

# Design Study of a Two-Parted Variable Inlet Guide Vane with an Optimized Axial Gap

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Knowledge for Tomorrow



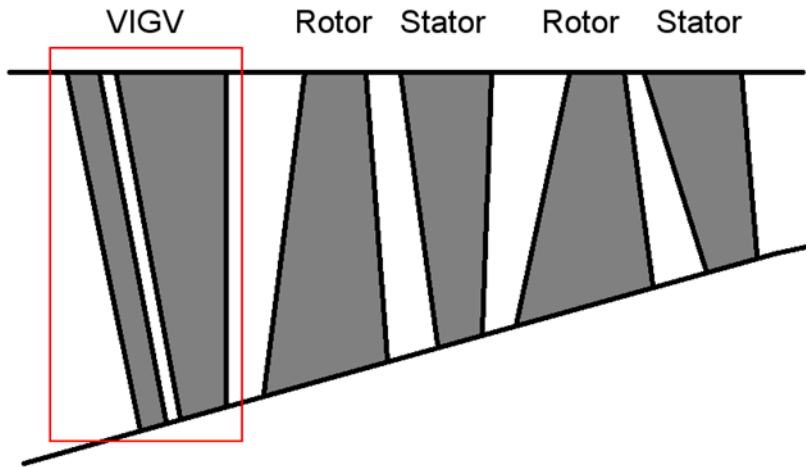
# Table of content

1. Introduction
2. Optimization description
3. Optimization 1: Axial gap only
4. Optimization 2: Whole VIGV
5. Conclusion and outlook

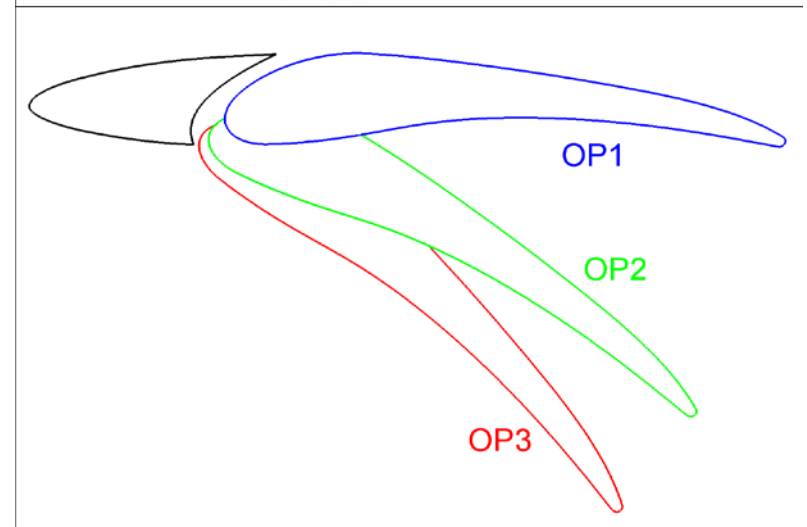
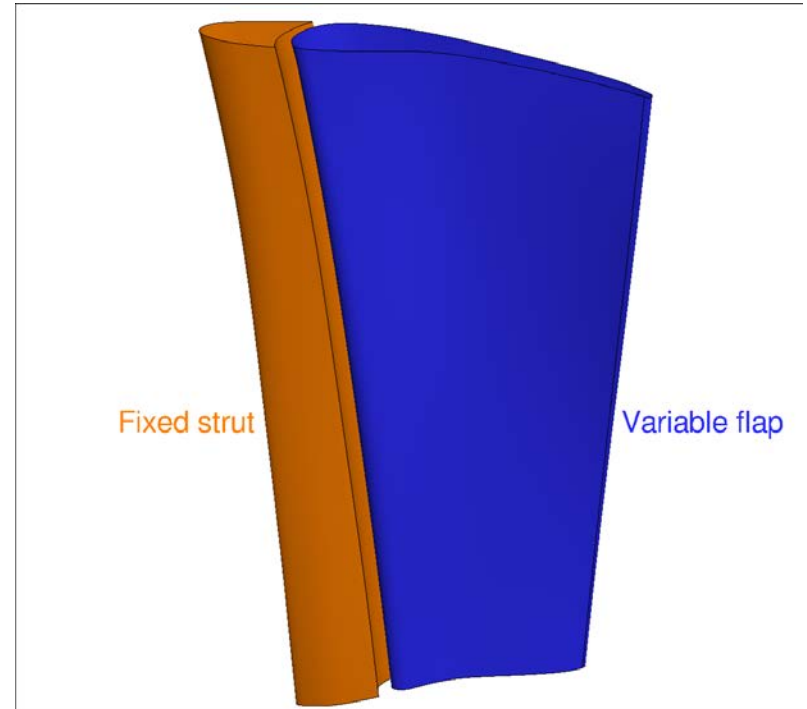


# 1. Introduction

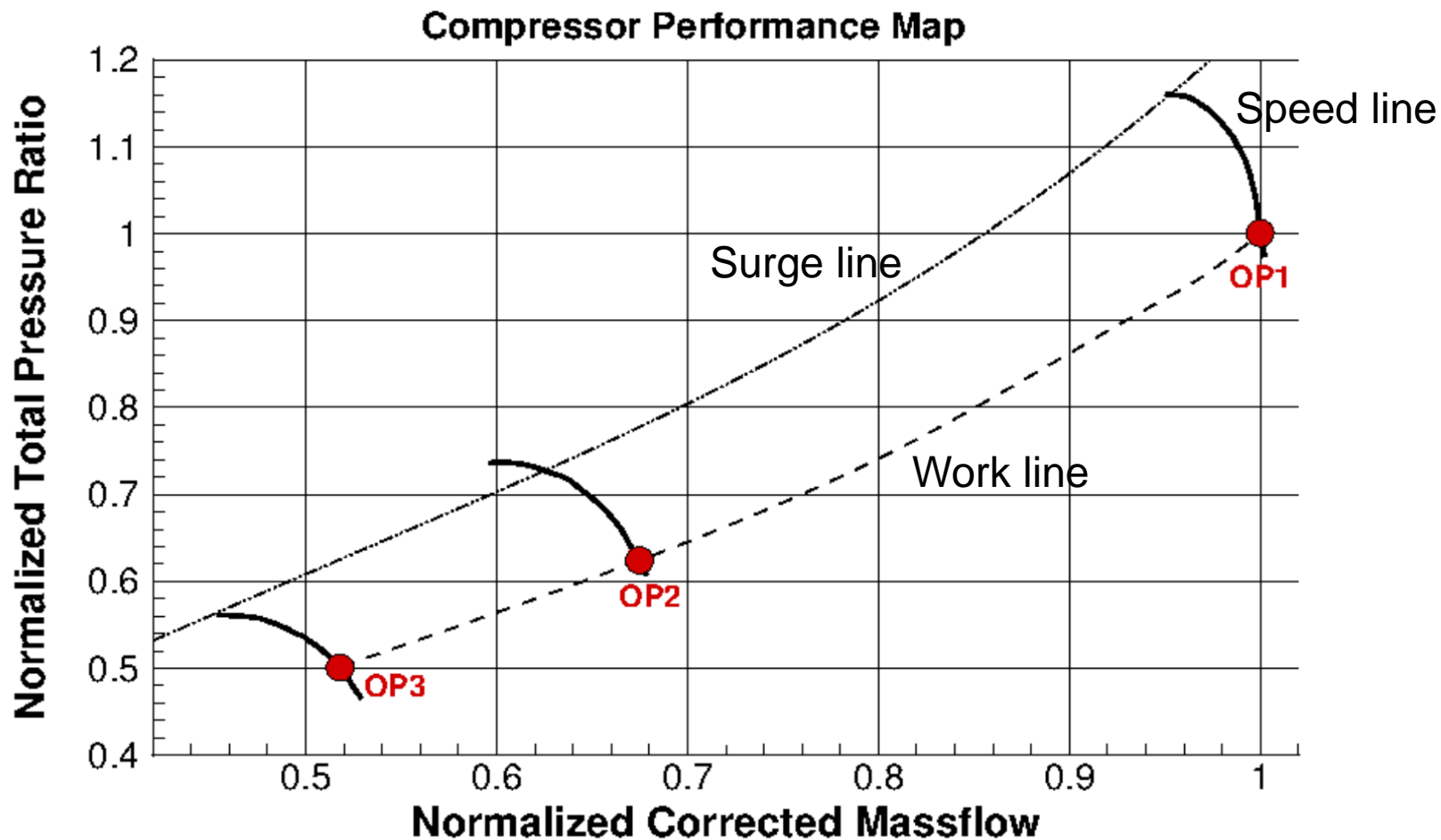
**Objectives: reduce the total pressure losses and maintain the deflection**



- VIGV is located at the beginning of the compressor channel
- VIGV is composed of two parts: the fixed strut and the variable flap
- VIGV is studied on 3 operating points



