

Methodology to simulate veneer-based structural components for static and crash load cases

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Knowledge for Tomorrow



DLR – Overview

Locations

- Approx. 8000 employees across 33 institutes and facilities at 16 sites
- Offices in Brussels, Paris, Tokyo and Washington

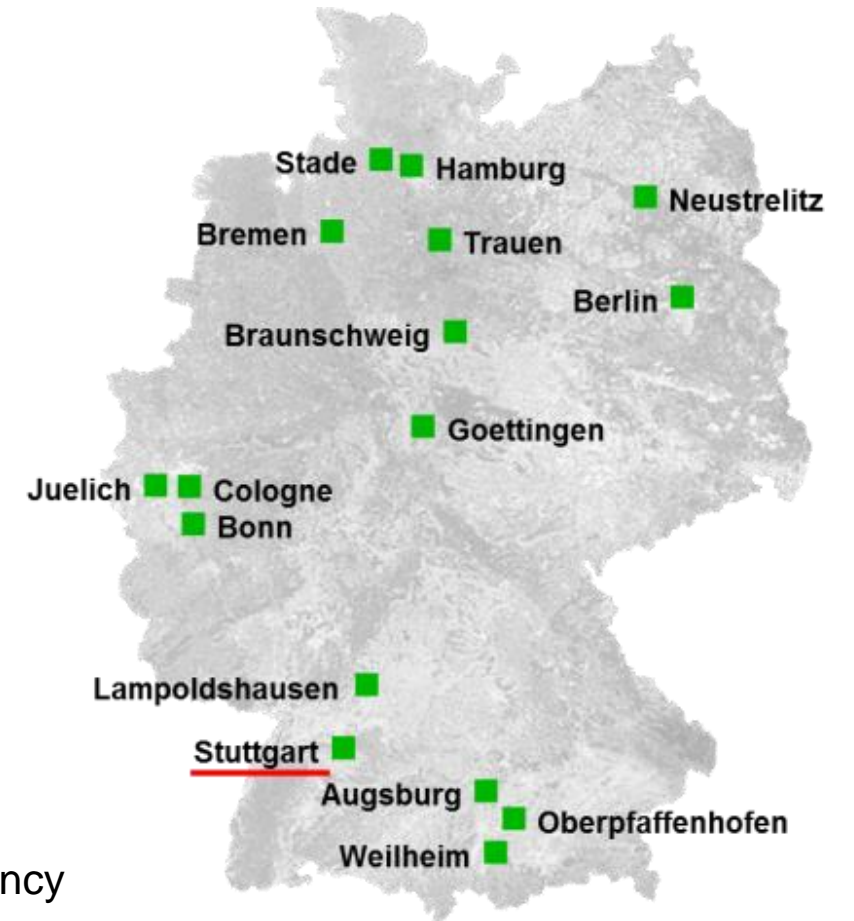


Research Areas

- Aeronautics
- Space Research and Technology
- **Transport**
- Energy
- Defence and Security

In addition

- Space Administration
- Project Management Agency



Outline

- Introduction
- Properties of veneer-based components
- Modelling of veneer-based layered composites
 - Material Model *MAT_054
 - Layered composites
- Simulation of generic veneer-based components
 - Calibration of intralaminar behavior
 - Validation of simulation and material model
 - Expansion of dynamic behavior
- Summary and outlook



Introduction

Project “For(s)tschritt”

Project duration:

- March 2017 – August 2020 (3.5 years)

Project content:

- **Development of veneer-based hybrid materials for vehicle structures**
 - Road vehicles
 - Rail vehicles
- Development, testing and **simulation**
- Demonstrators
(Front Door, Train Door, Train Side Panel)

Goal

- Qualification of wood for the usage in vehicle structures

For^(s)tschritt 

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Introduction

Motivation to use beech-based materials in structural components

Requirements:

- Lightweighting due to emission guidelines
- Reduction of GWP during production phase

Advantages of beech:

- Very good specific material characteristics especially for bending load cases (Ashby [1])
- Ecological with low CO₂-emissions
- High availability and renewable
- Economical
- High flexibility regarding production technologies

Challenges:

- Scatter of material characteristics
- Protection from environmental influences needed

Technical and economic comparison of materials [2] [3]

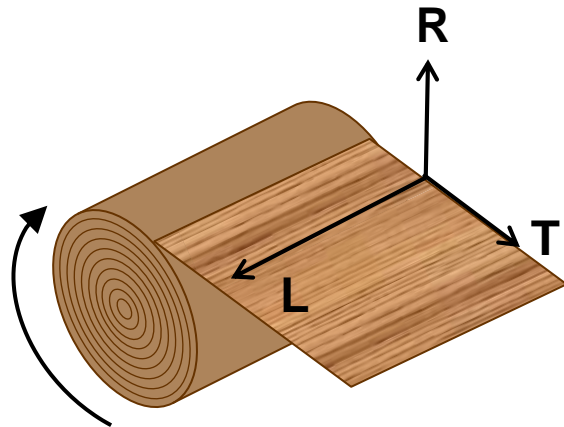
Parameter	Density	Young's Modulus	Ultimate strength	Spez. stiffness	Spez. strength
Symbol	ρ	E	UTS	$\frac{E^{1/2}}{\rho}$	$\frac{UTS^{2/3}}{\rho}$
Unit	[g/cm ³]	[MPa]	[MPa]	$\frac{MPa^{1/2}}{g/cm^3}$	$\frac{MPa^{2/3}}{g/cm^3}$
Aluminum	2,30-2,80	70.000	45-500	95 - 115	5 - 27
Beech	0,54-0,91	14.350(I)	100-140(I)	132 - 222	24 - 50
CFRP	~1,50	~140.000(I)	~1.700(I)	~250	~95
GFRP	~2,00	~44.500(I)	~1.100(I)	~105	~53
Magnesium	~1,74	45.000	100-300	122	12 - 26
Steel	~7,85	210.000	340-1.800	58	6 - 19



Properties of veneer-based components

Veneer:

- Produced by peeling of stem
- Orthotropic material properties
- Scatter in material characteristics



Longitudinal (L), Tangential (T)
and Radial direction of veneer

Veneer-based layered composites:

