

# 1 Impact of neutral boundary-layer turbulence on 2 wind-turbine wakes: A numerical modelling study

3 **Antonia Englberger · Andreas Dörnbrack**

4  
5 Received: DD Month YEAR / Accepted: DD Month YEAR

6 **Abstract** The wake characteristics of a wind turbine in a turbulent bound-  
7 ary layer under neutral stratification are investigated systematically by means  
8 of large-eddy simulations. A methodology to maintain the turbulence of the  
9 background flow for simulations with open horizontal boundaries, without the  
10 necessity of the permanent import of turbulence data from a precursor simula-  
11 tion, was implemented in the geophysical flow solver EULAG. These require-  
12 ments are fulfilled by applying the spectral energy distribution of a neutral  
13 boundary layer in the wind-turbine simulations. A detailed analysis of the  
14 wake response towards different turbulence levels of the background flow re-  
15 sults in a more rapid recovery of the wake for a higher level of turbulence. A  
16 modified version of the Rankine-Froude actuator disc model and the blade ele-  
17 ment momentum method are tested as wind-turbine parametrizations resulting  
18 in a strong dependence of the near-wake wind field on the parametrization,  
19 whereas the far-wake flow is fairly insensitive to it. The wake characteristics  
20 are influenced by the two considered airfoils in the blade element momentum  
21 method up to a streamwise distance of  $14D$  ( $D =$  rotor diameter). In addition,  
22 the swirl induced by the rotation has an impact on the velocity field of the  
23 wind turbine even in the far wake. Further, a wake response study reveals a  
24 considerable effect of different subgrid-scale closure models on the streamwise  
25 turbulent intensity.

26  
27 **Keywords** Atmospheric boundary layer · Large-eddy simulation · Turbu-  
28 lence · Wind-turbine wake

---

A. Englberger  
Institut für Physik der Atmosphäre, DLR Oberpfaffenhofen  
E-mail: antonia.englberger@dlr.de

A. Dörnbrack  
Institut für Physik der Atmosphäre, DLR Oberpfaffenhofen  
E-mail: andreas.doernbrack@dlr.de

## 1 Introduction

Wind turbines operate in the atmospheric boundary layer (ABL) where atmospheric turbulence arises from velocity shear (velocity change with height) and directional shear (wind direction change with height), thermal stratification, low-level moisture, as well as from the interaction of the airflow with vegetation, buildings or terrain (Naughton et al., 2011; Emeis, 2013, 2014). ABL turbulence affects the velocity deficit and the turbulence in the wake, having a large impact on energy production, on fatigue loading, and on the life expectancy of wind turbines. Numerical simulations of wind turbines in the ABL have become an important tool in the investigation of these complex processes. Different numerical approaches exist to simulate the impact of ABL turbulence on wind-turbine wakes. Here, we focus on a large-eddy simulation (LES), being an approved tool to study the turbulence in the ABL (Bellon and Stevens, 2012).

The influence of a turbulent flow on the structure of the wake has been investigated in experimental studies (Medici and Alfredsson, 2006; Chamorro and Porté-Agel, 2009; Zhang et al., 2012) as well as in numerical simulations (Troldborg et al., 2007; Wu and Porté-Agel, 2012). According to their investigations, the wake structure is strongly influenced by the presence of turbulence in the inflow and the wake recovers more rapidly for higher turbulence intensity levels of the incoming flow.

Different methods have been applied to generate a turbulent flow field upstream of the wind turbine. In wind-tunnel experiments, additional roughness elements in front of the wind turbine evoke a turbulent flow, which can be generated by turbulence grids (Medici and Alfredsson, 2006) or obstacles on the floor (Chamorro and Porté-Agel, 2009). Implementing this method in a numerical simulation requires a rather large upstream section, which is computationally expensive, leading to other approaches.

A simple synthetic method avoiding the simulation of atmospheric turbulence was proposed by Mann (1994), e.g. used in Troldborg et al. (2007). The resulting three-dimensional turbulence field is compact and provides turbulence spectra as expected in an ABL. This method, however, is not based on a physical model and only offers a synthetic turbulence field (Naughton et al., 2011). An alternative approach is to couple meteorological data (e.g. wind speed, wind direction, temperature) from a mesoscale simulation on the microscale LES of the wind turbine. However, the two-way coupling as well as the one-way coupling between mesoscale and microscale models, induces different problems (Mirocha et al., 2013; Muñoz-Esparza et al., 2014).

The necessity of synthetic or mesoscale atmospheric parameters can be avoided by the use of a precursor simulation. Wu and Porté-Agel (2012) created a neutral ABL flow forced by a streamwise pressure gradient. The main simulation is initialized with data from the precursor simulation. By applying streamwise periodic boundary conditions, a buffer zone prevented the turbulence in the wake from re-entering the domain and interacting with the wind turbine.

74 Open streamwise boundary conditions do not require a buffer zone. In-  
75 stead, the wind-turbine simulation has to be fed continuously with turbulence  
76 data from a precursor simulation to generate a fully developed turbulent flow  
77 field. Naughton et al. (2011) ensured a turbulent inflow by prescribing instan-  
78 taneous velocity components from the precursor simulation at the inflow plane  
79 at regular time intervals. Witha et al. (2014) realized a turbulent inflow for  
80 an array of wind turbines in a wind park based on a recycling method after  
81 Kataoka and Mizuno (2002). The main simulation used the data from the pre-  
82 cursor simulation for initialization and persistently extracted turbulence from  
83 a region upstream of the wind turbine adding it to the mean inflow profiles.

84 The first goal of our study is to develop and investigate a new method-  
85 ology to generate and maintain a realistic background turbulence field in the  
86 wind-turbine LES with open horizontal boundary conditions, and by avoiding  
87 a continuous turbulent inflow from a precursor simulation. At each timestep  
88 of the wind-turbine LES, the flow field shall be perturbed by velocity fluctu-  
89 ations extracted from a selected state of the precursor simulation of a neutral  
90 ABL. The aim is to maintain the spectral properties of realistic background  
91 turbulence and to control the energy of the applied perturbation fields. Here,  
92 we describe the new methodology and compare our numerical results with  
93 published results from previous simulations and measurements.

94 In addition to a realistic background turbulence field, an LES of wind-  
95 turbine wakes require a detailed knowledge and parametrization of the forces  
96 exerted by a wind turbine on the atmosphere. In a numerical model the wind-  
97 turbine forces can be parametrized as a disc that can either rotate or not.  
98 Alternatively, individual rotating lines represent the blades of the wind tur-  
99 bine. The respective approaches are termed the actuator disc model (ADM)  
100 and the actuator line model (ALM). The impact of wind-turbine parametriza-  
101 tions on the wake has been studied focusing on various aspects.

102 Mikkelsen (2003) investigated the parametrization of a wind turbine with  
103 the ADM and the ALM, extended for a multiplicity of rotor configurations,  
104 e.g. a coned or a yawed rotor. Numerous investigations validating the different  
105 wind-turbine parametrizations were performed by e.g. Ivanell et al. (2008),  
106 Porté-Agel et al. (2010), Wu and Porté-Agel (2011) and Tossas and Leonardi  
107 (2013). All of these studies resulted in a near-wake wind field, sensitive to the  
108 wind-turbine parametrization, whereas the far-wake structure depends mainly  
109 on the background turbulence. Mirocha et al. (2014) implemented the gener-  
110 alized actuator disc wind-turbine parametrization into the Weather Research  
111 and Forecasting (WRF-LES) model. This approach enabled the investigation  
112 of the interaction of a wind turbine with different ABL stratifications, result-  
113 ing in good agreement of the wake characteristics with observations under  
114 weakly convective conditions. Numerous studies explored the impact of the  
115 distribution of the forces. Ivanell et al. (2008) and Tossas and Leonardi (2013)  
116 studied the impact of different smearing parameters of the forces acting on  
117 the atmosphere, resulting in numerical instabilities for a tight volume-force  
118 distribution at the rotor position. Ivanell et al. (2008), Wu and Porté-Agel  
119 (2011) and Gomes et al. (2014) investigated the influence of the number of

120 grid points representing the disc on the wake structure with the result that  
121 the wake characteristics are independent of the resolution, if a minimum of ten  
122 grid points cover the rotor diameter in the spanwise and the vertical directions.  
123 Gomes et al. (2014) also analyzed the effect of the radial dependencies of the  
124 applied forces. A strong sensitivity of the near-wake wind field was found in  
125 contrast to the far-wake behaviour.

126 Here, we apply a modified version of the classical Rankine-Froude ADM  
127 and the blade element momentum (BEM) method for two different airfoils  
128 as wind-turbine parametrizations in our numerical simulations. In the second  
129 part, systematic investigations of the wake characteristics depending on the  
130 two parametrizations, the local blade characteristics, and the rotation of the  
131 disc are made.

132 We implement our turbulence preserving method and both wind-turbine  
133 parametrizations in the multiscale geophysical flow solver EULAG (Prusa  
134 et al., 2008). This LES model resolves all energy containing modes of the turbu-  
135 lent transport and scales larger than the spatial resolution of the computational  
136 grid. Only the turbulence of the smallest unresolved scales is parametrized  
137 using a subgrid-scale (SGS) closure model. The sensitivity of the numerical  
138 results towards different SGS closure models (turbulent kinetic energy (TKE)  
139 closure, Smagorinsky closure) as well as an implicit LES (Grinstein et al.,  
140 2007) constitute the third task investigated.

141 The outline of the paper is as follows: the LES model is presented in Sect. 2,  
142 while the turbulence preserving method is formulated in Sect. 3, and the wind-  
143 turbine models are described in Sect. 4. The results of the numerical simula-  
144 tions studying the influence of the intensity of background turbulence, the  
145 wind-turbine parametrizations, the rotation of the wind turbine and the SGS  
146 closure models on the wake characteristics follow in Sect. 5. Conclusions are  
147 given in Sect. 6.

## 148 **2 Numerical model framework**

149 An inviscid and incompressible flow through a wind turbine is simulated with  
150 the multiscale geophysical flow solver EULAG (Prusa et al., 2008). The geo-  
151 physical flow solver EULAG is at least second-order accurate in time and space  
152 (Smolarkiewicz and Margolin, 1998) and is well suited for massively-parallel  
153 computations (Prusa et al., 2008). It can be run parallel up to a domain de-  
154 composition in three dimensions. A comprehensive description and discussion  
155 of the geophysical flow solver EULAG can be found in Smolarkiewicz and  
156 Margolin (1998) and Prusa et al. (2008).

157 For the numerical simulations conducted herein, the Boussinesq equations  
158 for a flow with constant density  $\rho_0 = 1.1 \text{ kg m}^{-3}$  are solved for the Cartesian  
159 velocity components  $\mathbf{v} = (u, v, w)$  and for the potential temperature pertur-

160 bations  $\Theta' = \Theta - \Theta_0$  (Smolarkiewicz et al., 2007),

$$\frac{d\mathbf{v}}{dt} = -G\nabla\left(\frac{p'}{\rho_0}\right) + \mathbf{g}\frac{\Theta'}{\Theta_0} + \mathbf{v} + \mathbf{M} + \frac{\mathbf{F}}{\rho_0} \equiv \mathcal{R}^v, \quad (1)$$

$$\frac{d\Theta'}{dt} = \mathcal{H} \equiv \mathcal{R}^\Theta, \quad (2)$$

$$\nabla \cdot (\rho_0 \mathbf{v}) = 0, \quad (3)$$

161 where  $\Theta_0 = 301$  K. In Eqs. 1, 2, and 3,  $d/dt$ ,  $\nabla$ , and  $\nabla \cdot$  represent the total  
 162 derivative, the gradient and the divergence, respectively. The quantity  $p'$  rep-  
 163 represents the pressure perturbation with respect to the environmental state and  
 164  $\mathbf{g}$  is the vector of acceleration due to gravity. The factor  $G$  represents geometric  
 165 terms that result from the general, time-dependent coordinate transformation  
 166 (Wedi and Smolarkiewicz, 2004; Smolarkiewicz and Prusa, 2005; Prusa et al.,  
 167 2008; Kühnlein et al., 2012). The SGS terms  $\mathbf{v}$  and  $\mathcal{H}$  symbolise viscous dis-  
 168 sipation of momentum and diffusion of heat,  $\mathbf{M}$  denotes the inertial forces  
 169 of coordinate-dependent metric accelerations and  $\mathbf{F}$  additional external forces  
 170 related to the parametrization of the wind turbine in the geophysical flow  
 171 solver EULAG. The terms  $\mathcal{R}^v$  and  $\mathcal{R}^\Theta$  summarize symbolically all forces in  
 172 the corresponding equations.

