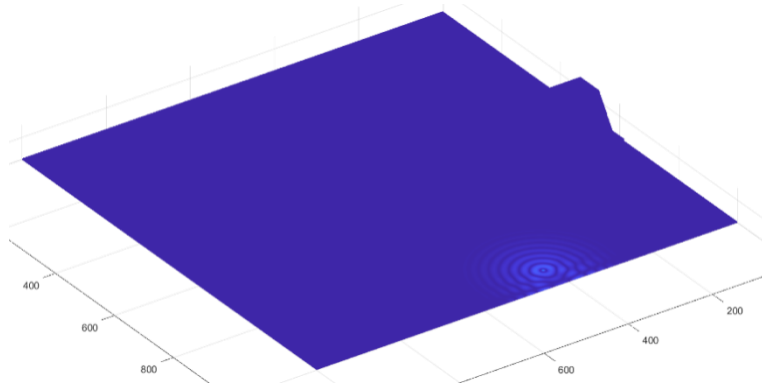
SHM plate – $f_c = 40$ kHz, Displacement amplitudes (normalized)

$t = 0.7E-4$ s

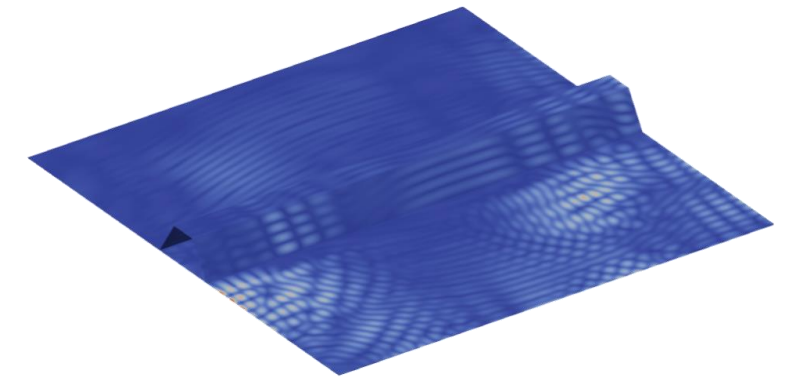
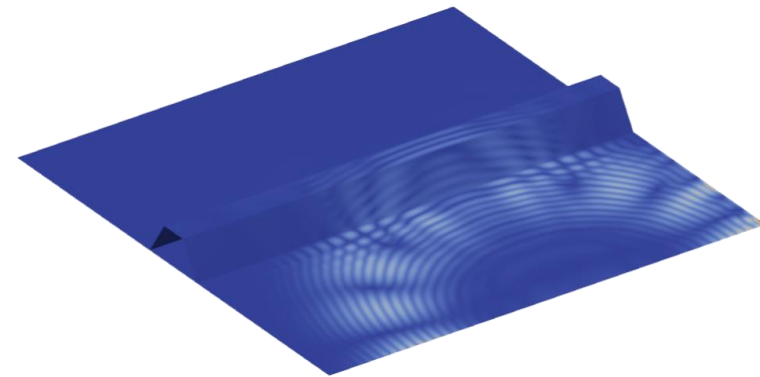
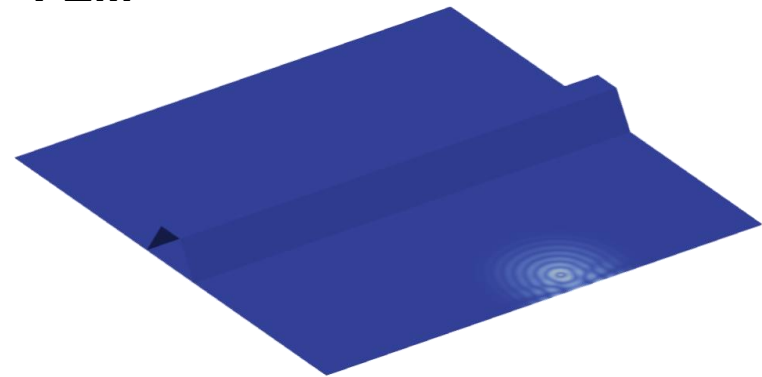
$t = 2.1E-4$ s

