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Cascaded Trajectory Tracking Control for Automated Vehicles

R. de Castro and J. Brembeck

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R. de Castro and J. Brembeck, " Cascaded Trajectory Tracking Control for Automated Vehicles," 2019 IEEE 90th Vehicular Technology Conference (VTC2019-Fall)
10.1109/VTCFall.2019.8891467

Cascaded Trajectory Tracking Control for Automated Vehicles

Ricardo de Castro, Jonathan Brembeck
 Institute of System Dynamics and Control, German Aerospace Center (DLR)
 Muenchener Str. 20, 82234 Wessling, Germany
 Email: Ricardo.deCastro@dlr.de

Abstract—This work deals with the design of a trajectory tracking controller (TTC) for highly automated vehicles. The proposed design follows a modular architecture, composed of an inner velocity loop and an outer trajectory tracking loop. The outer loop, which represents the main focus on this work, is further divided into two parts. The first uses input-output linearization methods to tackle model nonlinearities. The second part exploits Lyapunov methods and convex optimization to enforce physical constraints – e.g. friction circle – and minimize TTC’s loss of performance. Simulation results demonstrate that, in comparison with unconstrained TTC methods, the proposed approach is able to reduce tracking errors in up to 50% when the vehicle operates at the limit of friction.

I. INTRODUCTION

Automated driving is currently envisaged as a key technology for the future of road transportation. As discussed in [1], automated driving has the potential to decrease road accidents due to human errors, improve traffic flow, energy efficiency and passenger comfort. For the successful deployment of this technology several technical areas need to be tackled, including reliable perception of the environment, safe trajectory generation and accurate vehicle control. This work focuses on the vehicle control, with particular attention to the design of the *trajectory tracking controller* (TTC). From a control architecture perspective, there are two main design avenues that can be explored to develop TTCs. The first, called *integrated architecture*, implements the TTC using a single controller. The second, called *cascaded architecture*, splits the problem into two loops (see Fig. 1). The inner loop manipulates the vehicle actuators –steering, braking and traction motor– in order to track longitudinal velocity and yaw-rate references, while fulfilling vehicle stability limits. The outer loop generates velocity setpoints in order to execute the reference trajectory, computed by the trajectory planner.

The main goal of this work consists in designing the outer loop of the TTC. In the literature, several solutions can be found for this design problem. They range from nonlinear control laws, reliant on Lyapunov methods [2], input–output linearization [3], linear control based on disturbance observers (DOB) [4], sliding mode control [5] and model–predictive control (MPC) [6]. The majority of approaches usually assume that the vehicle operates with negligible wheel slips and/or distant from friction limits. In automotive applications these assumptions are not always easy to fulfill, especially when performing emergency maneuvers, such as obstacle avoidance

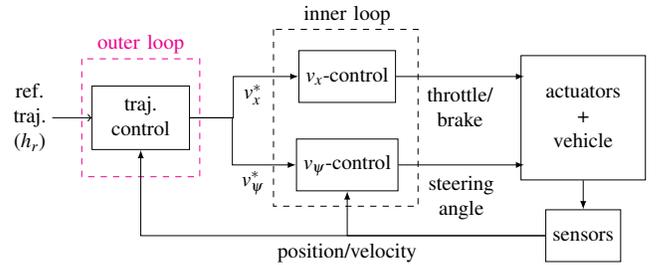


Fig. 1. Block diagram of the cascaded trajectory tracking controller (notation: v_x^* = longitudinal velocity reference; v_ψ^* =yaw-rate reference, h_r =reference trajectory.)

with aggressive braking and cornering accelerations. If the TTC’s outer loop neglects these physical limits, infeasible references might be requested to the inner loops, which – as will be shown in later sections – might increase trajectory tracking errors and compromise the safe execution of planned maneuvers. Integrating physical constraints into the design of the TTC’s outer loop is, however, a challenging task. This is due to nonlinear coupling between longitudinal and lateral tire force constraints – also known as friction circle [7] – and the dependence on tire-road friction coefficient, which is time-varying and usually uncertain.

