"Material Characterisation of Polyurethane-Based Paper Honeycomb Sandwich Structures"

16th International Conference on Experimental Mechanics 7th – 11th July 2014, Cambridge - UK

Authors:





DLR - Overview

Lampoldshausen

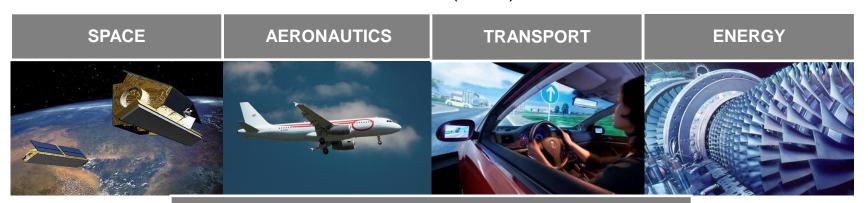
Oberpfaffenhofen

Stuttgart

Weilheim •

Köln-Porz

- 7.700 employee are working at 32 research institutes etw
 and facilities in
 - 9 locations
 - 7 branch offices
- International Branch Offices in Brussels, Paris and Washington
- Partner of
 - ♦ European Transsonic Wind Tunnel (ETW)
 - German Dutch Wind Tunnels (DNW)







DLR - Overview

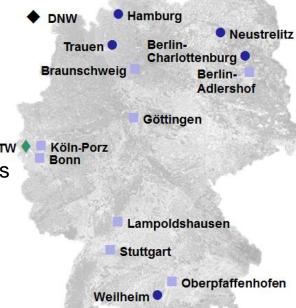
DLR Institute of Vehicle Concepts (DLR FK)

