

Parameterization of trimmed NURBS geometries for mesh deformation

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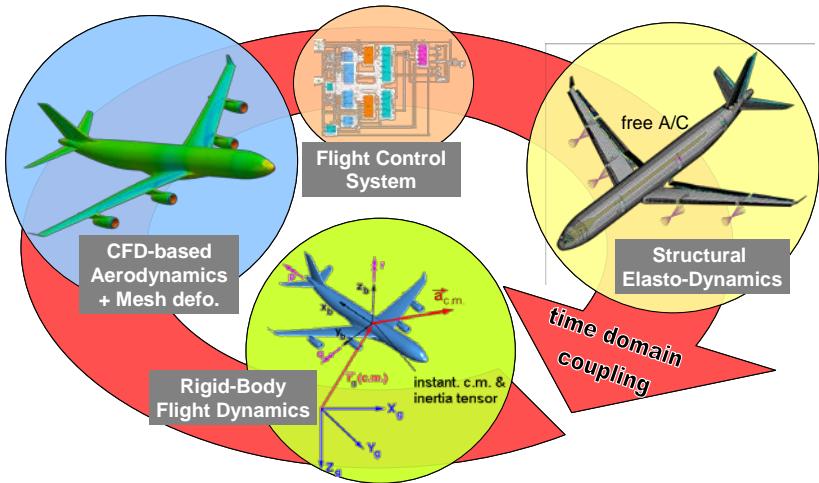