173 The acronym EULAG refers to the ability of solving the equations of motions  
 174 either in an Eulerian (flux form) (Smolarkiewicz and Margolin, 1993)  
 175 or in a semi-Lagrangian (advective form) (Smolarkiewicz and Pudykiewicz,  
 176 1992) mode, via

$$\psi^{\xi+1} = \text{LE} \left( \psi^\xi + \frac{1}{2} \Delta t \mathcal{R}^\psi |^\xi \right) + \frac{1}{2} \Delta t \mathcal{R}^\psi |^{\xi+1}, \quad (4)$$

177 where  $\psi = (u, v, w, \Theta)$ ,  $\xi$  denotes the timestep and LE is the corresponding  
 178 finite-difference operator (semi-Lagrangian/Eulerian). In general, the geophys-  
 179 ical flow solver EULAG owes its versatility to a unique design that combines a  
 180 rigorous theoretical formulation in generalized curvilinear coordinates (Smo-  
 181 larkiewicz and Prusa, 2005) with non-oscillatory forward-in-time (NFT) dif-  
 182 ferencing for fluids built on the multi-dimensional positive definite advection  
 183 transport algorithm (MPDATA), which is based on the convexity of upwind  
 184 advection (Smolarkiewicz and Margolin, 1998; Prusa et al., 2008) and a ro-  
 185 bust, exact-projection type, elliptic Krylov solver (Prusa et al., 2008). The flow  
 186 solver has been applied to a wide range of scales simulating various problems  
 187 like turbulence (Smolarkiewicz and Prusa, 2002), flow past complex or mov-  
 188 ing boundaries (Wedi and Smolarkiewicz, 2006; Kühnlein et al., 2012), gravity  
 189 waves (Smolarkiewicz and Dörnbrack, 2008; Doyle et al., 2011) or even solar  
 190 convection (Smolarkiewicz and Charbonneau, 2013). The turbulence closure  
 191 in the geophysical flow solver EULAG can be described by a TKE model, a  
 192 Smagorinsky model or an implicit LES, with no turbulence closure model due

193 to not considering the diffusion process. The implicit LES properties of nu-  
 194 merical solvers based on MPDATA are documented in e.g. Margolin and Rider  
 195 (2002), Margolin et al. (2002) and Margolin et al. (2006) for structured grids.  
 196 A detailed description of an implicit LES is given in Grinstein et al. (2007).

### 197 3 Turbulence preserving method

198 The basic idea of our new methodology that preserves the background tur-  
 199 bulence in an LES of a flow through a wind turbine is to extract velocity  
 200 perturbations from a precursor simulation of the neutral ABL. The velocity  
 201 fields are used to disturb the wind-turbine simulation in a special manner  
 202 as described below. For this purpose, a precursor simulation of the turbulent  
 203 neutral ABL has to be conducted.

#### 204 3.1 Precursor simulation

205 To drive the neutral ABL flow, an additional forcing  $-u_*^2/H$  is applied for the  
 206  $u$ -component of Eq. 1, where  $H$  is the height of the computational domain.  
 207 Sensitivity tests revealed that a value of the friction velocity  $u_* = 0.4 \text{ m s}^{-1}$   
 208 results in a realistic pressure gradient of the ABL. This forcing is comparable  
 209 to the streamwise mean pressure gradient force applied in Wu and Porté-Agel  
 210 (2012). The precursor simulation is performed with the same number of grid  
 211 points as the wind-turbine simulations, but with periodic boundary conditions  
 212 in the horizontal directions. The initial wind speed is set to zero, and the drag  
 213 coefficient in the surface parametrization is set to 0.1.

214 Applying only the above forcing, it is a long lasting process until the pre-  
 215 cursor simulation is in an equilibrium state. Additional velocity gradients in  
 216 the neutral flow can serve as a trigger, breaking the symmetry and acting as  
 217 a seed for turbulence to develop. Therefore, the precursor simulation is dis-  
 218 turbed by inserting an obstacle in the domain for a few timesteps. The flow  
 219 around this obstacle enhances the velocity gradients in the neutral ABL flow,  
 220 and the equilibrium state of the precursor simulation is attained more rapidly.

#### 221 3.2 Methodology

222 The perturbation velocities  $\mathbf{u}_p^*|_{i,j,k}^\xi$  are extracted from the precursor simulation  
 223 according to,

$$223 \mathbf{u}_p^*|_{i,j,k}^\xi = \alpha \cdot \beta \cdot \underbrace{\left( \mathbf{u}_p|_{i^*,j,k} - \underbrace{\frac{1}{n \cdot m} \sum_{i=1}^n \sum_{j=1}^m \mathbf{u}_p|_{i,j,k}}_I \right)}_{II}, \quad (5)$$

224 where  $\mathbf{u}_p|_{i^*,j,k}$  is the velocity vector of the precursor simulation in an equilib-  
 225 rium state and the term  $I$  in Eq. 5 denotes the height-averaged mean value of  
 226 the corresponding wind component at each grid point  $i$ ,  $j$ , and  $k$ . The indices  
 227 of the grid points are denoted by  $i = 1 \dots n$ ,  $j = 1 \dots m$ , and  $k = 1 \dots l$  in the  
 228  $x$ ,  $y$ , and  $z$  directions, respectively.

229 The perturbation velocity from Eq. 5 contributes to the velocity field of  
 230 the wind-turbine simulation  $\mathbf{u}|_{i,j,k}^\xi$  at the initial timestep  $\xi = 0$  and at each  
 231 following timestep  $\xi$ . The values of the precursor simulation  $\mathbf{u}_p|_{i^*,j,k}$  are shifted  
 232 in the streamwise direction by one grid point every timestep  $\xi$ , symbolized by  
 233  $i^* = i + \xi^*$ , with  $i^* \in [1, n]$  and  $\xi^*$  representing the number of timesteps since  
 234 the start of the simulation. Furthermore, the difference as denoted by  $II$  in  
 235 Eq. 5 is multiplied with a random number  $\beta$  ranging from  $-0.5$  to  $0.5$ . Both  
 236 the grid point shift and the random number multiplication are necessary to  
 237 only apply the spectral energy distribution of the precursor simulation instead  
 238 of impressing individual flow patterns onto the wind-turbine simulation. To  
 239 account for different magnitudes of the background turbulence, the term  $II$  in  
 240 Eq. 5 is additionally multiplied by a factor  $\alpha$ , representing the amplitude of the  
 241 turbulence perturbations (hereafter referred to as perturbation amplitude).

242 Applying this method maintains the spectral properties of the turbulent  
 243 fluctuations in the wind-turbine simulation. It offers several possibilities for  
 244 the numerical scheme:

- 245 1. Periodic boundary conditions and a buffer zone can be avoided, enabling  
 246 open inflow and outflow Neumann boundary conditions and minimising  
 247 the domain size of the simulation.
- 248 2. The perturbation data from the precursor simulation are imported only  
 249 once and are stored in three 3D fields ( $u$ ,  $v$ ,  $w$ ) during the wind-turbine  
 250 simulation.
- 251 3. The method is computationally very efficient, as it allows to reapply the  
 252 background turbulence of one precursor simulation to a variety of wind-  
 253 turbine simulations.
- 254 4. The response of a wind turbine to different intensities of the background  
 255 turbulence can be easily investigated by changing the parameter  $\alpha$  in Eq. 5.

### 256 3.3 Validation of the turbulence preserving method

257 We performed a simulation applying term  $I$  from Eq. 5 as wind field. In ad-  
 258 dition, the spectral energy distribution of the precursor simulation is applied  
 259 with the prescribed methodology. After integrating for the same amount of  
 260 time as in the following wind-turbine simulations, this simulation resulted in  
 261 the same values of  $\langle u \rangle_t$ ,  $\langle v \rangle_t$  and  $\langle w \rangle_t$ , as well as  $\sigma_u$ ,  $\sigma_v$  and  $\sigma_w$   
 262 with  $\sigma_i = \sqrt{i'^2}$  as the precursor simulation, validating the mechanism of the  
 263 turbulence preserving method.

## 4 Wind-turbine Parametrization

### 4.1 Parametrization of the forces

The classical Rankine-Froude theory is the simplest ADM representation of turbine-induced forces in a numerical model where the disc covers the span of the blades. It was introduced by Froude (1889) who continued the work of Rankine (1865) on the momentum theory of propellers. The forces induced by a wind turbine are basically parametrized as a 1D thrust force, which is constant over the disc. Despite its simplicity, this non-rotating ADM has been widely used in LES as it provides reliable results on coarse grids (Calaf et al., 2010; Porté-Agel et al., 2010; Wu and Porté-Agel, 2011; Tossas and Leonardi, 2013; Meyers and Meneveau, 2013). A wind turbine rotates and the incoming profiles of the horizontal wind speed are often vertically sheared ( $\partial \mathbf{u} / \partial z \neq 0$ ). Both processes limit the applicability of the simple ADM parametrization. To circumvent these limitations and to enable an investigation of the impact of the local blade characteristics by comparing to the results of the BEM parametrization (Manwell et al., 2002; Hansen, 2008), we apply a modified version of the Rankine-Froude ADM considering the axial force  $F_x(y, z)$  in the streamwise ( $x$ ) direction and the tangential force  $F_\Theta(y, z)$  perpendicular to  $F_x$  in the  $y$ - $z$  plane,

$$|F_x|_{x_0, y, z} = \frac{1}{2} \rho_0 c'_T A_{x_0, y, z} \langle u_{x_0, y, z}^2 \rangle_t, \quad (6)$$

$$|F_\Theta|_{x_0, y, z} = \frac{1}{2} \rho_0 c'_P A_{x_0, y, z} \langle u_{x_0, y, z}^2 \rangle_t \frac{u_{x_0, y, z}}{\Omega r_{x_0, y, z}}. \quad (7)$$

Both forces  $F_x$  and  $F_\Theta$  result in the total force  $\mathbf{F}|_{x_0, y, z}$  (Hansen, 2008), with

$$\mathbf{F}|_{x_0, y, z} = \mathbf{F}_x|_{x_0, y, z} + \mathbf{F}_\Theta|_{x_0, y, z}, \quad (8)$$

where the centre of the rotor is defined by the grid-point coordinates  $x_0$ ,  $y_0$  and  $z_h$  (hub height). In Eqs. 6 and 7,  $c_T$  represents the thrust coefficient ( $c'_T = c_T / (1 - a)^2$ ) and  $c_P$  the power coefficient ( $c'_P = c_P / (1 - a)^3$ ). The factor  $a$  corresponds to the axial induction factor and can be derived from the one-dimensional momentum theory to a value of  $1/3$  for an ideal rotor (Betz, 1926).  $A_{x_0, y, z}$  is the area of the rotor at position  $x_0$  covered by grid points in the  $y$ - $z$  plane,  $\Omega$  is the angular velocity of the turbine and  $r_{x_0, y, z}$  the radial position inside the rotor ( $0 \leq r_{x_0, y, z} \leq R$ ), with  $R = D/2$  and  $D$  representing the diameter of the wind-turbine rotor. The time-averaged value of the squared streamwise velocity component at the rotor position  $x_0$ ,  $y$ ,  $z$  is denoted by  $\langle u_{x_0, y, z}^2 \rangle_t$ .

