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Size Effect in Intermetallics: Experiments and Numerics

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Knowledge for Tomorrow

- 1. Motivation, background and overview
- 2. Experimental studies
- 3. Gradient-extended plasticity damage theory
- 4. Numerical simulations
- 5. Stochastic size effect
- 6. Conclusion & outlook





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Motivation

- Aviation accounts for ≈ 2.1% of global CO₂ emissions - roughly equivalent to Germany's total emissions^[1].
- EU aims to reduce its domestic CO₂ emissions by 80% by 2050 compared to 1990 levels ^[2].
- For aviation industry, goal is halving aviation CO₂ emissions relative to 2005 levels by 2050 ^[3].



[1] EPRS BRI(2017)603925_EN. [2] EUR-Lex - 52011DC0112 (2011) [3] EUR-Lex - 32017R2392.



Background

Why Intermetallics?



	σ_{UTS} [MPa]	σ_Y [MPa]	ho [g/cm ³]	σ_{UTS}/ ho [*]	E [GPa]	ε _f [%]	T _{oxid} [°C]
Stainless Steel AISI 304 ^[3]	505	215	8	63	200	70	850
Ni-based super alloy [4]	1500	1200	8.5	176	210	35	900
Titanium Aluminide ^[5]	600	450	3.8	158	170	<1	1000

[2] Appel et al. (2011) [3] asm.matweb.com [4] Pollock & Tim (2006) [5] Kim (1989)



Overview

Weakest link Weibull model

• Failure probability^[1] is written as

 $P = 1 - \exp\{-\int \mathfrak{s}(\sigma)dV\}$ $\mathfrak{s}(\sigma) = \frac{1}{V_0} \cdot \left(\frac{\sigma}{\sigma_0}\right)^m$

- Empirical relation for stochastic failure strength
- Empirical relation for volume <u>effect</u>
- FE implementation tells if not where failure occurs!
- Scaling may not be applicable to all materials e.g. TiAl, CMC
- Failure probability in terms of fracture energy more suitable for materials with SSY.



[1] Weibull (1951)



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Titanium Aluminide alloy as test material

- Titanium Aluminide (TiAl) Typ GE48-2-2
- Titanium Aluminium Chromium Niobium
 48:48:2:2 at% and 59.8:32.8:2.55:4.85 wt%
- TiAl 48-2-2 remelt stocks are manufactured by Vacuum Arc Remelting (VAR)
- Manufacturer provided properties:

Fracture toughness	15 – 18 MPa√m
Fracture strength	450 – 500 MPa
Density	3.97 g/cm ³
Hardness	285 HV10
Young's Modulus	160 GPa
Yield strength	430 MPa



Duplex with lamellar colonies





Specimen preparation and test configuration

- Three point bending tests are performed
- Specimens are cut using the Electrical Discharge Machining process.
- Dimensional tolerance and surface finish as per ASTM E399.
- For each specimen, force and DIC data are recorded. Strain rate fixed at 0.5% min⁻¹.
- Three geometrically identical sizes are selected.

	Length (L)	Height (H)	Width (B)	Count (n)
Size 1	25.2	6.0	6.0	11
Size 2	37.8	9.0	6.0	14
Size 3	50.4	12.0	6.0	4



All dimensions in mm





Force displacement data

Experiment data Upper and lower bounds for elastic modulus are computed using analytical relation

 $u = \frac{FL^3}{48EI} + \frac{FLH^2}{40GI}$

Averaged data

95 % confidence intervals are computed as

$$CI = \pm 1.96 \frac{sd}{\sqrt{n}}$$





Distribution in fracture stress



• Cumulative density function for failure probability is obtained from sorted sample of size <u>N</u> as ^[1]

$$P_f = \frac{i - 0.3}{N + 0.4}$$

• Analytical relation for flexural stress is only valid in linear elastic region

$$\sigma = \frac{3FL}{2BH^2}$$



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Phase field modeling of elastic-plastic fracture^[1]

• Define the space-time accumulated work density as

$$\mathcal{W} = \int_{V} \left(\int_{0}^{T} \boldsymbol{\sigma} : \dot{\boldsymbol{\varepsilon}} \, dt \right) dV = \int_{V} \left[\underbrace{\widehat{\mathcal{W}}(\underline{\mathfrak{C}})}_{\text{rate-indep.}} + \underbrace{\mathfrak{D}}_{\text{rate-dep.}} \right] dV$$

where $\mathfrak{C} = \{ \boldsymbol{\varepsilon}, \boldsymbol{\varepsilon}^{\mathrm{p}}, \alpha, \nabla \alpha, d, \nabla d \}$ is the constitutive set.