Department of Lightweight and Hybrid Design Methods

Pfaffenwaldring 38 – 40

D – 70569 Stuttgart

Germany







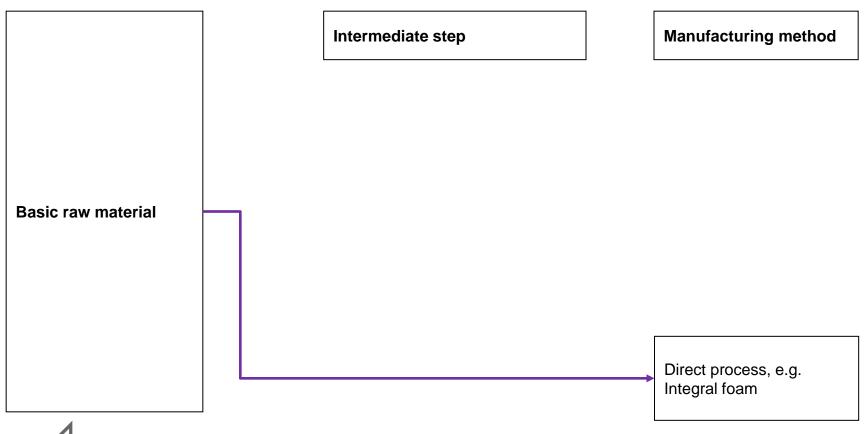
SECURITY

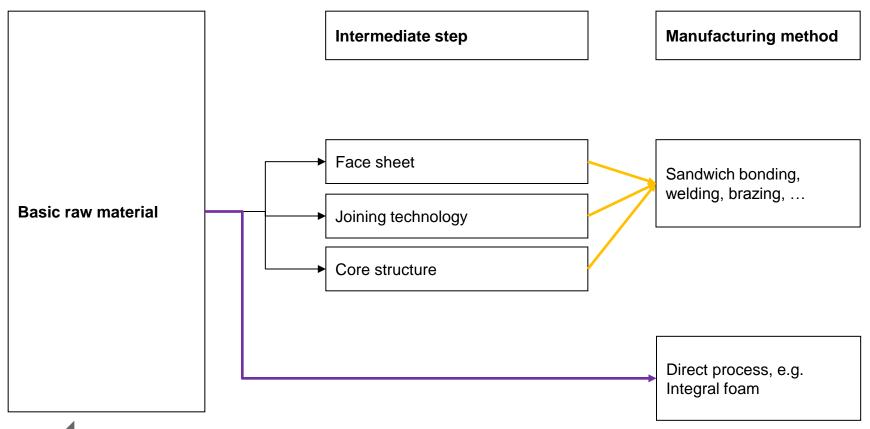
Outline

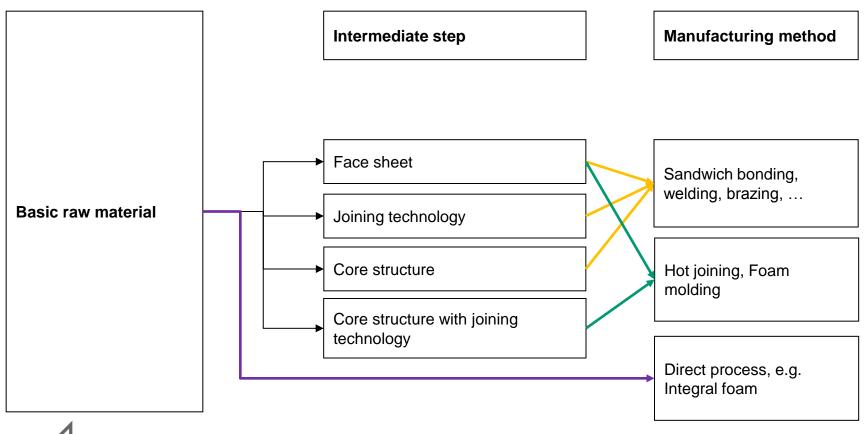
- Introduction and motivation
- Sandwich characterization
 - Approach
 - Face sheet material
 - Core material
 - Sandwich configuration in bending
- Conclusion