Knowledge for Tomorrow

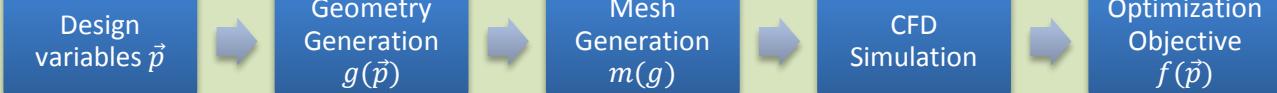


Motivation

Gradient based design optimization



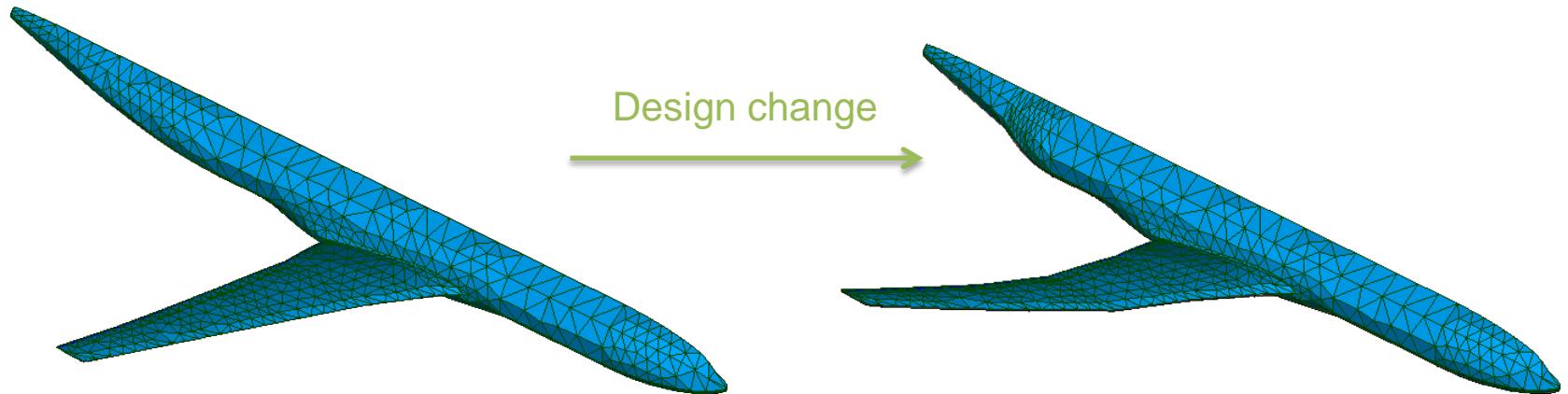
Optimization



Motivation

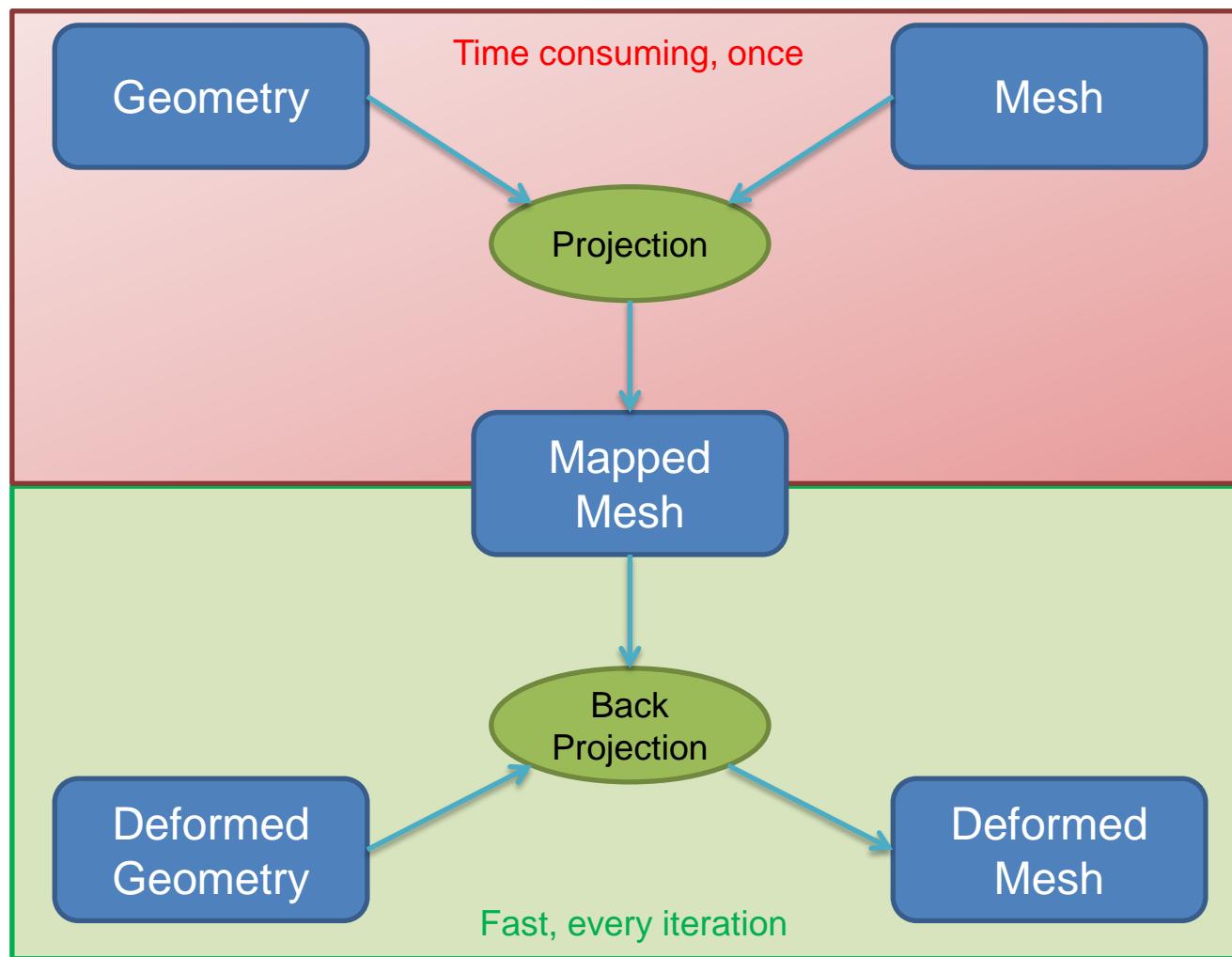
Geometry based mesh deformation

- Problem: fully automatic mesh creation is hard:
 - Structured meshes require fine tuning
 - Number of mesh points varies → Finite differences not possible
- Solution: deform manually created mesh on design changes
- Can be used for gradient based optimization to determine mesh gradient w.r.t design parameters