$t = 3.5E-4$ s

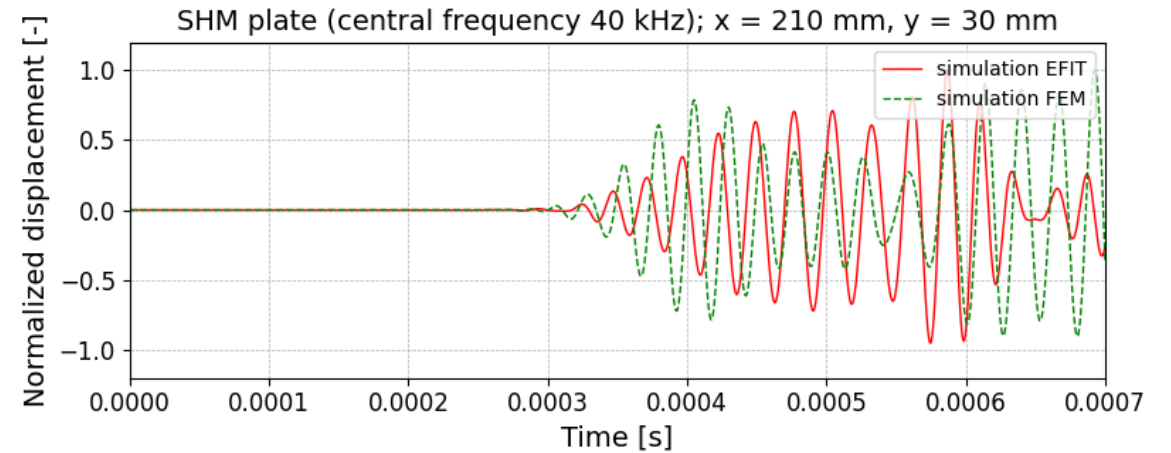
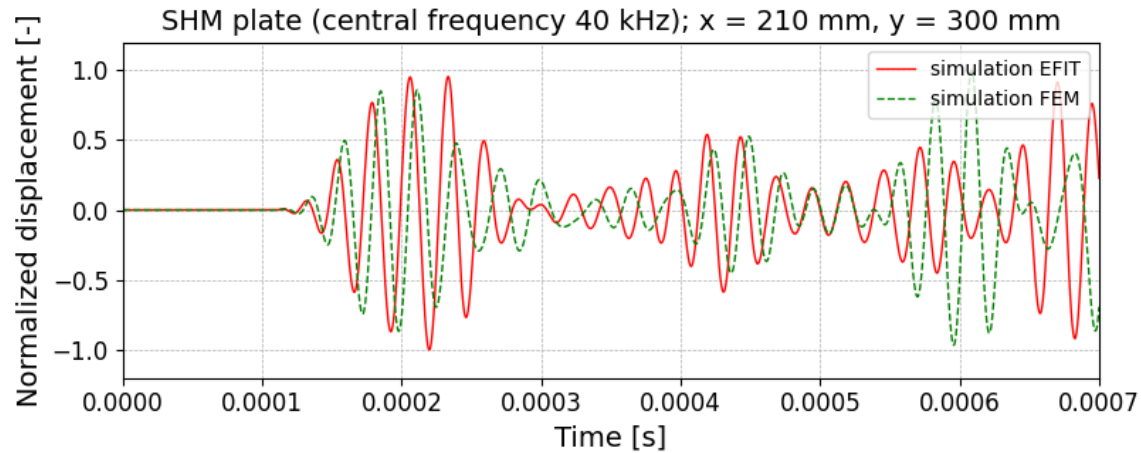