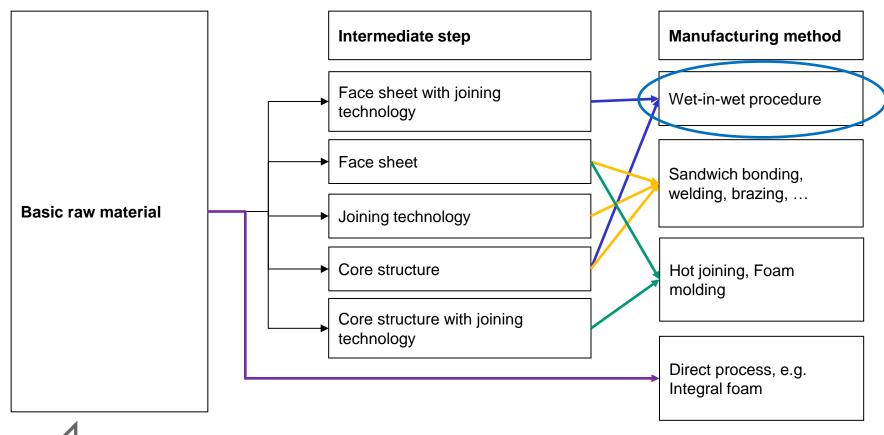












Overview of the polyurethane fibre spraying process

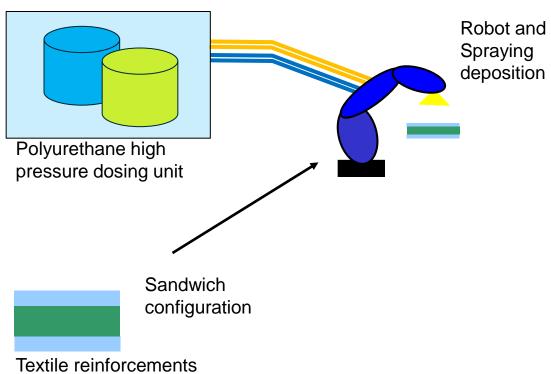


Sandwich configuration

Textile reinforcements and core structures



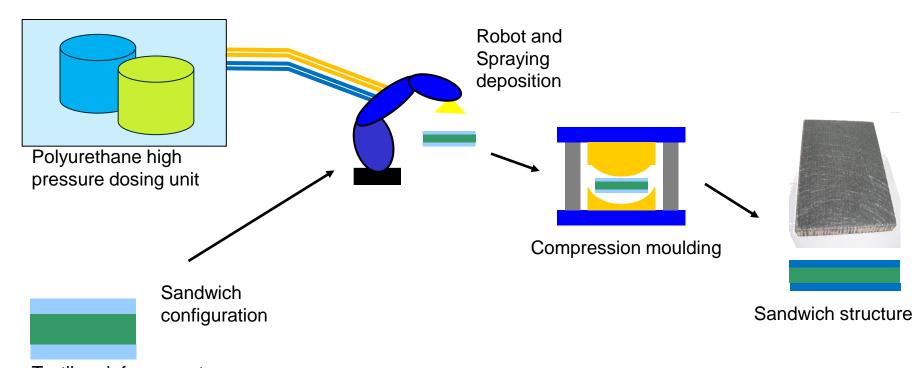
Overview of the polyurethane fibre spraying process



Textile reinforcements and core structures



Overview of the polyurethane fibre spraying process



Textile reinforcements and core structures



Challenges for the future development and application of the process

- Detail analyze of the sandwich and material properties, depending of the manufacturing process
- Examination of the relevant mode of failure





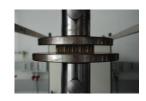
Approach

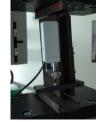


Approach

Property characterisation of face sheet and core structure in tension, compression and shear







Source: Fraunhofer ICT



Approach

Property characterisation of face sheet and core structure in tension, compression and shear

Derive correlations between the core densities and the core material properties

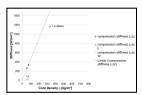






Source: Fraunhofer ICT

$$\frac{Prop_{C}}{Prop_{S}} = C \cdot \left(\frac{\rho_{C}}{\rho_{S}}\right)^{n}$$



Approach

Property characterisation of face sheet and core structure in tension, compression and shear

Derive correlations between the core densities and the core material properties

Derived failure mode maps with analytical formula and analyse the relevant modes of failure in bending

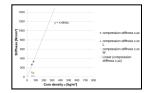




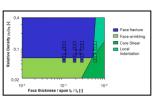


Source: Fraunhofer ICT

$$\frac{Prop_{C}}{Prop_{S}} = C \cdot \left(\frac{\rho_{C}}{\rho_{S}}\right)^{2}$$



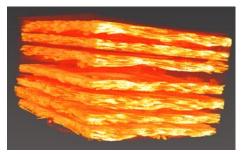






Face sheet material

- The face sheet thickness was measured with the computer tomograph (CT) also for different glass fiber mats
- In a wide range the material properties were independent of the mass of deposit polyurethane



Source: Fraunhofer ICT

	300 g/m²	450 g/m²	600 g/m²	900 g/m²
	300 g/111	450 g/111	000 g/111	900 g/111
Face sheet thickness [μm]	340	430	545	781
Standard deviation [µm]	31	26	32	38
Young modulus [N/mm²]	10300	12300	13200	13300
Standard deviation [N/mm²]	510	530	560	850
Tensile strength [N/mm²]	152	169	212	212
Standard deviation [N/mm²]	9,1	11,0	6,3	10,3



Source: Fraunhofer ICT

Variations of the sandwich structure

- Chopped glass strand mats with a mass per unit area of 600 g/m² in each cover as reinforcements for the faces
- Paper honeycomb core (A-wave, Testliner III,
 Density ca. 55 kg/m³ and A-wave, Testliner II,
 Density ca. 75 kg/m³)
- Polyurethane spraying deposition of 600 g/m² on the face sheets



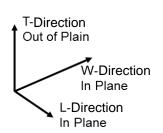


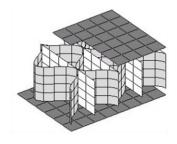




Core material

- The core properties were measured in plane and out of plane
- Measurement based on DIN53291,
 DIN53292 and DIN53294