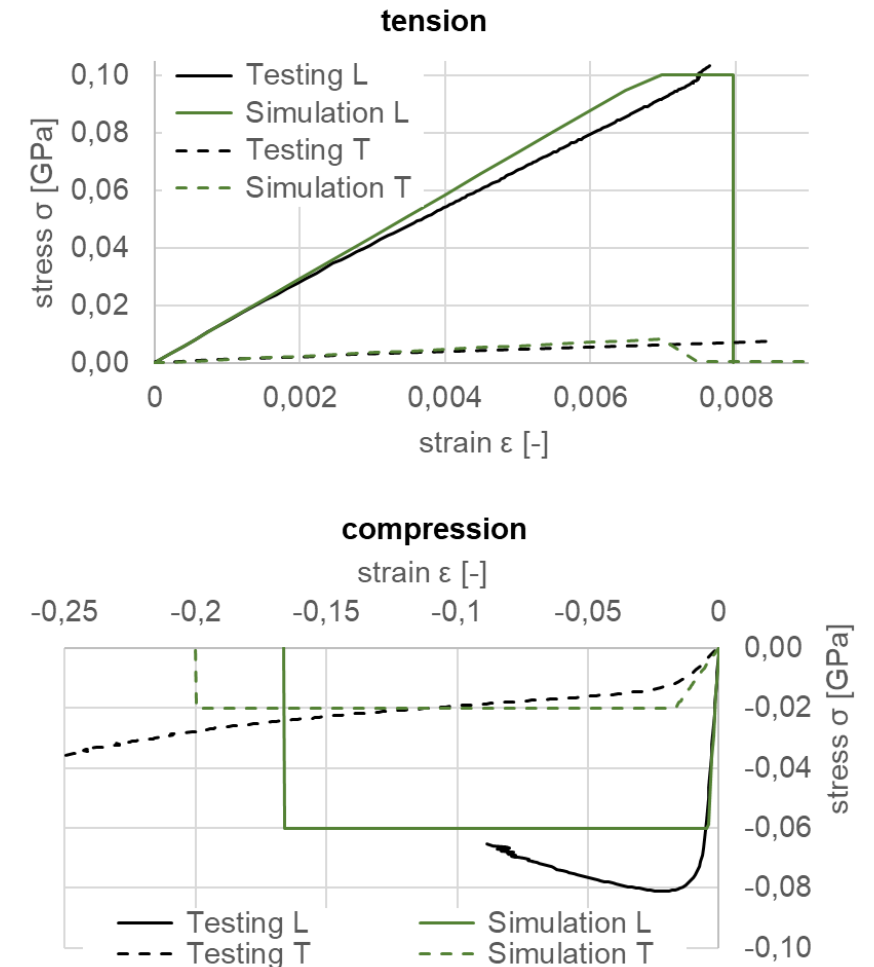
- Multiple veneer layers glued together
- Free selection of orientation, thickness, material of each layer
- Reduction of scatter:
 - Preselection of veneer with little to none imperfections
 - Statistically averaging due to usage of multiple veneers



Modelling of veneer-based layered composites

Material Model

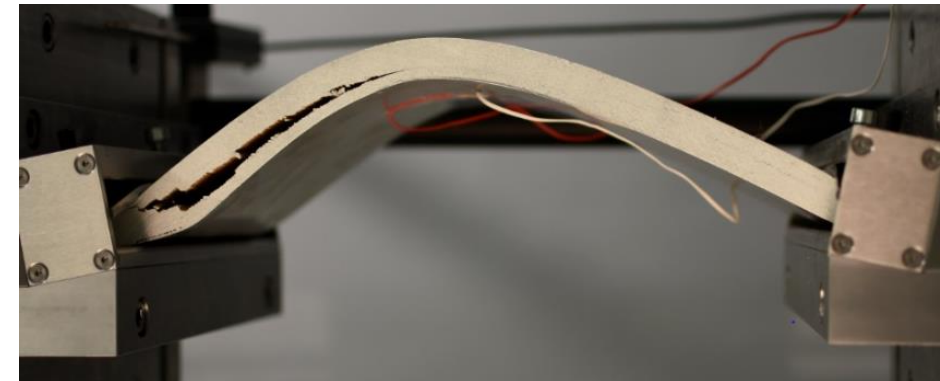
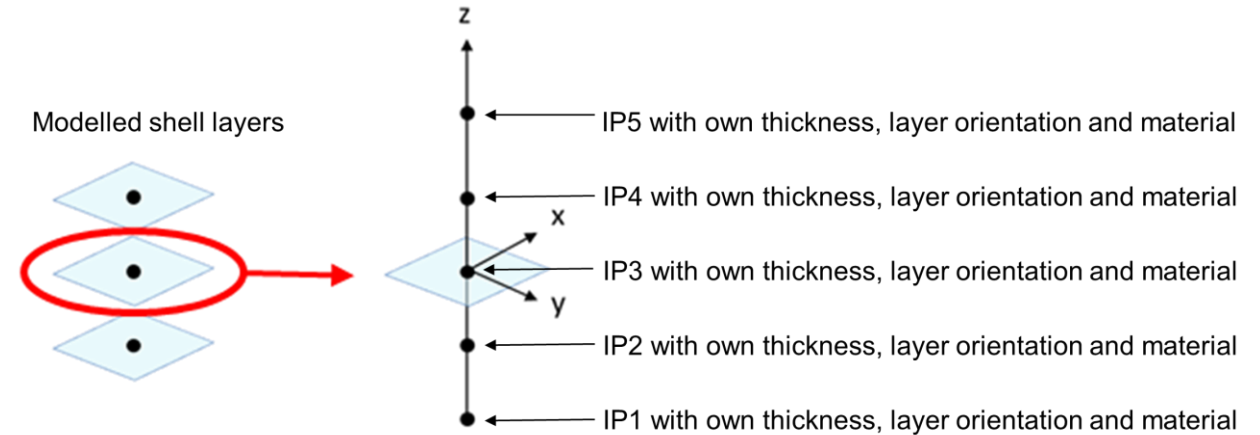
- FEM-Solver LS-Dyna
- **MAT_ENHANCED_COMPOSITE_DAMAGE (*MAT054) [4]*
 - Orthotropic material model
 - Failure Modell after Chang/Chang or Hashin
- Comparison with data from testing shows:
 - UTS and failure strain in tension \parallel met quite well
 - UTS and failure strain in tension \perp met quite well
 - UTS in compression \parallel met quite well
 - UTS in compression \perp met quite well
 - But: failure strain in compression \parallel and compression \perp is a tradeoff