Spurred by these challenges, we propose a novel method to handle physical constraints in the design of the TTC’s outer loop. The key idea is to augment a nominal control law –designed without explicit consideration of constraints– with a corrective control action that simultaneously enforces physical constraints and minimizes loss of performance when the vehicle operates close to friction limits. This augmentation relies on the combination of Lyapunov methods [8] and convex optimization methods [9] and represents the main contribution of the present manuscript.

II. MODEL AND PROBLEM FORMULATION

This section provides the control-oriented model for the design of the TTC’s outer loop, including physical actuation constraints, and the control problem formulation.

A. Vehicle Model

Let us consider the three degree-of-freedom planar model of the vehicle (see Fig. 2). In this model, $\mathbf{p} = [p_x \ p_y]^T$ is

the vehicle position represented in the XY coordinate system fixed with the road surface, and ψ the yaw angle between the X axis and the vehicle's x axis. The vector $\mathbf{v}_{xy} = [v_x \ v_y]^T$ contains the vehicle linear velocity – with v_x the longitudinal velocity and v_y the lateral velocity, defined in the vehicle's local frame (xy) – while v_ψ is the yaw-rate. The mathematical model is described as:

$$\dot{\mathbf{p}} = \mathbf{R}(\psi)\mathbf{v}_{xy}, \quad \dot{\psi} = v_\psi \quad (1)$$

$$\mathbf{R}(\psi) = \begin{bmatrix} \cos(\psi) & -\sin(\psi) \\ \sin(\psi) & \cos(\psi) \end{bmatrix} \quad (2)$$

As depicted in Fig. 1, the inner loop controls the v_x and v_ψ velocities. The longitudinal velocity is controlled through manipulation of the powertrain actuators (brakes and traction motor), while the yaw-rate is controlled through a steer-by-wire actuator, installed in the vehicle's front axle. Control Lyapunov functions are used to construct these inner loops (see [10] for details). Their dominant response is approximated here with first-order dynamics

$$\dot{v}_x = \alpha_x(v_x^* - v_x) + d_x \quad (3a)$$

$$\dot{v}_\psi = \alpha_\psi(v_\psi^* - v_\psi) + d_\psi \quad (3b)$$

where v_i^* is the velocity reference signal transmitted to the inner loop $i = \{x, \psi\}$, and $\alpha_i = 1/\tau_i$ the inverse of the dominant time constant. The disturbances d_x, d_ψ capture model mismatches between the real vehicle response and the control-oriented model (3). They are assumed to lie in the set $\mathcal{D} = \{(d_x, d_\psi) : |d_i| \leq \bar{d}_i, i \in \{x, \psi\}\}$, where \bar{d}_i is a known bound.

While v_x and v_ψ can be accurately controlled by the inner loops, the vehicle's lateral velocity v_y is usually more difficult to measure and control. In practice, the inner yaw-rate loop contains additional control mechanisms to keep the value of v_y small, which ensures that the vehicle remains within its safe operation envelope –see [10] for details. As a result, we will assume that inner loop guarantees $|v_y| \leq \bar{v}_y$, where \bar{v}_y is a small value. Based on this assumption, the lateral velocity dynamics can be represented as:

$$\dot{v}_y = w \quad (4)$$

where w is an exogenous signal that can be measured. For example, using the lateral acceleration (a_y) and the velocity signals provided by the vehicle's inertial measure unit (IMU), this exogenous signal can be obtained as $w = a_y - v_x v_\psi$ [11]. To further facilitate the outer-loop design, we will introduce the following virtual control inputs

$$u_i = \alpha_i(v_i^* - v_i), \quad i = \{x, \psi\} \quad (5)$$

From a physical point of view, u_x can be interpreted as a desired linear acceleration and u_ψ a desired yaw acceleration. Based on these elements, the combined vehicle planar model and inner control loops can be compactly represented as:

$$\dot{\mathbf{z}} = \begin{bmatrix} \mathbf{R}(\psi)\mathbf{v}_{xy} \\ v_\psi \\ \mathbf{F}_1 w \\ 0 \end{bmatrix} + \begin{bmatrix} \mathbf{0} & \mathbf{0} \\ 0 & 0 \\ \mathbf{0} & \mathbf{F}_2 \\ 0 & 1 \end{bmatrix} (\mathbf{u} + \mathbf{d}) \quad (6)$$