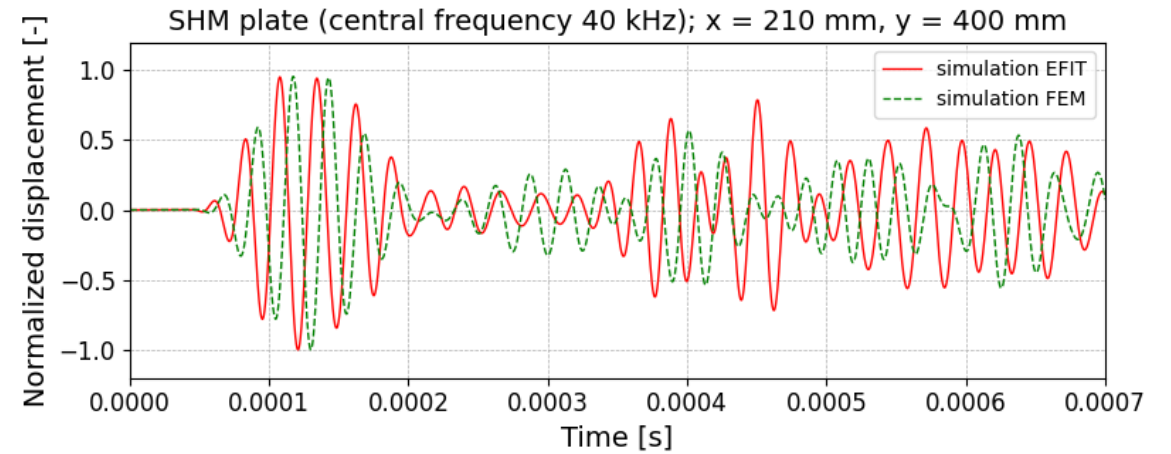
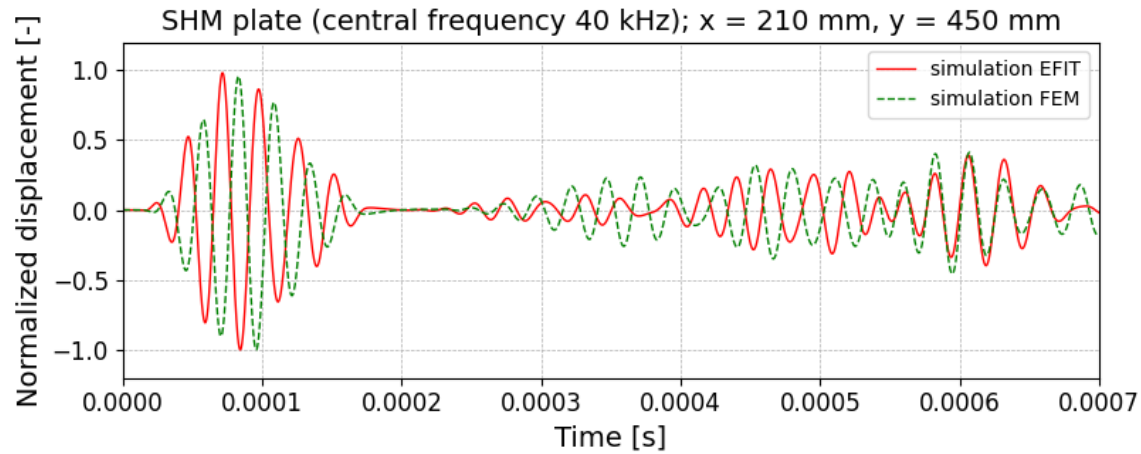
EFIT