Modelling of veneer-based layered composites

Layered Composites

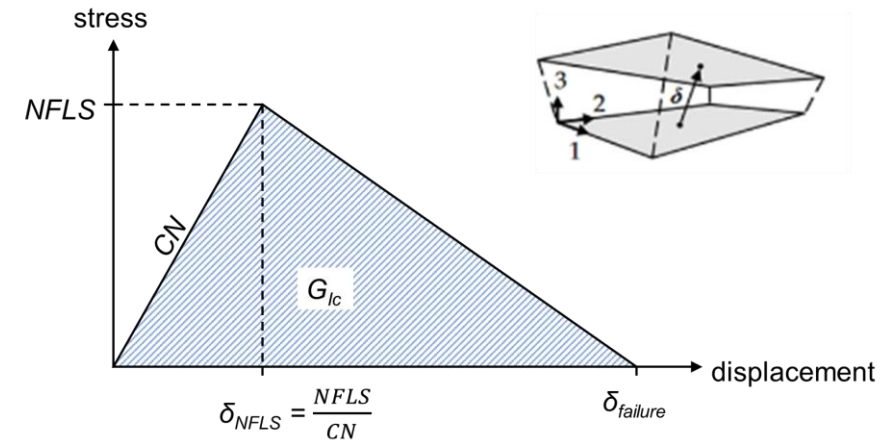
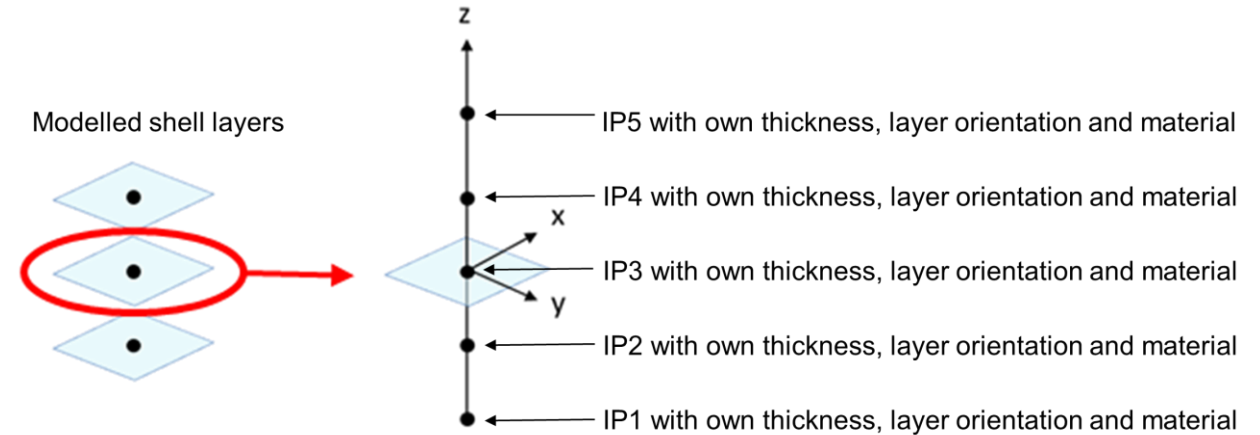
- Delamination in the middle due to exceeding of shear stress possible
- Component is discretized with 2 Shell-layers or more
- Each shell layer represents a sub-laminate
 - Characteristics from true layers stored in numerical integration points (IP)



Modelling of veneer-based layered composites

Layered Composites

- Delamination in the middle due to exceeding of shear stress possible
- Component is discretized with 2 Shell-layers or more
- Each shell layer represents a sub-laminate
- Characteristics from true layers stored in numerical integration points (IP)
- Delamination based on a bilinear traction-separation law



Simulation of veneer-based layered composites

Calibration of intralaminar behavior – setup

- **Generic structure comparable to a door impact beam:**

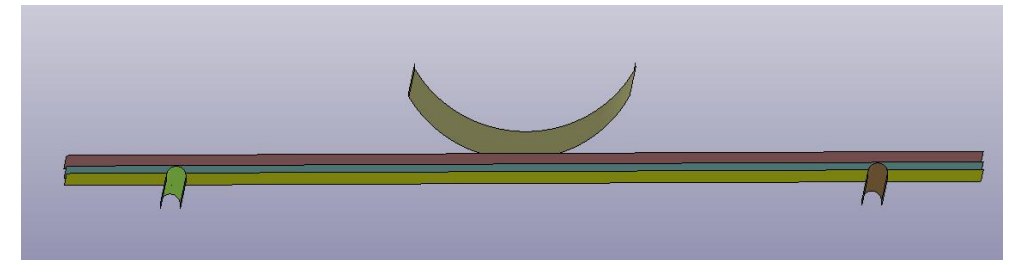
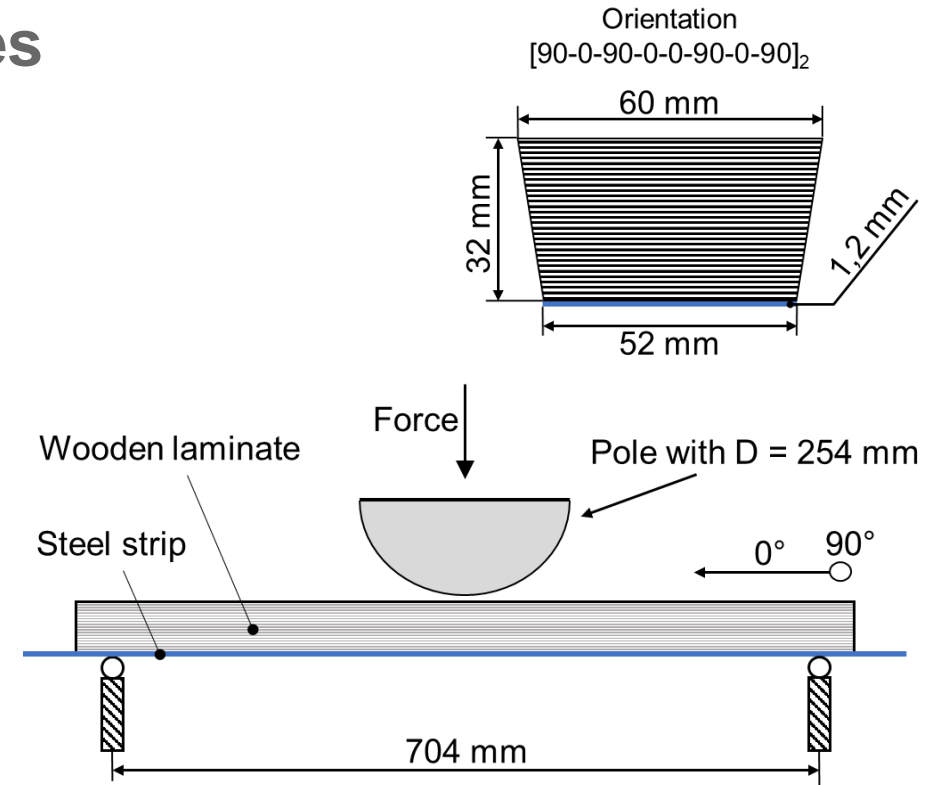
- Wooden part made of 16x 2,0 mm veneer layers
- Steel strip with 1,2 mm thickness