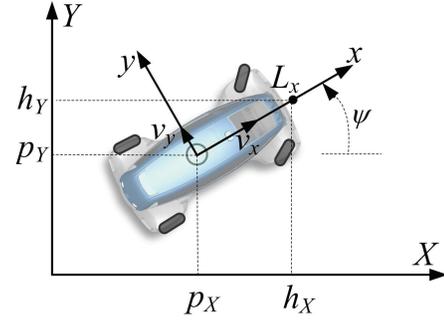


Fig. 2. Planar vehicle model. The point (p_x, p_y) represents the position of the vehicle's center of gravity, while (h_x, h_y) is the virtual point to be tracked.

where $\mathbf{z} = [\mathbf{p}^T \ \psi \ v_{xy} \ v_\psi]^T$ is the vehicle state, $\mathbf{u} = [u_x \ u_\psi]^T$ the virtual control input, $\mathbf{d} = [d_x \ d_\psi]^T$ the disturbances and $\mathbf{0}$ a zero matrix with 2 rows and 1 column. The model also depends on the auxiliary matrices $\mathbf{F}_1 = [0 \ 1]^T$, $\mathbf{F}_2 = [1 \ 0]^T$. It is assumed that the state \mathbf{z} is measured or estimated by the vehicle's on-board sensors.

B. Control Constraints

The vehicle control is constrained by two types of physical limits. The first type emerges due to constraints in the vehicle actuators, such as limits in braking and traction motor torques. The second type is due to constraints in the friction forces produced by the tires. These constraints can be represented through the *vehicle friction circle* [7]:

$$\mathcal{A} = \{(a_x, a_y) : a_x^2 + a_y^2 \leq \mu^2 g^2, \ \underline{a}_x \leq a_x \leq \bar{a}_x\} \quad (7)$$

where a_x is the vehicle's longitudinal acceleration, a_y the lateral acceleration, μ the tire-road adhesion conditions, g the gravity acceleration, and $\underline{a}_x, \bar{a}_x$ represent the minimum and maximal longitudinal acceleration.

The acceleration limits \mathcal{A} are relevant, not only for the inner loop, but also to the TTC's outer loop. If the outer loop neglects these limits, infeasible reference signals might be sent to the inner loops, compromising trajectory tracking performance. To mitigate these risks, the acceleration limits \mathcal{A} will be transformed into a set of admissible control inputs \mathcal{U} . More specifically, pragmatic approximations will be used to relate (a_x, a_y) with the virtual control inputs (u_x, u_ψ) . For example, neglecting the disturbances d_x , the longitudinal acceleration can be approximated as:

$$a_x = \dot{v}_x - v_\psi v_y = u_x + d_x - v_\psi v_y \approx u_x - v_\psi v_y \quad (8)$$

For the lateral acceleration mapping, we assume perfect tracking of the yaw-rate reference ($v_\psi \approx v_\psi^*$) and make use of (4),(5) to approximate the lateral acceleration as:

$$a_y = \dot{v}_y + v_\psi v_x \approx w + \left(\frac{u_\psi}{\alpha_\psi} + v_\psi \right) v_x \quad (9)$$

Replacing (8) and (9) into \mathcal{A} leads to the following input constraint set:

$$\mathcal{U} = \left\{ (u_x, u_\psi) : \begin{aligned} & \underline{a}_x \leq u_x - v_\psi v_y \leq \bar{a}_x, \quad \underline{u}_\psi \leq u_\psi \leq \bar{u}_\psi \\ & (u_x - v_\psi v_y)^2 + \left(w + \left(\frac{u_\psi}{\alpha_\psi} + v_\psi \right) v_x \right)^2 \leq \mu^2 g^2 \end{aligned} \right\}$$

which also contains a lower (\underline{u}_ψ) and upper (\bar{u}_ψ) bound on yaw-acceleration.

