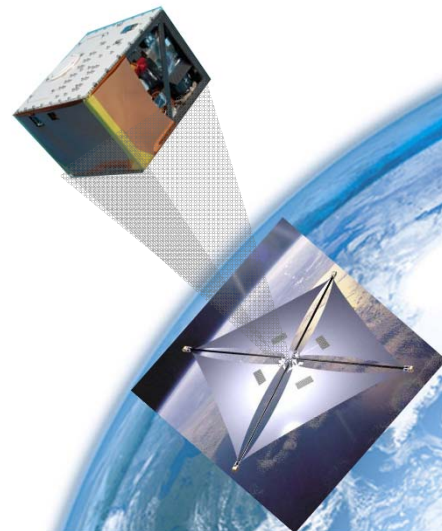


IAA-PDC-15-04-17

From Sail to Soil – Getting Sailcraft Out of the Harbour on a Visit to One of Earth's Nearest Neighbours

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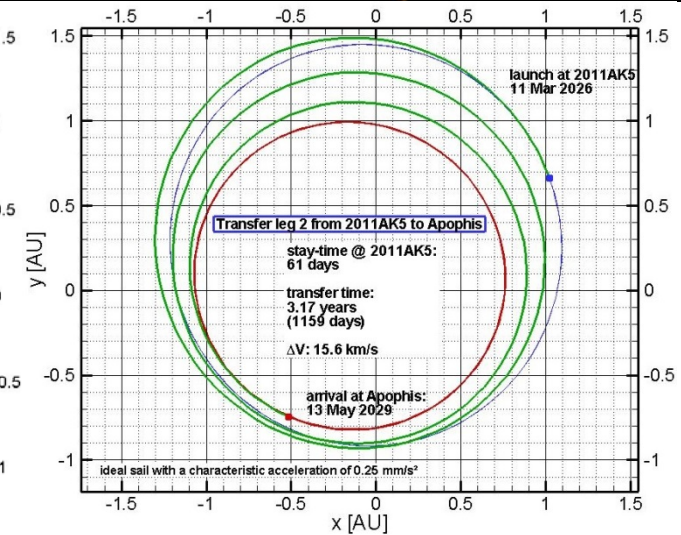
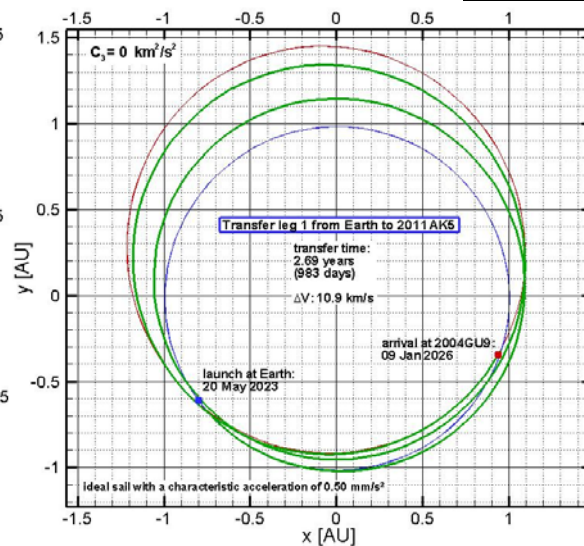
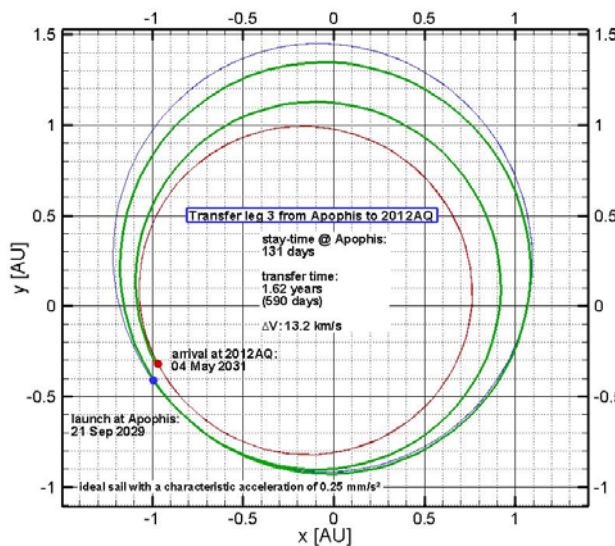
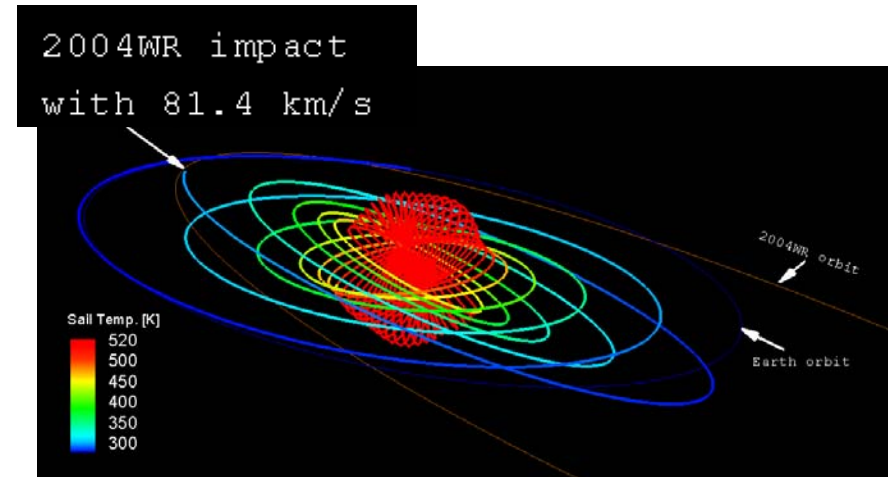
Knowledge for Tomorrow

