

Space Sustainability by Laser Propulsion

DLR Research from Launch to Post-Mission Disposal

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Laser Propulsion Studies

@ DLR Institute of Technical Physics



Spacecraft Propulsion Concepts

- Laser-ablative micropropulsion
- Remotely powered orbital transfer
- Remotely powered launch

Debris Propulsion Applications

- Collision avoidance by photon pressure
- Laser-ablative collision avoidance
- Laser-ablative debris removal by re-entry

Sustainability Issues of Laser Propulsion



Spacecraft Propulsion Concepts

- Laser-ablative micropropulsion
- Remotely powered orbital transfer
- Remotely powered launch

Eco-friendly exhaust	Propellant reduction
✓	✗
✓	✓
✓	✓

Debris Propulsion Applications

- Collision avoidance by photon pressure
- Laser-ablative collision avoidance
- Laser-ablative debris removal by re-entry

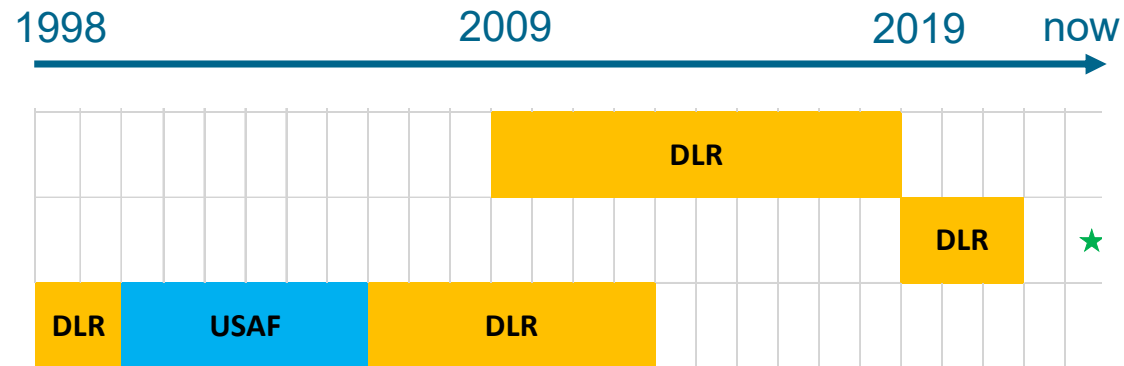
Risk mitigation	LEO protection
✓	✗
✓	✗
✓	✓

Activity status



Spacecraft Propulsion Concepts

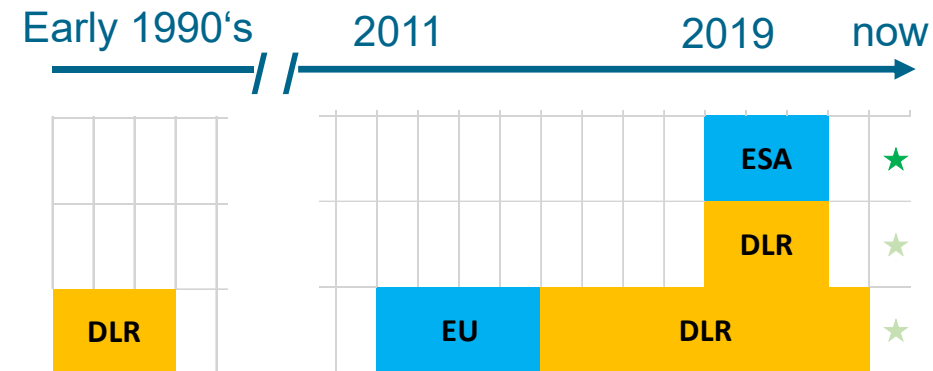
- Laser-ablative micropropulsion
- Remotely powered orbital transfer
- Remotely powered launch



Debris Propulsion Applications

- Collision avoidance by photon pressure
- Laser-ablative collision avoidance
- Laser-ablative debris removal by re-entry