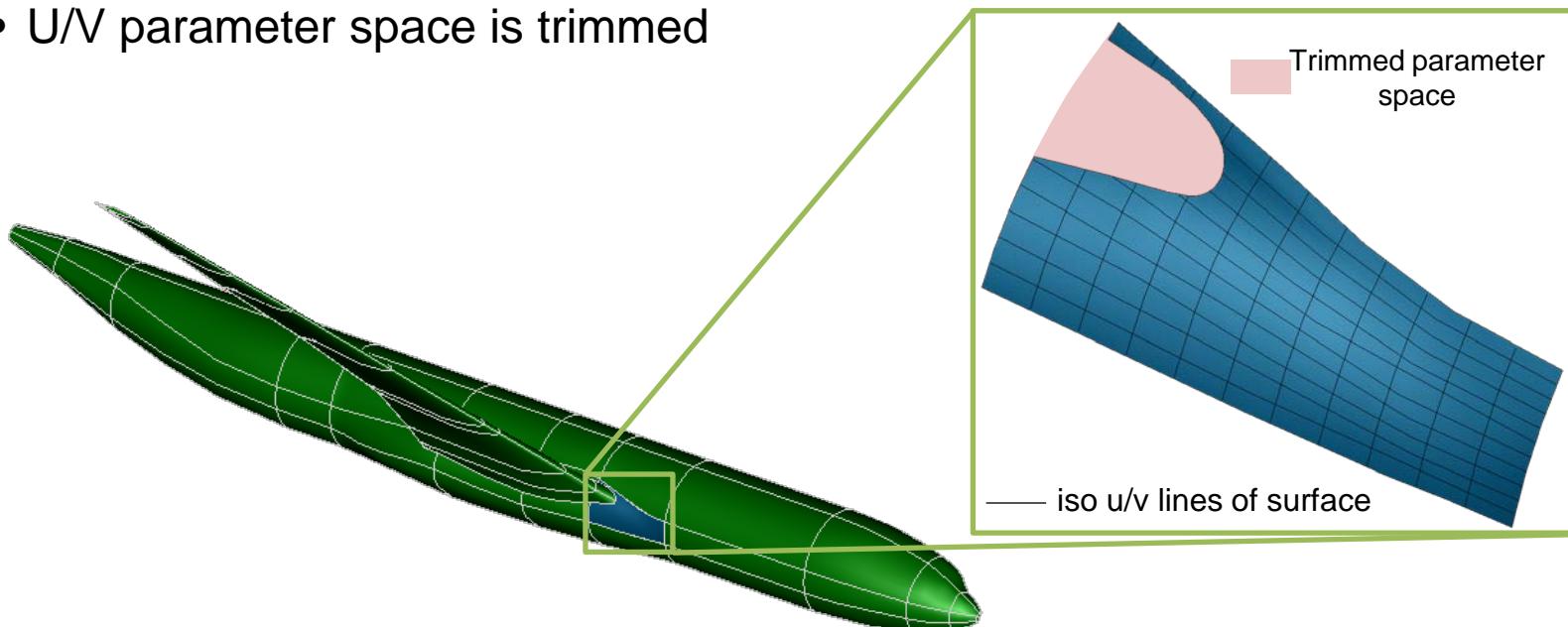
Geometry based mesh deformation

General idea



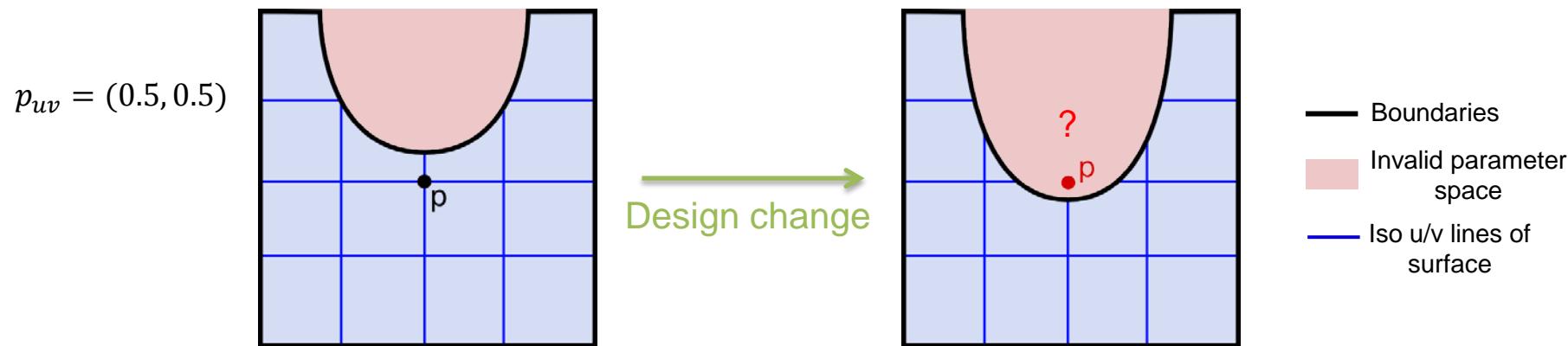
Trimmed NURBS based geometries

- Geometry consists of many panels
- Each panel is a trimmed NURBS
- U/V parameter space is trimmed