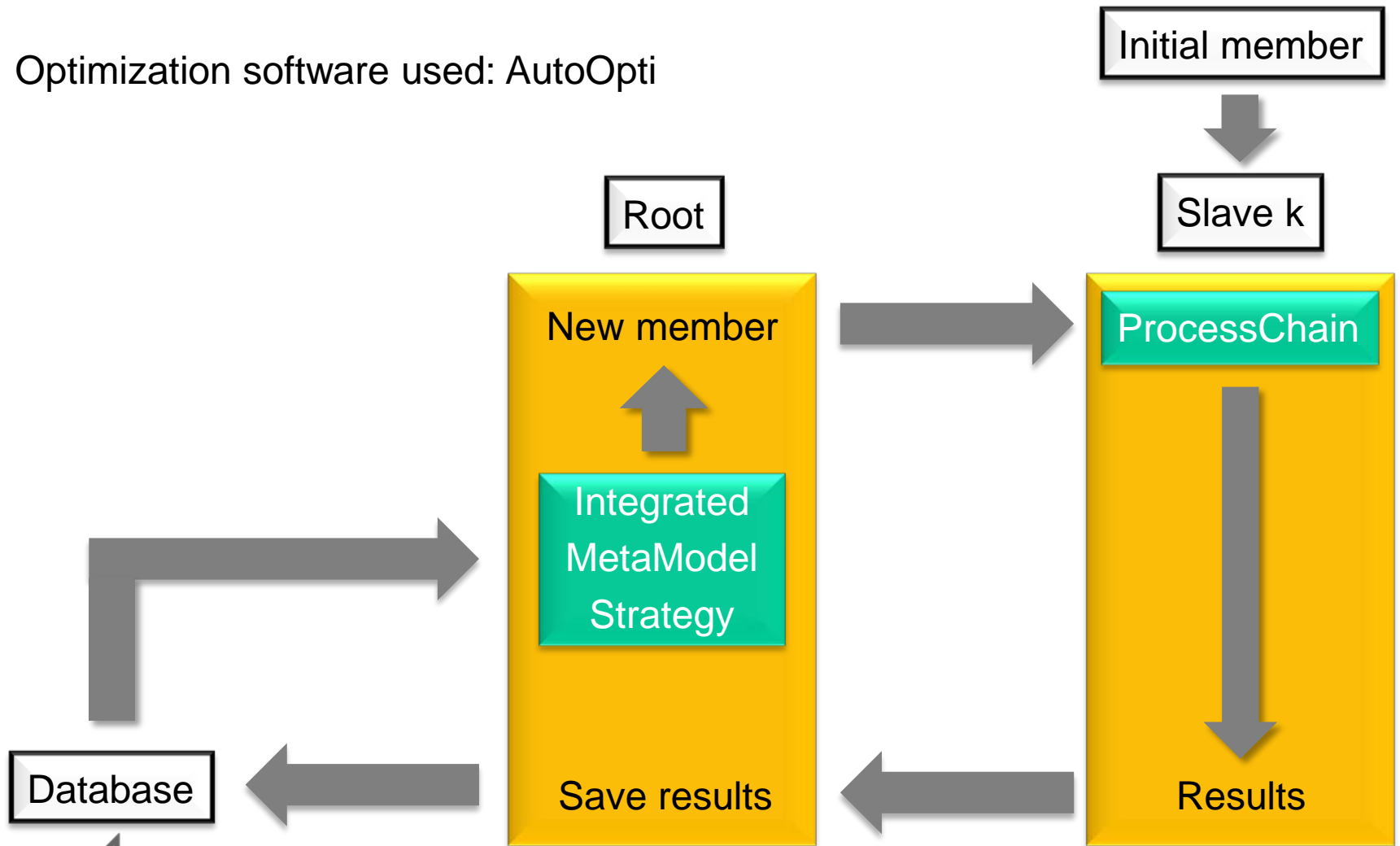
# 1. Introduction



## 2. Optimization description

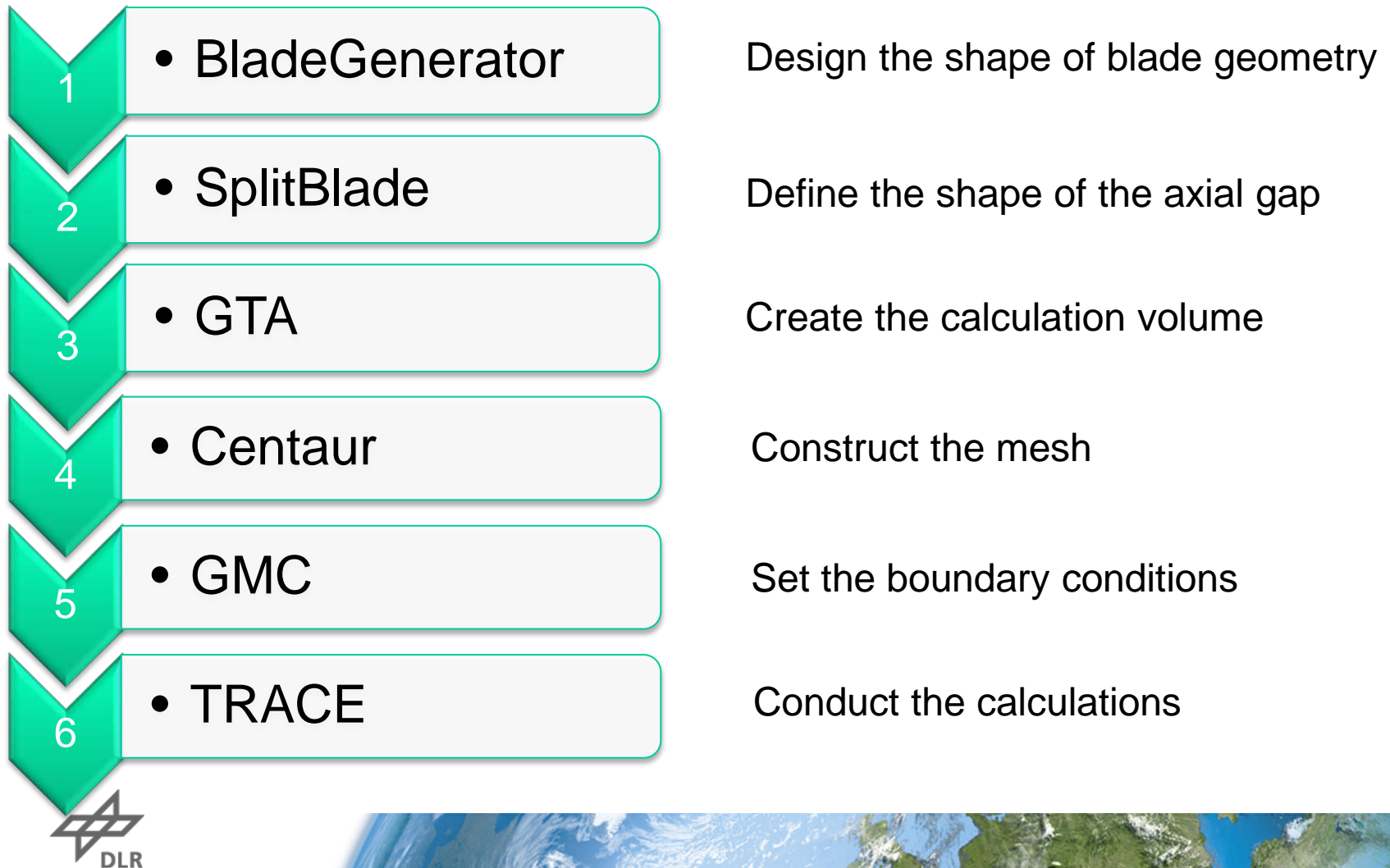
### Optimization model | Processchain | Reference Member