FEM

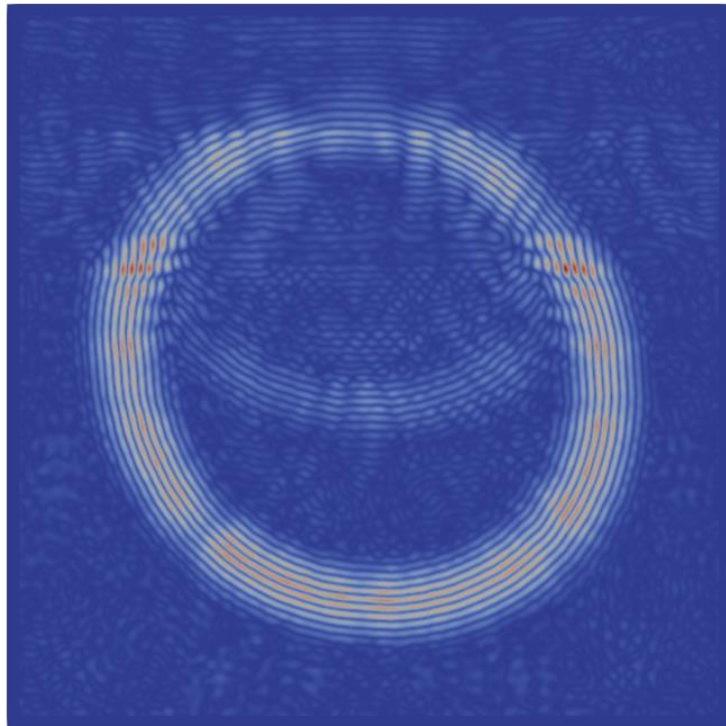


SHM plate – $f_c = 40$ kHz, Out-of-plane displacement in path T3 – T9

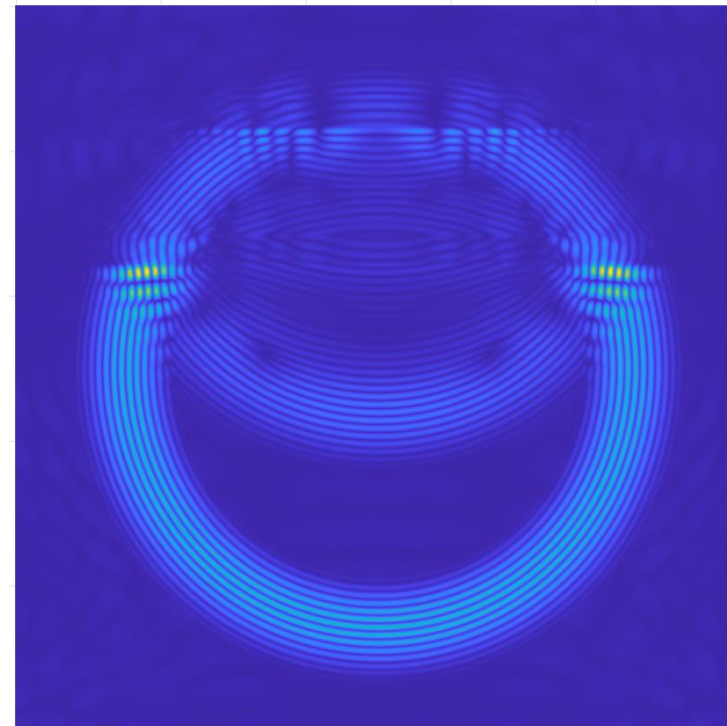


Wave field plate – $f_c = 100$ kHz, Out-of-plane velocity amplitudes

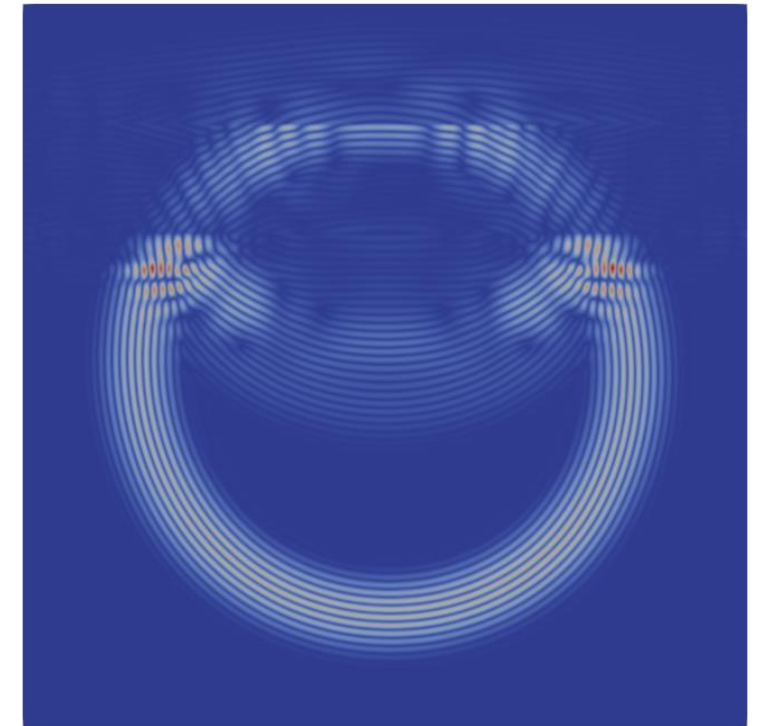
Scanning laser doppler
vibrometry [KUDELA2022]



Simulation (EFIT)