The trimmed NURBS parameterization problem

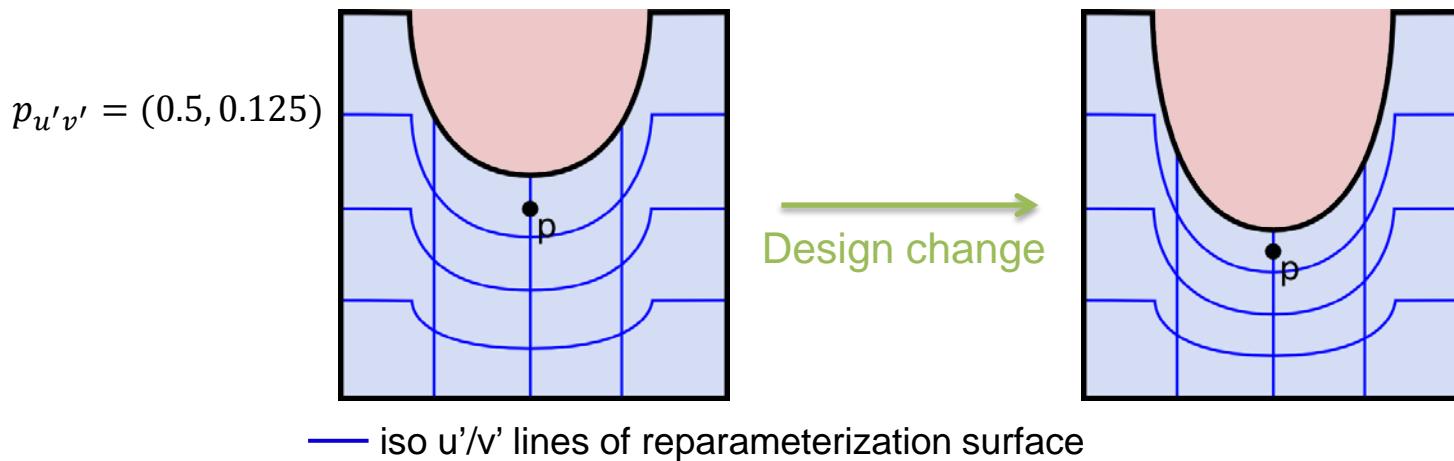
- Trimmed NURBS:
 - Parameter space of surface trimmed by boundary curves



- Mesh points get lost / are outside panel after design change
 - u/v surface coordinates no good choice for mapping