A great improvement of the simple momentum theory was the classical BEM method by Glauert (1963). This method accounts for local blade characteristics, as it enables calculation of the steady loads as well as the thrust and the power for different wind speeds, rotational speeds, and pitch angles of

**Table 1** The crucial characteristics of the three different parametrizations A, B, and C of the wind turbine used in this study.

Parametrization	A	B + C
Name	MMT	BEM
Characteristics	Eq. 6 Eq. 7	Eq. 9 Eq. 10

299 the blades. The axial and tangential forces of the BEM method are represented  
300 in Eqs. 9 and 10,

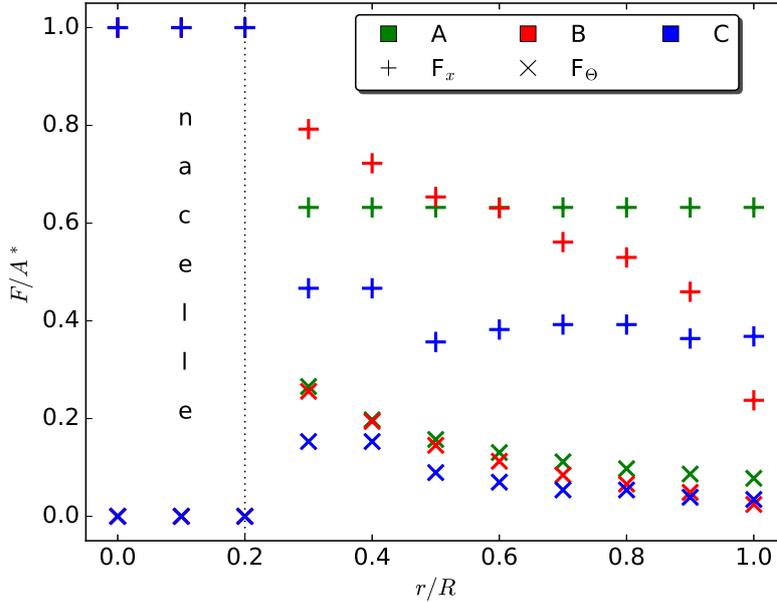
$$|F_x|_{x_0,y,z} = \frac{1}{2} \rho_0 \frac{Bc}{2\pi r_{x_0,y,z}} (c_L \cos \Phi + c_D \sin \Phi) \times A_{x_0,y,z} \frac{u_{x_\infty,y,z}^2 (1-a)^2}{\sin^2 \Phi} \quad (9)$$

$$|F_\theta|_{x_0,y,z} = \frac{1}{2} \rho_0 \frac{Bc}{2\pi r_{x_0,y,z}} (c_L \sin \Phi - c_D \cos \Phi) \times A_{x_0,y,z} \frac{u_{x_\infty,y,z} (1-a) \Omega r_{x_0,y,z} (1+a')}{\sin \Phi \cos \Phi}. \quad (10)$$

301 Here,  $B$  represents the number of blades,  $c$  is the chord length of the blade,  
302  $c_L$  is the lift coefficient,  $c_D$  is the drag coefficient,  $\Phi$  is the angle between the  
303 plane of rotation and the relative streamwise velocity, and  $a'$  is the tangential  
304 induction factor. Following Hansen (2008), we calculate  $a$  and  $a'$  by an iterative  
305 procedure from the airfoil data. The upstream velocity  $u_{x_\infty,y,z}$  is taken at the  
306 first upstream grid point in the  $x$ -direction and the corresponding  $y$  and  $z$   
307 coordinates. With the exception of  $\rho_0$  and  $B$ , all other parameters appearing  
308 in Eqs. 9 and 10 depend on the radius  $r_{x_0,y,z}$  and vary spatially.

309 In this work, the modified version of the Rankine-Froude ADM as well  
310 as the BEM parametrization are implemented via Eq. 8 in the geophysical  
311 flow solver EULAG. The forces are treated implicitly in the numerical scheme  
312 according to Eq. 4. In the geophysical flow solver EULAG, the rotor of a wind  
313 turbine is not implemented as a real circular obstacle (e.g. grid-point blocking  
314 as in Heimann et al. (2011)) or a permeable rotor (Witha et al., 2014; Tossas  
315 and Leonardi, 2013; Gomes et al., 2014). Instead, at every grid point covered by  
316 the rotor, the velocity field experiences the turbine-induced force  $\mathbf{F}$  according  
317 to Eq. 1. This implementation is inspired by the immersed boundary method,  
318 successfully applied in the geophysical flow solver EULAG by Smolarkiewicz  
319 and Winter (2010). The implicit treatment of the forces in Eq. 4 has a positive  
320 effect on the timestep, because there are no large velocity gradients between  
321 the rotor area and its surroundings.

322 Altogether, three different parametrizations of wind-turbine induced forces  
323 are implemented in the geophysical flow solver EULAG. The respective param-  
324 etrizations A, B, and C are listed together with their main characteristics in  
325 Table 1. It should be noted that the parametrizations B and C are essentially  
326 the same, however, the airfoil data applied in B and C differ. The radial distri-



**Fig. 1** Radial distributions of the axial and tangential forces  $F_x$  and  $F_\Theta$  normalized by the area  $A^*$  for the different wind-turbine parametrizations A, B, and C of Table 1. The values of  $F_x$  and  $F_\Theta$  are normalized by the maximum of the axial force at the nacelle, which is the same in all three parametrizations. The axial forces are represented by (+) and the tangential forces by ( $\times$ ). They are plotted for each discrete position of the rotor, assuming 21 grid points are covering the rotor with radius  $R$ . The nacelle covers 20 % of the blades, denoted by the dotted vertical line. For the calculation of the forces in these schematic illustration, a rotor diameter of 100 m is assumed, together with a rotation frequency  $\Omega = 7$  r.p.m. and a constant upstream velocity  $u_{x_\infty, y, z} = 8 \text{ m s}^{-1}$ .

327 butions of the respective axial and tangential forces are depicted in Fig. 1. In  
 328 each parametrization, a nacelle is represented within  $r/R \leq 0.2$  by a stronger  
 329 drag force in comparison to the blade values and no lift force. The size of the  
 330 parametrized nacelle is large compared to a real wind turbine, because the  
 331 numerical resolution demands enough grid points representing the nacelle to  
 332 avoid instabilities. The tower is not considered in our parametrizations as it  
 333 is not the major source of turbulence.

334 Parametrization A represents the modified version of the Rankine-Froude  
 335 ADM, hereafter referred to as modified momentum theory (MMT). It can be  
 336 applied for a rotating actuator with  $F_\Theta \neq 0$  or for a non-rotating actuator  
 337 with  $F_\Theta = 0$ . Parametrization A can be regarded as a simplified version of  
 338 parametrization B, as the values of  $c'_{T_{blade}} = 1.27$  and  $c'_{P_{blade}} = 0.87$  in Eqs.  
 339 6 and 7 are deduced from parametrization B. These prescribed values are  
 340 comparable to other studies (Meyers and Meneveau, 2013).

341 The BEM method is used to investigate the influence of the blade structure.  
 342 The airfoil data are taken from two different wind turbines. The 10 MW refer-  
 343 ence wind turbine from DTU (Technical University of Denmark) referred to as  
 344 parametrization B (Mark Zagar (Vestas), personal communication, 2015) and  
 345 the three-blade GWS/EP-6030x3 rotor (Wu and Porté-Agel, 2011) referred to  
 346 as parametrization C. For both wind turbines, the rotor radius as well as the  
 347 chord length of the blades are scaled to a rotor diameter of 100 m, to make the  
 348 results comparable to each other. The most relevant wind-turbine parameters  
 349 used for parametrizations B and C are listed in the Appendix.

350 For the nacelle,  $c'_{T_{nacelle}} = 1.48$  and  $c'_{P_{nacelle}} = 0$  are chosen in all three  
 351 parametrizations. The value of the drag coefficient of the nacelle of 1.0 agrees  
 352 with the drag coefficient interval of cylindrically shaped bluff bodies between  
 353 0.8 and 1.2 (Schetz and Fuhs, 1996), and has also been used e.g. in El Kasmi  
 354 and Masson (2008).

## 355 4.2 Application of the forces

356 The numerical simulations conducted in this study are performed on an equidis-  
 357 tant Cartesian mesh with grid spacings  $\Delta x$ ,  $\Delta y$  and  $\Delta z$ , in the stream-  
 358 wise, lateral and vertical directions, respectively. It must be noted, that all  
 359 parametrizations A, B, and C are coded to perform properly in terrain-following  
 360 coordinates with variable vertical grid spacings over hilly terrain.

361 To calculate the forces of the actuator, we use polar coordinates that serve  
 362 as a local mesh. The centre coordinate of the polar mesh is the centre of  
 363 the rotor. From this position, the polar mesh is described by a very fine grid  
 364 with  $\Delta r = R/1000$  as radial step size and  $\Delta\varphi = 1^\circ$  as azimuthal step size. The  
 365 step sizes in the radial and azimuthal directions are fine enough to minimize  
 366 the errors that would result from calculating the forces on a Cartesian mesh  
 367 (Ivanell et al., 2008). The computational costs arising from such a fine polar  
 368 mesh are insignificant, as the disc is always at the same position, making this  
 369 calculation of the actuator force in polar coordinates  $F_{r,\theta,z}$  only necessary  
 370 once.

371 The force acting on each polar grid point  $F_{r,\theta,z}$  is transformed to the  
 372 corresponding force in Cartesian coordinates  $F_{x,y,z}^* = \mathcal{M}_{x,y,z} \cdot F_{r,\theta,z}$  through  
 373 the transformation matrix  $\mathcal{M}_{x,y,z}$ . The force  $F_{x,y,z}^*$  contributes to a certain  
 374 fraction  $\mu \in [0, 1]$  to the actuator force  $F_{x,y,z} = \mu \cdot F_{x,y,z}^*$ . The fraction  $\mu$  is  
 375 determined by the ratio of the grid-cell volume of the polar coordinate and  
 376 the corresponding Cartesian coordinate, i.e.  $\mu = 1$  if the Cartesian grid point  
 377 is completely covered by the rotor and  $\mu = 0$  in case of a rotor-free grid point.  
 378 At the edge of the rotor, the fraction  $\mu < 1$ , because the Cartesian grid cell is  
 379 not completely covered by the local polar mesh representing the rotor.

380 A smearing of the turbine-induced forces in the axial as well as in the  
 381 radial direction is necessary to avoid numerical instabilities. As a first step,  
 382 the forces from Eq. 8 are additionally distributed in the streamwise direction.  
 383 This approach is performed for all parametrizations. The forces in Eq. 8 are

384 smeared with a 1D Gaussian function in the  $x$ -direction,

$$F_{s_x} = \frac{1}{\sqrt{\pi}\sigma} \exp\left(-\frac{(x-x_0)^2}{\sigma^2}\right). \quad (11)$$

385 Similar to other studies (Meyers and Meneveau, 2013), the value of  $\sigma$  is set to  
386 1.5 and is given in absolute values of the radius.

387 In parametrization A, the axial force  $F_x$  in the  $y$ - $z$  plane only varies with  
388 the incoming velocity across the rotor. A moderate velocity gradient results in  
389 very similar  $F_x$  values and generates large gradients at the edges of the rotor.  
390 An additional two-dimensional smearing  $F_{s_{y-z}}$  in the  $y$ - $z$  plane is introduced  
391 to avoid too sharp radial gradients in the turbine-induced forces between the  
392 rotor area and the immediate surroundings. The forces of the schematic illustra-  
393 tion in Fig. 1 decrease with a step function over the last three grid points  
394  $\in [0.8r/R, 1.0r/R]$ . The force at each of these outer region grid points is half  
395 of the force of the corresponding nearest inner neighbour grid point.  $F_{s_{y-z}}$   
396 is not applied for the forces in the BEM method, as the parameters in Eqs. 9  
397 and 10 already decrease with increasing  $r$ .