		In-house studies
		External funding
★	★	Planned studies



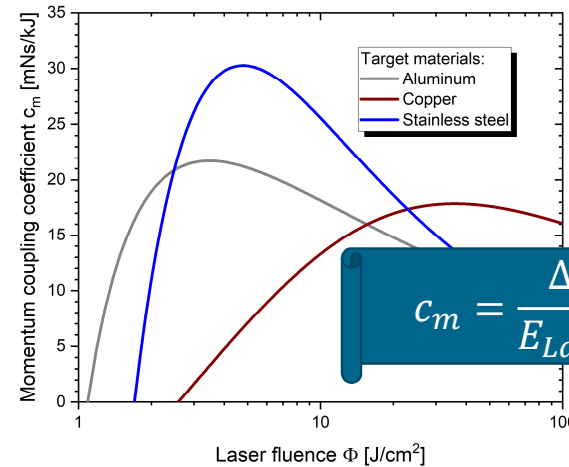
Micropropulsion w/o moving Components



Target applications

- Precise attitude control
- Replacement / complement for electric propulsion

- Very small impulse bits
- Laser pulse rate → thrust level

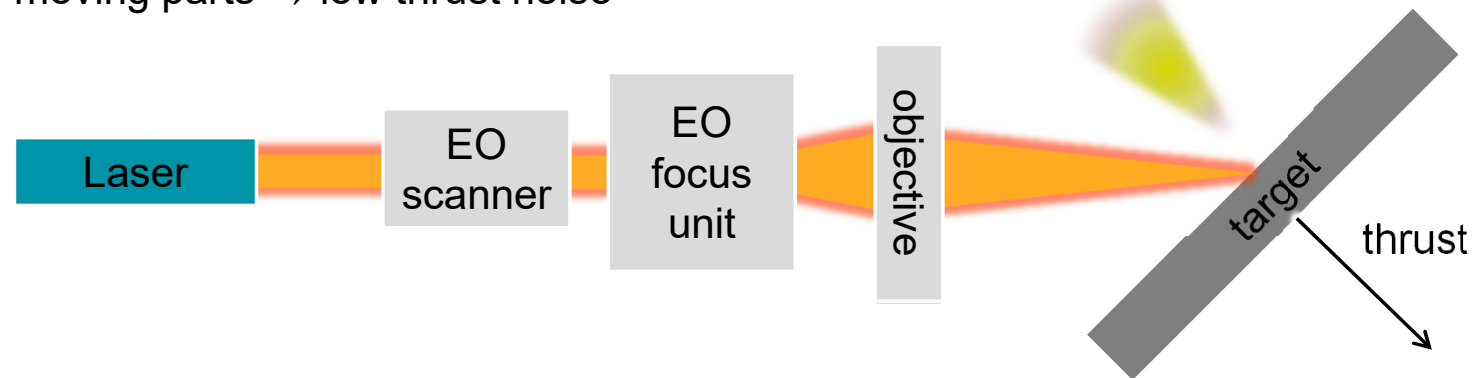


$$c_m = \frac{\Delta p}{E_{\text{Laser}}}$$

No moving parts → low thrust noise

DLR heritage

- Small ($\sim\mu\text{N}$) thrust measurement
- Thrust noise evaluation
- Design for micro thruster



Onboard laser → NOT remote propulsion

Micropropulsion w/o moving Components

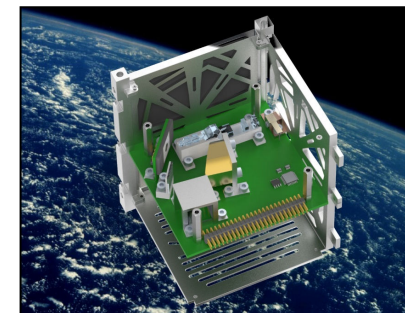
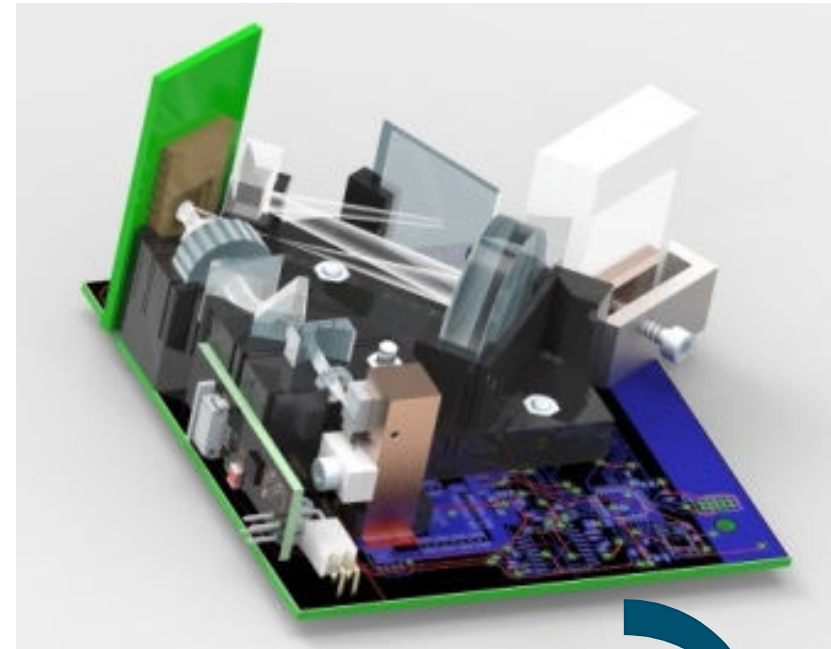


State of the art

- Demonstration module for micro thruster
 - 180 mW laser diode (10 μ s pulse, 1 kHz)
 - 2D MEMS scanner \rightarrow residual thrust noise
 - Liquid lens (electro-optical)
 - Propellant: graphite
 - USB-powered

Development needs

- Maturation of core technologies for in space use:
 - ns-pulsed Laser
 - Optical Scanner
 - Optics cleaning