Why solar sailing?

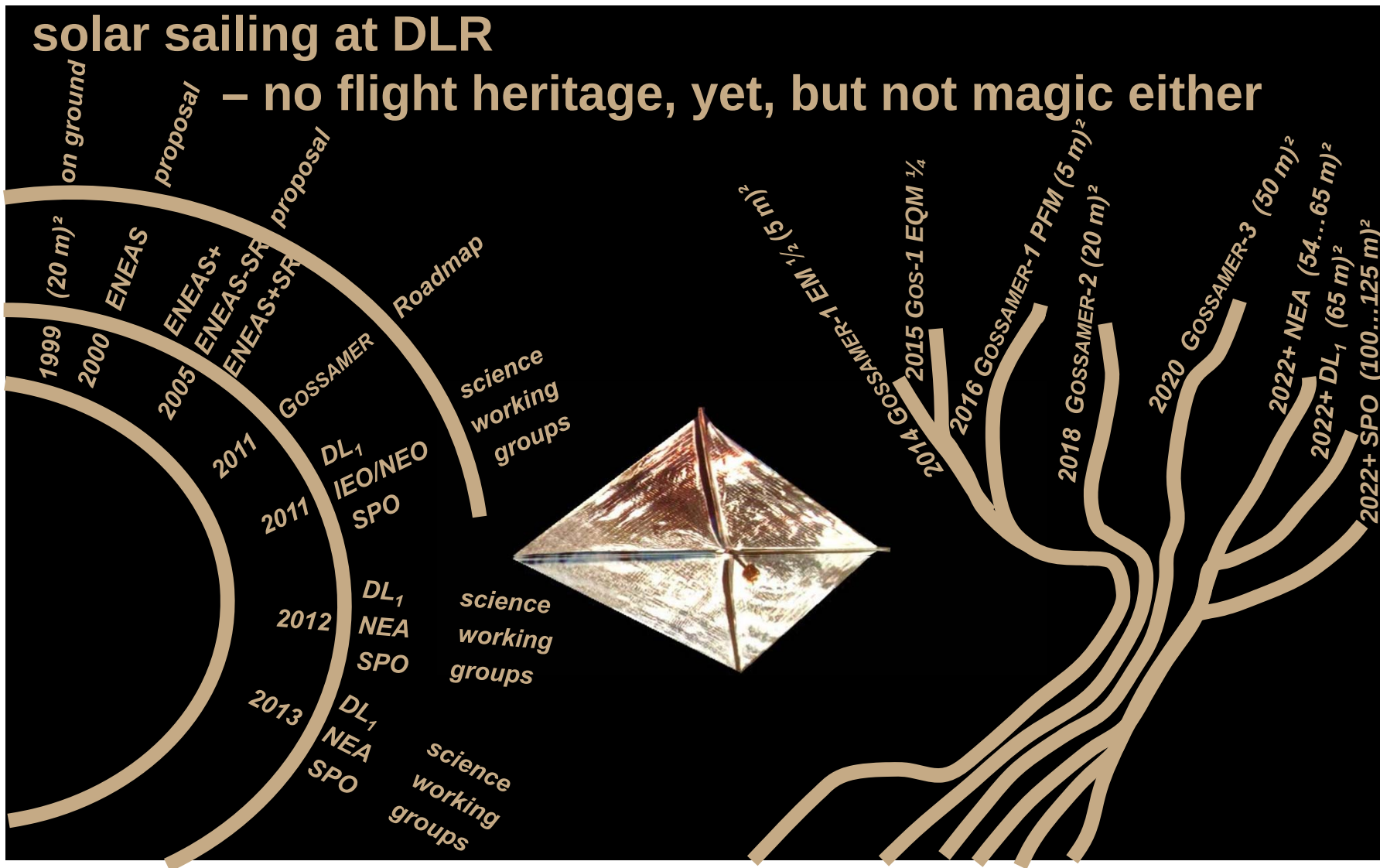
– well, you might as well ask...