Optimization software used: AutoOpti



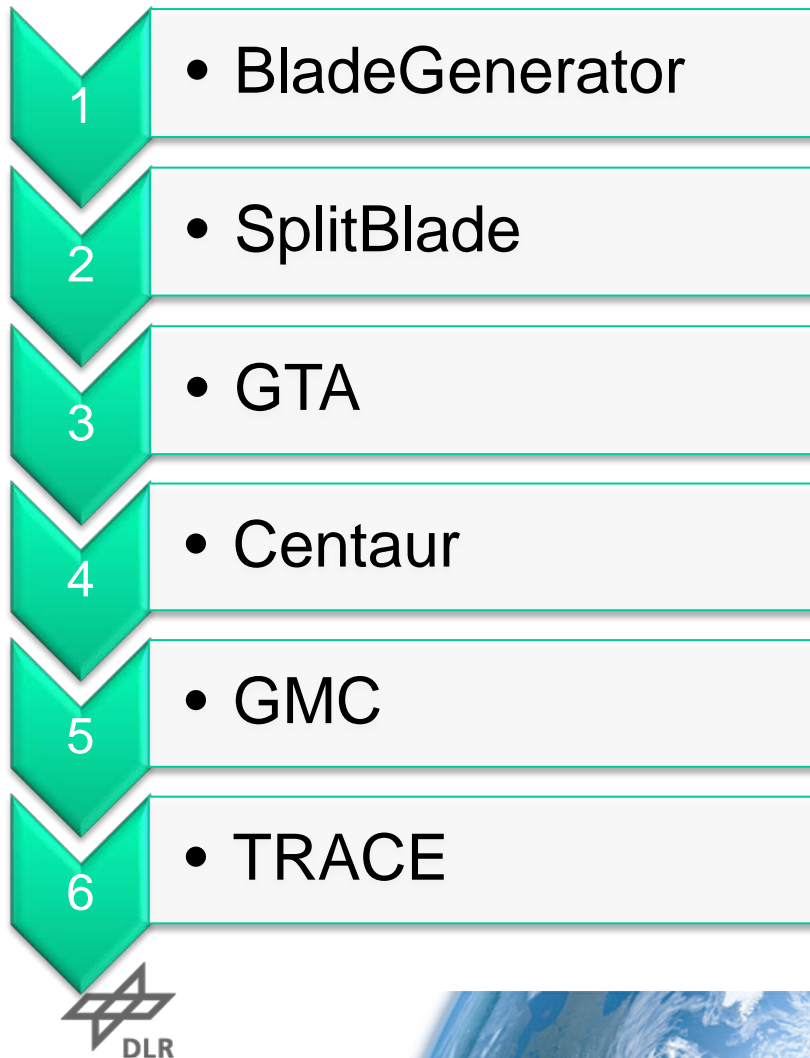
## 2. Optimization description

Optimization model | **Processchain** | Reference Member



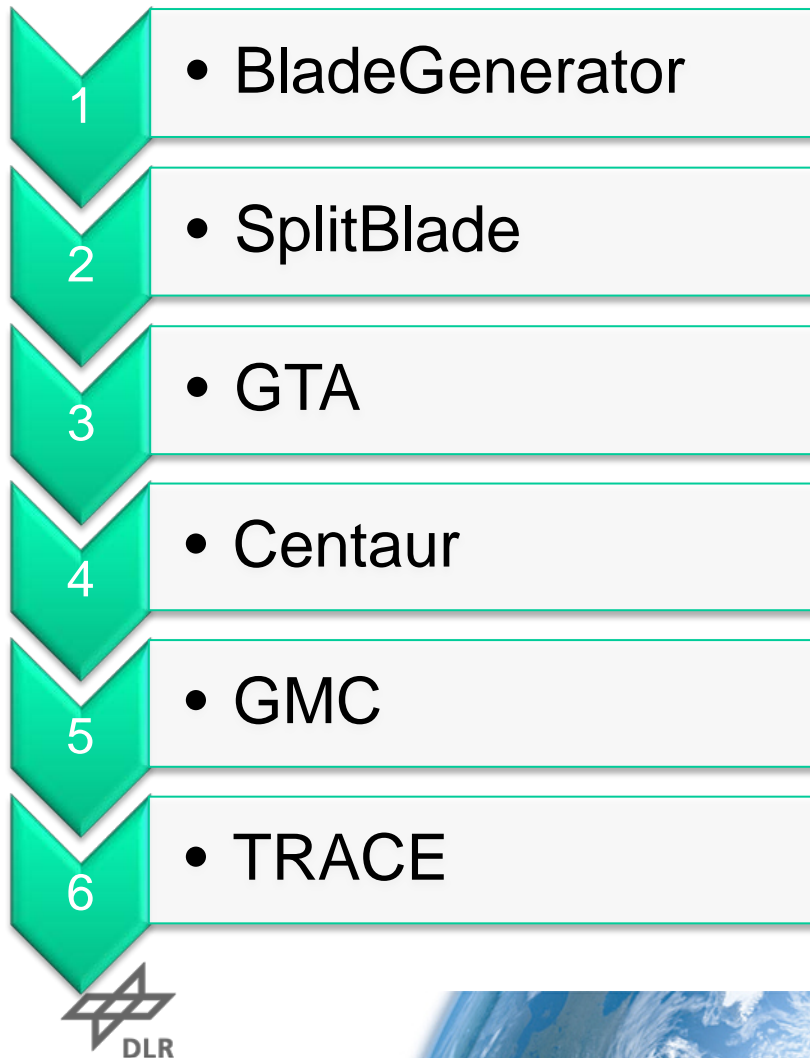
## 2. Optimization description

Optimization model | Processchain | Reference Member



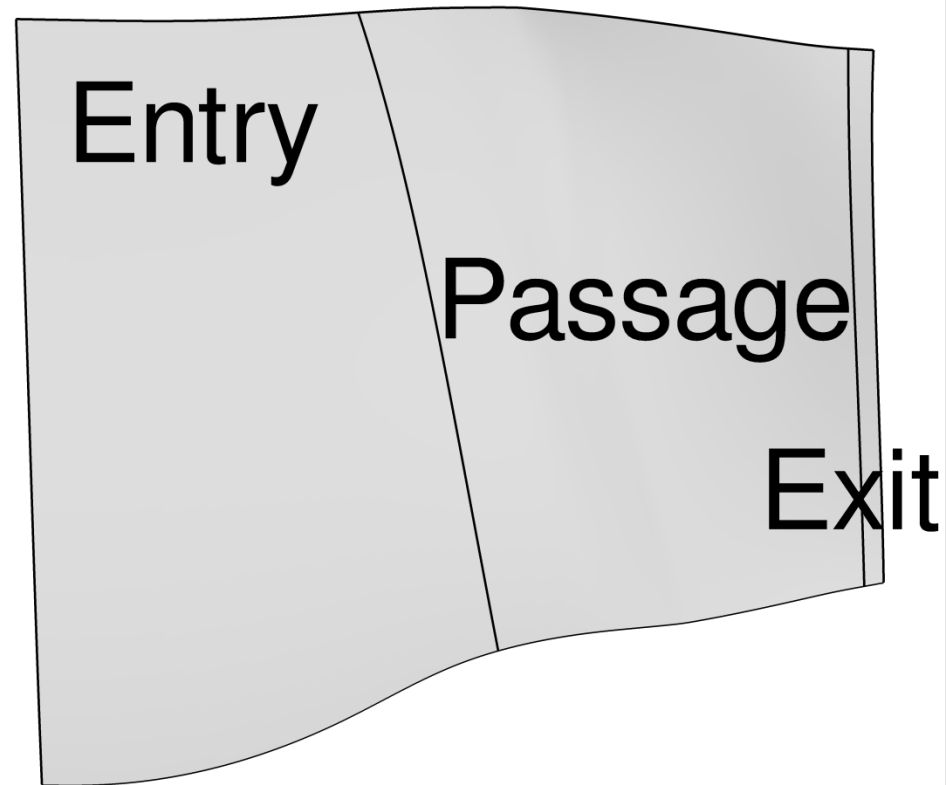
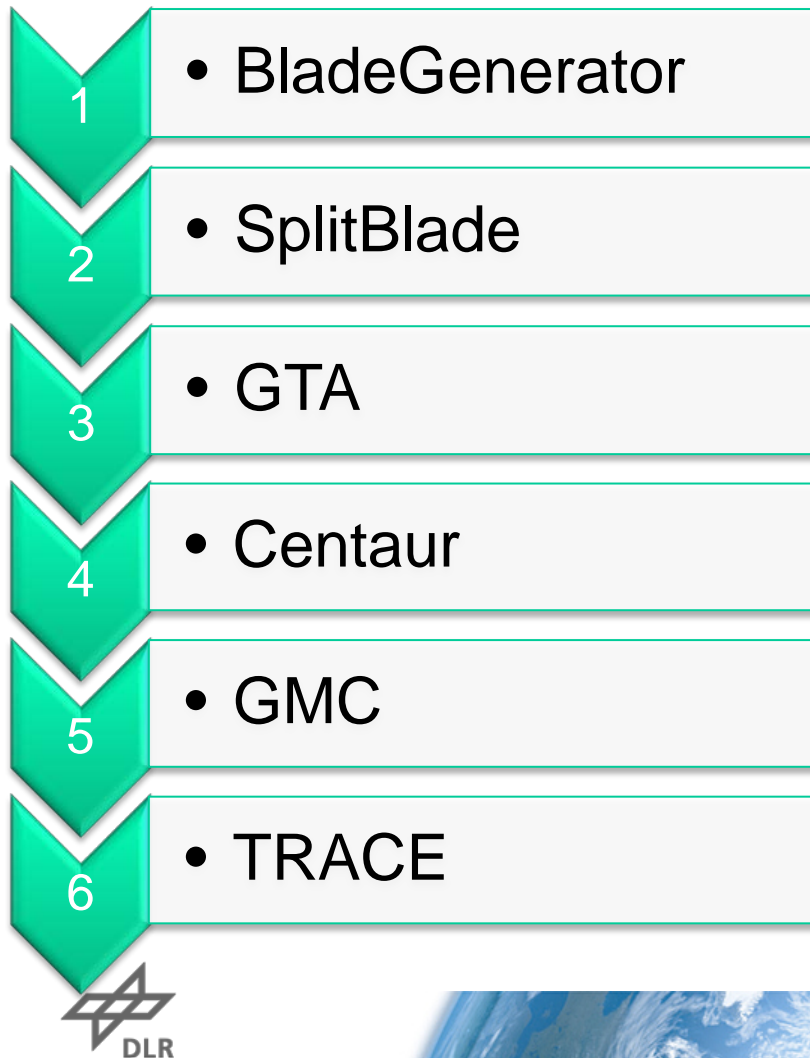
## 2. Optimization description