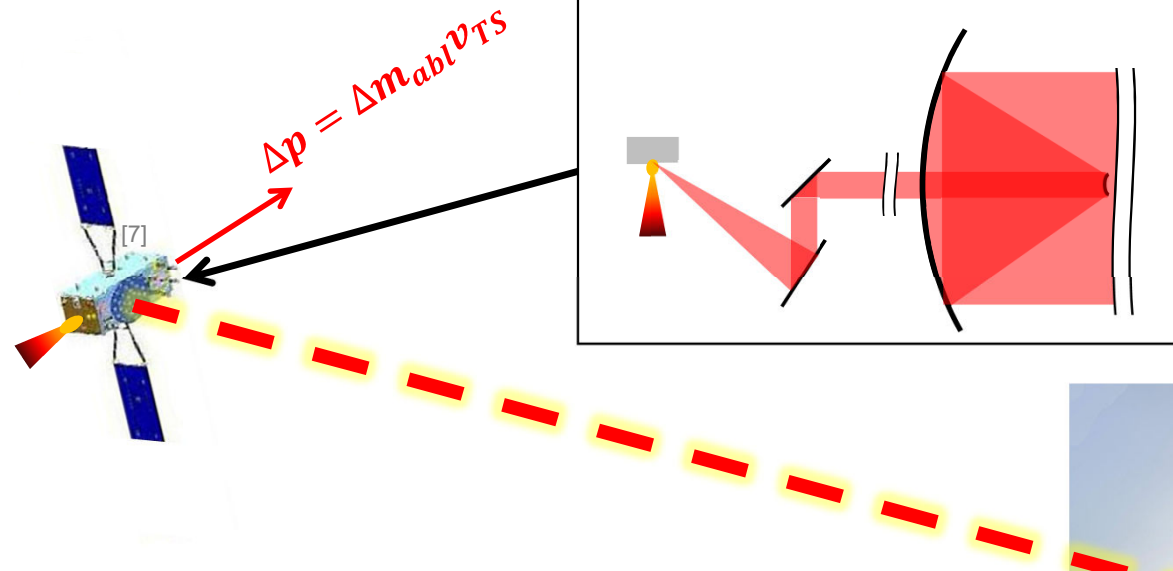
10 x 10 x 5 cm³

Post-mission Disposal powered from Ground



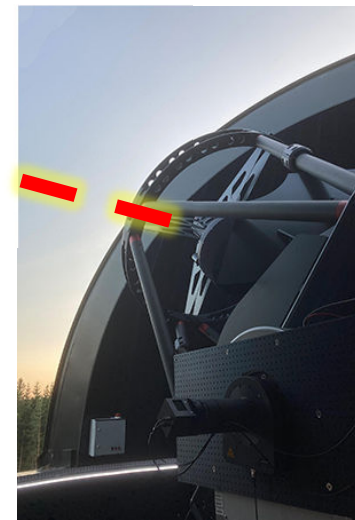
Target applications

- Deorbiting of space craft post mission.
- Business case: Laser-based deorbiting service



DLR heritage

- Conceptual studies comprising:
 - Power to thrust converter
 - Power beaming
 - Orbital maneuvers



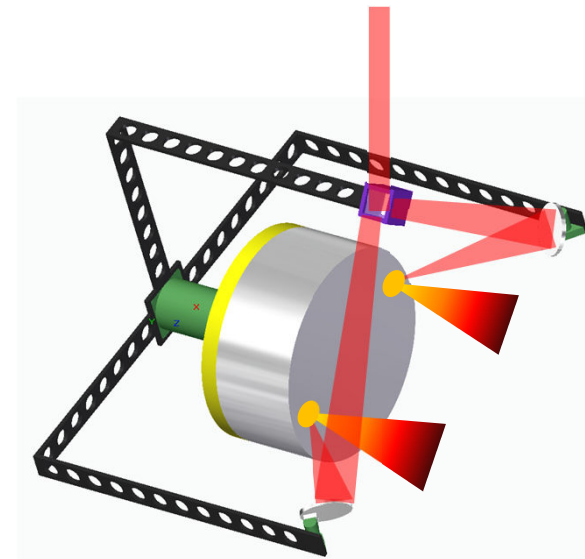
Post-mission Disposal powered from Ground

State of the art

- Detailed calculations and concepts for
 - Power to thrust converter
 - Power beaming
 - Orbital maneuvers

Development needs

- Technologies for ground Station:
 - Laser-system, adaptive optics, large aperture power-beaming
- Technologies on S/C:
 - Large aperture laser-power receiver, beam steering, power to thrust conversion



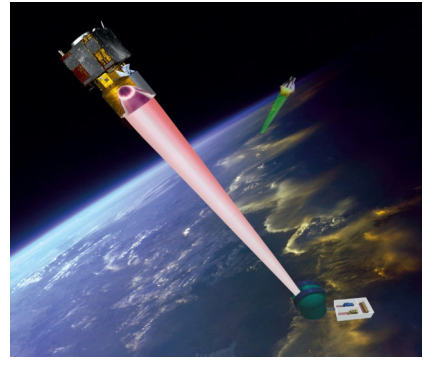
Laser Lightcraft: Launch by Photons + Air

„ $I_{sp} \rightarrow \infty$ “



Target applications

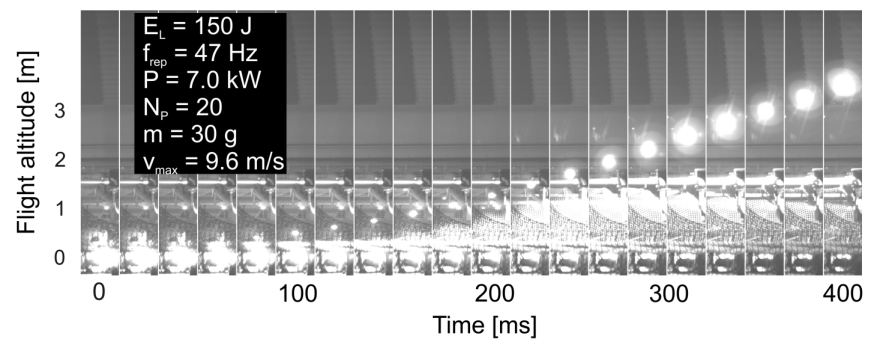
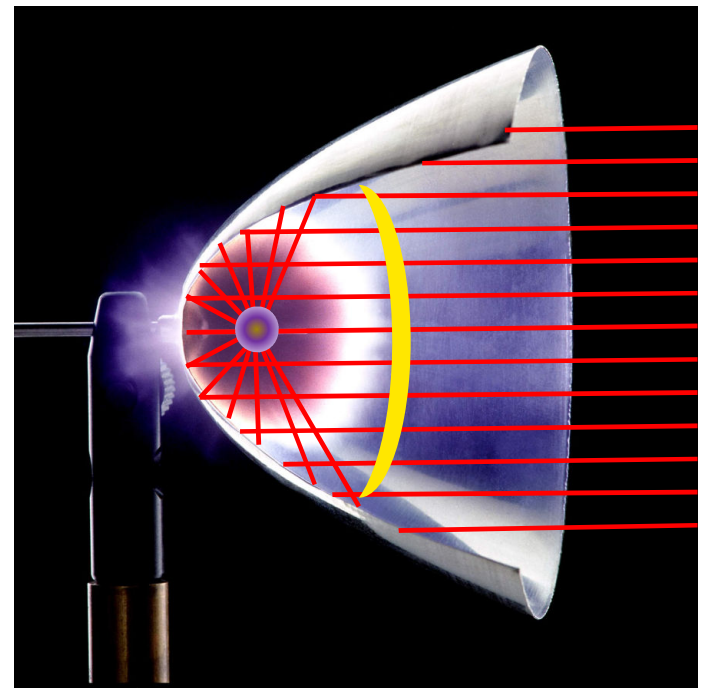
- Picosat launchers
- Ground station for sustainable supply of propulsive energy
- ... or from space-based station:
 - Sample return missions (tractor beam)
 - In-orbit logistics