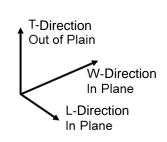
In Plane: Out of Plane:

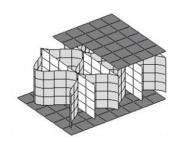
 $\begin{aligned} &\sigma_{xx}, \ \sigma_{yy}, \tau_{xy} \\ &\sigma_{zz}, \ \tau_{xz}, \ \tau_{yz} \end{aligned}$



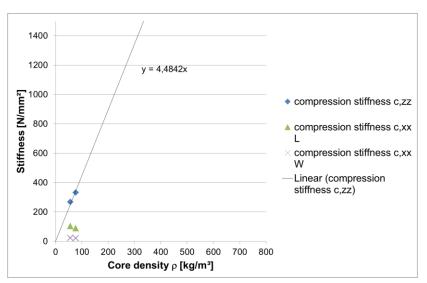
Core material

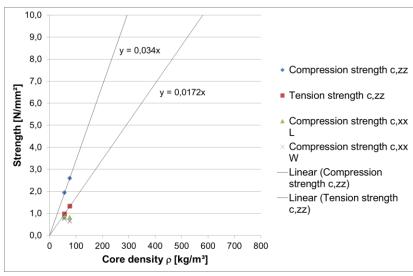
- The core properties were measured in plane and out of plane
- Measurement based on DIN53291,
 DIN53292 and DIN53294





In Plane: Out of Plane: $\sigma_{xx}, \ \sigma_{yy}, \tau_{xy}$ $\sigma_{zz}, \ \tau_{xz}, \ \tau_{yz}$

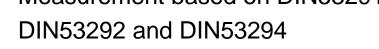


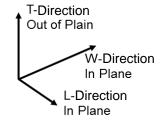


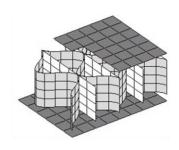


Core material

- The core properties were measured in plane and out of plane
- Measurement based on DIN53291,







In Plane: Out of Plane:

 $\sigma_{xx}, \ \sigma_{yy}, \tau_{xy}$ $\sigma_{zz},~\tau_{xz},~\tau_{yz}$

- Examples:

Material properties	Correlation in tension and compression	
Tension strenght out of plane	$\frac{\sigma_{C,tension,zz}}{\sigma_{S}} = \left(\frac{\rho_{C}}{\rho_{S}}\right)$	
Strenght in compression out of plane	$\frac{\sigma_{C,Pressure,zz}}{\sigma_S} = \left(\frac{\rho_C}{\rho_S}\right)$	

Material properties	Correlation in shear
Shear strenght out of plane L-direction	$\frac{\tau_{C,xz,L}}{\tau_S} = 1{,}348 \cdot \left(\frac{\rho_C}{\rho_S}\right)$
Shear stiffness out of plane L-direction	$\frac{G_{C,xz,L}}{G_S} = 0.763 \cdot \left(\frac{\rho_C}{\rho_S}\right)$



Sandwich configuration in bending

- Relevant mode of failure:
 - Face sheet failure
 - Core shear failure
 - Locale indentation