**Why a 300 kg piggy-back launchable
>75 km/s Kinetic Impactor?**

**Why rendezvous
3 NEAs in <10 years?
(sample-return option included)**



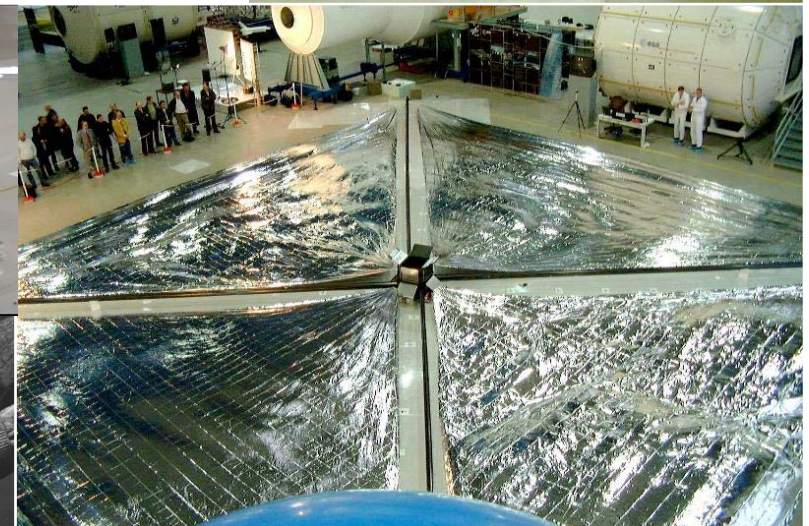
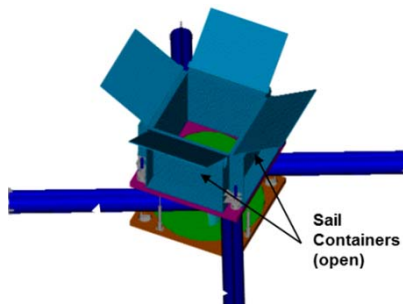




The December 7th, 1999 DLR/ESA Solar Sail Ground Demonstration

(20 m)² push-out boom, hoisted sail

- you need space – the European Astronaut Center hall next to the ISS model
- Deployment Module: 24 kg
- CFRP booms (4 x 14m, 101 g/m): 6 kg
- Sails (20 m)², 4-12 μ m foil: 5 kg
- Dimensions: 60cm x 60 cm x 65cm

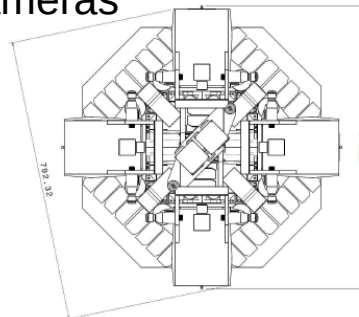
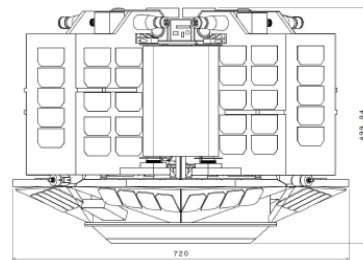


The GOSSAMER Roadmap

step 1 – deployment

GOSSAMER-1 – in-orbit deployment demonstrator

- (5 m)² sail area, all deployment-related mechanisms
- 1-boom, 2-quadrant EM in operation * →
- 1-boom EQM under construction now for extensive qualification testing to be finished by end of 2015
- proven MASCOT-style concurrent AIV approach **
- PFM detailed design progressed beyond PDR
- free-flyer independent spacecraft (really 5-in-1)
- “piggy-back” launch to LEO, <50 kg total ↓
- extensive instrumentation: 6 hi-res video cameras