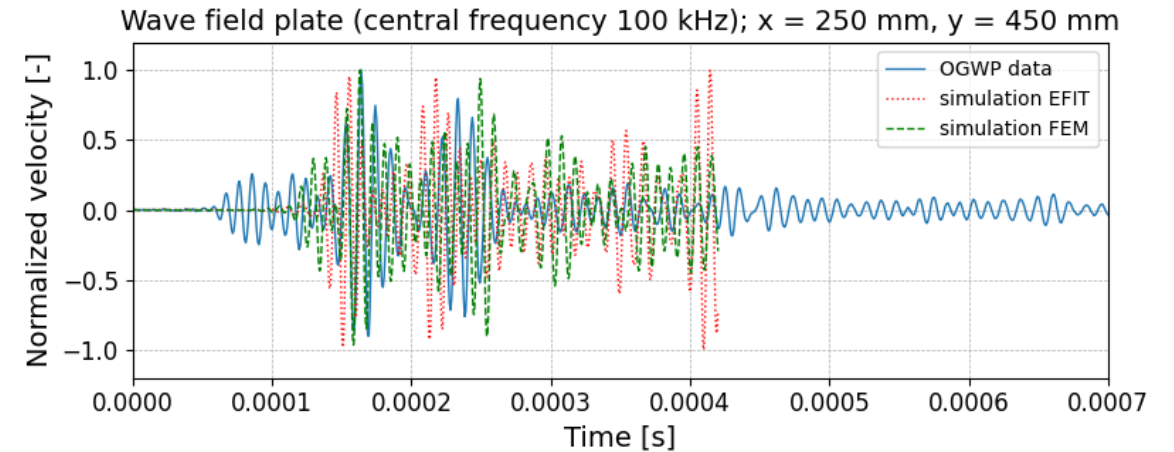
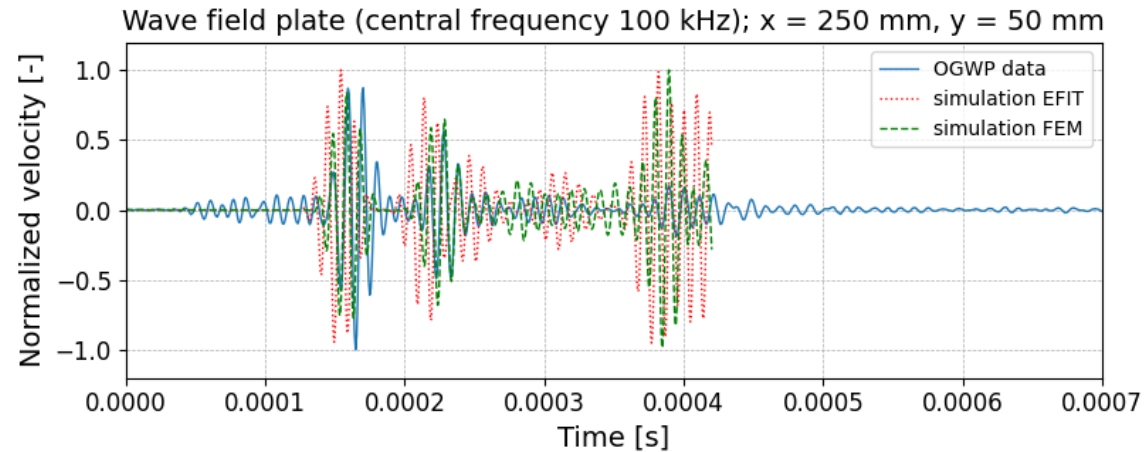
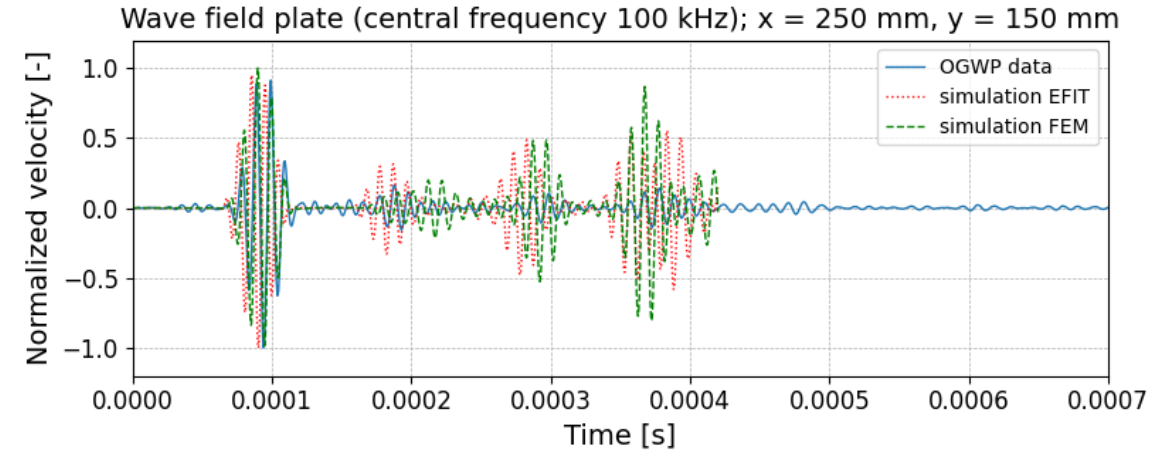
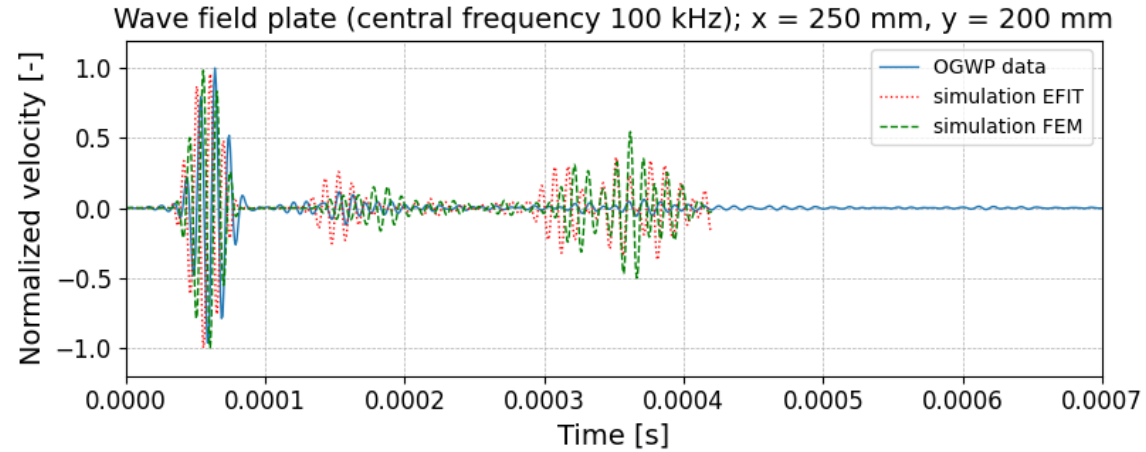
Simulation (FEM)