398 The values of the smearing parameters and of the step function applied  
399 on the forces in the  $y$ - $z$  plane in parametrization A are chosen in such a way  
400 that the integrated force distributed in three dimensions is the same as in the  
401 two-dimensional case without smearing. By combining the smearing in the  $x$ -  
402 direction  $F_{s_x}$  and the smearing in the  $y$ - $z$  plane  $F_{s_{y-z}}$ , the difference of the  
403 forcings between a 2D and a 3D disc is less than 1% for 21 grid points per  
404 disc and decreases for a finer resolution.

405 The parametrization  $\mathbf{F}|_{x_0,y,z}$  (Eq. 8) together with the coordinate trans-  
406 formation  $F_{x,y,z}$  and the applied smearing in the axial  $F_{s_x}$  and radial  $F_{s_{y-z}}$   
407 directions result in a total parametrized force,

$$\mathbf{F}|_{x,y,z} = \mathbf{F}|_{x_0,y,z} \cdot F_{x,y,z} \cdot F_{s_x} \cdot F_{s_{y-z}}, \quad (12)$$

408 where the wind-turbine induced force  $\mathbf{F}|_{x,y,z}$  corresponds to the force  $\mathbf{F}$  in Eq.  
409 1.

#### 410 4.3 Validation of the wind-turbine parametrization

411 We validate our numerical results for the wind-turbine parametrizations A,  
412 B, and C at the rotor position  $(x_0, y, z)$  and in the wake  $(x_w, y, z)$ , whereby  
413  $x_w \geq x_0$ , with theoretical wind predictions from the one-dimensional momen-  
414 tum theory,

$$u_{x_0,y,z} = u_{x_\infty,y,z}(1-a), \quad (13)$$

$$u_{x_w,y,z} = u_{x_\infty,y,z}(1-2a), \quad (14)$$

415 where  $a$  is the axial induction factor defined as

$$a := \frac{u_{x_\infty,y,z} - u_{x_0,y,z}}{u_{x_\infty,y,z}}. \quad (15)$$

**Table 2** Parameters for the two different wind turbines (wind turbine 1 in laminar flow and wind turbine 2 in the turbulent ABL) with the rotor diameter  $D$ , the hub height  $z_h$ , the spatial resolution  $\Delta$ , the rotation frequency of the blades  $\Omega$  in revolutions per minute (r.p.m.), the location of the centre of the rotor in the simulated domain, as well as the velocity at the hub height  $u_{x,y,z_h}$  of the wind turbine and the vertical profile of the incoming velocity  $u_{x_\infty,y,z}$ . In the prescribed logarithmic wind profile,  $u_*$  represents the friction velocity,  $\kappa$  is the von Karman constant ( $\kappa=0.4$ ), and  $z_0$  is the roughness length. All simulations are performed on an equidistant grid with the spacing  $\Delta = \Delta x = \Delta y = \Delta z$ .

parameters	wind turbine 1	wind turbine 2
grid points	512x128x128	512x64x64
$D$ (m)	4	100
$z_h$ (m)	4	100
$\Delta$ (m)	0.1	5
$\Omega$ (r.p.m.)	0	7
rotor centre	$x_0 = 120\Delta$ $y_0 = 64\Delta$ $z_h = 40\Delta$	$x_0 = 60\Delta$ $y_0 = 32\Delta$ $z_h = 20\Delta$
$u_{x,y,z_h}$ (m s <sup>-1</sup> )	0.08 and 0.10	8.0
$u_{x_\infty,y,z}$	constant wind profile $\frac{du}{dz} = 0$	logarithmic wind profile $u_{x_\infty,y,z} = \frac{u_*}{\kappa} \ln\left(\frac{z}{z_0}\right)$ $u_* = 0.45 \text{ m s}^{-1}$ ; $z_0 = 0.1 \text{ m}$

**Table 3** Theoretically predicted velocities for different axial induction factors  $a$  at the rotor position  $u_{x_0,y,z}$  and in the wake  $u_{x_w,y,z}$  scaled with the upstream velocity  $u_{x_\infty,y,z}$  according to Eqs. 13 and 14 and the deviations obtained from the numerical simulations. The deviations are calculated as an average over the disc area.

$a$	$\frac{u_{x_0,y,z}}{u_{x_\infty,y,z}}$ expected	$\frac{u_{x_0,y,z}}{u_{x_\infty,y,z}}$ deviation	$\frac{u_{x_w,y,z}}{u_{x_\infty,y,z}}$ expected	$\frac{u_{x_w,y,z}}{u_{x_\infty,y,z}}$ deviation
1/3	0.67	2 %	0.33	5 %
1/4	0.75	0 %	0.50	0 %
1/5	0.80	2 %	0.60	4 %

416 Equation 13 follows directly from Eq. 15, and Eq. 14 can be derived from  
 417 the Bernoulli equation and Newton's second law of motion (Hansen, 2008).  
 418 This comparison is strictly applicable only for laminar and uniform inflow  
 419 conditions  $u_{x_\infty,y,z}$ .

420 Numerical simulations with the set-up as listed in Table 2 for wind turbine  
 421 1 are performed with different axial induction factors  $a=1/3, 1/4, 1/5$  for  
 422 all parametrizations. Exemplary, the results for parametrization A, a non-  
 423 rotating disc and  $u_{x_\infty,y,z} = 0.08 \text{ m s}^{-1}$  are listed in Table 3. The results for  
 424 parametrizations B and C and for  $u_{x_\infty,y,z} = 0.10 \text{ m s}^{-1}$  are quantitatively  
 425 similar and therefore not shown here.

426 The simulated ratios of  $u_{x_0,y,z}/u_{x_\infty,y,z}$  and  $u_{x_w,y,z}/u_{x_\infty,y,z}$  for a realistic  
 427 value of the axial induction factor of 1/4 are in complete agreement with the  
 428 one-dimensional momentum theory. For larger ( $a=1/3$ ) and smaller ( $a=1/5$ )  
 429  $a$  values, the simulation results deviate by less than 5 % from the theoretical  
 430 predictions.

**Table 4** List of all performed simulations with information of the perturbation amplitude, the type of the wind-turbine parametrization, the tangential force and the SGS closure model used in the LES model.

simulation	perturbation amplitude $\alpha$	wind turbine parametrization	tangential force $F_\Theta$	SGS closure model
B_1	1	B	$\neq 0$	TKE
B_5	5	B	$\neq 0$	TKE
B_10	10	B	$\neq 0$	TKE
A_1	1	A	$\neq 0$	TKE
C_1	1	C	$\neq 0$	TKE
A_NR	1	A	$= 0$	TKE
B_S	1	B	$\neq 0$	Smagorinsky
B_I	1	B	$\neq 0$	no (implicit LES)

431 Summarizing, we successfully validated our LES model EULAG for the  
 432 non-rotating disc of parametrization A and realistic values of the axial induc-  
 433 tion factor against the one-dimensional momentum theory.

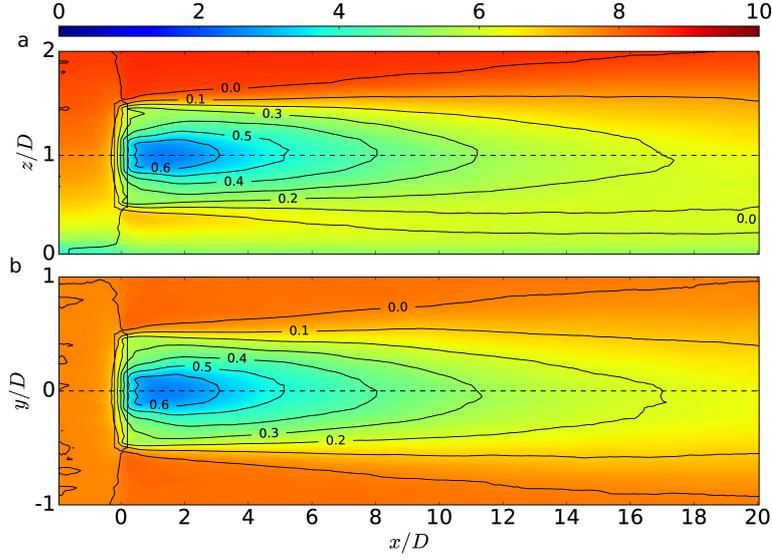
## 434 5 Numerical Experiments and Results

435 In this section, a detailed investigation of the reference simulation B\_1 (base  
 436 case) with  $\alpha = 1$  and wind turbine 2 (Table 2) is given to confirm the applica-  
 437 tion of the turbulence preserving model in a wind-turbine simulation. Details  
 438 of the simulation set-up are listed in Table 2. Further, the dependence of the  
 439 wake characteristics of the reference simulation B\_1 are investigated regarding  
 440 the impact of,

- 441  $a$ , the perturbation amplitude
- 442  $b$ , the wind-turbine parametrization
- 443  $c$ , the rotation of the disc
- 444  $d$ , the SGS closure model.

445 The corresponding parameters of B\_1 and of all other simulations are listed in  
 446 Table 4.

447 All simulations are performed for 60 min, a period long enough for the  
 448 wake to reach an equilibrium state with statistical convergence of the results.  
 449 All mean values are averaged over the last 50 min. The temporal average  
 450  $\langle \Psi_{x,y,z} \rangle_t$  of a quantity  $\Psi$  for a time period  $t$  is calculated online in the  
 451 numerical model and updated at every timestep according to the method of  
 452 Fröhlich (2006, Eq. 9.1). In the following numerical simulations, the rotor  
 453 covers 21 grid points. This leads to a high enough resolution according to  
 454 investigations of Ivanell et al. (2008), Wu and Porté-Agel (2012) or Gomes  
 455 et al. (2014) to avoid any dependence of the wake on the resolution. Generally,  
 456 the numerical simulation results are plotted in dimensionless coordinates as a  
 457 function of the rotor diameter  $D$ . The contour of the actuator in the cross-  
 458 sections represents the transition to a force of zero. Furthermore, only a sector  
 459 of the complete computational domain is shown in most of the following plots.



**Fig. 2** Streamwise wind field in a vertical  $x-z$  cross-section at  $y_0$  in (a) and in a horizontal  $x-y$  cross-section at  $z_h$  in (b). The contours represent the velocity deficit  $(u_{\infty,y_0,k} - u_{i,y_0,k})/u_{\infty,y_0,k}$  in (a) and  $(u_{\infty,j,z_h} - u_{i,j,z_h})/u_{\infty,j,z_h}$  in (b). Note, that in these cross-sections, the scale in the  $z$  or  $y$ -direction is exaggerated compared to the horizontal scale the in  $x$ -direction.

460 Now, we investigate the following characteristics of the wake of a wind  
461 turbine:

- 462 – The spatial distribution of the velocities  $u$ ,  $v$  and  $w$ .
- 463 – The streamwise velocity ratio

$$VR_{x,y,z} = \frac{\langle u_{x,y_0,z_h} \rangle_t}{\langle u_{x_\infty,y_0,z_h} \rangle_t}, \quad (16)$$

464 as it is related to the power loss of a wind turbine.

- 465 – The streamwise turbulent intensity

$$I_{x,y,z} = \frac{\sigma_{u_{x,y,z}}}{\langle u_{x,y,z_h} \rangle_t}, \quad (17)$$

466 with  $\sigma_{u_{x,y,z}} = \sqrt{\langle u_{x,y,z}'^2 \rangle_t}$  and  $u_{x,y,z}' = u_{x,y,z} - \langle u_{x,y,z} \rangle_t$ , as it affects  
467 the flow-induced dynamic loads on downwind turbines.