• Make the energetic-dissipative split

$$\widehat{W} = \underbrace{\psi^{e}(\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^{p}, d)}_{\text{stored energy}} + \underbrace{\mathcal{D}^{pf}(\alpha, \nabla \alpha, d, \nabla d)}_{\text{dissipation}}$$

• Stored energy takes the form

$$\psi^{e}(\boldsymbol{\varepsilon} - \boldsymbol{\varepsilon}^{\mathrm{p}}, d) = g(d)w_{e} = g(d)\left[\frac{\kappa}{2}\mathrm{tr}^{2}[\boldsymbol{\varepsilon}^{e}] + \mu\,\mathrm{tr}[\mathrm{dev}(\boldsymbol{\varepsilon}^{e})^{2}]\right]$$

• Dissipation due to gradient-plasticity and fracture takes the form

$$\mathcal{D}^{pf}(\alpha, \nabla \alpha, d, \nabla d) = g(d)w_p(\alpha, \nabla \alpha) + (1 - g(d))w_c + 2\frac{l_f\gamma(d, \nabla d)}{w_c}w_c$$



 ε , total strain ε^{p} , plastic strain α , equi. Plastic strain d, crack phase field

Phase field modeling of elastic-plastic fracture [1] ...

- Define a dissipation potential function $\widehat{V}(\dot{\mathfrak{C}}) = \sup_{f^{p}, r^{p}, f^{f} - r^{r}} \sup_{\lambda^{p}, \lambda^{f}} \begin{bmatrix} f^{p} : \dot{\boldsymbol{\varepsilon}}^{p} - r^{p} \dot{\alpha} + (f^{f} - r^{r}) \dot{d} - \lambda^{p} \phi^{p} - \lambda^{f} \phi^{f} \end{bmatrix} \begin{array}{c} f^{p} = -\partial_{\varepsilon^{p}} \psi^{e} \\ r^{p} = \delta_{\alpha} \mathcal{D}^{pf} \\ f^{f} = -\partial_{d} \psi^{e} \\ r^{r} = \delta_{d} \mathcal{D}^{pf} \end{array}$
- Define the global rate potential of coupled gradient-plasticity-gradient damage

$$\Pi = \int_{V} \left(\int_{t} \frac{d}{dt} \widehat{W}(\mathfrak{C}) + \widehat{V}(\dot{\mathfrak{C}}) \right) - \mathcal{P}_{ext}(\dot{\boldsymbol{u}})$$

- The global *minimization* of the *multi-field problem* yields the primary unknowns $\{\dot{u}, \dot{\alpha}, \dot{d}, \dot{\epsilon}^p\} = \operatorname{Arg}\{\inf_{\dot{u}, \dot{\alpha}, \dot{d}, \dot{\epsilon}^p}\}$
- Of particular interest is the Euler equation derived for the evolution of phase field fracture $\dot{d} = (1 - d)\mathcal{H} - [d - l_f^2 \Delta d] \qquad \text{where} \qquad \mathcal{H} = \max_{s \in [0, t]} \mathfrak{P} \ge 0$
- The dimensionless *crack driving state function* is

$$\mathfrak{P} = \left\langle \frac{w_e(\boldsymbol{\varepsilon}^e) + w_p(\alpha, \nabla \alpha)}{w_c} - 1 \right\rangle$$

[1] C. Miehe, F. Aldakheel, A. Raina (2016)

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Simulations of average 3PB response



Validation with DIC data



5.2E-04 4.2E-04

3.1E-04 2.1E-04

1.0E-04

0.0E+00

-1.0E-04 -2.1E-04

-3.1E-04 -4.2E-04 -5.2E-04

> 5.2E-04 4.2E-04 3.1E-04

2.1E-04

1.0E-04

0.0E+00

-1.0E-04 -2.1E-04

-3.1E-04 -4.2E-04 -5.2E-04



Fracture with crack phase field





We propose an empirical relation for the size effect

• Volume dependent critical fracture energy is proposed as

$$w_c = \frac{w_0}{\sqrt{\left(1 + \left(\frac{V}{V_0}\right)^{\frac{1}{3}}\right)}}$$

• Rearranging the terms, we get

$$\left(\frac{w_0}{w_c}\right)^2 = V_0^{-\frac{1}{3}} \cdot \left(V^{\frac{1}{3}}\right) + 1$$
$$V_0 = 1092.7 \text{ mm}^3$$

$$v_0 = 1092.7 \text{ mm}$$

 $w_0 = 3.78 \text{ Mpa}$

• See Bažant and Kazemi (1990).





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To be done ...



- Can we simulate the stochastic nature of observed properties during tests to make the design and analysis of components more efficient?
- We start with finding an alternative to the weakest link Weibull model.



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Conclusion & Outlook

Conclusion

- Titanium Aluminide alloys present a potential alternative to traditional Ni-basis super alloys.
- An empirical relation for size dependent fracture energy density is proposed.
- Crack initiation to final fracture accompanied by SSY is accurately captured by gradient-extended plasticity damage theory.

Outlook

- A multiscale model accounting for statistical descriptors of duplex microstructure in a SRUC.
- A crystal-plasticity coupled phase-field fracture theory for micromechanical motivation of size effect.
- An alternative to the weakest link Weibull model to account for failure in presence of SSY.







Thank you for your attention!