Solution: reparameterization of the parameter space

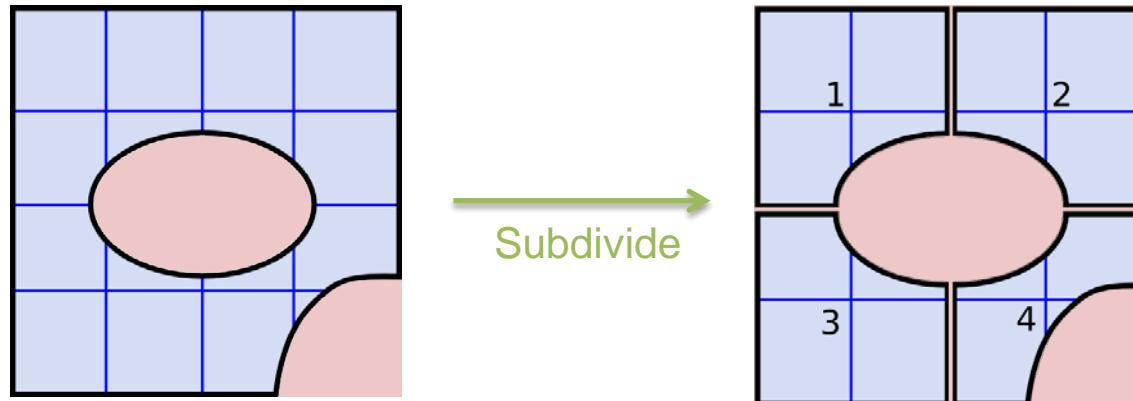
- Create 2D reparameterization surface that respects boundaries



- No loss of mesh points anymore
- Next problem: what to do with holes or complicated boundaries?

The trimmed NURBS parameterization problem

- Panels with complicated boundaries can not be reparameterized with one surface!



- Solution: Subdivision of the panel into multiple sub-surfaces
 - Reparameterize each sub-surface

Projection / Back projection

- **Projection:**

$$\vec{P} \in \mathbb{R}^3 \rightarrow i, j, u', v'$$

with:

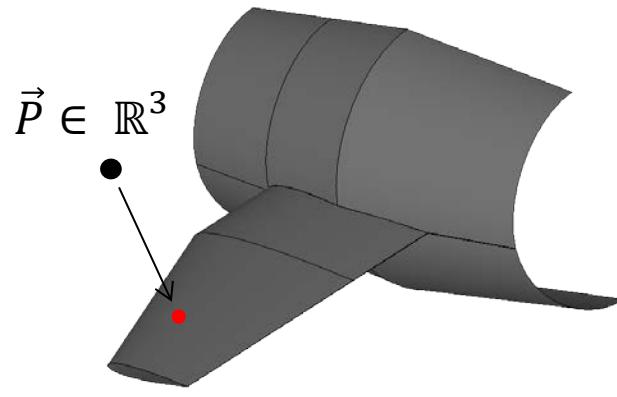
i : panel index

j : sub-surface index

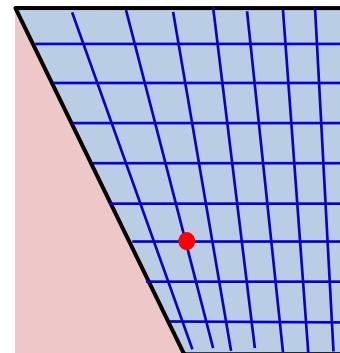
$u', v' \in [0,1]$

- **Back projection:**

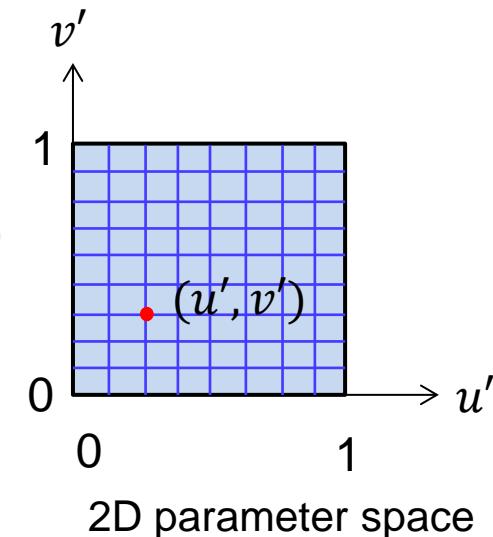
$$i, j, u', v' \rightarrow \vec{P} \in \mathbb{R}^3$$



3D



reparametrized 2D surface



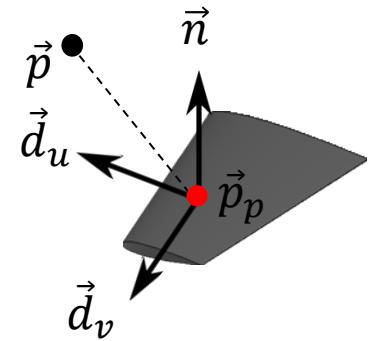
Improvement: Exact back projection

- Motivation:
 - Geometry changes are very small for finite differences
 - Mesh points may not exactly lie on the original geometry
- But: Invariant geometries should result in exactly the same mesh!