## 468 5.1 Reference simulation B.1

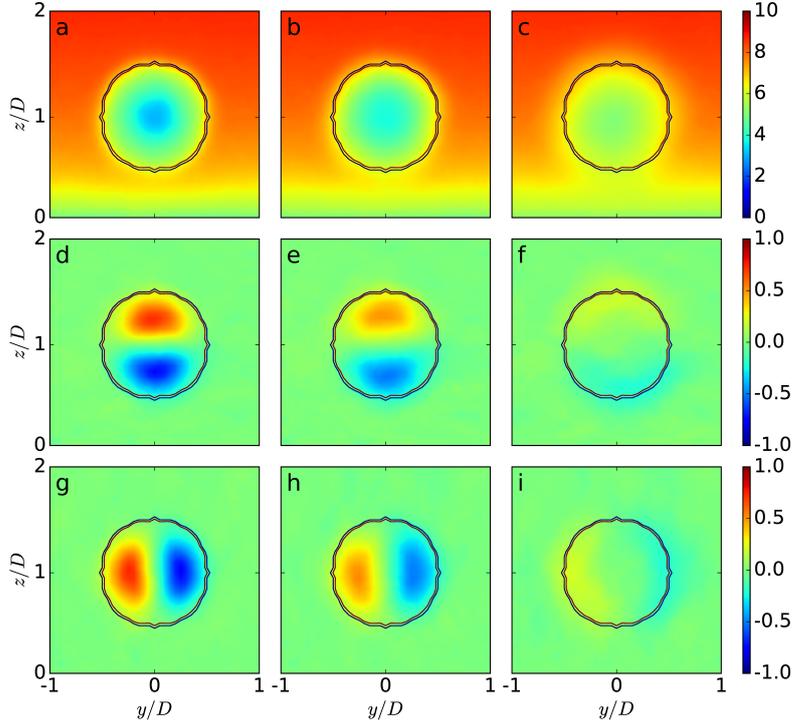
469 Figure 2 shows the vertical (Fig. 2a) and horizontal (Fig. 2b) cross-sections of  
470 the streamwise wind field of simulation B.1. The general wake structure reveals

471 a minimum of the velocity right behind the rotor with a velocity increase in the  
 472 radial and streamwise directions. This pattern results from the entrainment of  
 473 surrounding air with higher velocity values, it is observed prevalently in field  
 474 experiments in the atmosphere (Heimann et al., 2011, Fig. 3) or in wind-tunnel  
 475 measurements (Zhang et al., 2012, Fig. 4) as well as simulated numerically  
 476 (Porté-Agel et al., 2010, Fig. 5; Wu and Porté-Agel, 2012, Fig. 3; Aitken et al.,  
 477 2014, Fig. 5; Mirocha et al., 2014, Fig. 5).

478 The  $x$ - $y$  cross-section of  $u$  shows a nearly axisymmetric distribution (Fig. 2b),  
 479 whereas the  $x$ - $z$  cross-section of  $u$  displays a non-axisymmetric mean velocity  
 480 profile (Fig. 2a) as a consequence of the vertically sheared upstream wind  
 481 profile and the effect of the surface. Another feature in the  $x$ - $z$  cross-section  
 482 (Fig. 2a) represents the region of higher velocity air at the lowest part of  
 483 the rotor in comparison to the surroundings. The velocity deficit plotted as  
 484 contour lines in Fig. 2 enables a comparison with lidar measurements (Iungo  
 485 et al., 2013; Käsler et al., 2010) or with remotely piloted aircraft measurements  
 486 (Wildmann et al., 2014). These measurements for similar sized turbines and  
 487 wind speeds result in a wind speed deficit of about 50 to 60 % at  $x = 4D$ , which  
 488 is in line with the contours of the reference simulation in Fig. 2.

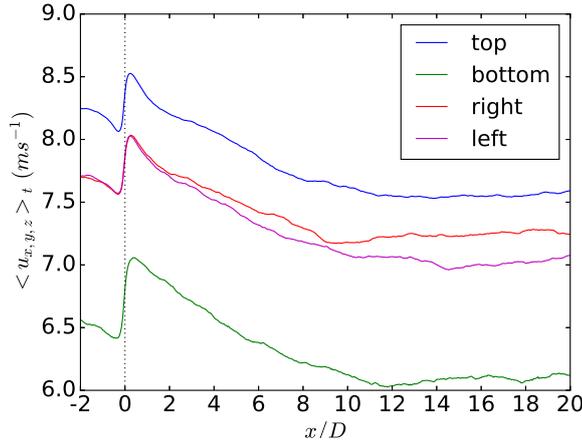
489 In Fig. 3, the mean values of  $u$ ,  $v$  and  $w$  are plotted in a  $y$ - $z$  cross-sections  
 490 for selected downstream positions at  $x = 3D$ ,  $x = 5D$  and  $x = 10D$ . With in-  
 491 creasing streamwise distance from the rotor, the flow field  $u$  recovers and starts  
 492 to converge towards the upstream wind profile. The general structure of the  
 493 position of the velocity minimum as well as the recovery of the wind field  
 494 is comparable to published results (e.g., Wu and Porté-Agel, 2012, Fig. 4;  
 495 Mirocha et al., 2014, Fig. 4). Depending on the implementation of a nacelle,  
 496 the flow field directly behind the centre of the wind turbine changes. Among  
 497 others, Wu and Porté-Agel (2011) and Meyers and Meneveau (2013) include  
 498 the nacelle, whereas it is neglected in Aitken et al. (2014) and Mirocha et al.  
 499 (2014). The slices of the lateral wind component  $v$  reveal a maximum at the  
 500 upper rotor part and a minimum at the lower part, which corresponds to the  
 501 vertical velocity field  $w$  with a maximum for  $y/D \in [-1, 0]$  and a minimum for  
 502  $y/D \in [0, 1]$ . The intensity of this rotational effect decreases with increasing  
 503 streamwise distance from the rotor. The regions with the maximum swirl of  
 504 the flow are veering away from the rotor centre for an increasing downstream  
 505 distance. The pattern in  $v$  and  $w$  is comparable to Mirocha et al. (2014, Fig. 4).  
 506 In contrast to our results, the  $y$ - $z$  cross-sections in Mirocha et al. (2014) are  
 507 asymmetric, which is most likely induced by the weakly convective ABL in  
 508 their simulations.

509 In Fig. 4, the temporally averaged velocity component in streamwise di-  
 510 rection  $\langle u_{x,y,z} \rangle_t$  is plotted as a function of streamwise distance for different  
 511 positions (top, bottom, right ( $y/D \in [0, 1]$ ), left ( $y/D \in [-1, 0]$ )) 60 m away  
 512 from the centre of the rotor. These positions, although located outside of the  
 513 actuator ( $R = 50$  m), are still close enough to represent the effect of the forces  
 514 resulting from Eq. 8 on the flow field. In the upstream region, the velocities  
 515 at the top and the bottom locations differ due to the incoming logarithmic  
 516 wind profile whereas the wind speeds right and left of the rotor are the same.



**Fig. 3** The averaged values of the base-case simulation (B.1) of  $\langle u_{x,y,z} \rangle_t$  in (a)-(c),  $\langle v_{x,y,z} \rangle_t$  in (d)-(f) and  $\langle w_{x,y,z} \rangle_t$  in (g)-(i) in  $y$ - $z$  cross-sections at downstream positions  $x = 3D$  ((a), (d), (g)),  $x = 5D$  ((b), (e), (h)) and  $x = 10D$  ((c), (f), (i)).

517 Approaching the rotor, the flow is decelerated in front of the wind turbine and  
 518 accelerated behind it. This behaviour is induced by the flow deceleration due  
 519 to the axial force  $F_x$ , which causes a pressure increase in front of the rotor  
 520 and a decrease behind (Bernoulli equation) (Hansen, 2008). The difference of  
 521 the flow in the spanwise direction for  $x/D > 2$  results from the rotation of the  
 522 actuator, leading to an accelerated (decelerated) flow on the right (left) due to  
 523 downward (upward) transport of air with higher (lower) momentum. The flow  
 524 recovers with increasing distance and the velocity values start to approach the  
 525 values of the incoming wind field for  $x \geq 10D$ . The effect of the wind turbine on  
 526 the wake is not negligible even at a streamwise distance of  $x = 20D$  in Fig. 2,  
 527 therefore we expect a full recovery in Fig. 4 at positions  $x > 20D$ .



**Fig. 4** The velocity component in streamwise direction  $\langle u_{x,y,z} \rangle_t$  averaged over the last  $t = 50$  min of the base-case simulation B\_1 at four positions, which are located 60 m away from the rotor centre ( $R = 50$  m), in both spanwise (left and right) and vertical (top and bottom) directions. The spanwise directions correspond to Figs. 2 and 3 with right  $\equiv y/D \in [0, 1]$  and left  $\equiv y/D \in [-1, 0]$ .

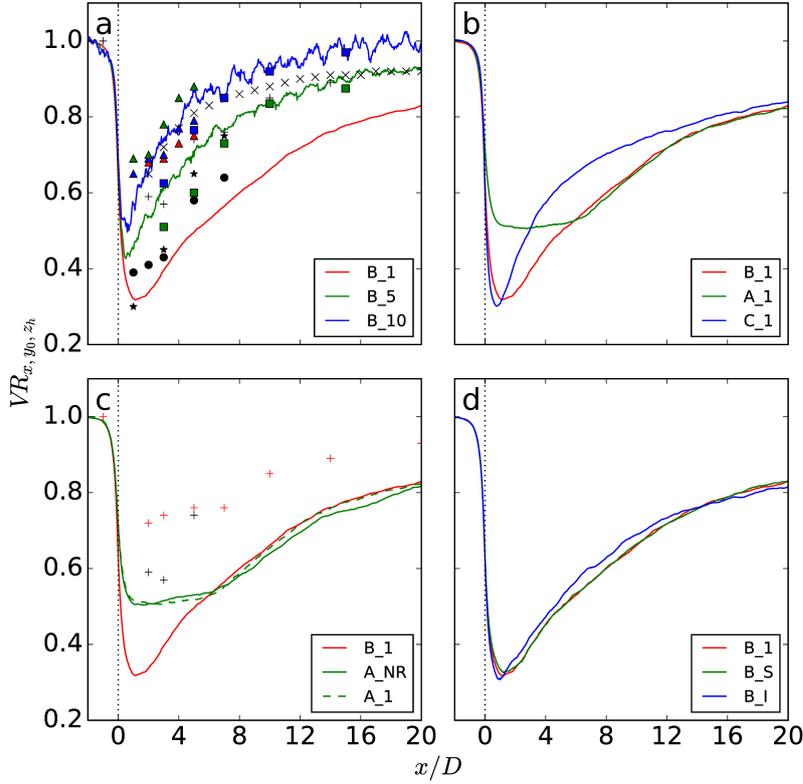
## 528 5.2 Impact of the perturbation amplitude

529 The method of preserving the background turbulence includes the factor  $\alpha$  in  
 530 Eq. 5, which was introduced as the amplitude of the perturbation. The impact  
 531 of  $\alpha$  is studied in simulations B\_5 ( $\alpha = 5$ ) and B\_10 ( $\alpha = 10$ ) and compared to  
 532 the reference simulation B\_1 ( $\alpha = 1$ ).