- **Testing:**

- Quasi-static three-point bending flexural test
- Usage of 10" pole (254 mm diameter)
- Force measurement at pole
- Displacement measured at pole

- **Simulation:**

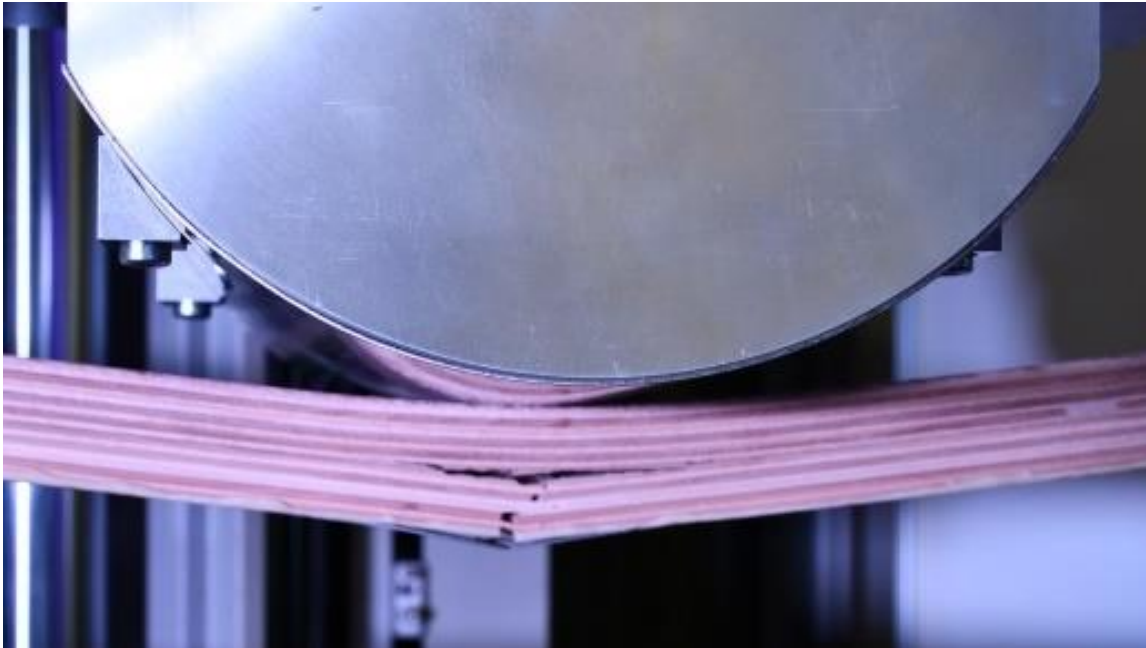
- 2 shell layers for wooden part and 1 shell layer for strip
- Material model as developed
- **Aim: Calibration of intralaminar behavior (delamination)**



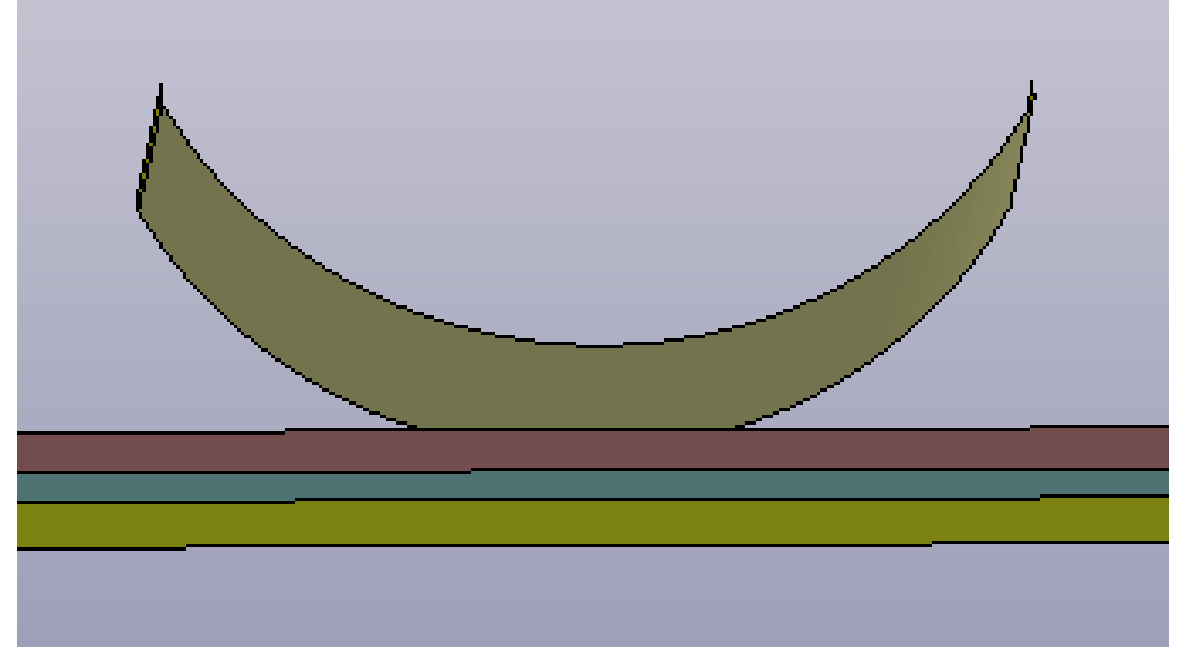
Simulation of veneer-based layered composites