H.-A. Eckel et al., Concept for a Laser Propulsion Based Nanosat Launch System, AIP Conf. Proc. 702, 263 – 273 (2004), <https://doi.org/10.1063/1.1721006>

DLR heritage

- Parabolic lightcraft with ignition and steering device
- CO₂ high energy laser experiments: plasma diagnostics, impulse pendulum, lab flight experiments
- Beam-riding analysis

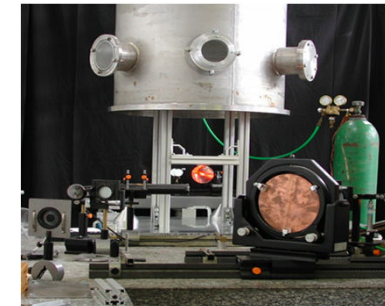
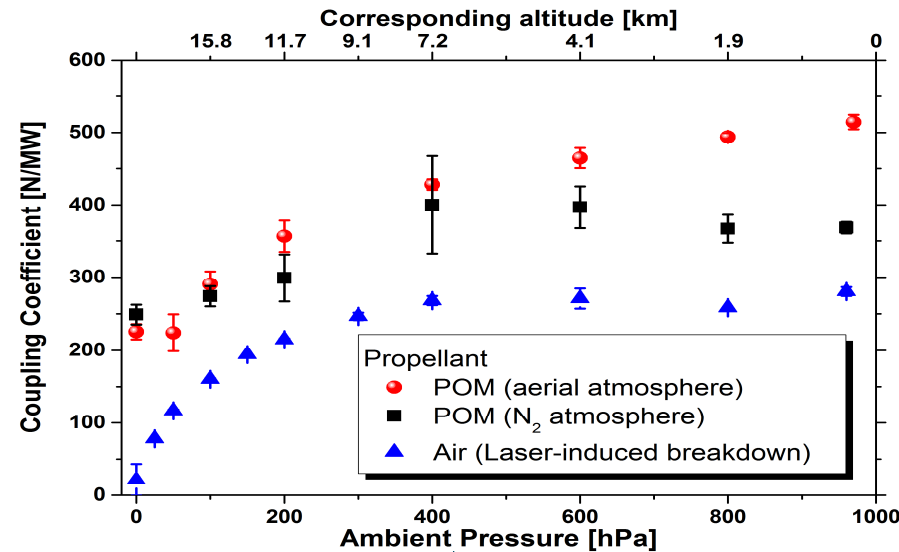


Picosat Launchers with laser-ablative propellant



State of the art

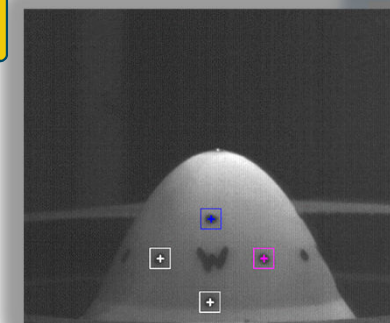
- Free flight in the laboratory (8 m) with air breakdown (no propellant)
- 10 kW pulsed laser → 30 g vehicle
- Beam-riding constraints identified
- Analysis of laser ablation & detonation



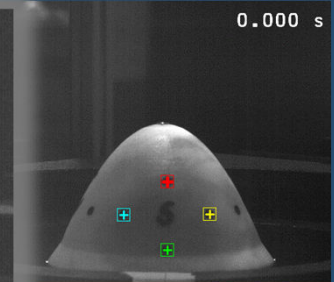
W.L. Bohn and W.O. Schall, Laser Propulsion Activities in Germany, AIP Conf. Proc. 664, 79 – 91 (2003), <https://doi.org/10.1063/1.1582098>

Development needs

- High energy ground laser ($m_{payload}/P_{Laser} \approx 1 \text{ kg/MW}$) (10 – 100 kJ, 10 – 1000 Hz, near IR)
- High energy transmitter + adaptive optics
- Beam riding design + steering loop
- Propellant optimization



Front view



Side view

LEO Space Debris Nudging by Photon Pressure



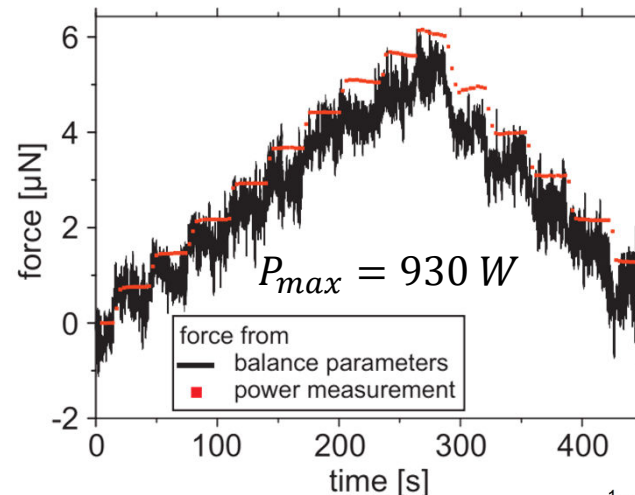
Target applications

Collision avoidance:

- Debris vs. debris
- Deb. vs. non-maneuvrable S/C
- Service for propellant saving

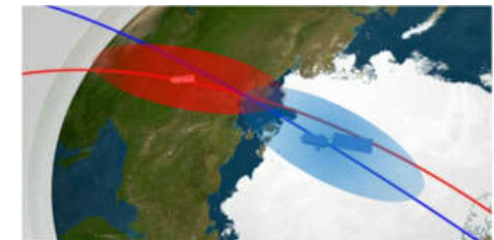
DLR heritage

- Laser ranging and lightcurve analysis
- Development of high power solid-state lasers
- Measurement of photon pressure at kW laser power level
- Light force raytracing computations for arbitrarily shaped targets
- Network analysis for laser tracking and momentum transfer (LTMT) including weather constraints
- High-level LTMT station design
- Prime of ESA Phase 0 Study (2019 – 2021)

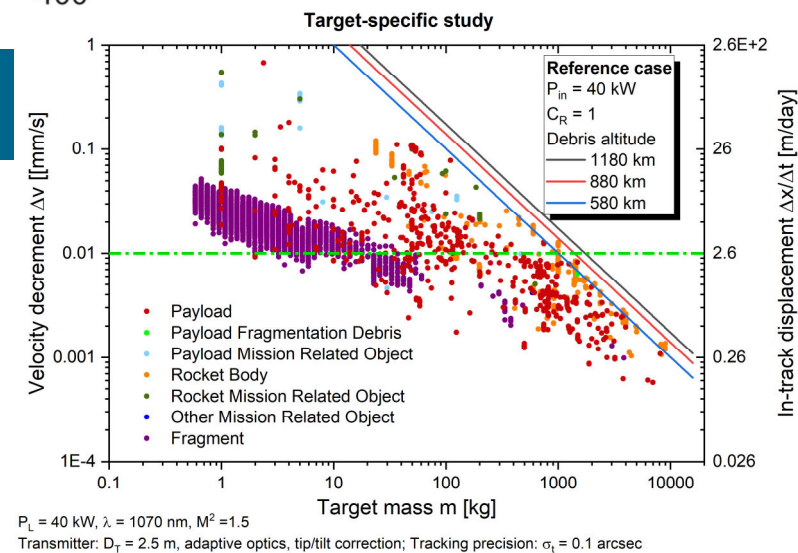


S. Karg et al., Laser Propulsion Research Facilities at DLR Stuttgart, HPLA 2014, <https://elib.dlr.de/89162/>

$c_m \approx 5 \mu\text{N/kW}$



Collision probability from covariance overlap
Image: ESA / DLR



Space Debris Nudging by a LTMT Station Network



$$\Delta x / \Delta t \approx 2.6 \text{ m/d from } \Delta v = 10 \text{ } \mu\text{m/s}$$

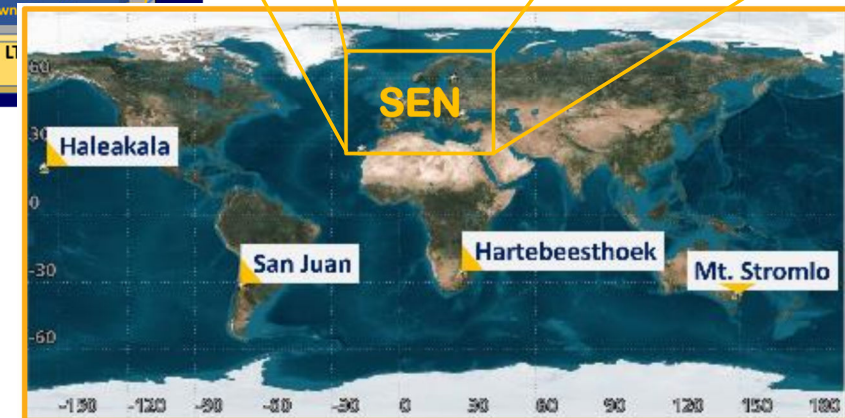
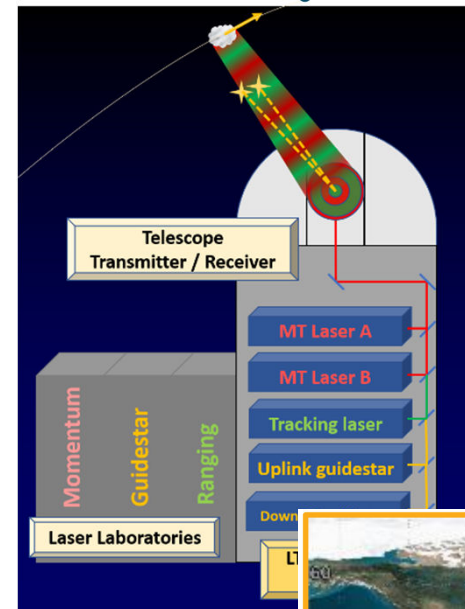
State of the art

Feasible station requirements found:

- 0.1" precision tracking with adaptive optics
- Adaptive focusing supported by laser ranging
- 40 kW by beam combining of solid-state lasers
- 2.5 m transmitter
- 0.1" pointing precision

Development needs

- High power beam combining
- High power adaptive optics
- High precision tracking and pointing
- Risk mitigation (thermal, reflections, political)