Optimization model | Processchain | Reference Member



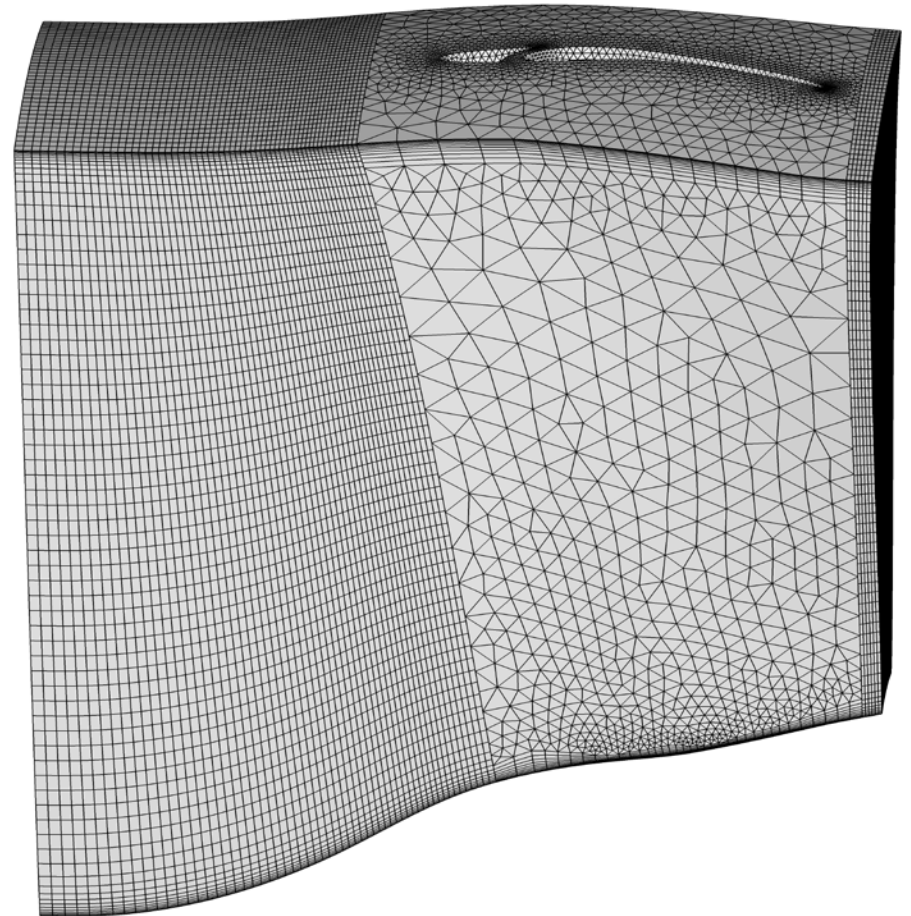
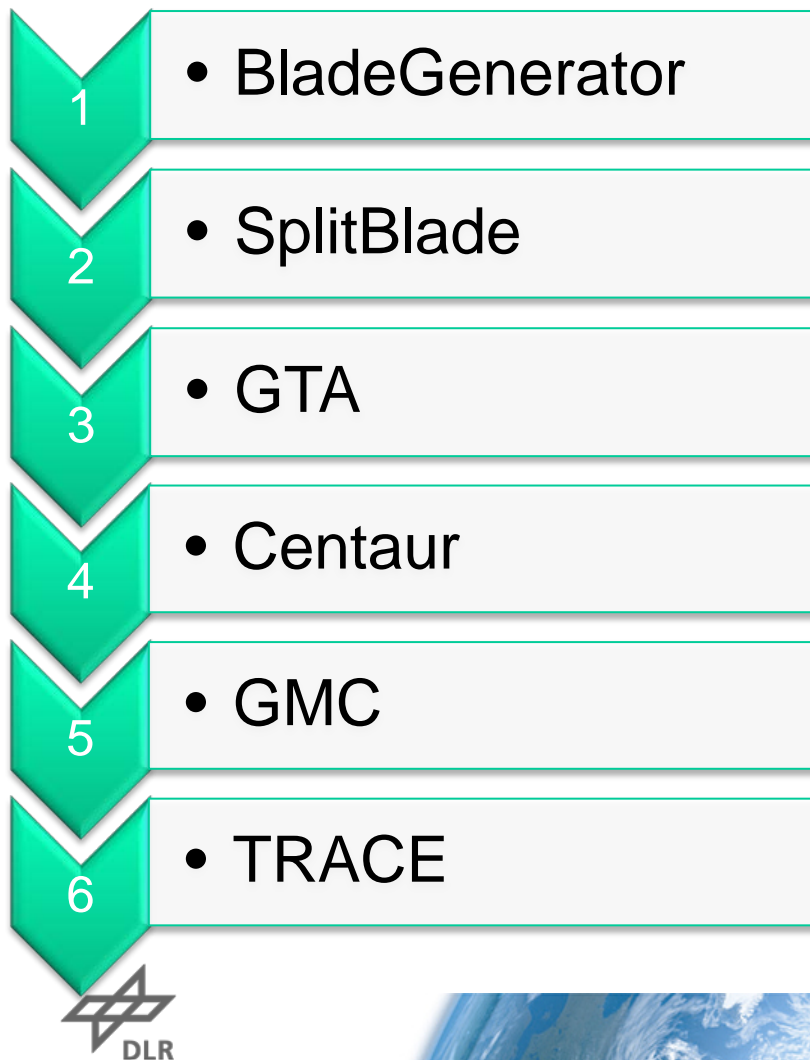
## 2. Optimization description

Optimization model | Processchain | Reference Member



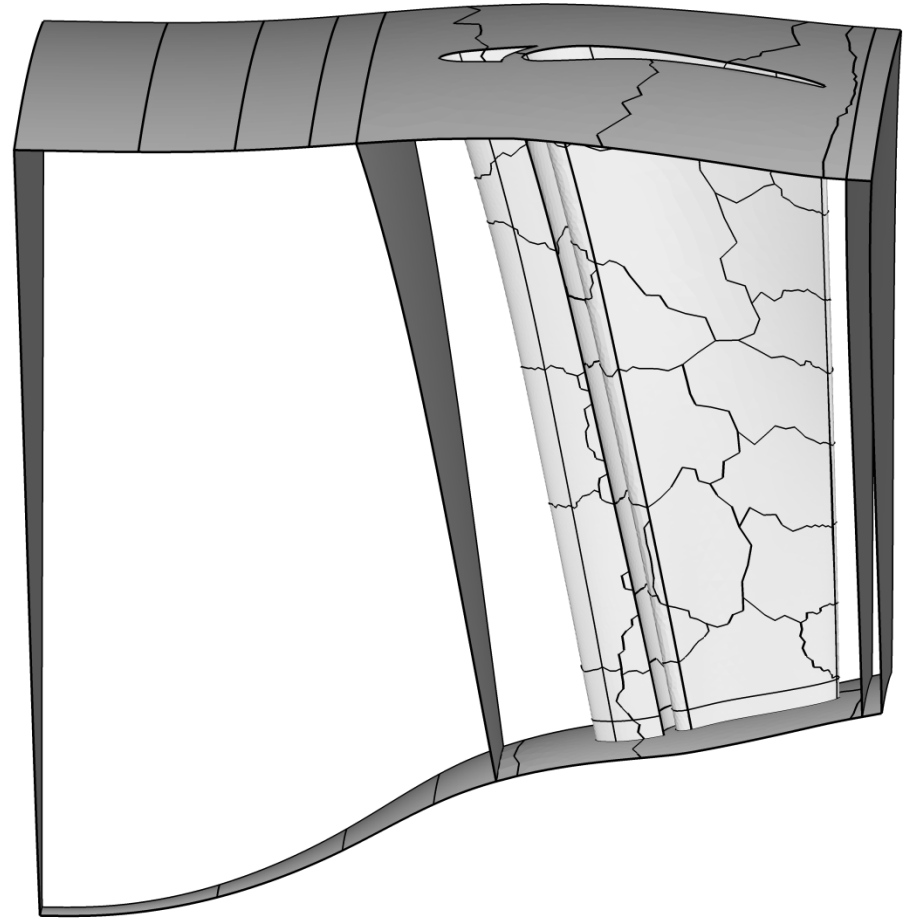
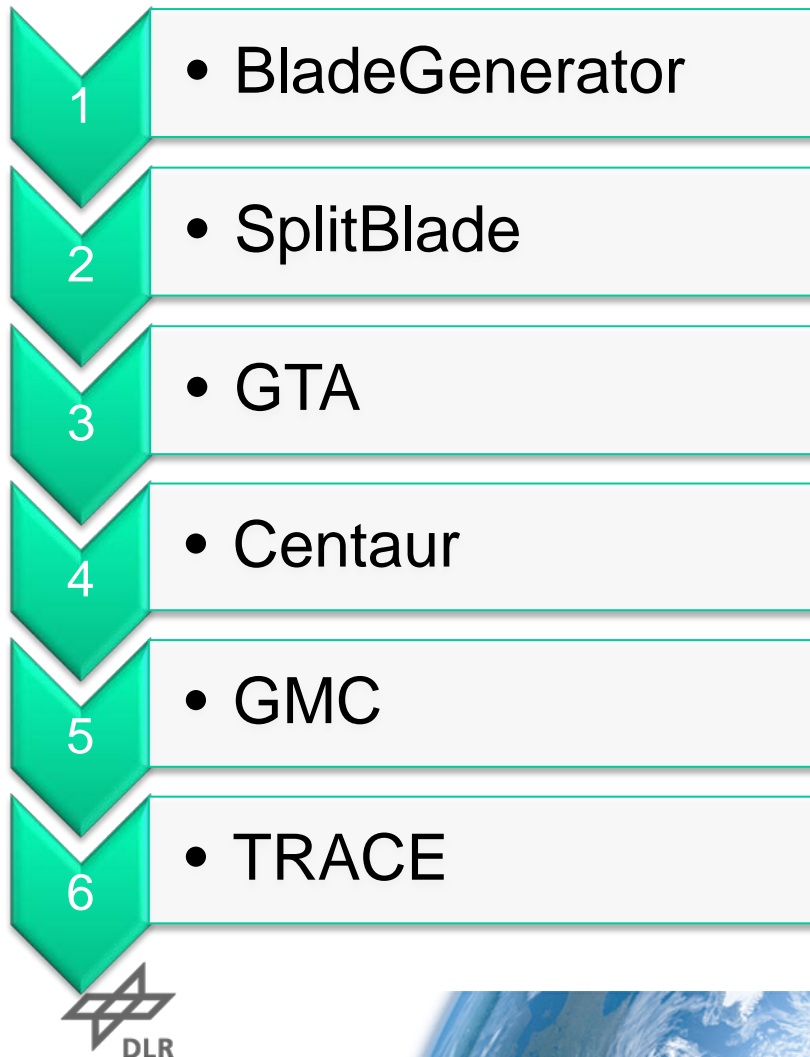
## 2. Optimization description