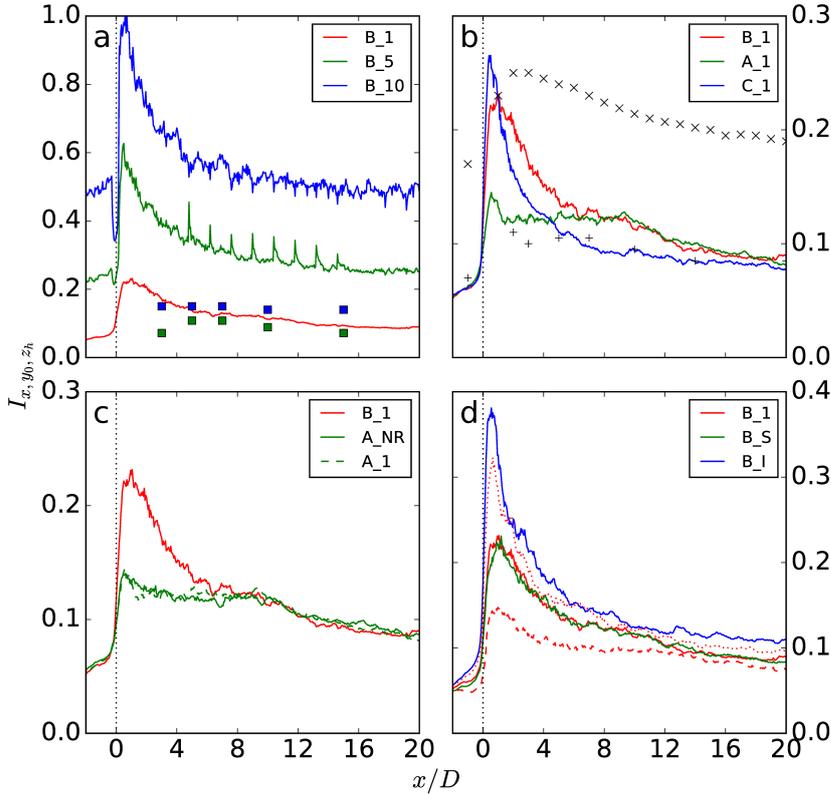
533 Figure 5a shows the streamwise profiles of the velocity ratio from Eq. 16 for  
 534 different values of the perturbation amplitude  $\alpha$ . A larger  $\alpha$  value leads to a  
 535 progressively shorter streamwise extension of the wake, induced by a stronger  
 536 entrainment of ambient air. Further, the minimum of the velocity ratio in the  
 537 near wake directly behind the nacelle increases.

538 The markers in Fig. 5a correspond to different wind-turbine studies, as  
 539 described in detail in the caption of Fig. 5. The simulation results of B\_1 are  
 540 comparable to lidar measurements and WRF-LES model results for a stable  
 541 ABL (Aitken et al., 2014). By increasing the value of  $\alpha$ , the velocity ratio ap-  
 542 proaches values found in observations and simulations of cases with enhanced  
 543 turbulence. The numerical results of simulation B\_5 correspond to a neutral  
 544 ABL (Wu and Porté-Agel, 2011; Gomes et al., 2014), whereas the results of  
 545 simulation B\_10 are almost comparable to measurements and WRF-LES model  
 546 results in a convective ABL (Mirocha et al., 2014). This comparison with other  
 547 studies leads to the hypothesis that the factor  $\alpha$  from Eq. 5 could be related  
 548 quantitatively to different levels of atmospheric turbulence.

549 We also tested various precursor simulations (convection or Coriolis force  
 550 as trigger to excite turbulence) resulting in different spectral energy densities.  
 551 The velocity ratio for a larger amount of the spectral energy density is in



**Fig. 5** The streamwise dependence of the velocity ratio from Eq. 16 at  $y_0$  and  $z_h$  for all simulations listed in Table 4, grouped together regarding the wake impact of the perturbation amplitude in (a), the wind-turbine parametrization in (b), the rotation of the disc in (c), and the SGS closure model in (d). The markers in (a) and (c) correspond to the results of the velocity ratio from the wake of a wind turbine in various studies: the values marked by a plus sign are extracted out of the LES from Wu and Porté-Agel (2011, Fig. 4) for a neutral ABL. The crosses correspond to the neutral ABL RANS simulation by Gomes et al. (2014, Fig. 1). The circles are extracted from lidar measurements in a stable ABL and the asterisks from the corresponding WRF-LES model simulation of a stable ABL, see Aitken et al. (2014, Fig. 6). The red triangles are extracted from convective ABL measurements, the blue triangles correspond to the WRF-LES model simulation of a convective ABL characterized by a heat flux of  $20 \text{ W m}^{-2}$ , and the green triangles correspond to the WRF-LES model simulation of a convective ABL characterized by a heat flux of  $100 \text{ W m}^{-2}$ , investigated in Mirocha et al. (2014, Fig. 8). The green squares correspond to a neutral ABL with a roughness length  $z_0 = 1 \times 10^{-5} \text{ m}$ , and the blue squares to a value of  $z_0 = 1 \times 10^{-1} \text{ m}$  (Wu and Porté-Agel, 2012, Fig. 5). The red plus signs in (c) correspond to the results of the non-rotating disc in Wu and Porté-Agel (2011, Fig. 4) opposed to their rotating results in black.



**Fig. 6** The streamwise dependence of the turbulence intensity from Eq. 17 at  $y_0$  and  $z_h$  for all simulations listed in Table 4, grouped together regarding the wake impact of the perturbation amplitude in (a), the wind-turbine parametrization in (b), the rotation of the disc in (c), and the SGS closure model in (d). The markers in (a) and (b) result from the streamwise turbulent intensity in the wake of a wind turbine in various studies: the green squares in (a) correspond to a neutral ABL with a roughness length  $z_0 = 1 \times 10^{-5}$  m, and the blue squares to a value of  $z_0 = 1 \times 10^{-1}$  m (Wu and Porté-Agel, 2012, Fig. 8). The values marked by a plus sign in (b) are extracted out of the LES from Wu and Porté-Agel (2011, Fig. 7) for a neutral ABL. The crosses correspond to the neutral ABL RANS simulation by Gomes et al. (2014, Fig. 1). The dotted line in plot (d) represents simulation B.1 with 1/2 times the length scale in the SGS closure model, whereas the dashed line represents simulation B.1 with twice the length scale in the SGS closure model.

552 agreement with a larger value of  $\alpha$  (not shown here). The parameter  $\alpha$  is  
 553 also comparable to the different roughness lengths used in Wu and Porté-Agel  
 554 (2012), with a larger roughness length corresponding to a higher perturbation  
 555 amplitude.

556 The streamwise profiles of the turbulent intensity in Eq. 17 are presented  
 557 in Fig. 6a for different  $\alpha$  values. The turbulent intensity  $I_{x, y_0, z_h}$  increases

558 with increasing  $\alpha$ . In the upstream as well as in the downstream region, the  
 559 streamwise distribution of  $I_{x,y_0,z_h}$  is proportional to  $\alpha$ . Wu and Porté-Agel  
 560 (2012) investigate an increase of  $I_{x,y_0,z_h}$  for increasing  $z_0$ . We also result in an  
 561 increase of  $I_{x,y_0,z_h}$  for increasing  $\alpha$ , reinforcing our assumption that larger  $\alpha$   
 562 values are comparable to a surface with an increased roughness length.

563 We conclude that the entrainment in the wake can be easily modified by  
 564 adjusting the value of  $\alpha$  in the numerical simulations. In this way, a realistic  
 565 level of atmospheric background turbulence intensity corresponding to various  
 566 atmospheric stratifications or different roughness lengths can be parametrized  
 567 by applying our turbulence preserving model.

### 568 5.3 Impact of the wind-turbine parametrization

569 The impact of the three wind-turbine parametrizations A, B, and C on the  
 570 wake is studied for  $\alpha = 1$  in simulations A\_1, B\_1 and C\_1. The different  
 571 parametrizations influence the velocity ratio in the wake as documented in  
 572 Fig. 5b.

573 A comparison between simulation A\_1 and simulation B\_1 focuses on the  
 574 difference between the MMT and the BEM method. Approaching a down-  
 575 stream distance of  $x = 5D$ , the difference in the wake structure becomes mar-  
 576 ginal. Therefore, we define a streamwise distance of  $x = 5D$  as the transition  
 577 between the near wake and the far wake. Further, the value of the minimum  
 578 of the velocity ratio in the near wake is larger for parametrization A in A\_1  
 579 due to no radial dependence of the thrust and power coefficients in Eqs. 6 and  
 580 7.

581 The difference between parametrizations B and C are the local blade char-  
 582 acteristics of the two airfoils. In parametrization C the velocity field in the  
 583 streamwise direction recovers more rapidly up to approximately  $x = 14D$  in  
 584 comparison to type B. This is caused by the sharper gradient in the axial  
 585 force at the edge of the nacelle between  $0.2 r/R$  and  $0.3 r/R$  in Fig. 1.

586 The different parametrizations also have an impact on the value of the  
 587 maximum of the turbulent intensity in Fig. 6b. The maximum is larger for  
 588 parametrization B in comparison to parametrization A. This is caused by the  
 589 radial gradient of the axial force in parametrization B, which contrasts a con-  
 590 stant force in parametrization A, as shown in Fig. 1. The streamwise turbulent  
 591 intensities of parametrizations A and B are very similar in the far wake. The  
 592 difference in the maximum between parametrizations B and C correlates with  
 593 the gradient of the axial force close to the nacelle in Fig. 1. A larger maximum  
 594 corresponds to a sharper gradient. A sharper gradient also results in a more  
 595 rapid decline in parametrization C in comparison to parametrization B up to  
 596 approximately  $x = 14D$ .

597 Comparing these results to other studies, the turbulent intensity values  
 598 of all three parametrizations are rather small in comparison to the RANS  
 599 simulation of Gomes et al. (2014) approaching  $x \geq 2D$ . A comparison with  
 600 the LES of Wu and Porté-Agel (2011) results in a rather good agreement in the

601 near wake for parametrization A and in the far wake for parametrization C.  
 602 The agreement of parametrization C is referable to a similar radial distribution  
 603 of the forces yielded from the same blade characteristics.

604 We conclude that the MMT is sufficient as simplification of the BEM  
 605 parametrization if only the far wake is of interest. In the near wake the ra-  
 606 dial dependence of the axial force becomes important. Further, the local blade  
 607 characteristics influence the wake up to a downstream distance of  $x = 14D$ .

608 In the scope of this work, we also implemented an advanced version of the  
 609 MMT. It considers the radial distribution of the forces in Eqs. 6 and 7, which is  
 610 adopted from the radial chord length dispersion in Micallef et al. (2013). The  
 611 forces in Eqs. 6 and 7 are modified similarly to the procedure in Gomes et al.  
 612 (2014). Numerical simulations using this approach led to a better agreement  
 613 of the near-wake structure with the BEM method in parametrization B in  
 614 comparison to the MMT approach (not shown here).

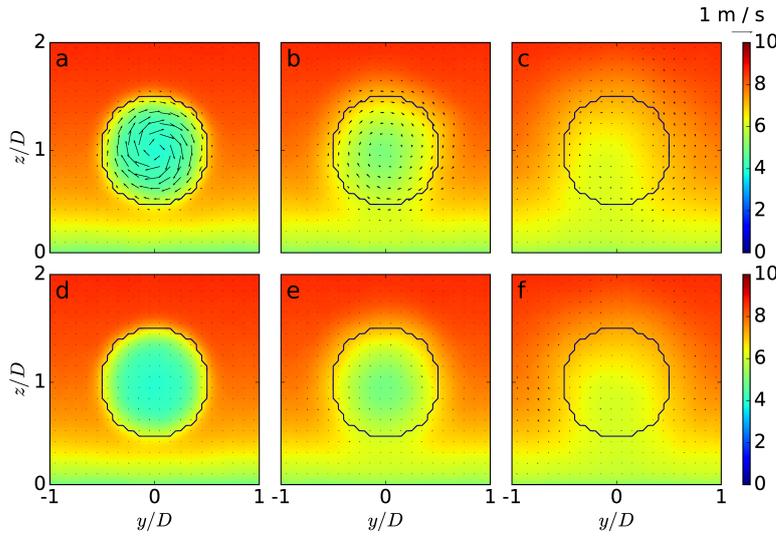
#### 615 5.4 Impact of the rotation of the disc

616 To investigate the impact of the rotation of the actuator on the wake structure,  
 617 simulation A\_NR with parametrization A, no rotation of the disc ( $F_\theta = 0$  in  
 618 Eq. 7) and  $\alpha = 1$  is performed and compared to simulations A\_1 and B\_1.