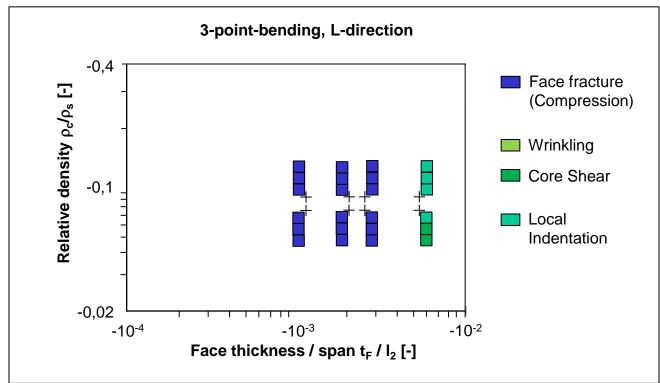




Sandwich configuration in bending

- Failure mode maps



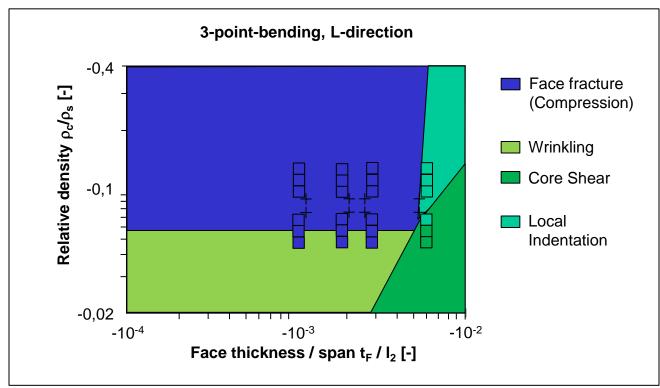




Sandwich configuration in bending

- Failure mode maps



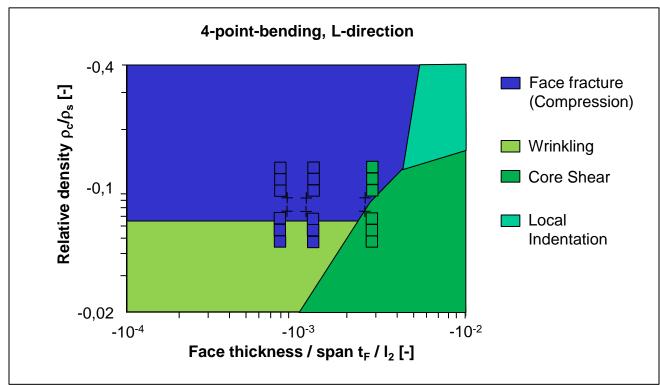




Sandwich configuration in bending

- Failure mode maps



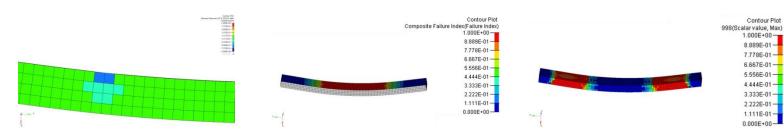




Conclusion

Material Characterisation of Polyurethane-Based Paper Honeycomb Sandwich Structures

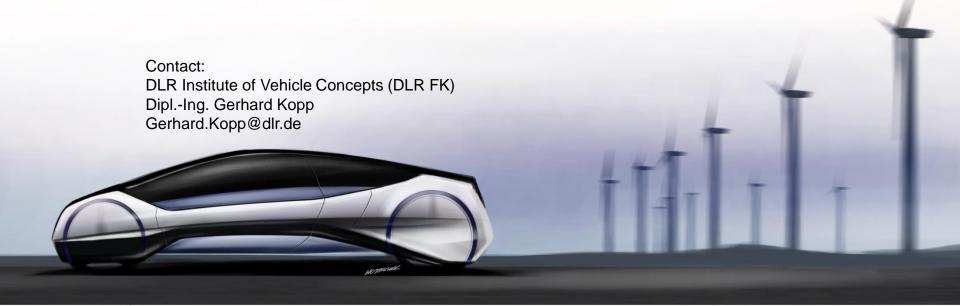
- Basic material parameters were determined from sandwich structures produced using the polyurethane fibre spraying process
- Furthermore, the critical failure modes were determined using bending tests
- These results were used to design and simulate such sandwich components with FEM







Thank you very much for your attention.





Source

- [1] Cageao, R.A.; Lorenzo, J.M.; Franken, K.: Studies of Composites made with Baypreg F: Component Selection of Optimal Mechanical Properties, in: Polyurethanes, 2004
- [2] Triantafillou, T.; Gibson, L.: Failure Mode Maps for Foam Core Sandwich Beams, in: Materials Science and Engineering, Vol. 98, 1987, p. 37 53
- [3] Petras, A.; Sutcliffe, M.: Failure mode maps for honeycomb sandwich panels, in: Composite Structures, Vol 44, 1999, p. 237 252