Normalized magnitudes of velocity at $t = 1.4E-4$ s



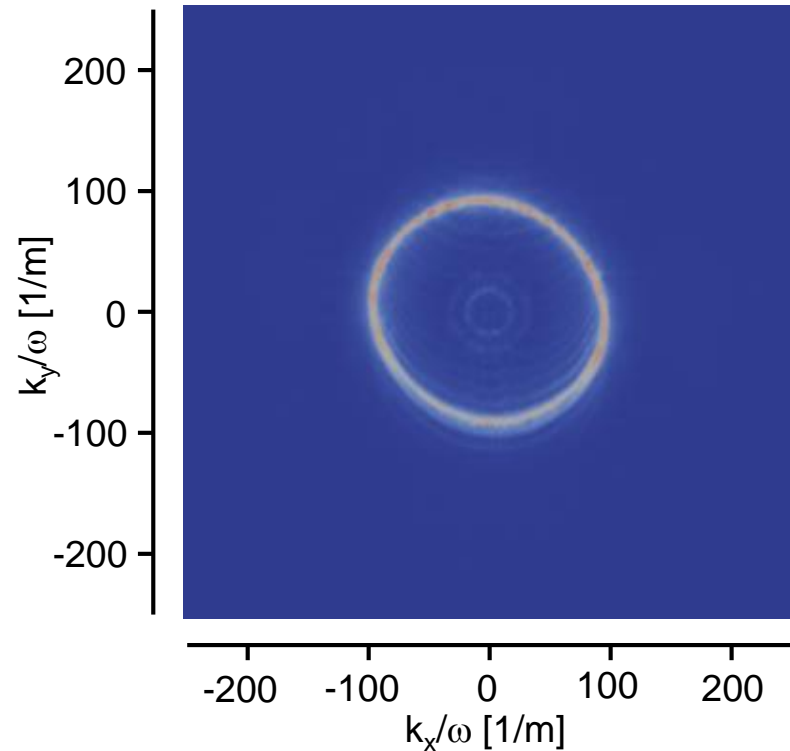
Wave field plate – $f_c = 100$ kHz, out-of-plane velocity