619 The minimum of the velocity ratio in simulation B\_1 is smaller in compar-  
 620 ison to simulation A\_NR. This finding is in agreement with the results of Wu  
 621 and Porté-Agel (2011) (markers in Fig. 5c). A comparison between simulation  
 622 A\_1 and simulation A\_NR results in a marginal impact of the tangential force  
 623 on the streamwise velocity ratio according to Fig. 5c. Therefore, the difference  
 624 between simulation B\_1 and simulation A\_NR is evoked by the uniform thrust  
 625 force distribution over the disc, which has a larger impact on the velocity ratio  
 626 than the marginal effect of rotation.

627 Wu and Porté-Agel (2011) show an increase of the turbulence intensity  
 628 applying the BEM method instead of the classical Rankine-Froude approach.  
 629 The streamwise turbulent intensity at the centre line in Fig. 6c is also larger for  
 630 the BEM parametrization in the near wake. The effect of rotation is marginal.  
 631 Consequently, not the swirl, but the non-uniform distribution of the axial force  
 632 in the BEM method (Fig. 1) is responsible for the near-wake difference in the  
 633 streamwise turbulent intensity in Fig. 6c.

634 The rotation of the disc in simulation A\_1 leads to a swirl in the wake as  
 635 shown in Figs. 7a-c. The rotational effect of the disc is evident at  $x = 3D$ . Ap-  
 636 proaching  $x = 10D$ , the swirl in the disc region decays while it is transported  
 637 outwards. Both effects originate from entrainment processes. At a downstream  
 638 position of  $x = 20D$ , the rotation in the disc region approaches zero, whereas  
 639 there is still some swirl in the air around the disc. In contrast to this rotational  
 640 behaviour, there is no swirl of the air downstream of the non-rotating disc of  
 641 simulation A\_NR in Figs. 7d-f. The pattern of the streamwise velocity  $u$  in  
 642 the rotor region as well as in the surroundings are comparable in both simu-  
 643 lations at  $x = 3D$  and  $10D$ . At  $x = 20D$ , the wake pattern in simulation A\_NR



**Fig. 7** The averaged value of  $\langle u_{x,y,z} \rangle_t$  in a  $y$ - $z$  cross-section at downstream positions  $x = 3D$  ((a), (d)),  $x = 10D$  ((b), (e)) and  $x = 20D$  ((c), (f)) for simulation A.1 ((a)-(c)) and simulation A\_NR ((d)-(f)). The arrows represent the wind vectors ( $\langle v_{x,y,z} \rangle_t$ ,  $\langle w_{x,y,z} \rangle_t$ ). The magnitude of  $1 \text{ m s}^{-1}$  is shown at the right edge of the plot.

644 is symmetric, whereas in simulation A.1 it is shifted towards  $y/D \in [-1, 0]$ .  
 645 This asymmetric streamwise velocity field results from the rotation of the disc  
 646 and is also prevalent in the study of Wu and Porté-Agel (2012, Fig. 4).

647 This investigation leads to the conclusion that the rotation has a minor  
 648 effect on the velocity ratio and on the streamwise turbulent intensity at the  
 649 centre line. However, the effect of the tangential force on the  $v$  and  $w$  wind  
 650 components is prevailing even in the far-wake region, with an influence on the  
 651 streamwise velocity field in the  $y$ - $z$  plane.

## 652 5.5 Impact of the SGS closure model

653 The impact of the SGS closure models is investigated by comparing the TKE  
 654 SGS closure model simulation B.1 with the Smagorinsky SGS closure model  
 655 simulation B.S. The geophysical flow solver EULAG provides a reliable numerical  
 656 testbed to study the SGS closure model sensitivities. Further, it depends  
 657 on the NFT integrations of Eqs. 1 to 3 and therefore offers the possibility to  
 658 integrate these equations without an explicit SGS closure model by setting  
 659  $\mathcal{V} = 0$  and  $\mathcal{H} = 0$  in Eqs. 1 and 2 in the implicit LES B.I.

660 The streamwise dependence of the velocity ratios in Fig. 5d agrees quan-  
 661 titatively very well for simulation B.1 and simulation B.S. The contrast to  
 662 simulation B.I is insignificant.

663 The turbulent intensities in Fig. 6d are also rather similar for the TKE and  
 664 the Smagorinsky SGS closure model. For the implicit LES, the maximum of  
 665  $I_{x,y_0,z_h}$  is roughly 1.7 times larger than in the simulations with the SGS closure  
 666 model. In the far wake the difference becomes rather small. The dependency of  
 667 the difference in the turbulent intensity in the near wake between an implicit  
 668 LES and a simulation using an explicit SGS closure model is verified with two  
 669 further simulations, modifying the SGS closure model of simulation B\_1. In the  
 670 first simulation, the length scale of the TKE SGS closure model is multiplied  
 671 by a factor of 1/2, resulting in the dotted red line in Fig. 6d, whereas in the  
 672 second simulation, the length scale is multiplied by a factor of 2, resulting  
 673 in the dashed red line. Decreasing (increasing) the length scale of the closure  
 674 model results in a weaker (stronger) damping. A weaker damping induces  
 675 larger turbulence, approaching the turbulent intensity behaviour of the implicit  
 676 LES, whereas a stronger damping results in a weaker turbulent behaviour.  
 677 The streamwise velocity ratios are nearly unaffected by the length scale of the  
 678 closure model (not shown here).

679 The agreement between the established SGS schemes (TKE and Smagorin-  
 680 sky) is a remarkable result and confirms earlier findings by Smolarkiewicz et al.  
 681 (2007). The possibility of an implicit LES of wind-turbine flows enables nu-  
 682 merical simulations with stretched or adaptive meshes, where an explicit SGS  
 683 parametrization might be difficult and troublesome.

684 The length scale of the closure model offers another tuning parameter in  
 685 addition to  $\alpha$ , which can explain the difference in the streamwise turbulent in-  
 686 tensity in comparison to other simulation results of Wu and Porté-Agel (2011),  
 687 Wu and Porté-Agel (2012) and Gomes et al. (2014).

## 688 6 Conclusion

689 The wake characteristics of a wind turbine in a turbulent and neutral ABL flow  
 690 were investigated by means of LES. Besides reliable wind-turbine parametriza-  
 691 tions, an effective method to preserve the atmospheric background turbulence  
 692 was applied successfully in the numerical solver. The numerical simulations  
 693 using these two ingredients result in realistic wake structures, which are quan-  
 694 titatively comparable with previous observations and numerical simulation  
 695 results.

696 The atmospheric background turbulence field was simulated by a precu-  
 697 sor simulation of the neutral ABL using cyclic boundary conditions. Velocity  
 698 perturbations were extracted once from the equilibrium state of the precursor  
 699 simulation. These perturbation velocities were superimposed on the flow field  
 700 of the wind-turbine simulations by a new method suitable for open horizontal  
 701 boundaries. This method preserves the atmospheric background turbulence by  
 702 applying the spectral energy distribution at every timestep taken from three  
 703 3D fields ( $u$ ,  $v$ ,  $w$ ) of the precursor simulation. The newly developed turbu-  
 704 lence preserving method uses an empirical factor  $\alpha$ , which controls the energy  
 705 content of the background turbulence. Larger  $\alpha$  values refer to more turbulent

706 flow regimes, e.g. under convective conditions or for flows over a surface with  
707 an increased roughness length. An increase of the atmospheric background  
708 turbulence, i.e. larger  $\alpha$  values, enhance the entrainment of air into the wake,  
709 resulting in a shorter streamwise wake extension and an increase of the stream-  
710 wise turbulent intensity. The turbulence preserving method as presented here  
711 provides a simple and numerically very effective tool for studying the inter-  
712 action of ABL flow of different thermal stratifications with a wind turbine  
713 by applying the same spectral energy distribution and varying the parameter  
714  $\alpha$ . Considering different stratifications of the atmosphere is important, as a  
715 near-neutral stratification occurs only with a frequency of roughly 10 % ac-  
716 cording to data from a field experiment (SWiFT Facility Representation and  
717 Preparedness; 730 days of measurement in the period from 2012 to 2014 (Sue  
718 Ellen Haupt (NCAR), personal communication, 2015)).

719 Furthermore, the wake structure was investigated for different wind-turbine  
720 parametrizations. We considered the MMT and the BEM method as wind-  
721 turbine parametrizations, varied the local blade characteristics in the BEM  
722 method and studied the effect of rotation of the actuator. The BEM method  
723 yields a more accurate prediction of the near-wake characteristics if the air-  
724 foil data of the wind turbine are known. Considering how sparse information  
725 on detailed blade geometries is available, the MMT offers an alternative. It  
726 was found that the MMT is a reasonable simplification of the BEM model  
727 for studies of the far wake, when near-wake characteristics are of secondary  
728 importance. The wake structure for the two considered airfoils in the BEM  
729 model differs up to a streamwise distance of  $14D$ . The very far wake is not  
730 affected by the blade characteristics. The rotation of the wind turbine leads  
731 to a swirl in the wake and impacts on the streamwise velocity field in the  $y$ - $z$   
732 plane even in the far wake.

733 The sensitivity of the wake to two SGS closure models (TKE and Smago-  
734 rinsky-type models) and numerical simulations without an explicit SGS closure  
735 model (implicit LES) was studied. The choice of the SGS closure models has  
736 a rather small impact on the wake characteristics. Even the implicit LES re-  
737 sults of the streamwise velocity ratio agree surprisingly well with the former  
738 simulations reinforcing the suitability of this approach to study a wide class  
739 of ABL flows. However, there is a remarkable impact on the streamwise tur-  
740 bulent intensity in the near wake, which is strongly affected by the amount of  
741 damping in the SGS closure model.

742 In this study, we presented a simple and numerically effective method to  
743 perform LES of wind turbines with a realistic background turbulence field.  
744 Our turbulence preserving model as well as the wind-turbine models, both  
745 implemented in the numerical model EULAG, allow for subsequent future  
746 applications for a wide range of scales, for different thermal stratifications, as  
747 well as for flows over heterogeneous and hilly terrains.

748 **Acknowledgements** The authors thank Mark Zagar for providing the airfoil data of the  
749 10 MW reference wind turbine from DTU and Fernando Porté-Agel for the constructive  
750 discussion on our work in a previous state. This research was performed as part of the LIPS

751 project, funded by the Federal Ministry for the Environment, Nature Conservation, Building  
752 and Nuclear Safety by a resolution of the German Federal Parliament. The authors gratefully  
753 acknowledge the Gauss Centre for Supercomputing e.V. ([www.gauss-centre.eu](http://www.gauss-centre.eu)) for funding  
754 this project by providing computing time on the GCS Supercomputer SuperMUC at Leibniz  
755 Supercomputing Centre (LRZ, [www.lrz.de](http://www.lrz.de)).