Optimization model | Processchain | Reference Member



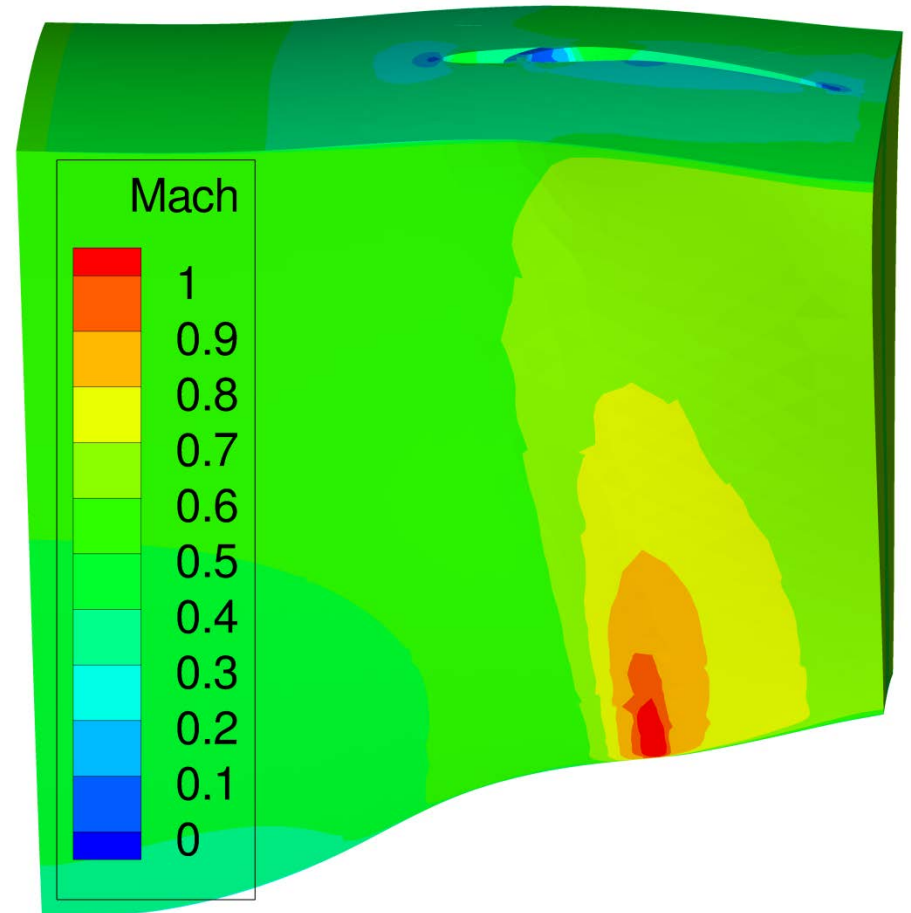
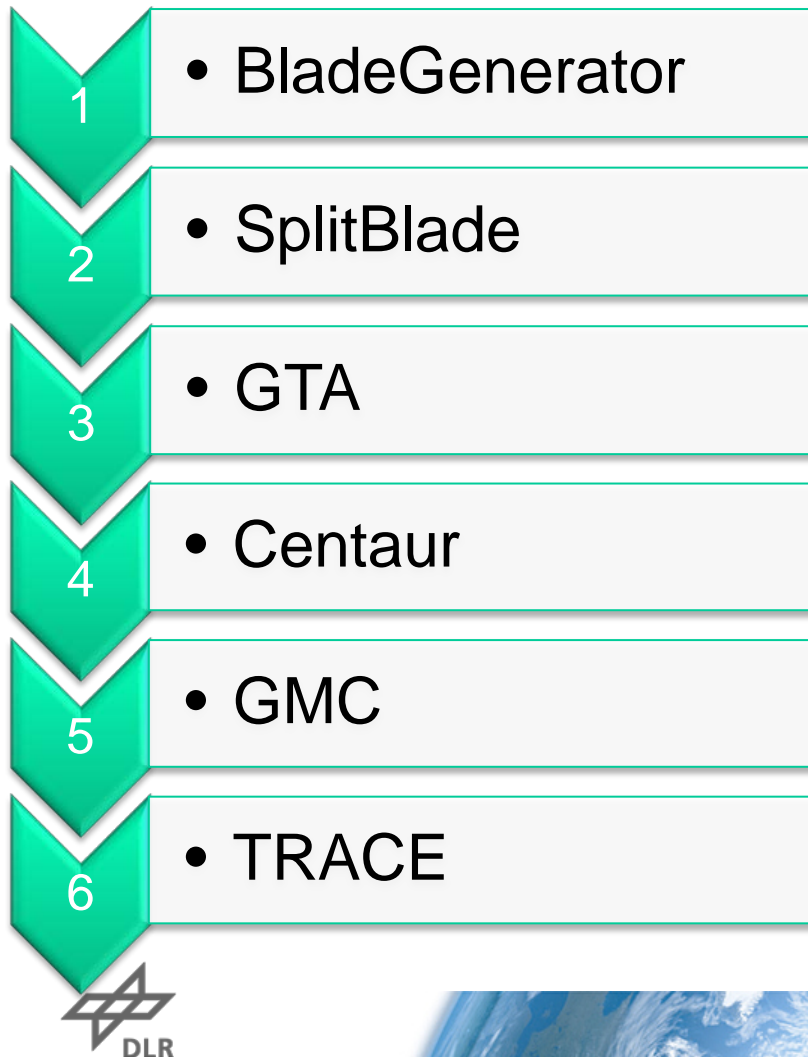
## 2. Optimization description