Wave field plate – $f_c = 100$ kHz, Experimental out-of-plane vs. simulation radial in-plane wavenumber profiles

- Transformation from time-space domain to frequency-wavenumber domain by applying 3DFFT

Data from scanning laser doppler vibrometry [Kudela2022]



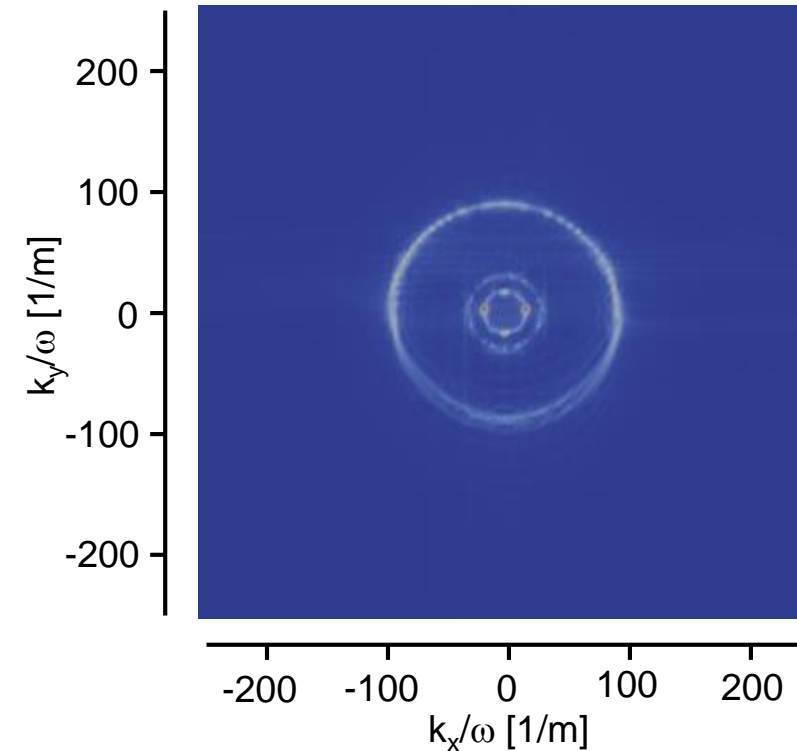
$f = 100$ kHz

A_0 - mode

SH - mode

S_0 - mode

Data from simulation (FEM)

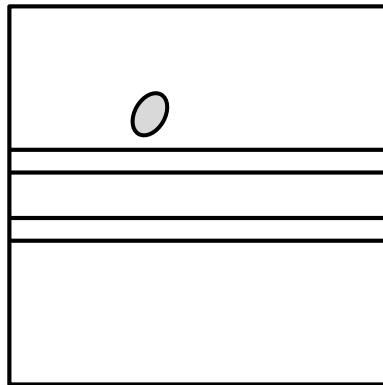


SHM plate – $f_c = 40$ kHz, Artificial reference damage

- Elliptical steel disk of 1 mm thickness and different axes sizes fixed on plate's surface
- Elliptical shape corresponds to the characteristics of impact damages on composites



[MOLL2020]