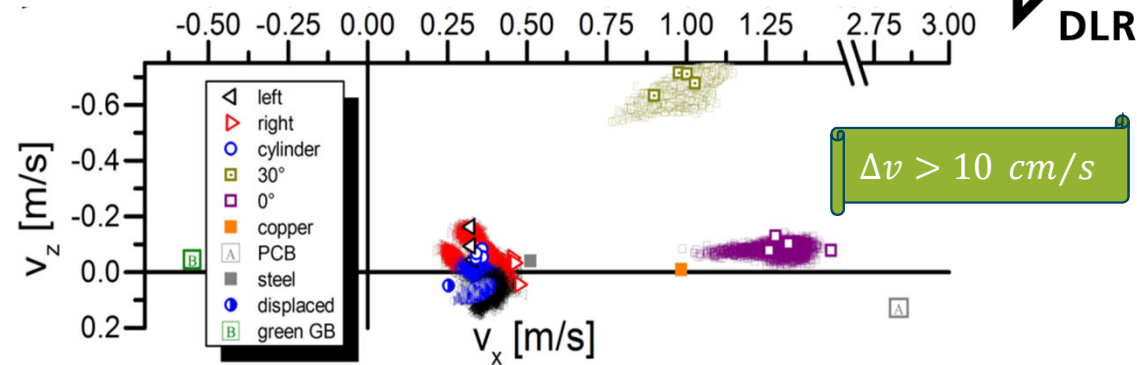
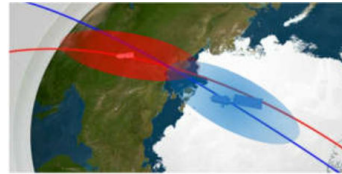


Ground-based Collision Avoidance by a few Laser Pulses



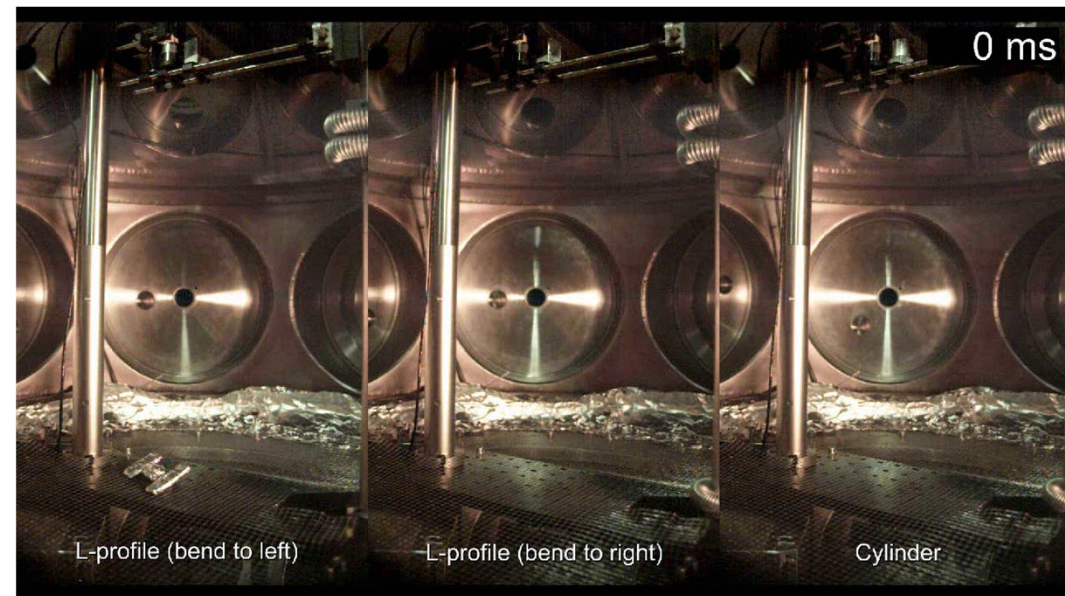
Target applications

- Collision avoidance:
 - Debris vs. debris
 - Debris vs. non-maneuvrable satellites
 - Service for propellant saving
 - Supplement / upgrade for LTMT by photon pressure



DLR heritage

- Single pulse experiments on laser- induce momentum and heat to cm-sized targets
- FEM and raytracing simulations on thermo-mechanical laser-matter interaction
- Simulations of atmospheric beam propagation and turbulence compensation

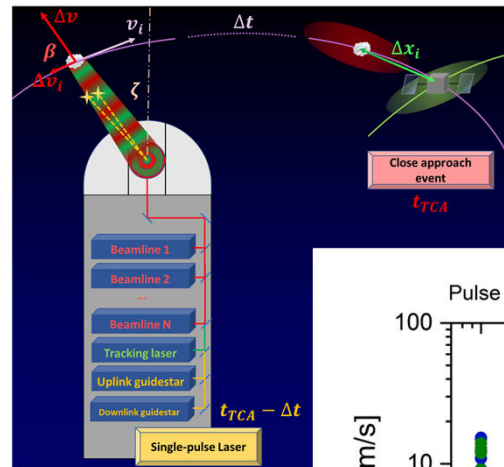


Laser pulse energy: 80 J, single pulse (nHelix, GSI Darmstadt)