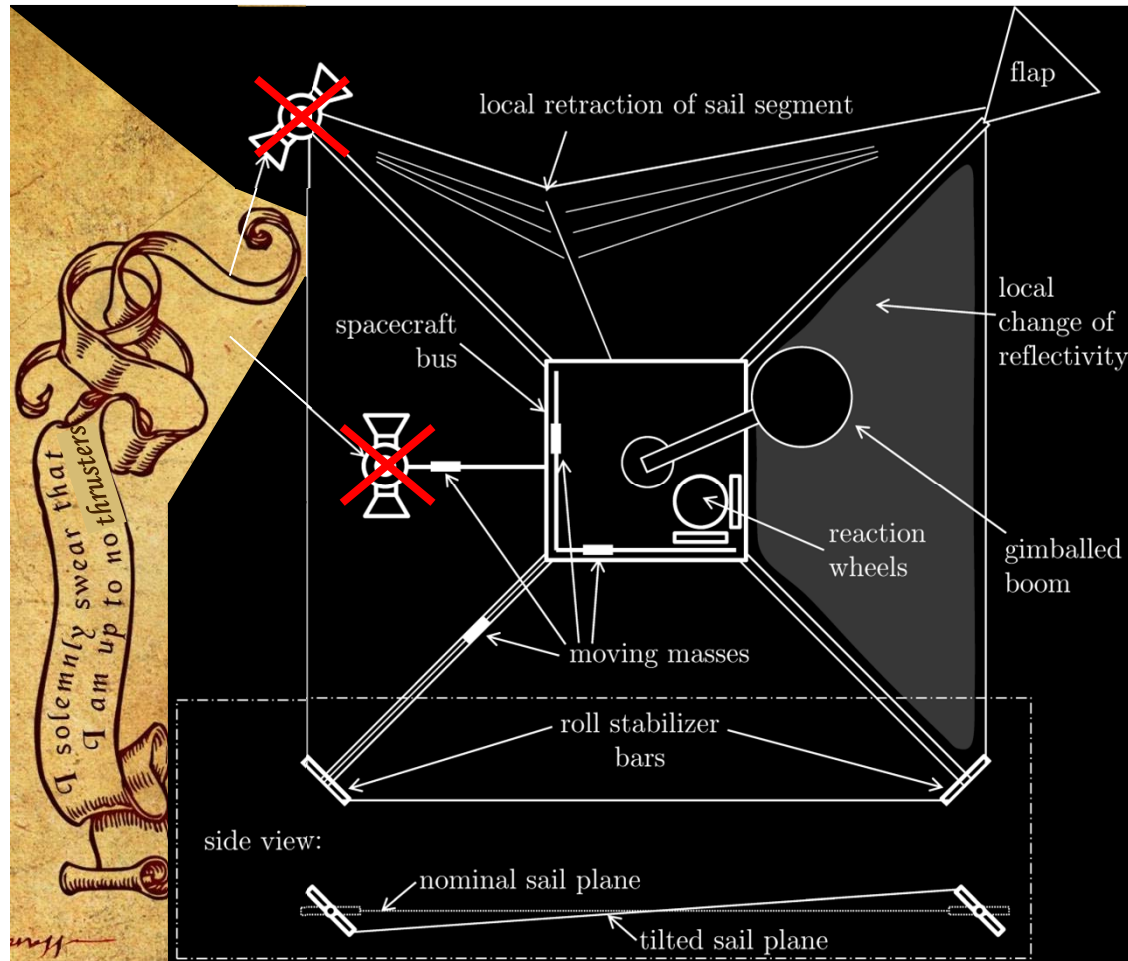
* IAA-PDC-15-P-20, P. Seefeldt et al, Large Lightweight Deployable Structures ...