Calibration of intralaminar behavior – comparison

Testing

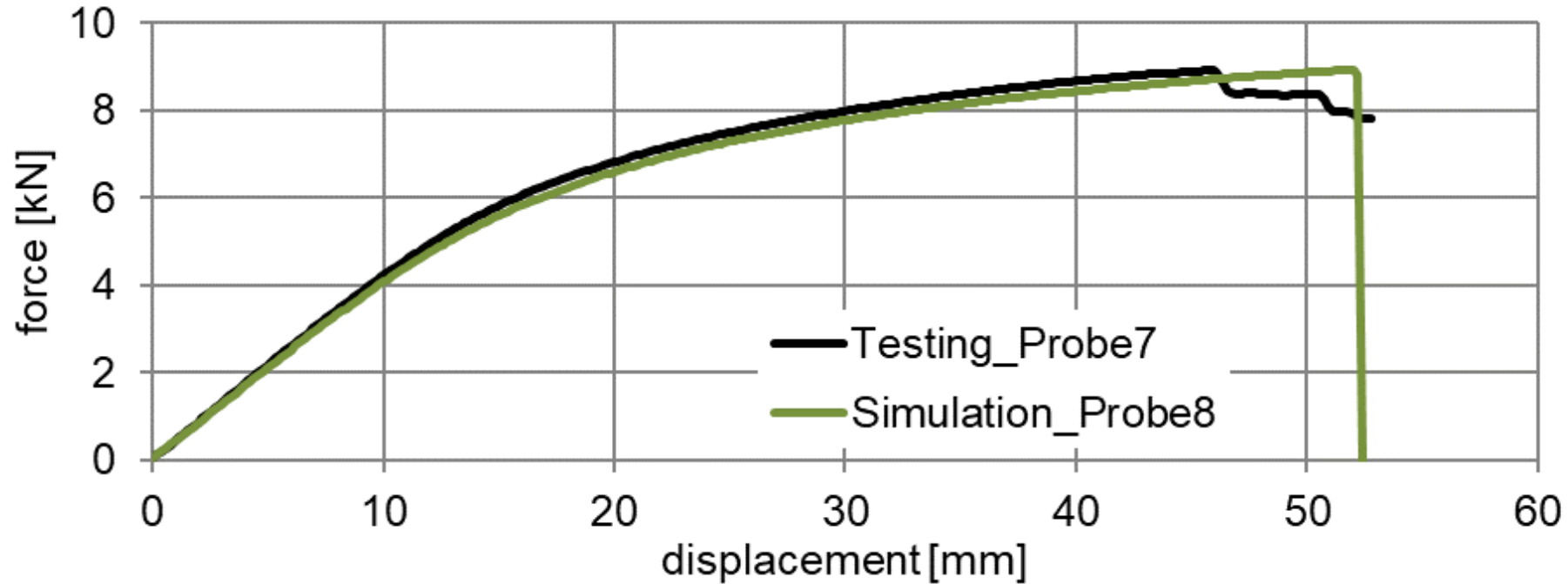


Simulation



Simulation of veneer-based layered composites

Calibration of intralaminar behavior – results



Calibration of intralaminar failure:

- similar course of the force-displacement curve
- fracture pattern could be mapped to good approximation



Simulation of veneer-based layered composites

Validation of simulation and material model – setup

- **Generic beam:**

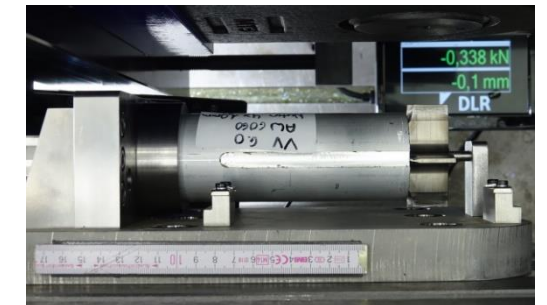
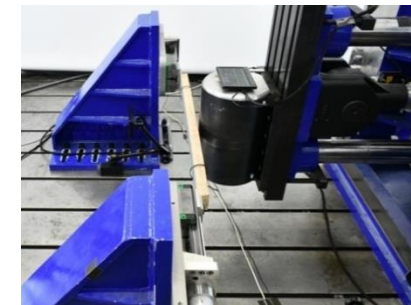
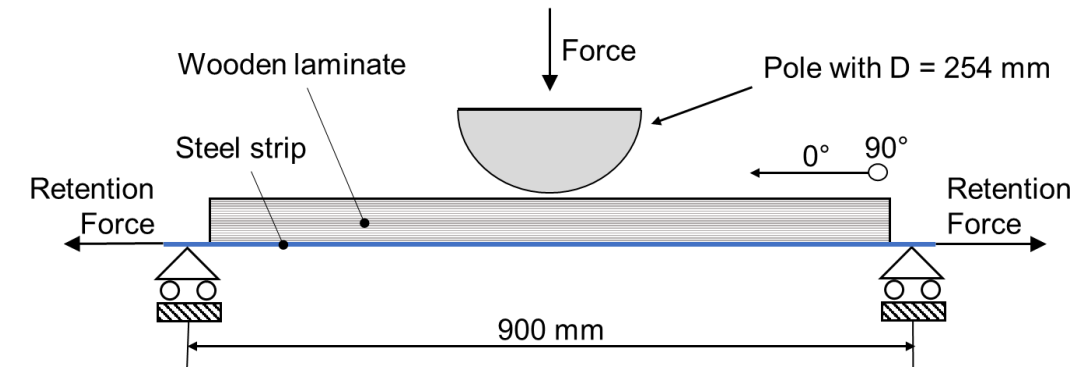
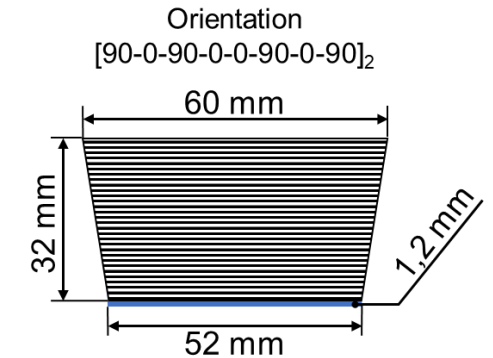
- Same geometry
- Different orientation of layers

- **Testing:**

- Quasi-static three-point bending flexural test
- Usage of absorber units (AU) for retention force
- Force measurement at pole and at AU
- Displacement measurement at pole and at AU

- **Simulation:**

- 2 shell layers for wooden part and 1 shell layer for strip
- Material model and sim. model as developed
- **Aim: Validation of simulation and material model**



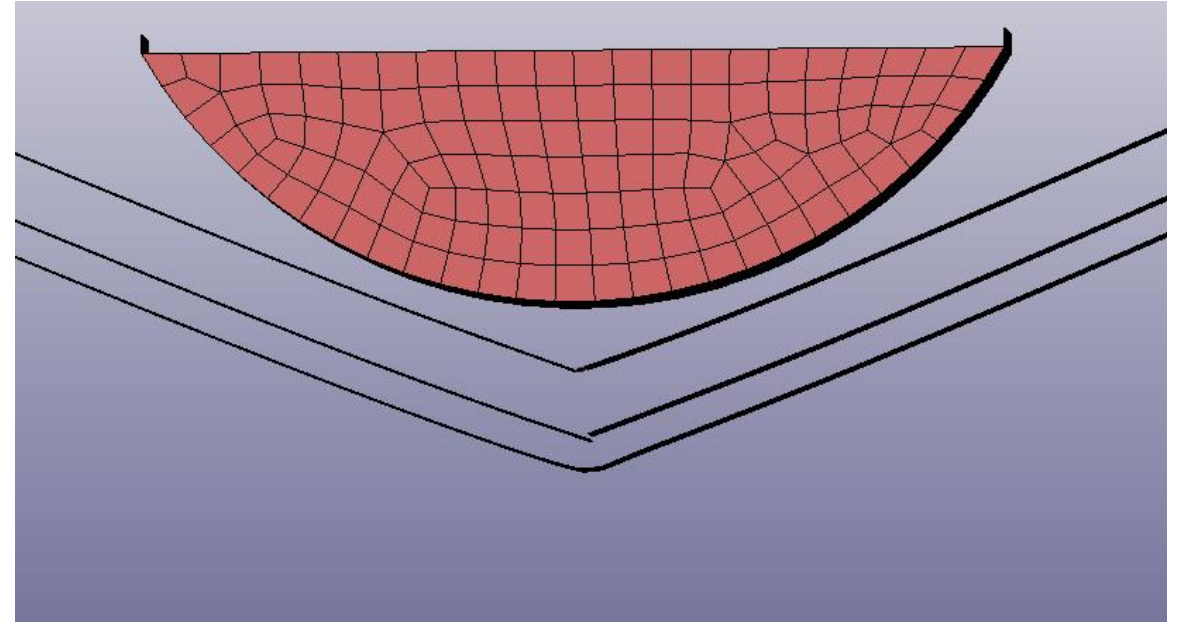
Simulation of veneer-based layered composites