Optimization model | Processchain | Reference Member



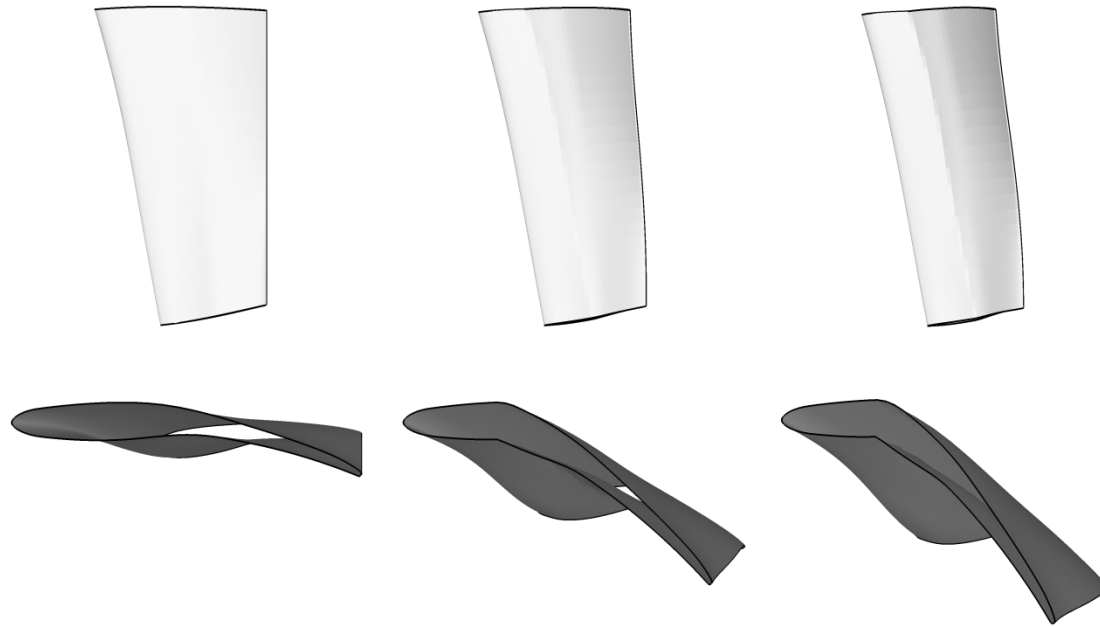
## 2. Optimization description

Optimization model | Processchain | Reference Member



## 2. Optimization description

Processchain | Optimization model | Reference Member \*



OP1

OP2

OP3

**$\omega$  OP1**

**$\omega$  OP2**

**$\omega$  OP3**

Ref. Member

0.1192

0.0968

0.0548

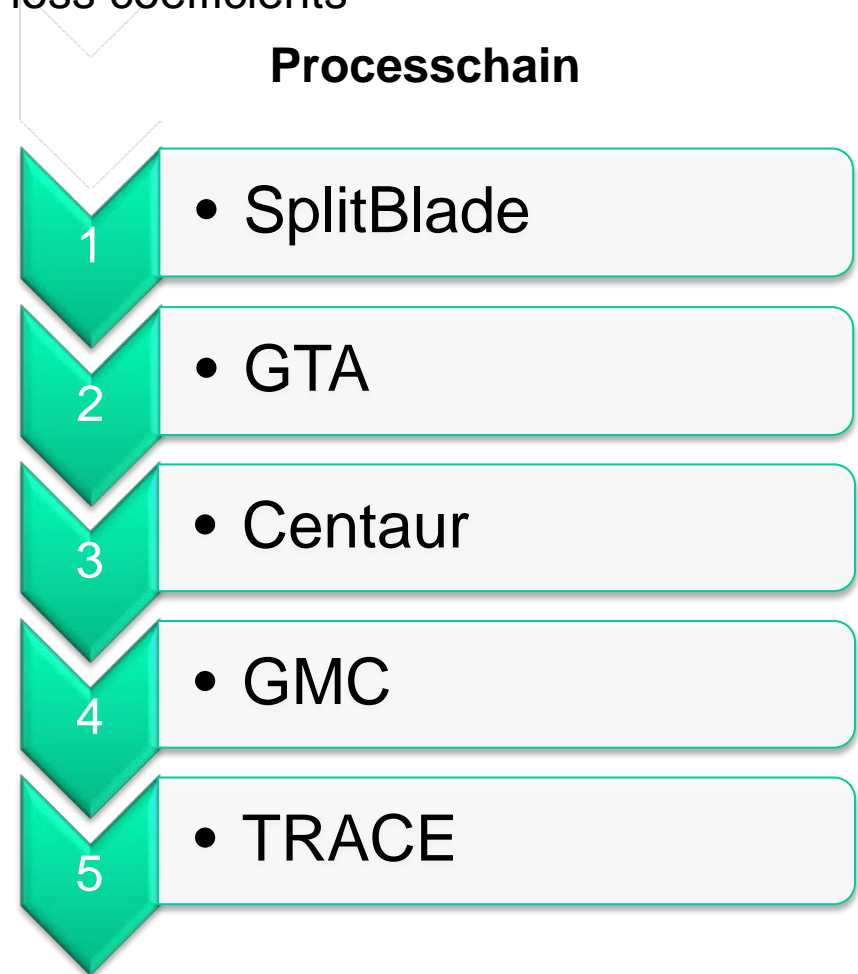
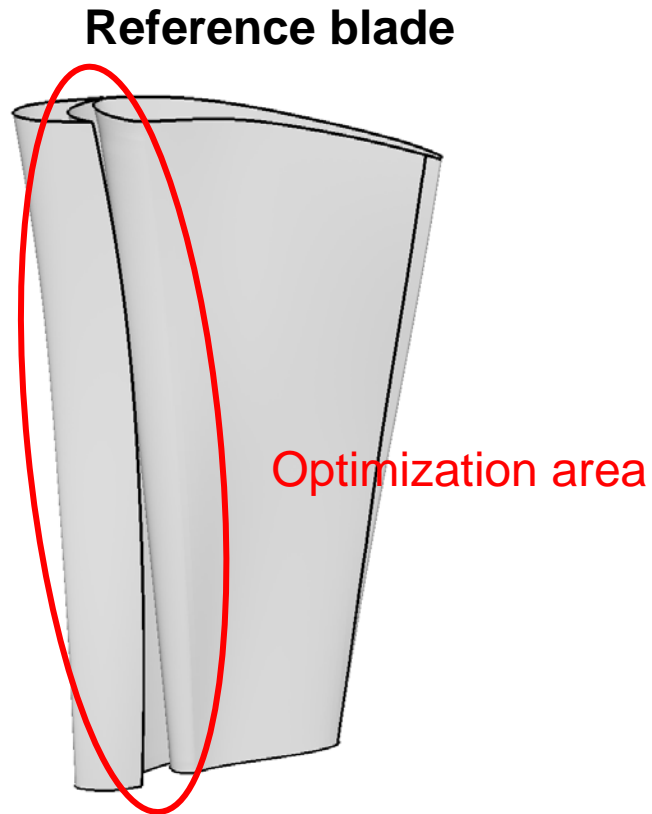
\* from „Design studies of a variable inlet guide vane in a 2.5-stage low pressure compressor, Gouezou, DLR report 2016“



# 3. Optimization 1: Axial gap only

Description | Free parameters | Results

**Objective:** Reduce the total pressure loss coefficients



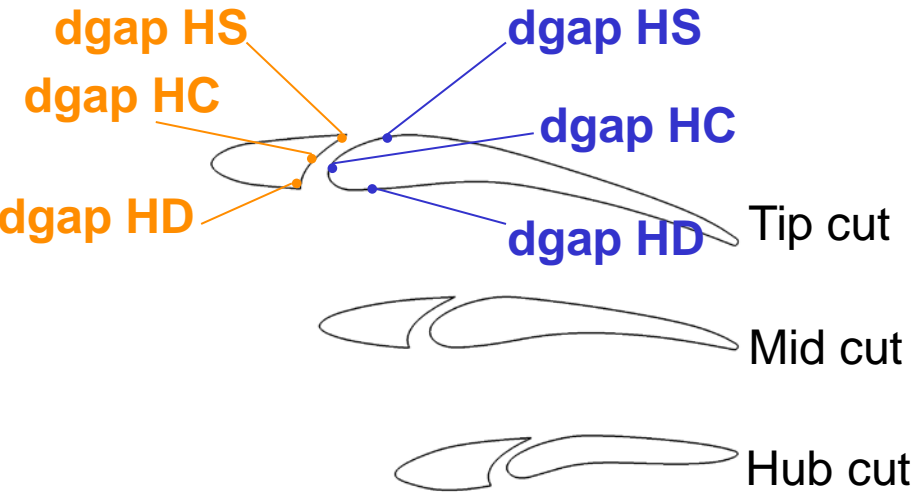
# 3. Optimization 1: Axial gap only

Description | Free parameters | Results

## SplitBlade

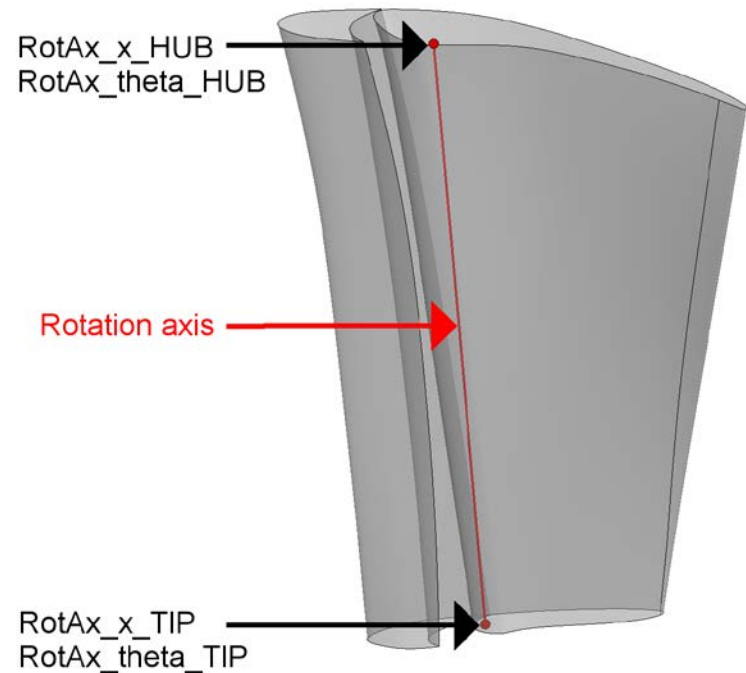
- The gap is constructed with 3 cuts (hub, middle and tip)
- There are 6 free parameters per cut
- There are 4 free parameters for the rotation axis