** IAA-PDC-15-P-66, C.D. Grimm et al, On Time, On Target – ...



The GOSSAMER Roadmap

step 2 – control



GOSSAMER-2 – in-orbit attitude & thrust vector control demo

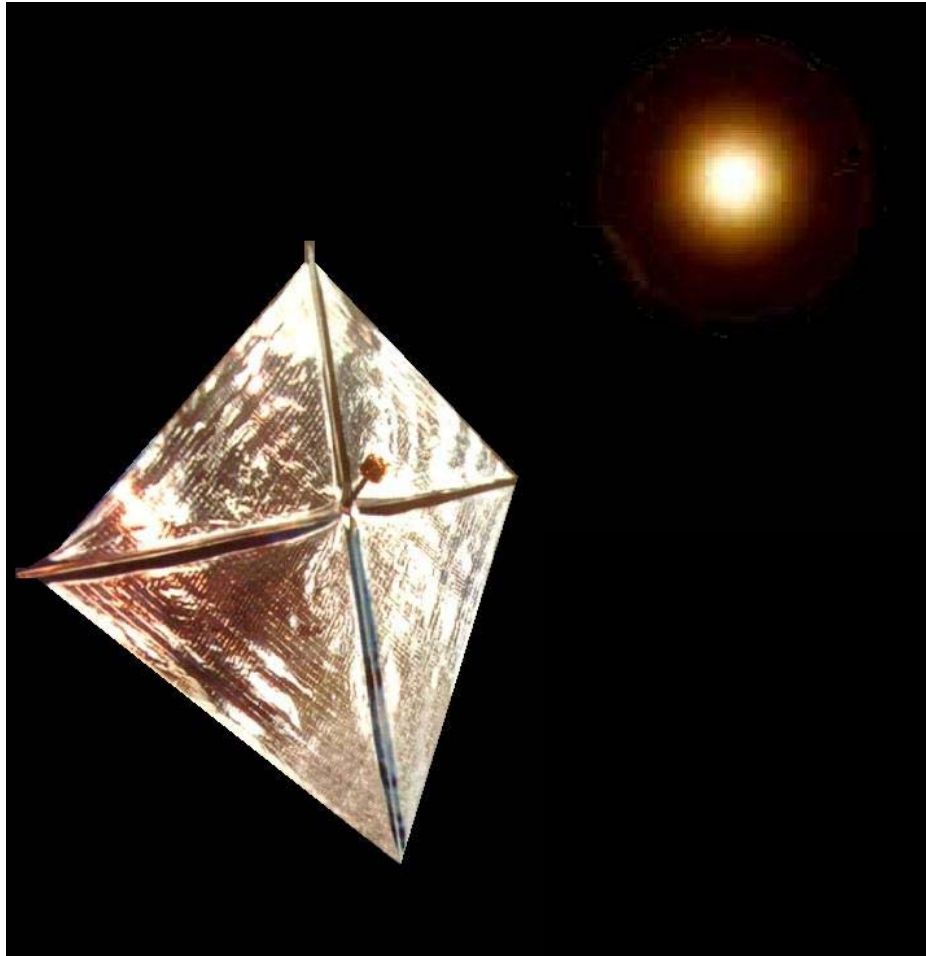
- $(20 \text{ m})^2$ sail area
- orbit where solar radiation pressure is dominant – high LEO, MEO, GTO
- implementation of several (all?) control methods and all relevant mechanisms
- **find out what's the best**
– 1, 2,..., many combined?
- ~2 years after GOSSAMER-1 flight: requires MASCOT-style concurrent AIV & project management *
- PFM free-flyer for “piggy-back” launch
- mass & undeployed size compatible with ASAP-micro & ESPA envelopes

* IAA-PDC-15-P-66, C.D. Grimm et al, On Time, On Target – How the Small Asteroid Lander MASCOT Caught a Ride ...



The GOSSAMER Roadmap

step 3 – proving the principle

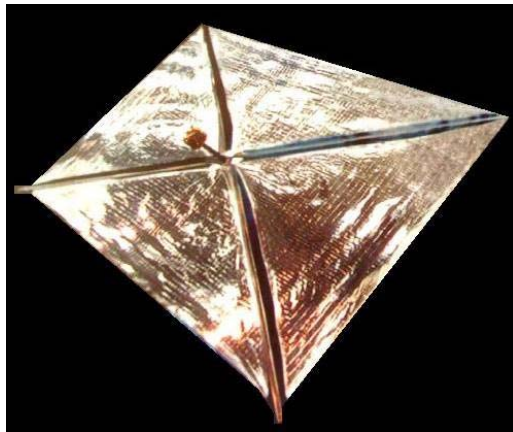


GOSSAMER-3 – all-up proof test science mission readiness demonstrator

- (50 m)² sail area
- initial orbit high enough to spiral out (sail up)
– high LEO, MEO, GTO, LTO, L_{1/2}TO
- applies best control method(s)
- **prove that sails can operate science missions successfully**
 - ***tiny*** science payload: imager & sail-environment interaction
- ~2 years after GOSSAMER-2 flight: again, MASCOT-style concurrent AIV*
- PFM free-flyer for “piggy-back” launch
- mass & undeployed size compatible with ASAP-micro & ESPA envelopes
- **instrumentation to observe sail ageing in space, as long as it lasts**



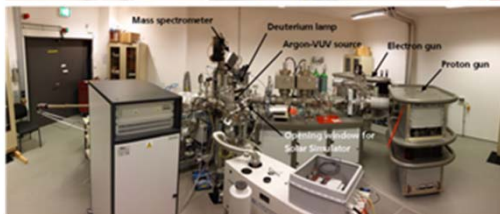
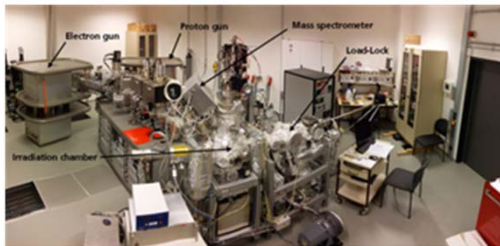
GOSSAMER-3 quality assurance – sail monitoring, pointing verification



GOSSAMER-3 – instruments wish list

a *tiny science*(...-like) payload to **observe sail ageing** *

- wide-angle camera to observe **sail deployment and long-term foil behaviour** and provide **proof images for attitude control, pointing stability & accuracy**
- sensor to observe **interaction of a sail with solar wind's & geomagnetic field**
- sensor to observe **plasma, particle and energetic radiation sail environment**
- sensor to observe **large area foil reflectivity ageing**, e.g. **thermal equilibrium**
- sensor to observe **small-scale space weathering mechanisms of foil ageing**
- sensor to observe **core spacecraft (electronics) electromagnetic signature**
- sensor to observe **illumination changes and Sun glints off the sail surface**
- sensor to register **space debris and natural dust impacts on the sail foil**



* IAA-PDC-15-P-20, P. Seefeldt et al, Large Lightweight Deployable Structures for Planetary Defence: ...