756 **Appendix: BEM parameters**

**Table 5** List of the BEM method parameters used in parametrization of type B (10 MW reference wind turbine from DTU) (Mark Zagar (Vestas), personal communication, 2015) and type C (three-blade GWS/EP-6030x3 rotor) (Wu and Porté-Agel, 2011). The radius  $r$  and the chord length  $c$  of the two rotors are scaled to a rotor diameter of 100 m.

parametrization B			parametrization C		
$r / \text{m}$	$c(r) / \text{m}$	$\Theta(r) / ^\circ$	$r / \text{m}$	$c(r) / \text{m}$	$\Theta(r) / ^\circ$
5.0	5.3	13.3	6.7	9.3	20.5
10.0	6.0	13.2	13.3	9.8	20.9
15.0	6.2	10.5	20.0	9.8	19.8
20.0	5.8	9.0	26.6	9.4	16.9
25.0	5.0	7.3	33.3	8.7	13.2
30.0	4.5	5.5	40.0	7.9	10.7
35.0	3.5	3.8	46.7	6.8	9.1
40.0	3.0	2.5	50.0	4.0	6.7
45.0	2.3	1.3			
50.0	1.0	0.2			

**References**

- 757 **References**
- 758 Aitken ML, Kosović B, Mirocha JD, Lundquist JK (2014) Large eddy simula-  
759 tion of wind turbine wake dynamics in the stable boundary layer using the  
760 Weather Research and Forecasting Model. *J Renew Sust Energy* 6:1529–  
761 1539
- 762 Bellon G, Stevens B (2012) Using the sensitivity of large-eddy simulations to  
763 evaluate atmospheric boundary layer models. *J Atmos Sci* 69:1582–1601
- 764 Betz A (1926) Windenergie und ihre Ausnutzung in Windmühlen
- 765 Calaf M, Meneveau C, Meyers J (2010) Large eddy simulation study of fully  
766 developed wind-turbine array boundary layers. *Phys Fluids* 22:015110
- 767 Chamorro LP, Porté-Agel F (2009) A wind-tunnel investigation of wind-  
768 turbine wakes: Boundary-layer turbulence effects. *Boundary-Layer Meteorol*  
769 132:129–149
- 770 Doyle JD, Gaberšek S, Jiang Q, Bernardet L, Brown JM, Dörnbrack A, Filaus  
771 E, Grubišić V, Kirshbaum DJ, Knoth O, et al. (2011) An intercomparison of  
772 t-rex mountain-wave simulations and implications for mesoscale predictabil-  
773 ity. *Mon Weather Rev* 139:2811–2831
- 774 El Kasmi A, Masson C (2008) An extended model for turbulent flow through  
775 horizontal-axis wind turbines. *J Wind Eng Ind Aerodyn* 96:103–122
- 776 Emeis S (2013) *Wind energy meteorology: Atmospheric physics for wind power*  
777 *generation*. Springer Science & Business Media, 196 pp
- 778 Emeis S (2014) Current issues in wind energy meteorology. *Meteorol Appl*  
779 21:803–819
- 780 Fröhlich J (2006) *Large Eddy Simulation turbulenter Strömungen*. Teubner  
781 Verlag / GWV Fachverlage GmbH, Wiesbaden, 414 pp
- 782 Froude RE (1889) On the part played in propulsion by difference of fluid  
783 pressure. *Trans RINA* 30:390
- 784 Glauert H (1963) Airplane propellers. In: *Aerodynamic theory*, W. F. Durand,  
785 Dover, New York, pp 169–360.
- 786 Gomes VMMGC, Palma JMLM, Lopes AS (2014) Improving actuator disk  
787 wake model. In: *The science of making torque from wind*. Conference series,  
788 vol 524, p 012170.
- 789 Grinstein FF, Margolin LG, Rider WJ (2007) *Implicit Large Eddy Simulation*.  
790 Cambridge university press, 546 pp
- 791 Hansen MO (2008) *Aerodynamics of wind turbines*, vol 2. Earthscan, London  
792 and Sterling, UK and USA, 181 pp
- 793 Heimann D, Käsler Y, Gross G (2011) The wake of a wind turbine and its  
794 influence on sound propagation. *Meteorol Z* 20:449–460
- 795 Iungo GV, Wu YT, Porté-Agel F (2013) Field measurements of wind turbine  
796 wakes with lidars. *J Atmos Ocean Technol* 30:274–287
- 797 Ivanell BS, Mikkelsen R, Henningson D (2008) Validation of methods using  
798 EllipSys3D. Technical report, KTH, TRITA-MEK 2008:12, pp 183–221.
- 799 Käsler Y, Rahm S, Simmet R, Kühn M (2010) Wake measurements of a multi-  
800 MW wind turbine with coherent long-range pulsed doppler wind lidar. *J*  
801 *Atmos Ocean Technol* 27:1529–1532

- 802 Kataoka H, Mizuno M (2002) Numerical flow computation around aeroelastic  
803 3D square cylinder using inflow turbulence. *Wind and Structures* 5:379–392
- 804 Kühnlein C, Smolarkiewicz PK, Dörnbrack A (2012) Modelling atmospheric  
805 flows with adaptive moving meshes. *J Comput Phys* 231:2741–2763
- 806 Mann J (1994) The spatial structure of neutral atmospheric surface-layer tur-  
807 bulence. *J Fluid Mech* 273:141–168
- 808 Manwell J, McGowan J, Roger A (2002) *Wind Energy Explained: Theory,*  
809 *Design and Application*. Wiley: New York, NY, USA, 134 pp
- 810 Margolin L, Rider W (2002) A rationale for implicit turbulence modelling.  
811 *International Journal for Numerical Methods in Fluids* 39:821–841
- 812 Margolin L, Smolarkiewicz PK, Wyszogrodzki A (2002) Implicit turbulence  
813 modeling for high reynolds number flows. *Journal of Fluids Engineering*  
814 124:862–867
- 815 Margolin L, Rider W, Grinstein F (2006) Modeling turbulent flow with implicit  
816 les. *Journal of Turbulence* 7:1–27
- 817 Medici D, Alfredsson PH (2006) Measurements on a wind turbine wake: 3D  
818 effects and bluff body vortex shedding. *Wind Energy* 9:219–236
- 819 Meyers J, Meneveau C (2013) Flow visualization using momentum and energy  
820 transport tubes and applications to turbulent flow in wind farms. *J Fluid*  
821 *Mech* 715:335–358
- 822 Micallef D, Bussel GV, Sant T (2013) An investigation of radial velocities  
823 for a horizontal axis wind turbine in axial and yawed flows. *Wind Energy*  
824 16:529–544
- 825 Mikkelsen R (2003) Actuator disc methods applied to wind turbines. PhD  
826 thesis, Technical University of Denmark
- 827 Mirocha J, Kirkil G, Bou-Zeid E, Chow FK, Kosović B (2013) Transition and  
828 equilibration of neutral atmospheric boundary layer flow in one-way nested  
829 large-eddy simulations using the weather research and forecasting model.  
830 *Mon Weather Rev* 141:918–940
- 831 Mirocha J, Kosovic B, Aitken M, Lundquist J (2014) Implementation of a  
832 generalized actuator disk wind turbine model into the weather research and  
833 forecasting model for large-eddy simulation applications. *J Renew Sust En-*  
834 *ergy* 6:013104
- 835 Muñoz-Esparza D, Kosović B, Mirocha J, van Beeck J (2014) Bridging the  
836 transition from mesoscale to microscale turbulence in numerical weather  
837 prediction models. *Boundary-Layer Meteorol* 153:409–440
- 838 Naughton JW, Heinz S, Balas M, Kelly R, Gopalan H, Lindberg W, Gundling  
839 C, Rai R, Sitaraman J, Singh M (2011) Turbulence and the isolated wind  
840 turbine. In: 6th AIAA Theoretical Fluid Mechanics Conference, Honolulu,  
841 Hawaii, pp 1–19.
- 842 Porté-Agel F, Lu H, Wu YT (2010) A large-eddy simulation framework for  
843 wind energy applications. In: *The Fifth International Symposium on Com-*  
844 *putational Wind Engineering*, vol 23-27 May 2010, Chapel Hill, North Car-  
845 *olina, USA*
- 846 Prusa JM, Smolarkiewicz PK, Wyszogrodzki AA (2008) EULAG, a computa-  
847 tional model for multiscale flows. *Computers & Fluids* 37:1193–1207

- 848 Rankine WJM (1865) On the mechanical principles of the action of propellers.  
849 Trans RINA 6:13
- 850 Schetz JA, Fuhs AE (1996) Handbook of fluid dynamics and fluid machinery.  
851 Wiley, New York, 2776 pp
- 852 Smolarkiewicz PK, Charbonneau P (2013) EULAG, a computational model  
853 for multiscale flows: An MHD extension. J Comput Phys 236:608–623
- 854 Smolarkiewicz PK, Dörnbrack A (2008) Conservative integrals of adiabatic  
855 durran’s equations. Int J Numer Methods Fluids 56:1513–1519
- 856 Smolarkiewicz PK, Margolin LG (1993) On forward-in-time differencing for  
857 fluids: extension to a curvilinear framework. Mon Weather Rev 121:1847–  
858 1859
- 859 Smolarkiewicz PK, Margolin LG (1998) MPDATA: A Finite-Difference Solver  
860 for Geophysical Flows. J Comput Phys 140:459–480
- 861 Smolarkiewicz PK, Prusa JM (2002) Forward-in-time differencing for flu-  
862 ids: simulation of geophysical turbulence. In: Turbulent Flow Computation,  
863 Kluwer Academic Publishers, Boston, USA, pp 279-312
- 864 Smolarkiewicz PK, Prusa JM (2005) Towards mesh adaptivity for geophysical  
865 turbulence: continuous mapping approach. Int J Numer Methods Fluids  
866 47:789–801
- 867 Smolarkiewicz PK, Pudykiewicz JA (1992) A class of semi-Lagrangian approx-  
868 imations for fluids. J Atmos Sci 49:2082–2096
- 869 Smolarkiewicz PK, Winter CL (2010) Pores resolving simulation of darcy flows.  
870 J Comput Phys 229:3121–3133
- 871 Smolarkiewicz PK, Sharman R, Weil J, Perry SG, Heist D, Bowker G (2007)  
872 Building resolving large-eddy simulations and comparison with wind tunnel  
873 experiments. J Comput Phys 227:633–653
- 874 Tossas LAM, Leonardi S (2013) Wind turbine modeling for computational  
875 fluid dynamics: December 2010-December 2012. NREL Technical Monitor:  
876 Pat Moriarty, pp 1–48.
- 877 Troldborg N, Sørensen JN, Mikkelsen R (2007) Actuator line simulation of  
878 wake of wind turbine operating in turbulent inflow. In: The science of making  
879 torque from wind. Conference series, vol 75, p 012063.
- 880 Wedi NP, Smolarkiewicz PK (2004) Extending gal-chen and somerville terrain-  
881 following coordinate transformation on time-dependent curvilinear bound-  
882 aries. J Comput Phys 193:1–20
- 883 Wedi NP, Smolarkiewicz PK (2006) Direct numerical simulation of the plumb-  
884 McEwan laboratory analog of the QBO. J Atmos Sci 63:3226–3252
- 885 Wildmann N, Hofsäß M, Weimer F, Joos A, Bange J (2014) MASC—a small re-  
886 motely piloted aircraft (RPA) for wind energy research. Advances in Science  
887 and Research 11:55–61
- 888 Witha B, Steinfeld G, Heinemann D (2014) High-resolution offshore wake sim-  
889 ulations with the LES model PALM. In: Wind energy - Impact of turbulence,  
890 Spring 2012, Oldenburg, Germany, pp 175–181.
- 891 Wu YT, Porté-Agel F (2011) Large-Eddy Simulation of Wind-Turbine Wakes:  
892 Evaluation of Turbine Parametrisations. Boundary-Layer Meteorol 138:345–  
893 366

- 
- 894 Wu YT, Porté-Agel F (2012) Atmospheric turbulence effects on wind-turbine  
895 wakes: An LES study. *Energies* 5:5340–5362
- 896 Zhang W, Markfort CD, Porté-Agel F (2012) Near-wake flow structure down-  
897 wind of a wind turbine in a turbulent boundary layer. *Exp Fluids* 52:1219–  
898 1235