➔ 22 free parameters



Rotation angle OP2: 28.3°

Rotation angle OP3: 40°



# 3. Optimization 1: Axial gap only

## Description | Free parameters | Results

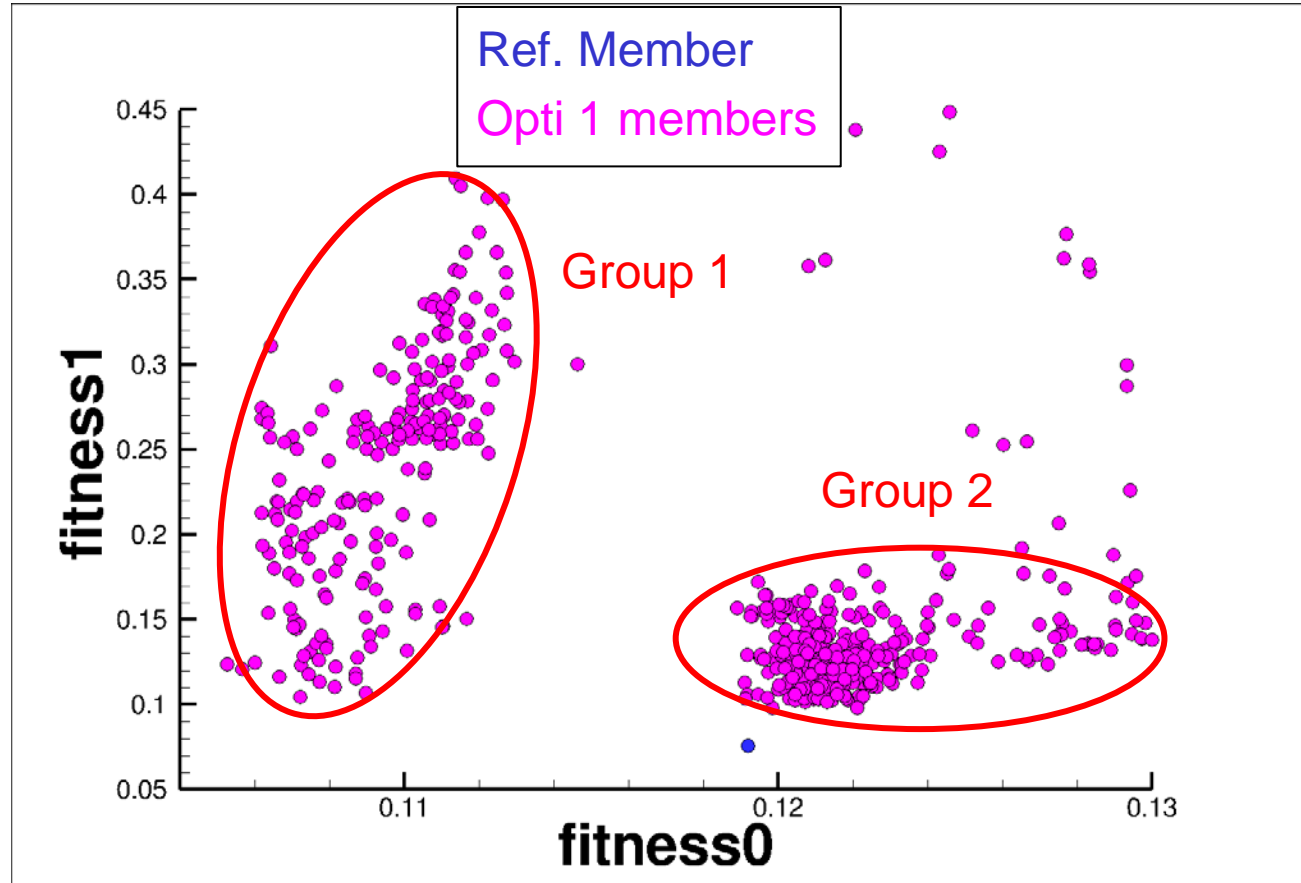
### Database

#### Objectives

- $fitness0 = \omega_{OP1}$
- $fitness1 = \frac{\omega_{OP2} + \omega_{OP3}}{2}$

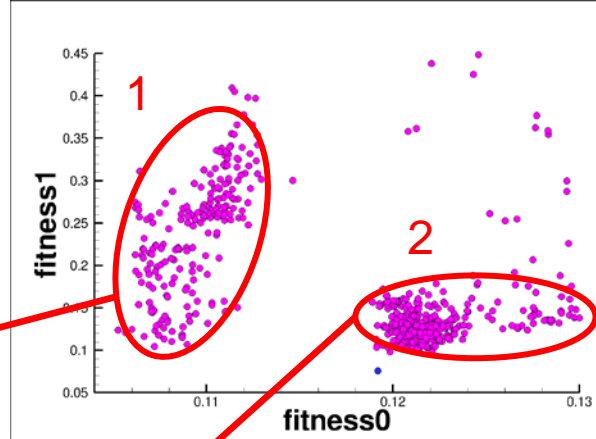
$\omega$  = total pressure loss coefficient

- Two groups of optimized members appear
- Reference member dominates opti-members in fitness1
- Group 1 members show better fitness0



# 3. Optimization 1: Axial gap only

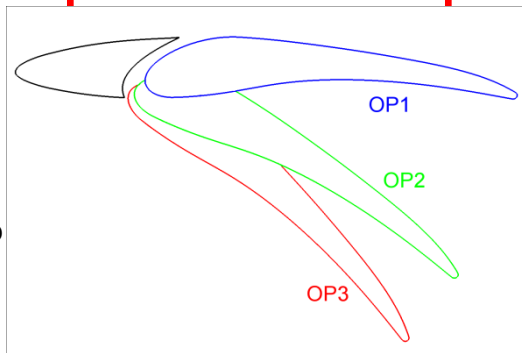
## Description | Free parameters | Results



### Groups of geometries

Three line drawings of airfoil geometries are shown, each with a percentage label to its right. The top drawing is labeled '90%', the middle '50%', and the bottom '10%'. These drawings represent different stages or members of Group 1.

Group 1 selected member



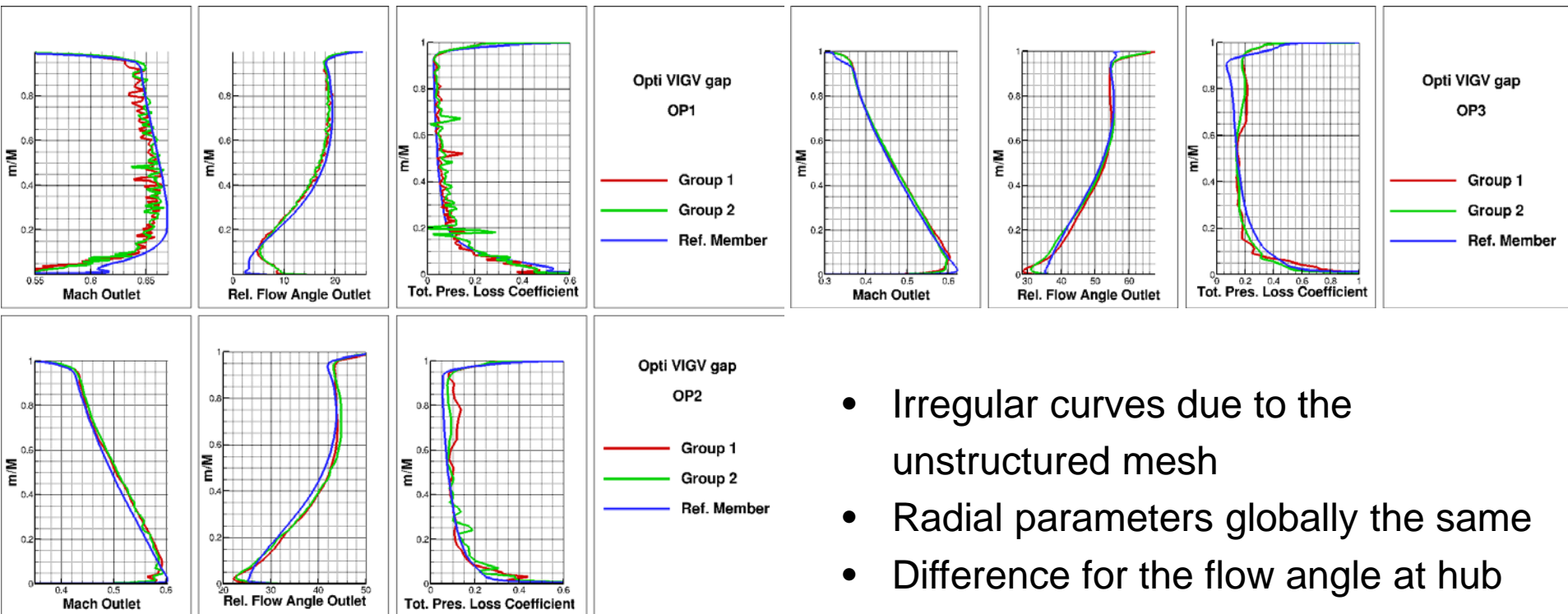
Three line drawings of airfoil geometries are shown, each with a percentage label to its right. The top drawing is labeled '90%', the middle '50%', and the bottom '10%'. These drawings represent different stages or members of Group 2.

Group 2 selected member



# 3. Optimization 1: Axial gap only

## Description | Free parameters | Results



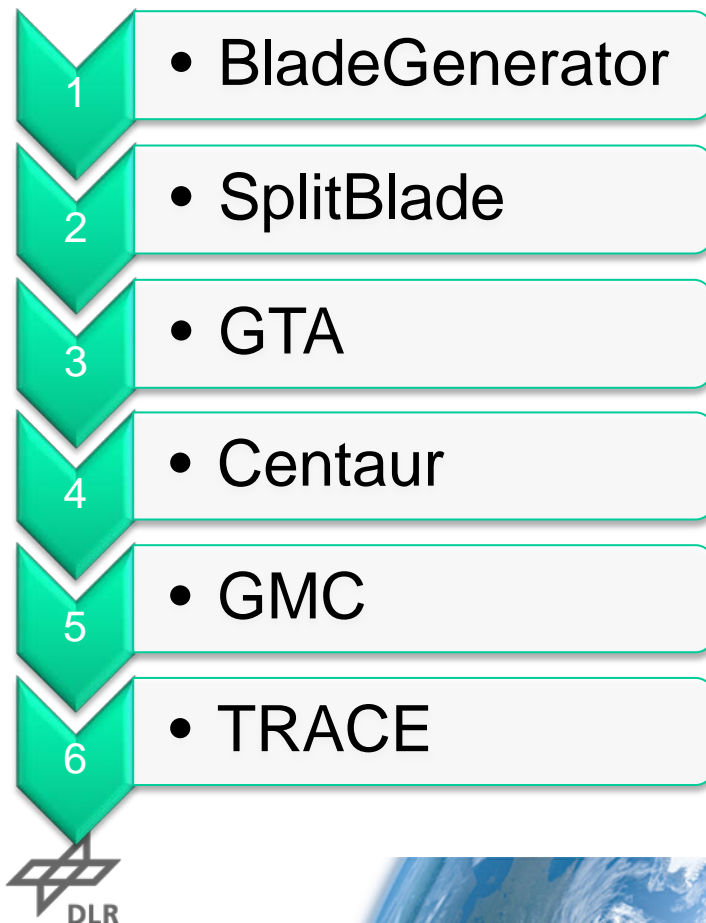
- Irregular curves due to the unstructured mesh
- Radial parameters globally the same
- Difference for the flow angle at hub