Gossamer-3 quality assurance



MASCOT Flight Spare Instruments capabilities

a tiny(...-like ;-) science payload*** to observe sail ageing *

- wide-angle camera to observe **sail deployment and long-term foil behaviour** and provide **proof images for attitude control, pointing stability & accuracy**

- 
- GNC PEC
- interaction of a sail with solar wind's & geomagnetic field
- plasma, particle and energetic radiation sail environment

The diagram shows a 3D model of the GNC system components. Labels include: GNC PEC, GNC OPS, thermal IR radiometer, and large area foil reflectivity ageing, e.g. thermal equilibrium. A red circle highlights the large area foil reflectivity ageing mechanism, and a red arrow points to the thermal IR radiometer.

GNC OPS

(MARA)

core spacecraft (electronics) electromagnetic signature

illumination changes and Sun glints off the sail surface

MicroOmega

Camera (CAV)

space debris and natural dust impacts on the sail foil



GOSSAMER-3 instruments left to be added

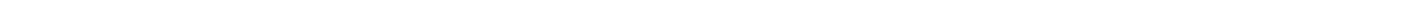
- ⊕ additional MAG instrument, 2 sensors spaced by 1 m at 1 or 2 boom ends

⊕ e.g. Spherical EUV and Plasma Spectrometer (SEPS), 2 on opposite booms

⊕ e.g. acoustic pick-ups at rigging nodes, structural vibration accelerometers

- * IAA-PDC-15-P-20, P. Seefeldt et al, Large Lightweight Deployable Structures for Planetary Defence: ...
- ** IAA-PDC-15-P-64, J.T. Grundmann et al & the MASCOT Team, Mobile Asteroid Surface Scout (MASCOT) – ...
- *** IAA-PDC-15-P-65, C. Lange et al, Technology and Knowledge Base Concepts to Enable Responsive NEO Characterization

- 



GOSSAMER-3 & MASCOT FS mission events wish list



minimum mission

- get launched cheap, deploy** & spiral up
- explore & improve sailing skills
- fly-by visit a target on time & look at it right

nominal mission

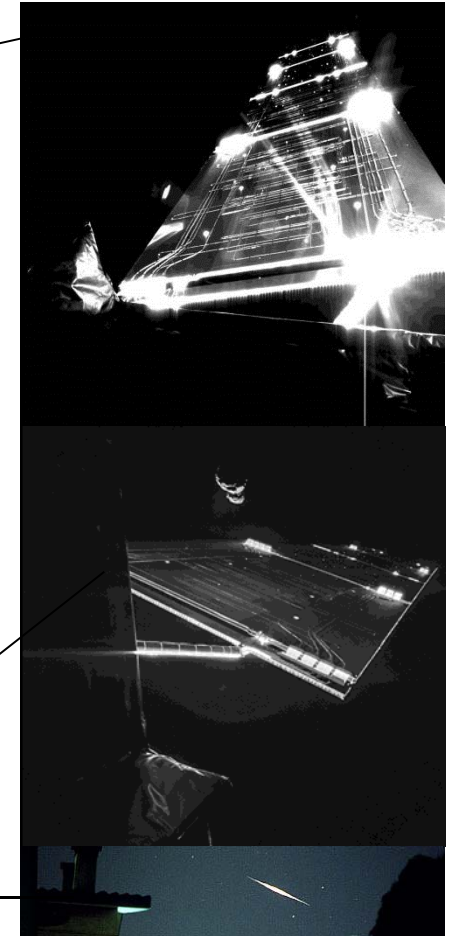
- explore practical flying in Earth-Moon system
- “all-up” navigation accuracy proof test
- low-altitude lunar gravity-assist fly-by

extended mission

- transfers to Earth-Sun L_1 , L_2 & pole-sitter
- demonstrate spaceweather Displaced- L_1

extended extended mission

- fly out to a convenient NEA – coorbital?
- rendezvous & drop MASCOT FS*,***,****
- ...some grand finale...



* IAA-PDC-15-04-08, S. Ulamec et al, Relevance of PHILAE and MASCOT in-situ investigations for Planetary Defense

** IAA-PDC-15-P-20, P. Seefeldt et al, Large Lightweight Deployable Structures for Planetary Defence: ...

*** IAA-PDC-15-P-64, J.T. Grundmann et al & the MASCOT Team, Mobile Asteroid Surface Scout (MASCOT) –...

**** IAA-PDC-15-P-65, C. Lange et al, Technology and Knowledge Reuse Concepts to Enable Responsive NEO Characterization ...