C. Formulation of the Control Problem

The main goal of the outer-loop controller is to track a virtual point \mathbf{h} , located at a constant distance L_x from the vehicle's center-of-gravity (see Fig. 2)

$$\mathbf{h} = [h_x \quad h_y]^T = \mathbf{p} + \mathbf{R}(\psi)\mathbf{L} \quad (10)$$

where $\mathbf{L} = [L_x \quad 0]^T$. The control problem consists in designing a feedback law for \mathbf{u} such that the virtual point \mathbf{h} follows, as closely as possible, the reference trajectory $\mathbf{h}_r(t) : \mathbb{R} \rightarrow \mathbb{R}^2$, while fulfilling the input constraint set \mathcal{U} . It is assumed that the reference trajectory \mathbf{h}_r complies with the model

$$\dot{\mathbf{h}}_r = \begin{bmatrix} \dot{h}_{r,x} \\ \dot{h}_{r,y} \end{bmatrix} = \begin{bmatrix} v_r \cos(\psi_r) \\ v_r \sin(\psi_r) \end{bmatrix}, \quad \dot{\psi}_r = v_{\psi,r}, \quad \dot{v}_r = a_r \quad (11)$$

where v_r , $v_{\psi,r}$ are the velocities and yaw-rate references, a_r represents the reference tangential acceleration and ψ_r the reference yaw angle.

D. Controller Structure

The control law for the TTC's outer loop is divided into two components :

$$\mathbf{u} = \mathbf{u}_{NL} + \Delta\mathbf{u} \quad (12)$$

The first component (\mathbf{u}_{NL}) is a nominal control law, obtained through input-output linearization [8]. It attempts to cancel the dominant nonlinearities in the model, which – under the assumption of perfect model ($\mathbf{d} = 0$) and unconstrained input set ($\mathcal{U} = \mathcal{R}^2$) – would be sufficient to fulfill the tracking requirements. The second component ($\Delta\mathbf{u}$) is a correction control action that enforces the input constraints \mathcal{U} with minimum loss of performance.

III. NOMINAL CONTROLLER

To derive the nominal control law \mathbf{u}_{NL} , let us consider the tracking error $\mathbf{e} = [e_x \quad e_y]^T = \mathbf{h} - \mathbf{h}_r$. Since this control law depends on input-output linearization [8], it is necessary to first compute the relative degree of the tracking error, i.e., differentiate \mathbf{e} until the control input \mathbf{u} explicitly appears in the derivatives of \mathbf{e} . Accordingly, computing the first and second time derivative of the tracking error, one obtains:

$$\ddot{\mathbf{e}} = \overbrace{\mathbf{TR}(\psi)v_\psi \mathbf{v}_{xy} - \mathbf{R}(\psi)\mathbf{L}v_\psi^2 - \ddot{\mathbf{h}}_r + \mathbf{R}(\psi)\mathbf{F}_1 w}^{\mathbf{f}(\mathbf{z}, w)} + \underbrace{[\mathbf{R}(\psi)\mathbf{F}_2 \quad \mathbf{TR}(\psi)\mathbf{L}]}_{\mathbf{G}(\psi)}(\mathbf{u} + \mathbf{d}) \quad (13)$$

$$= \mathbf{f}(\mathbf{z}, w) + \mathbf{G}(\psi)(\mathbf{u} + \mathbf{d}) \quad (14)$$

where \mathbf{T} is defined in Appendix. This result enables us to conclude that the relative degree¹ of the tracking error \mathbf{e} is $[2 \quad 2]^T$ for $L_x \neq 0$. When $L_x = 0$, the virtual point is located at the vehicle's center of gravity and the matrix $\mathbf{G}(\psi)$ becomes singular ($\det(\mathbf{G}(\psi)) = L_x = 0$), which prevents the cancellation of the model nonlinearities.