## 4. Optimization 2: Whole VIGV

Description | Free parameters | Results

### Processchain



### Fitness functions and restrictions

The objectives are the same:

$$\text{fitness0} = \omega_{OP1}$$
$$\text{fitness1} = \frac{\omega_{OP2} + \omega_{OP3}}{2}$$

7 constraints are added:

- outflow angle (OP1) at 5 radial positions
- outflow angle (OP2) at midspan
- outflow angle (OP3) at midspan



# 4. Optimization 2: Whole VIGV

## Description | Free parameters | Results

The initialization is based on group 2 of optimization 1

### Envelope parameters (BladeGenerator)

- Angles and radii : 18 parameters
- Control Points Suction side : 24
- Control Points Pressure side : 24

66 free parameters

### Gap parameters (Changes for SplitBlade)

- Same RotAx\_theta for the two rotation points
- The RotAx\_x\_TIP is calculated in function of RotAx\_x\_HUB
- Rotation angles OP2/OP3

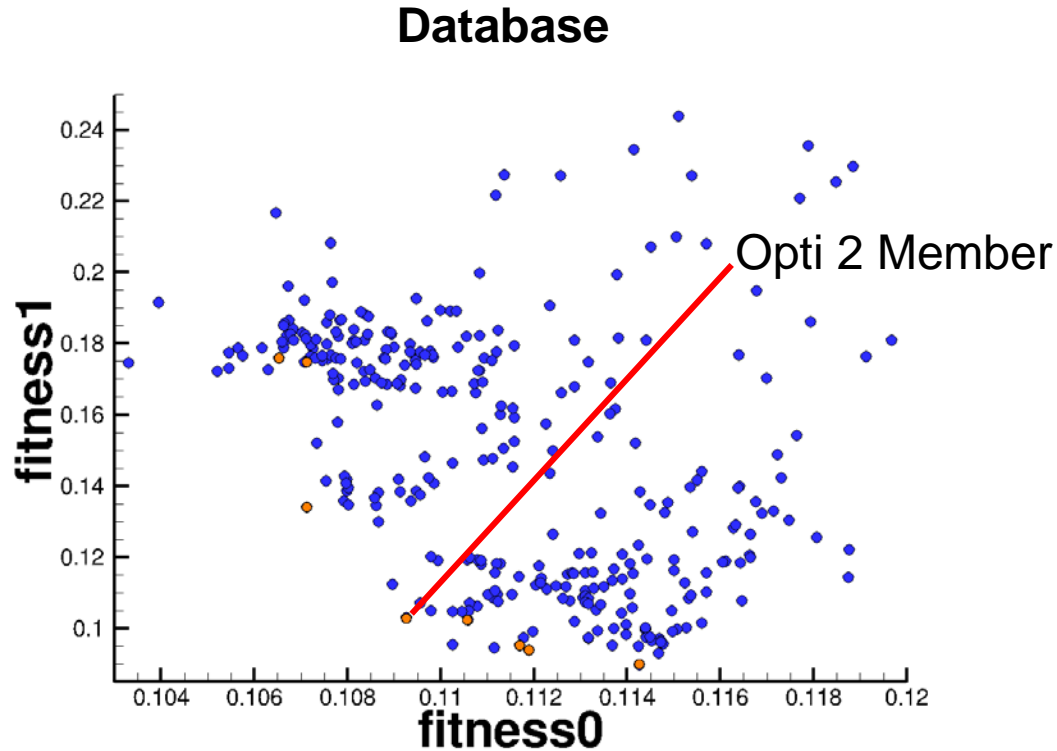
22 free parameters

88 free parameters



# 4. Optimization 2: Whole VIGV

## Description | Free parameters | Results



Opti 2	$\omega$ OP1	$\omega$ OP2	$\omega$ OP3
Opti 2 Member	0.1093	0.1108	0.0952
Opti 1 Member	0.1198	0.1188	0.0774
Ref. Member	0.1192	0.0968	0.0548

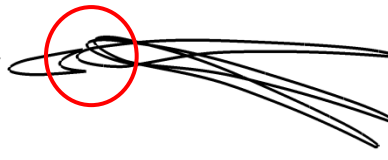
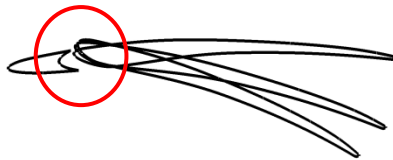


# 4. Optimization 2: Whole VIGV

Description | Free parameters | Results

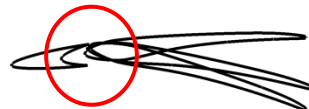
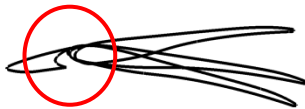
Opti 2 member      Opti 1 member      Reference member

90%



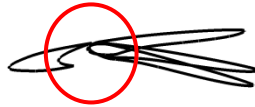
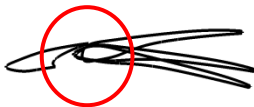
➤ At 90%, step made by opti 1 member is smaller

50%



➤ At 50%, the fixed strut of opti 1 member is more symmetric

10%

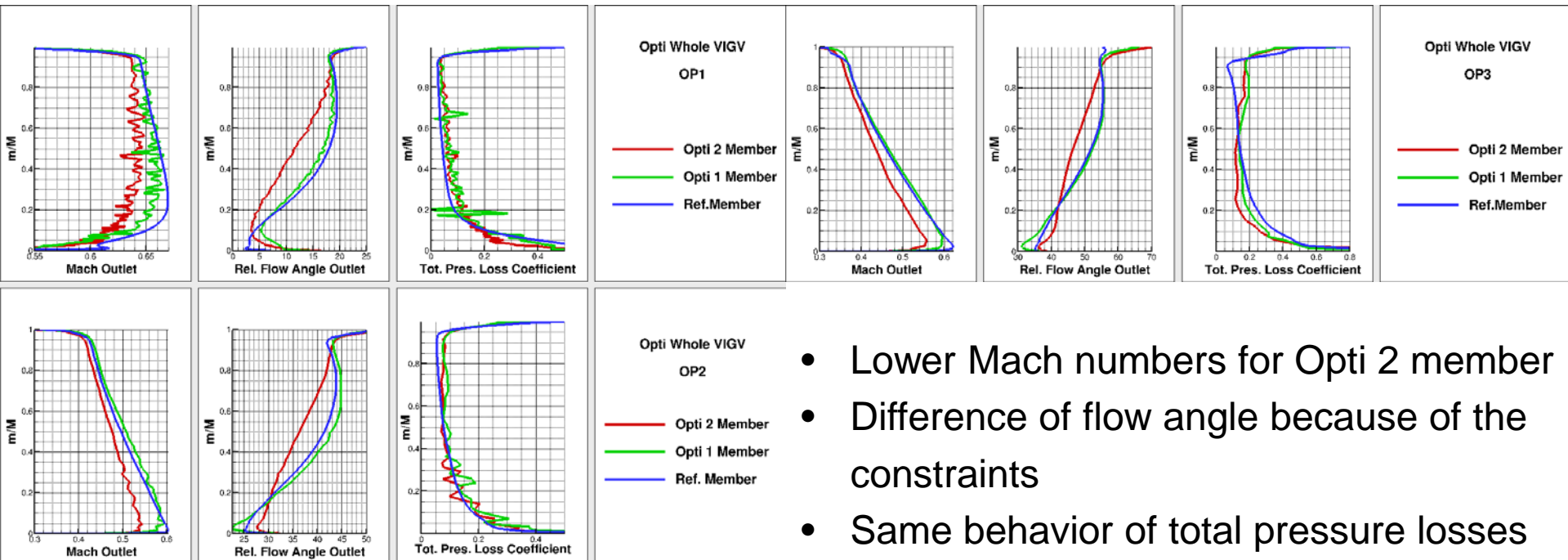


➤ At 10%, the fixed strut of opti 1 member has a pear form



# 4. Optimization 2: Whole VIGV

## Description | Free parameters | Results



- Lower Mach numbers for Opti 2 member
- Difference of flow angle because of the constraints
- Same behavior of total pressure losses for OP1 and OP2



## 5. Conclusion

- A tool to create an axial gap between strut and flap has successfully been integrated in the design process
- 2 VIGV optimizations have been conducted:
  - Optimization 1 has resulted in two groups with big geometric differences
  - The second optimization has given better results than the first optimization only in terms of total pressure losses in OP2 and OP3
  - The geometric differences between optimization 1 and optimization 2 are small, even though the free parameters were increased significantly
- Optimized geometries with axial gap show lower losses in OP1 and higher losses in OP2 and OP3 compared to the reference blade

## Outlook

- Optimization with an initial geometry from the first group of optimization 1
- Study about the secondary axial gap flow
- CFD calculations with the 2.5 stage compressor



# Thank you for your attention!

## Questions?