Validation of simulation and material model – comparison

Testing

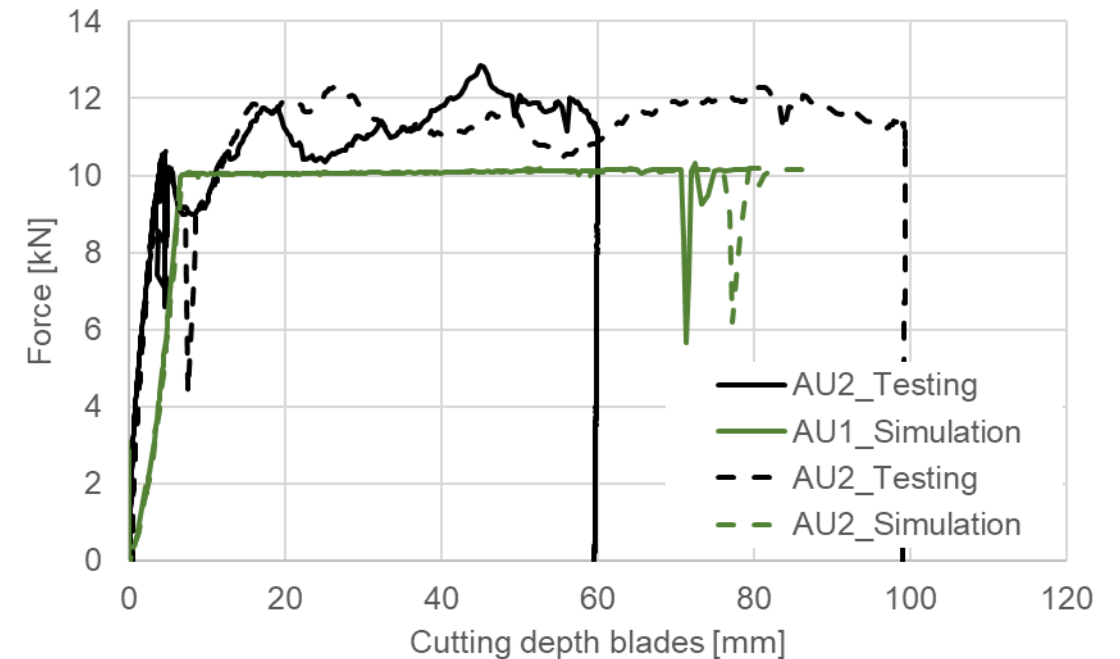
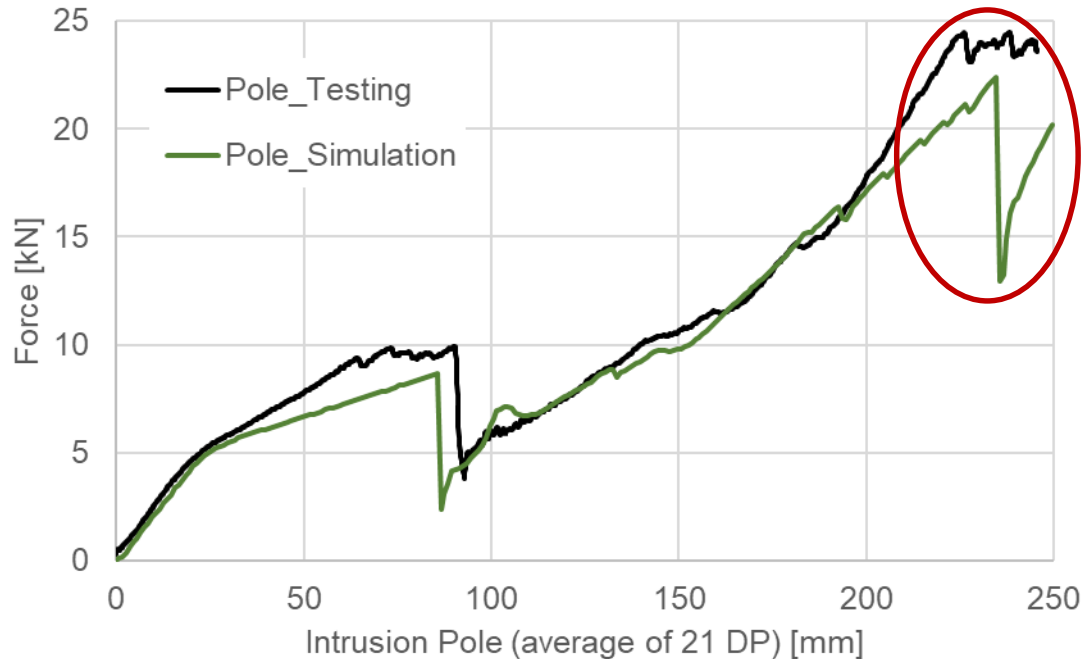


Simulation



Simulation of veneer-based layered composites

Validation of simulation and material model – results



- qualitative course is similar
- absolute values in simulation tend to be smaller
- Sum of displacement is similar
- Kinematics in good approximation to testing