- Solution:
 1. Project point \vec{p} onto surface
 2. Create local coordinate system (CS) from \vec{d}_u , \vec{d}_v , $\vec{n} = \vec{d}_u \times \vec{d}_v$
 3. Store also the deviation of projected point \vec{p}_p in the local CS

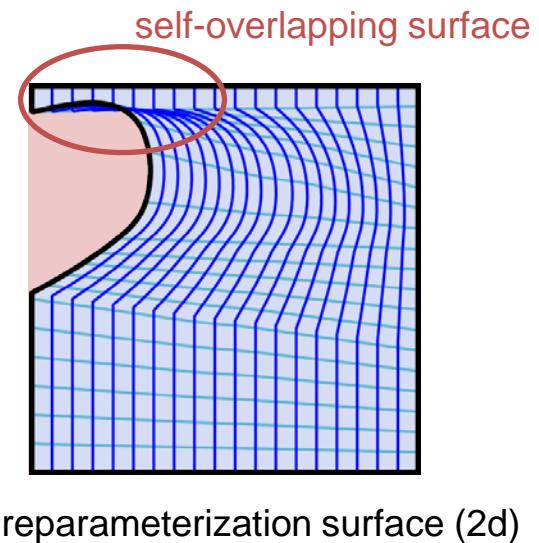
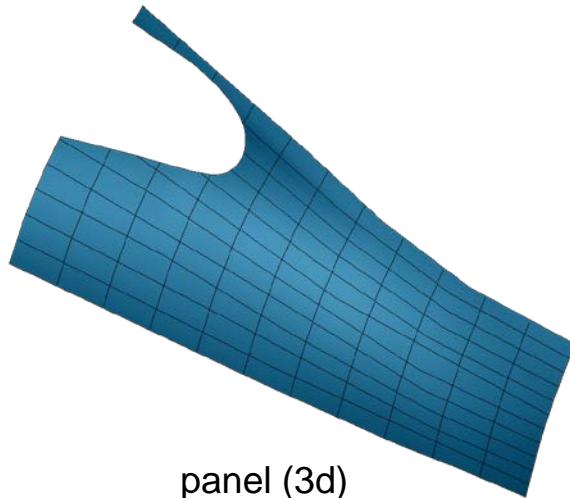
$$\vec{p} - \vec{p}_p = dn \cdot \vec{n} + du \cdot \vec{d}_u + dv \cdot \vec{d}_v$$

- (Back)Projection: $\vec{P} \in \mathbb{R}^3 \leftrightarrow i, j, u', v', dn, du, dv$



Uniqueness of projection

- Reparameterization surfaces can self-overlap



- Issue: Projection is not unique → multiple solutions !
- Resulting mesh might be invalid



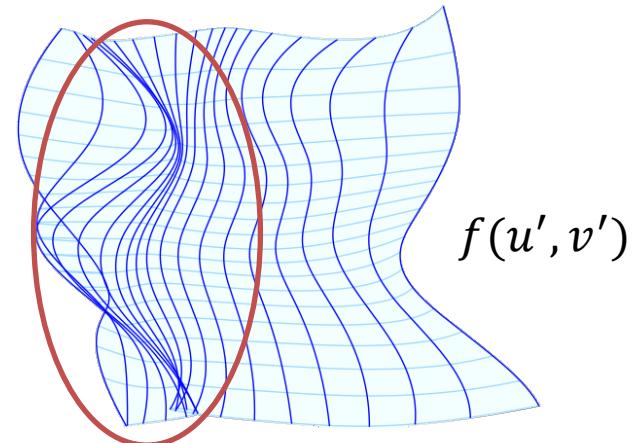
Uniqueness of projection

Invertibility of reparameterization surfaces

- Reparameterization surface $f(u', v') \rightarrow (u, v)$ must be invertible

➤ there must exist an inverse function f' , with

$$f'(u, v) \rightarrow (u', v'), u', v' \in [0,1], u, v \text{ inside trimmed parameter space}$$



- If invertible, projection is unique 
- How to check existence of $f'(u, v)$?