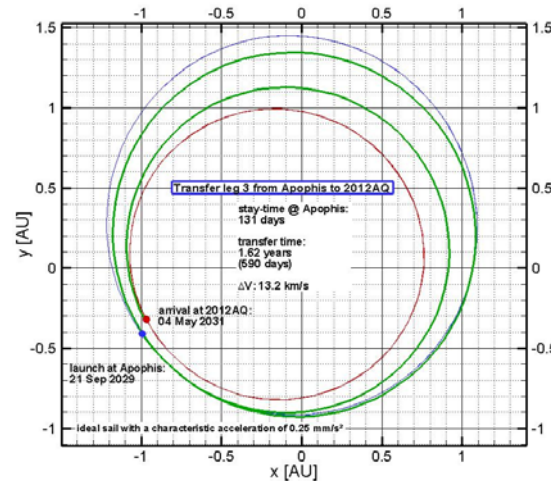
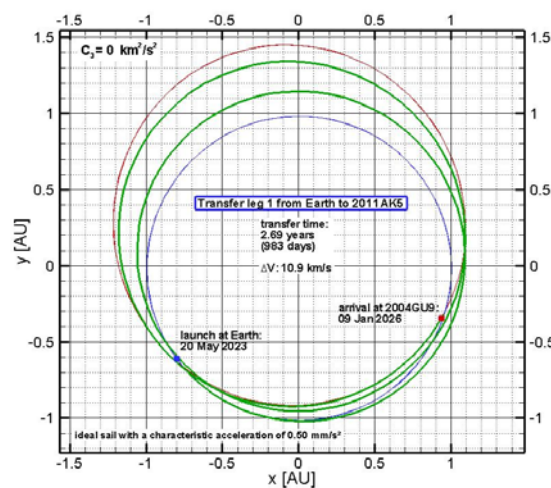
the GOSSAMER Roadmap for Solar Sailing – an Epilogue



*so bring on the lasers, bring on solar-electric propulsion! ☺
... and don't be shy about megawatts!! ;-)*



EXERCISE – EXERCISE – EXERCISE – EXERCISE PDC 2015

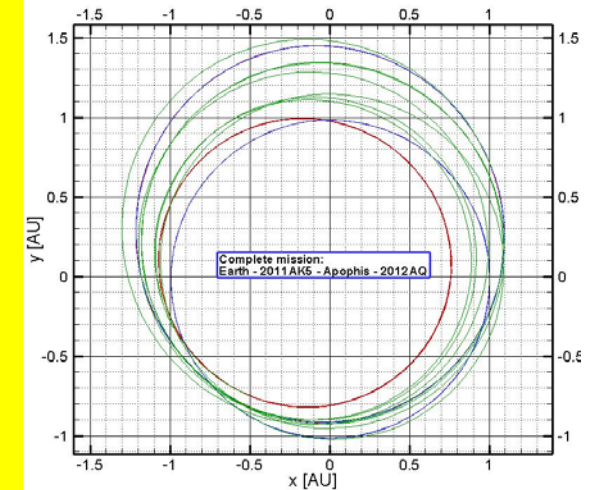
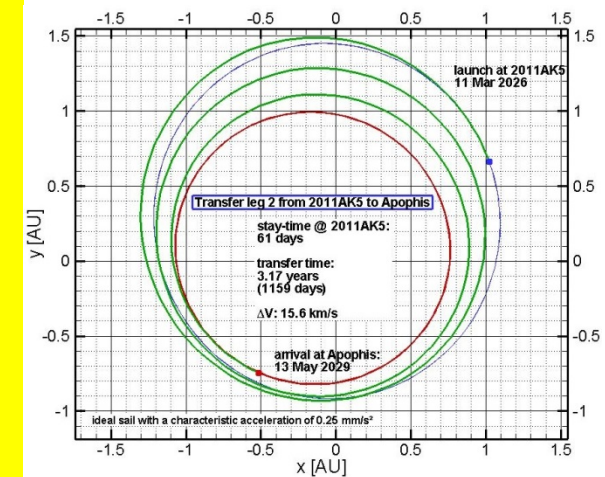


If GOSSAMER-3 had been launched according to the last Roadmap baseline in ~2020 we'd be on the way to 2015 PDC.