Let us now consider the following change of variable:

$$\eta = \begin{bmatrix} \eta_1 \\ \eta_2 \end{bmatrix} = \begin{bmatrix} \psi - \psi_r \\ v_y \end{bmatrix}, \quad \zeta = [\zeta_1 \quad \zeta_2 \quad \zeta_3 \quad \zeta_4]^T = \begin{bmatrix} \mathbf{e} \\ \dot{\mathbf{e}} \end{bmatrix}$$

which leads to the following dynamics:

$$\dot{\eta}_1 = l(\eta, \zeta), \quad \dot{\eta}_2 = w \quad (15a)$$

$$\dot{\zeta} = \mathbf{A}\zeta + \mathbf{B}(\mathbf{f}(\mathbf{z}, w) + \mathbf{G}(\psi)(\mathbf{u} + \mathbf{d})) \quad (15b)$$

where \mathbf{A} , \mathbf{B} and $l(\cdot)$ are defined in Appendix. In this representation, the system is separated into an internal state $\eta = [\eta_1 \quad \eta_2]^T$ (not directly controllable by the TTC) and an external state ζ , composed of the tracking error and its time derivatives. By driving the external state ζ toward the origin, the TTC's control goal is fulfilled. This can be achieved by adopting the following control law:

$$\mathbf{u}_{NL} = -\mathbf{G}^{-1}(\psi)\mathbf{f}(\mathbf{z}, w) - \mathbf{G}^{-1}(\psi)\mathbf{K}\zeta \quad (16)$$

where $\mathbf{K} \in \mathbb{R}^{2 \times 4}$ is a matrix selected by the designer such that $(\mathbf{A} - \mathbf{B}\mathbf{K})$ is Hurwitz, e.g. using pole-placement methods [8]. Furthermore, it can also be shown that the zero-dynamics η , in particular the yaw-angle error, are bounded (omitted due to space constraints) .

IV. AUGMENTED CONTROLLER

The design of the nominal control law (16) does not take into account the input constraint set \mathcal{U} . If the virtual control is not feasible, i.e. $\mathbf{u}_{NL} \notin \mathcal{U}$, then a correction term $\Delta\mathbf{u}$ needs to be incorporated into the control law in order to enforce the input constraint set. However, this correction term might have a negative impact in the performance and stability of the nominal controller. To attenuate these risks, we propose in this section a systematic method to compute the correction term that can simultaneously enforce input constraints ($\mathbf{u}_{NL} + \Delta\mathbf{u} \in \mathcal{U}$) and minimize loss of performance. The combination of the nominal control law and the corrective term is referred hereafter as the augmented controller.

A. Loss of Performance

To better understand the loss of performance due to control constraints, let us treat the corrective term $\Delta\mathbf{u}$ as a perturbation and investigate its effect in the closed-loop system using Lyapunov methods. Inserting (16) into (12) leads to the following closed-loop dynamics:

$$\dot{\zeta} = (\mathbf{A} - \mathbf{B}\mathbf{K})\zeta + \underbrace{\mathbf{B}\mathbf{E}(\eta_1 + \psi_r)}_{\mathbf{g}(\eta)}(\Delta\mathbf{u} + \mathbf{d}) \quad (17)$$

$$= (\mathbf{A} - \mathbf{B}\mathbf{K})\zeta + \mathbf{g}(\eta)(\Delta\mathbf{u} + \mathbf{d}) \quad (18)$$

¹recall that \mathbf{e} has a dimensionality of 2, consequently the relative degree has to be computed for all elements in \mathbf{e}