Uniqueness of projection

Invertibility criterion

- Jacobi determinant (i.e. Z-component of normal vector) must not be negative, i.e.

$$\det J(u', v') \stackrel{\text{def}}{=} \left[\frac{\partial}{\partial u'} f(u', v') \times \frac{\partial}{\partial v'} f(u', v') \right]_z \geq 0, \forall u', v' \in [0,1]$$

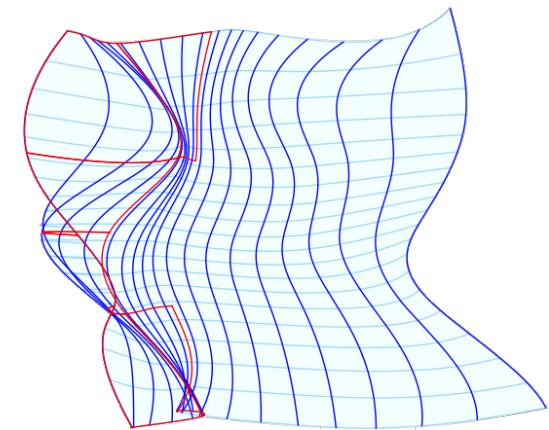
- Jacobi determinant of Bezier patch is 1d Bezier surface:

- control points T_{pq} can be computed according to *, eq. (11)
- Convex hull property: $T_{pq} > 0 \Rightarrow \det J(u', v') \geq 0$

➤ Split reparameterization surface f into Bezier patches

➤ Inspect control points of Bezier patches for positivity

- But: False positives possible!



Non-invertible Bezier patches

Implementation details

- Written in C++ as a library
- Library offers functions to
 - Perform projection and back-projection of points
 - Import CAD models (IGES, Step, BREP)
 - Check invertibility of each sub-surface
 - Distribute the geometry to each node of a cluster (Parallelization)
 - Compare the topology of two CAD models
- Uses the OpenCASCADE Technology* CAD library