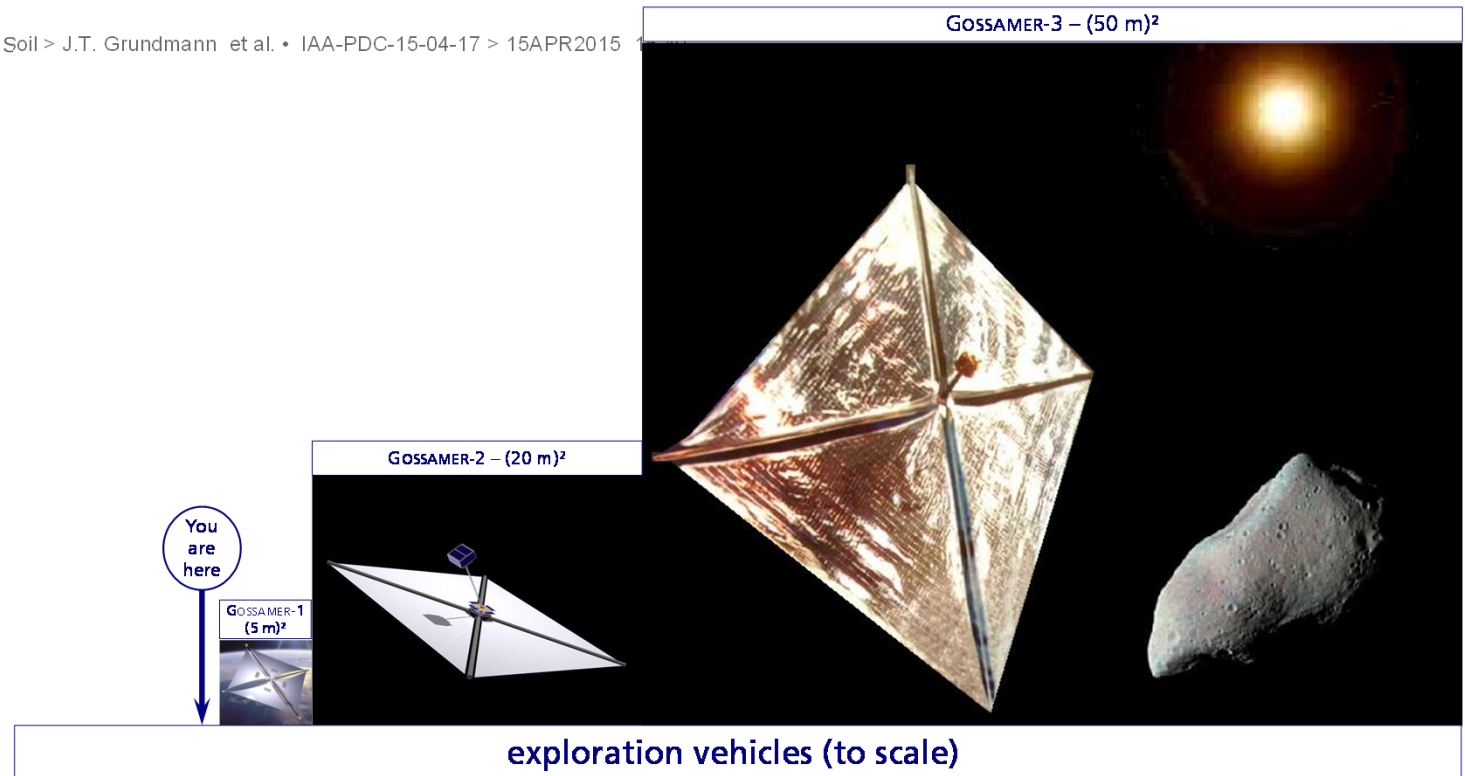
If it had been **2021 PDC**, we'd also be on the way.

why?

only Solar Sails can **change** interplanetary rendezvous **targets after launch** during cruise phase or other rendezvous in all cases



Questions?



Thank you for your attention

– take a look at these posters! 😊

our posters & talks – DLR @ PDC 2015

- IAA-PDC-15-04-08 *S. Ulamec et al*
Relevance of PHILAE and MASCOT in-situ Investigations for Planetary Defense
- IAA-PDC-15-04-17 *J.T. Grundmann et al*
From Sail to Soil – Getting Sailcraft Out of the Harbour on a Visit to One of Earth's Nearest Neighbours
- IAA-PDC-15-P-20 *P. Seefeldt et al*
Large Lightweight Deployable Structures for Planetary Defence: Solar Sail Propulsion, Solar Concentrator Payloads, Large-scale Photovoltaic Power
- IAA-PDC-15-P-64 *J.T. Grundmann et al & the MASCOT Team*
Mobile Asteroid Surface Scout (MASCOT) – Design, Development and Delivery of a Small Asteroid Lander aboard HAYABUSA2
- IAA-PDC-15-P-65 *C. Lange et al*
Technology and Knowledge Reuse Concepts to Enable Responsive NEO Characterization Missions based on the MASCOT Lander
- IAA-PDC-15-P-66 *C.D. Grimm et al*
On Time, On Target – How the Small Asteroid Lander MASCOT Caught a Ride Aboard HAYABUSA2 in 3 Years, 1 Week and 48 Hours