Under undisturbed ($\mathbf{d} = 0$) and unconstrained ($\Delta\mathbf{u} = 0$) conditions, the previous dynamics is dominated by the linear term $(\mathbf{A} - \mathbf{BK})\zeta$. Thus, closed-loop stability can be investigated through the quadratic Lyapunov function $V(\zeta) = \zeta^T \mathbf{P}\zeta$, where \mathbf{P} is a positive definite matrix². Computing the time derivative of V one obtains

$$\dot{V}(\zeta, \Delta_{\dot{V}}, \mathbf{d}) = -\zeta^T \mathbf{Q}\zeta + 2\zeta^T \mathbf{P}\mathbf{g}(\eta)\mathbf{d} + \Delta_{\dot{V}} \quad (19)$$

where $\Delta_{\dot{V}} = 2\zeta^T \mathbf{P}\mathbf{g}(\eta)\Delta\mathbf{u}$ represents a perturbation introduced by the correction term.

Recall that, ultimate boundedness of ζ , i.e. convergence of ζ to a small neighborhood around the origin, is achieved if \dot{V} fulfills [8]

$$\dot{V}(\zeta, \Delta_{\dot{V}}, \mathbf{d}) \leq -W_V(\zeta) < 0, \quad \forall \|\zeta\| \geq \varepsilon_\zeta, \mathbf{d} \in \mathcal{D} \quad (20)$$

where $W_V(\eta)$ is a positive definite function and $\varepsilon_\zeta > 0$ a small positive constant that affects the size of the ultimate bound. It can be verified that the nominal control law, with input constraints inactive ($\Delta_{\dot{V}}=0$), fulfills condition (20) – see [12] for details. On the other hand, the perturbation introduced by the correction term, $\Delta_{\dot{V}}$, can have a “good” or a “bad” effect in the ultimate boundedness condition. For example, a “good” effect is obtained when $\Delta_{\dot{V}} < 0$, which makes \dot{V} more negative, helping the fulfillment of (20). “Bad” effects occur when $\Delta_{\dot{V}} > 0$. In this case, ε_ζ might need to be increased in order to fulfill (20), leading to higher ultimate bounds and – in the worst case scenario – compromise the closed-loop stability.

B. Optimal Correction Term

In light of the above discussion, the correction term $\Delta\mathbf{u}$ should be selected in order to avoid positive values of $\Delta_{\dot{V}}$. With this goal in mind, we start by upper bounding the perturbation $\Delta_{\dot{V}}$ with a non-negative constant s , i.e.

$$\Delta_{\dot{V}} = 2\zeta^T \mathbf{P}\mathbf{g}(\eta)\Delta\mathbf{u} \leq s, \quad s \geq 0 \quad (21)$$

From a practical perspective, s can be interpreted as Lyapunov-based performance loss. When $s = 0$, no perturbation is introduced in the Lyapunov stability condition and no performance loss occurs. On the other hand, when $s > 0$, “bad” perturbations are induced, which might lead to tracking performance loss. The design of the correction term can now be re-phrased as: find $\Delta\mathbf{u}$ that minimizes s , while complying with the input constraint set \mathcal{U} . This can be translated into the following optimization problem:

$$\min_{\Delta\mathbf{u}, s} \quad w_s s^2 + \Delta\mathbf{u}^T \mathbf{W}_u \Delta\mathbf{u} \quad (22a)$$

$$s.t. \quad 2\zeta^T \mathbf{P}\mathbf{g}(\eta)\Delta\mathbf{u} \leq s \quad (22a)$$

$$s \geq 0 \quad (22b)$$

$$\begin{bmatrix} u_x & u_\psi \end{bmatrix}^T = \mathbf{u}_{NL} + \Delta\mathbf{u} \quad (22c)$$

$$\underline{a}_x \leq u_x - v_\psi v_y \leq \bar{a}_x, \quad \underline{u}_\psi \leq u_\psi \leq \bar{u}_\psi, \quad (22d)$$

$$\left(u_x - v_\psi v_y \right)^2 + \left(w + \left(\frac{u_\psi}{\alpha_\psi} + v_\psi \right) v_x \right)^2 \leq \mu^2 g^2$$