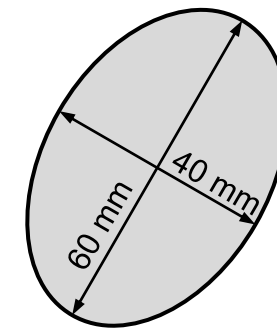
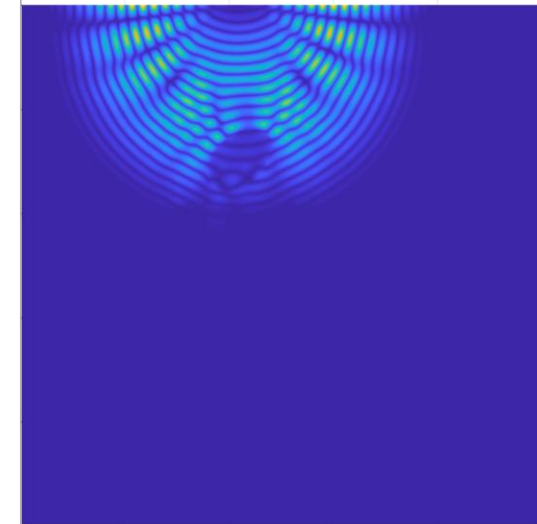


FEM

$t = 1.5E-4$ s



EFIT



Conclusions

- A concept to simulate the propagation of Ultrasonic Guided Waves in **shell-like composite structures** was developed by introducing the Elastodynamic Finite Integration Technique (EFIT) and finite shell-elements (FEM).
- Time-transient simulations were performed regarding **different excitation frequency levels**.
- Test structures and related data sets coming from the **Open Guided Waves Project** (wave field plate and SHM plate with stringer) were taken into account.
- **3DFFT** was used to transform results from space-time domain to wavenumber-frequency domain.
- For further visualization purposes **Matlab** as well as the DLR-tool **vampire** in combination with ParaView was used.

Vampire – Versatile mesh and postprocessing interface (https://gitlab.dlr.de/fa_sw/vampire)



Further steps

- Improvement of b2000++pro finite shell-elements regarding Lamb waves simulation purposes (linear varying displacement in **out-of-plane direction**)
- Simulation performance improvement by using CARA capabilities in **parallelization** (CARA – Cluster for Advanced Research in Aerospace – TU Dresden / ZIH)
- Augmenting model complexity
 - ✓ Geometry and discretization (elements and time steps)
 - ✓ Physical phenomena and interactions → **imperfections, damages**
- **Model assisted** probability of detection



References

- [BATHE2002] BATHE, K. J.: *Finite-Elemente-Methoden*. Springer-Verlag Berlin Heidelberg, (2002).
- [KUDELA2022] KUDELA, P.; RADZIENSKI, M.; MOIX-BONET, M.; WILLBERG, CH.; LUGOVTSOVA, Y.; BULLING, J.; TSCHÖKE, K.; MOLL, J.: *Dataset on full ultrasonic guided wavefield measurements of a CFRP plate with fully bonded and partially debonded omega stringer*. Data in Brief, (2022).
- [MOLL2020] MOLL, J.; KEXEL, CH.; KATHOL, J.; FRITZEN, C.-P.; MOIX-BONET, M.; WILLBERG, CH.; RENNOCH, M.; KOERDT, M.; HERRMANN, A.: *Guided Waves for Damage Detection in Complex Composite Structures: The Influence of Omega Stringer and Different Reference Damage Size*. Applied Science, 10, 3068, (2020).
- [REDDY1999] Reddy, J. N.: *Theory and Analysis of Elastic Plates*. TAYLOR & FRANCIS, Philadelphia, (1999).
- [SCHUBERT2004] SCHUBERT F.: *Numerical time-domain modeling of linear and nonlinear ultrasonic wave propagation using finite integration techniques - theory and applications*, Ultrasonics, 42, (2004).
- [TSCHÖKE2018] TSCHÖKE, K.; GRAVENKAMP, H.: *On the numerical convergence and performance of different spatial discretization techniques for transient elastodynamic wave propagation problems*, Wave Motion, 82, (2018).
- [WILLBERG2013] WILLBERG, CH.: *Development of a new isogeometric finite element and its application for Lamb wave based structural health monitoring*. Ph.D. thesis, Otto-von-Guericke-University of Magdeburg, Magdeburg, (2013).
- [WILLBERG2015] WILLBERG, CH.; DUCZEK, S.; VIVAR-PEREZ, J. M.; AHMAD, Z. A. B.: *Simulation Methods for Guided Wave-Based Structural Health Monitoring: A Review*. Applied Mechanics Reviews, 67, (2015).



Thank You!

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