→ Validated simulation and material model



Simulation of veneer-based layered composites

Expansion of dynamic behavior – setup

- **Generic beam:**

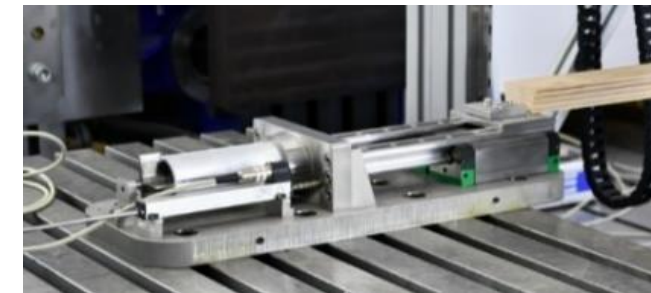
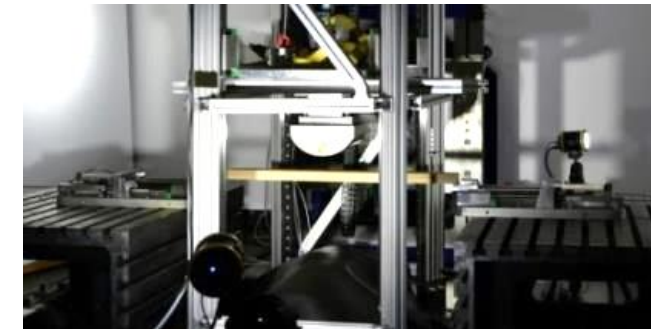
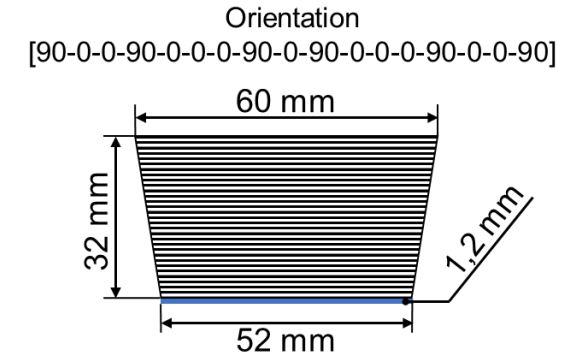
- Same geometry
- Different orientation of layers

- **Testing:**

- High-speed three-point bending flexural test
- Test setup comparable to validation
- Usage of drop tower in combination with AU
- Force measurement at pole and at AU
- Displacement measurement at pole and at AU

- **Simulation:**

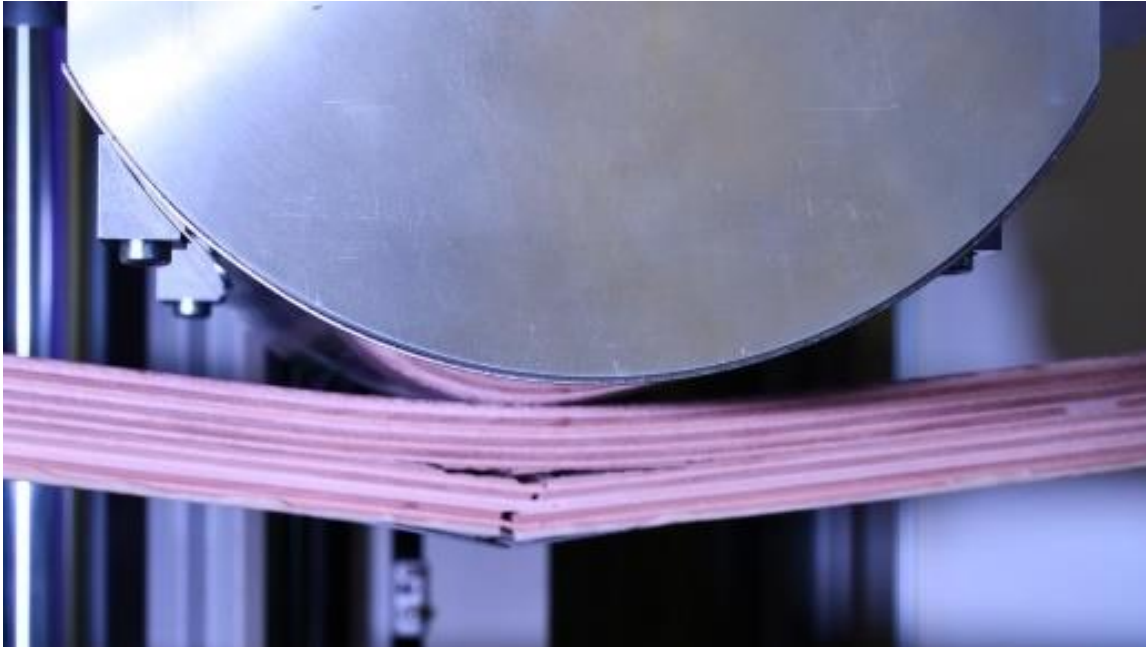
- 2 shell layers for wooden part & 1 shell layer for strip
- Material model with dynamic behavior
- **Aim: Expansion to dynamic material model**



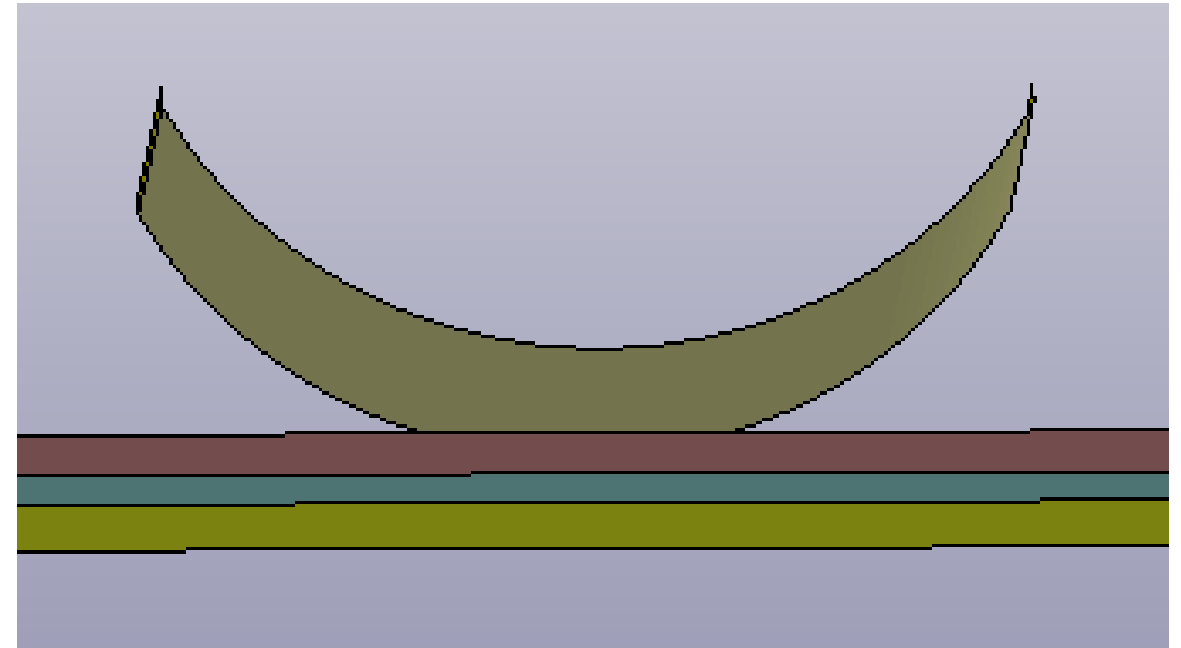
Simulation of veneer-based layered composites