Ground-based Collision Avoidance by a few Laser Pulses

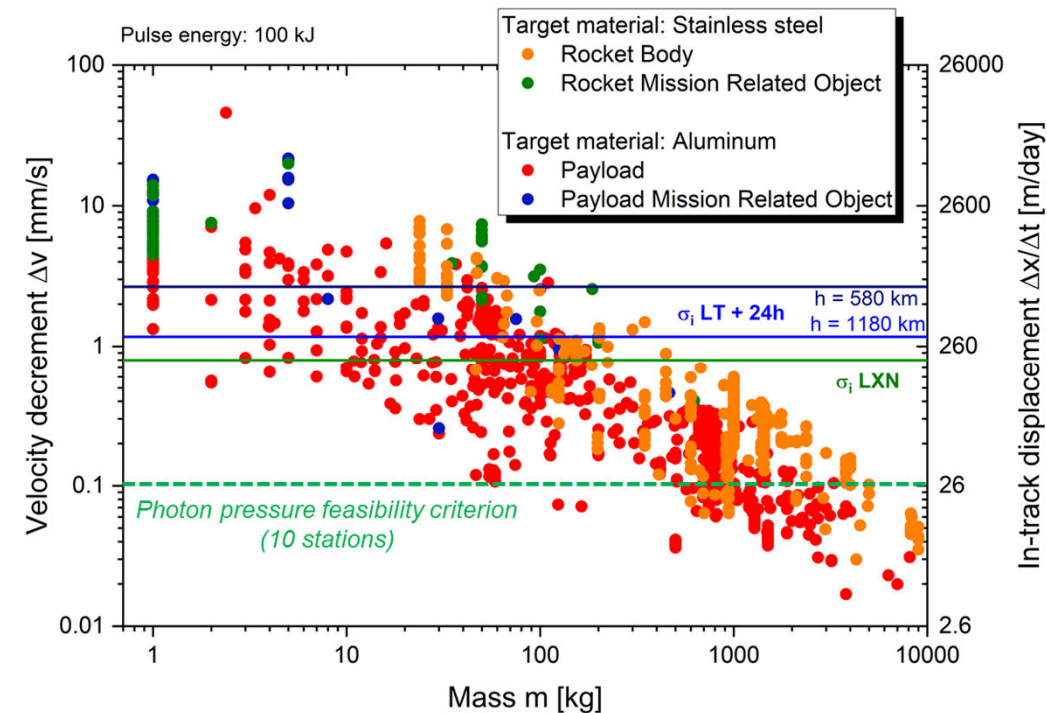
State of the art

- Experimental proof-of-principle in vacuum drop experiment
- Analysis of adaptive optics configuration to achieve relevant fluences in LEO



Development needs

- Adaptation of kJ beamline technology from inertial fusion (or coherent beam coupling)
- High energy transmitter with adaptive optics



De-orbiting of small Space Debris

Target applications

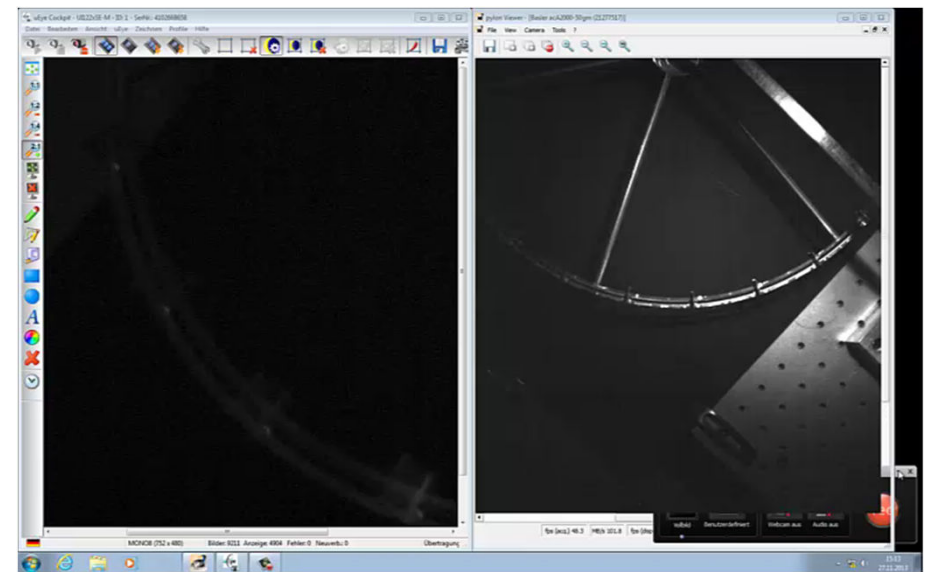
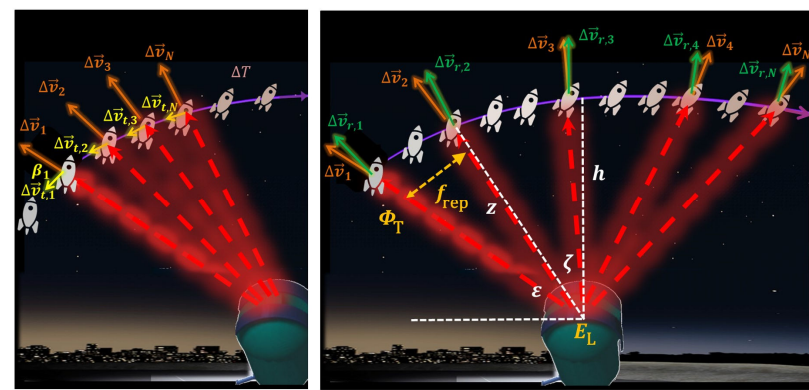
- Removal of debris fragments
- Collision avoidance as integrated business case

DLR heritage

- Development of pulsed high energy solid-state lasers
- Research on coherent beam coupling
- Laser-ablative thrust measurements for space debris materials
- Simulation of laser-momentum coupling to arbitrarily shaped targets
- Propagation of laser-modified orbits
- Analysis of operational safety (thermo-mechanical, legal...)