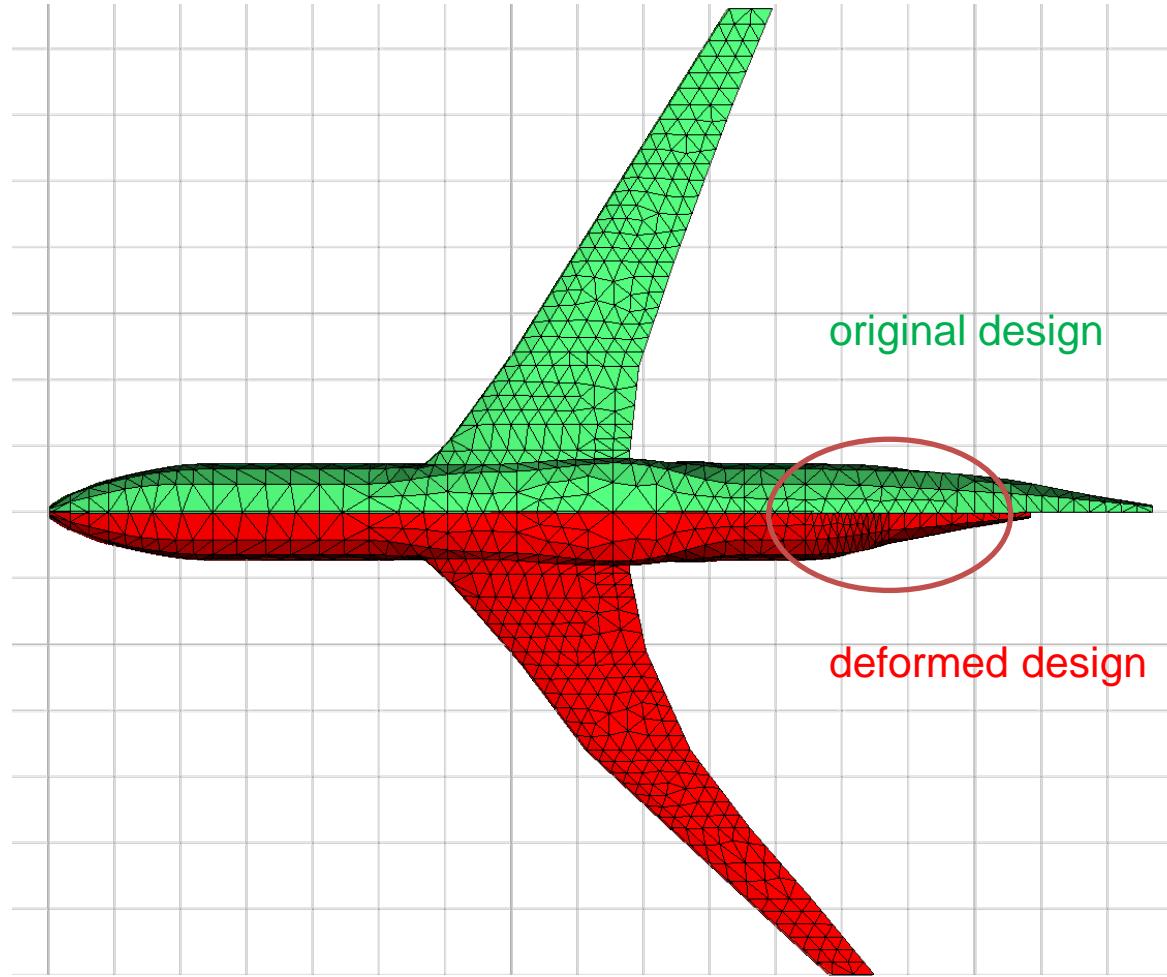
*<http://www.opencascade.com>

Results

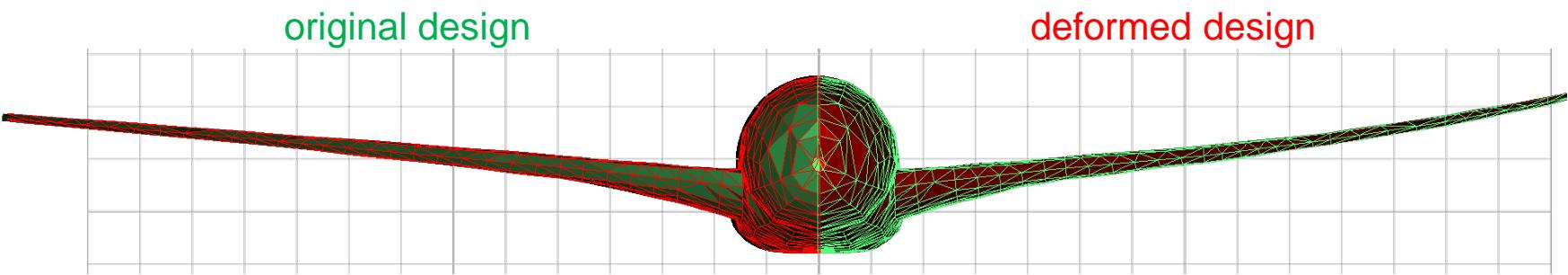
- Algorithm currently used at Airbus D&S for FEM based structural analysis
- Currently used in DLR for CFD simulations with structured meshes in the DLR internal project VicToria*:
 - Adjoint CFD solver TAU requires mesh gradients
(i.e. how does the mesh change with some design parameter change)
 - This method suitable to compute the gradients
- Method not suitable for large design changes:
 - Topology of panels and boundaries must not change!

*http://www.dlr.de/as/en/desktopdefault.aspx/tabcid-11460/20078_read-47033/

Results



Results



Summary + Outlook

- Presented a method to globally parameterize points on a trimmed NURBS based CAD geometry
 - Re-parameterization of surfaces
 - Surface splitting
 - Check for uniqueness of projection
- The method is already successfully used for large scale aircraft optimization
- In future:
 - Improve automatic splitting into sub-surfaces for different scenarios (hole(s), multiple boundaries ...)



Questions



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