Expansion of dynamic behavior – comparison

Testing

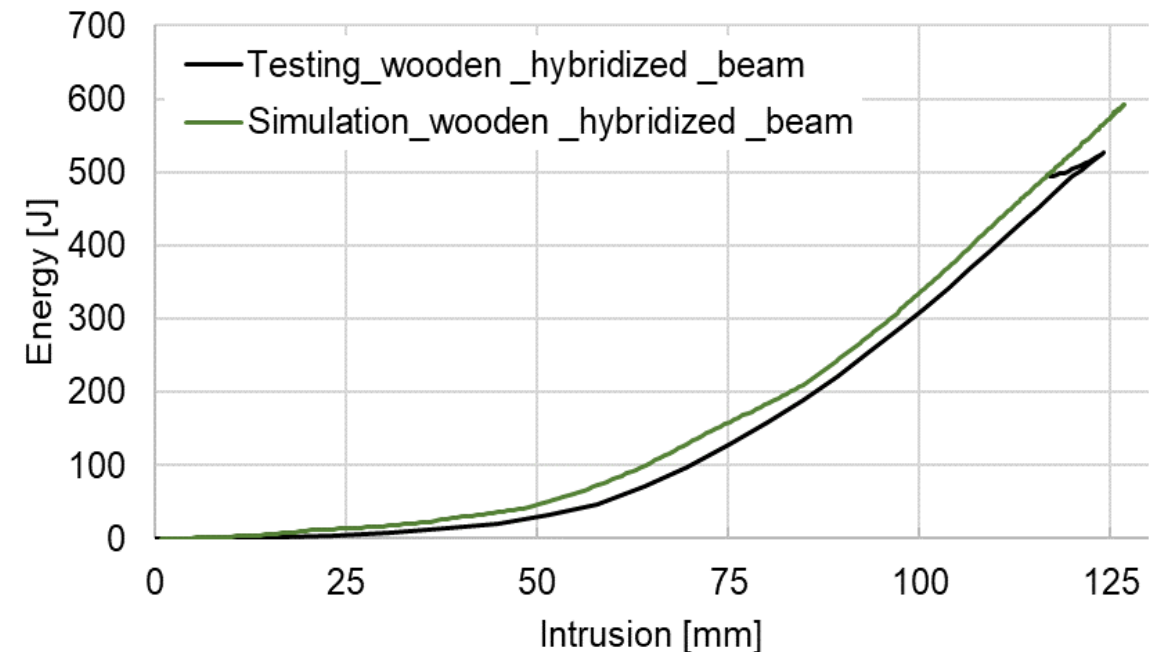
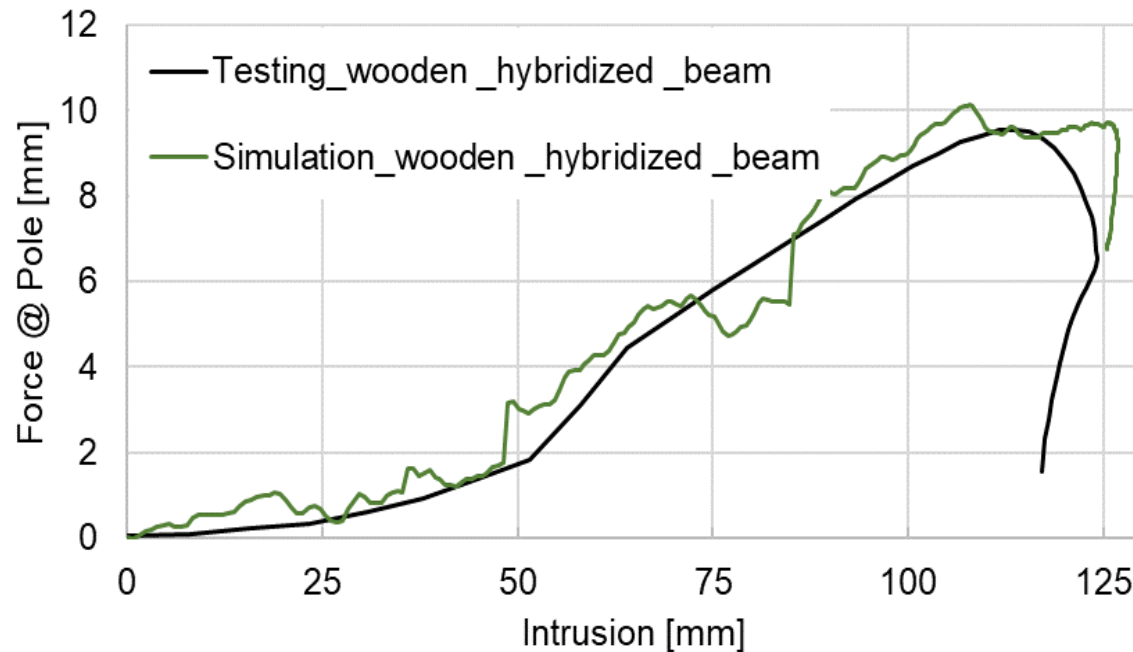


Simulation



Simulation of veneer-based layered composites

Expansion of dynamic behavior – results



- qualitative course is quite similar
- absolute values in simulation tend to be higher
- Higher absorbed energy in simulation (~15%)
- No spring back in simulation

→ **Successful expansion of dynamic behavior**



Summary and outlook

Summary

- Material model created for quasi-static and high-speed loads
- Intralaminar behavior modeled
- Modelling approach with layered shells validated

Outlook

- Refinement of dynamic parameters in order to better match absorbed energy
- Simulation of a specific assembly in a real use case, like Euro-NCAP Side Pole Impact
- Comparison of specific assembly with a reference impact beam to determine potential of veneer-based structure

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Further information
about the project

For(s)tschritt:
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Literature

- [1] Ashby, M.F.: “Materials Selection in Mechanical Design“, 3rd edition, Elsevier, 2004
- [2] Bergman, R., Alanya Rosenbaum, S.: “Cradle-to-gate life cycle assessment of laminated veneer lumber production in the United States”, Forest Products Journal, vol.67, November 2016
- [3] Klein, B.: Lightweight-Construction. Wiesbaden, 2011
- [4] LS-DYNA, Keyword user's manual, Volume II: material models, Livermore Software Technology Corporation (LSTC), 2013.