S. Scharring and R.-A. Lorbeer, DLR Institute of Technical Physics, Jun 29, 2023

$h_{perigee} \rightarrow 200 \text{ km}$



Laser power: 33 W, Thrust: 700 μN

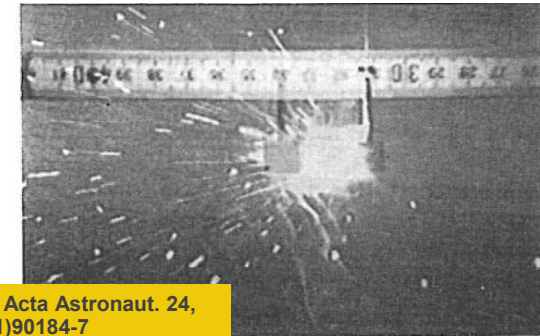
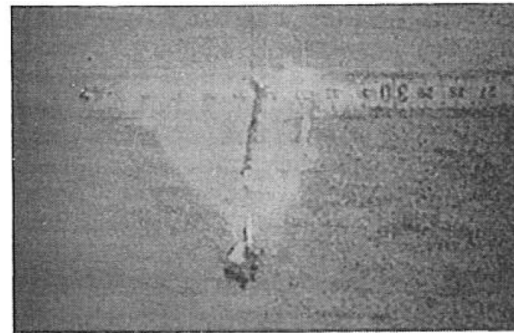
B. Esmiller et al., Space debris removal by ground-based lasers: main conclusions of the European project CLEANSPACE, Appl. Opt. 53(31): 145 – 154 (2014), <https://doi.org/10.1364/AO.53.000145>

De-orbiting of small Space Debris



State of the art

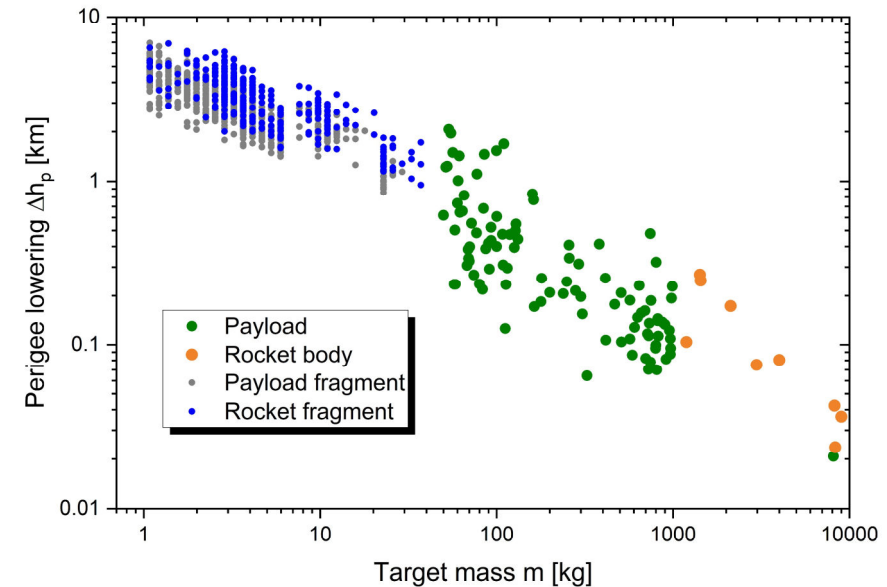
- Detailed analysis on momentum coupling
- Identification of thermal constraints
- Pulsed laser sources on the 1 Joule level
- Coherent coupling of a few laser emitters

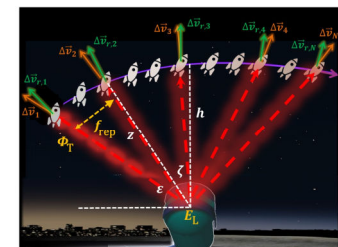
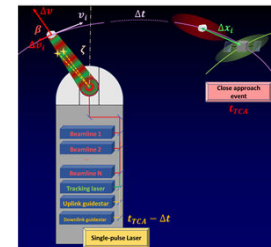
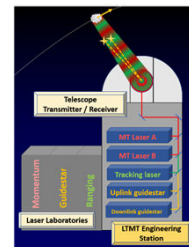
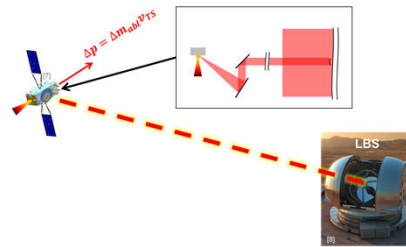
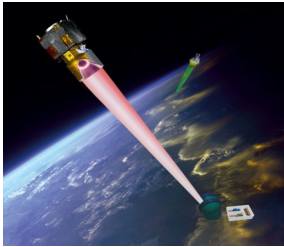
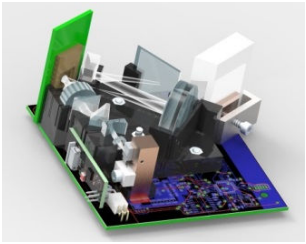


W. Schall, Orbital debris removal by laser radiation, Acta Astronaut. 24, 343–351 (1991), [https://doi.org/10.1016/0094-5765\(91\)90184-7](https://doi.org/10.1016/0094-5765(91)90184-7)

Development needs

- Predictive avoidance for orbit modification
- Debris reconnaissance for target selection/exclusion
- Pulsed laser sources on the 10 – 20 J level
- Coherent coupling of a few thousand laser emitters
- Transmitters and adaptive optics for high pulse energies





Thank you for your kind attention

Lasers and Space



*The things we see are the result of our past,
but the way we act is the result of our future.*

S. Scharring and R.-A. Lorbeer, DLR Institute of Technical Physics, Jun 29, 2023



Gravitational lensing smiley, credit: NASA & ESA under CC BY 4.0 license