²obtained by solving the equation $\mathbf{P}(\mathbf{A} - \mathbf{BK}) + (\mathbf{A} - \mathbf{BK})^T \mathbf{P} = -\mathbf{Q}$, with \mathbf{Q} a positive definite matrix defined by the designer

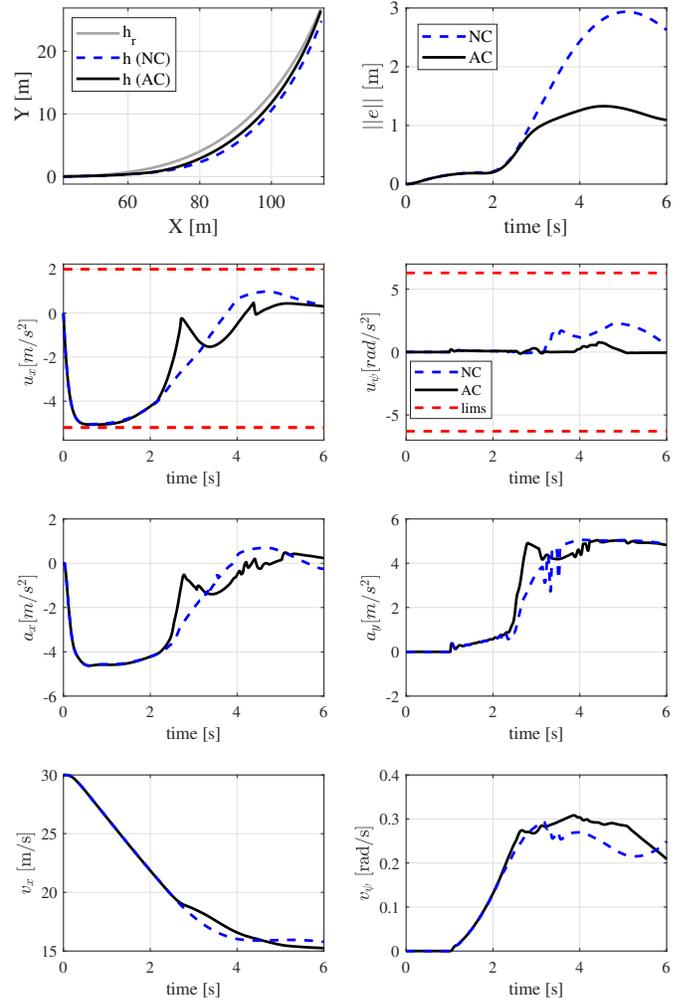


Fig. 3. Simulation results of the TTCs (NC=nominal controller; AC=augmented controller; red dashed lines represent range-limits in the linear and yaw acceleration).

The above cost function, besides penalizing loss of performance (s^2) also allows minimization of magnitude of the corrective term ($\Delta\mathbf{u}$); trade-off between these two terms is weighted by scalar w_s and matrix \mathbf{W}_u , which are defined by the designer.

V. VALIDATION

The trajectory controller was validated through several simulations carried out in a co-simulation environment between Matlab/Simulink and Dymola. The former is responsible for the controller’s implementation, while the latter provides a nonlinear double track model, with 14 degree-of-freedom, of the vehicle dynamics model. The vehicle model was implemented in Modelica and parameterized with experimental data of a Volkswagen Passat B6, adapted with a steer-by-wire actuator [13].

Two trajectory controllers were simulated. The first, called *nominal controller (NC)*, relies on the nonlinear control

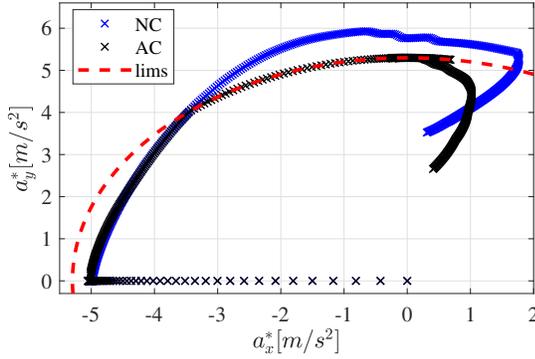


Fig. 4. Acceleration requested by the TTC's outer loop (NC =nominal controller without saturation; AC =augmented controller); a_x^* and a_y^* are computed using the mapping (8),(9); the red dashed line represents the friction circle constraint (7).

law (16), which is posteriorly saturated³ to enforce input constraint \mathcal{D} . The second controller, called *augmented controller* (AC), employs the optimization-based method described in (22) and solved with the embedded conic solver (ECOS) [14]. In both cases, the inner-loop controller relies on input-output methods, combined with control barrier functions, described in [10]. Fig. 3 shows the simulation results of the trajectory controllers obtained while negotiating a left-handed corner with a combined maximum acceleration of $0.45g$ and a known friction coefficient of $\mu = 0.55$. The analysis of this maneuver can be divided into two parts. In the first part ($t \in [0, 2.5]s$) the vehicle brakes aggressively in straight line; since the requested accelerations are feasible, both NC and AC have similar responses (see u_x, u_y signals). In the second part of the maneuver ($t \in [2.5, 4]s$) the vehicle transits from full braking $(a_x, a_y) = (-4.5, 0) m/s^2$ to full cornering $(a_x, a_y) = (0, 4.5) m/s^2$, demanding vehicle control close to friction limits with combined braking and cornering. The NC , because of its disregard for friction limits at design stage, generates an aggressive yaw-rate acceleration (u_ψ) as well as excessive braking demand (u_x), which introduces a significant deviation from the reference trajectory (maximum error $\|\mathbf{e}\|$ of 3m). These infeasible control requests are also visible in Fig. 4, which depicts the acceleration references generated by the NC (before saturation). On the other hand, the AC provides a more coordinated control; it reduces the amount of braking around 3s, which then allows a quicker build-up of lateral acceleration (a_y) and yaw-rate (v_ψ). As a result of this intervention, the ANC reduces the maximum tracking error $\|\mathbf{e}\|$ from 3m to 1.3m, i.e. a 56% reduction.

VI. CONCLUSION

A two-loop trajectory tracking controller, composed of inner-velocity and outer-position loops, was developed. Particular focus was given to the design of the outer loop in the presence of actuation and friction limits. We have shown how unconstrained control laws, based on input-output linearization, can be augmented with corrective actions that fulfill

control constraints and minimize loss of performance. The key idea was to treat the correction actions as perturbations in a Lyapunov sense, which then allowed us to relate corrective actions with Lyapunov-based performance loss. Simulation results demonstrate that, in comparison with unconstrained trajectory tracking controllers, the proposed control method is able to decrease the tracking errors in up to 1.7m when the vehicle operates close to friction limits.

APPENDIX I - DEFINITION OF VARIABLES

The control-oriented model of the vehicle uses the following variables

$$\mathbf{A} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}, \quad \mathbf{B} = \begin{bmatrix} 0 & 0 \\ 0 & 0 \\ 1 & 0 \\ 0 & 1 \end{bmatrix}, \quad \mathbf{T} = \begin{bmatrix} 0 & -1 \\ 1 & 0 \end{bmatrix} \quad (23)$$

$$l(\eta, \zeta) = -\frac{1}{L_x} \left(\eta_2 + \sin(\eta_1 + \psi_r)(\dot{h}_{r,x} + \zeta_3) - \cos(\eta_1 + \psi_r)(\dot{h}_{r,y} + \zeta_4) \right) - v_{\psi,r} \quad (24)$$

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³obtained by solving (22) without the stability constraint (22a)