

1 Impact of neutral boundary-layer turbulence on 2 wind-turbine wakes: A numerical modelling study

3 **Antonia Englberger · Andreas Dörnbrack**

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6 **Abstract** The wake characteristics of a wind turbine in a turbulent bound-
7 ary layer under neutral stratification are investigated systematically by means
8 of large-eddy simulations. A methodology to maintain the turbulence of the
9 background flow for simulations with open horizontal boundaries, without the
10 necessity of the permanent import of turbulence data from a precursor simula-
11 tion, was implemented in the geophysical flow solver EULAG. These require-
12 ments are fulfilled by applying the spectral energy distribution of a neutral
13 boundary layer in the wind-turbine simulations. A detailed analysis of the
14 wake response towards different turbulence levels of the background flow re-
15 sults in a more rapid recovery of the wake for a higher level of turbulence. A
16 modified version of the Rankine-Froude actuator disc model and the blade ele-
17 ment momentum method are tested as wind-turbine parametrizations resulting
18 in a strong dependence of the near-wake wind field on the parametrization,
19 whereas the far-wake flow is fairly insensitive to it. The wake characteristics
20 are influenced by the two considered airfoils in the blade element momentum
21 method up to a streamwise distance of $14D$ ($D =$ rotor diameter). In addition,
22 the swirl induced by the rotation has an impact on the velocity field of the
23 wind turbine even in the far wake. Further, a wake response study reveals a
24 considerable effect of different subgrid-scale closure models on the streamwise
25 turbulent intensity.

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27 **Keywords** Atmospheric boundary layer · Large-eddy simulation · Turbu-
28 lence · Wind-turbine wake

A. Englberger
Institut für Physik der Atmosphäre, DLR Oberpfaffenhofen
E-mail: antonia.englberger@dlr.de

A. Dörnbrack
Institut für Physik der Atmosphäre, DLR Oberpfaffenhofen
E-mail: andreas.doernbrack@dlr.de

1 Introduction

Wind turbines operate in the atmospheric boundary layer (ABL) where atmospheric turbulence arises from velocity shear (velocity change with height) and directional shear (wind direction change with height), thermal stratification, low-level moisture, as well as from the interaction of the airflow with vegetation, buildings or terrain (Naughton et al., 2011; Emeis, 2013, 2014). ABL turbulence affects the velocity deficit and the turbulence in the wake, having a large impact on energy production, on fatigue loading, and on the life expectancy of wind turbines. Numerical simulations of wind turbines in the ABL have become an important tool in the investigation of these complex processes. Different numerical approaches exist to simulate the impact of ABL turbulence on wind-turbine wakes. Here, we focus on a large-eddy simulation (LES), being an approved tool to study the turbulence in the ABL (Bellon and Stevens, 2012).

The influence of a turbulent flow on the structure of the wake has been investigated in experimental studies (Medici and Alfredsson, 2006; Chamorro and Porté-Agel, 2009; Zhang et al., 2012) as well as in numerical simulations (Troldborg et al., 2007; Wu and Porté-Agel, 2012). According to their investigations, the wake structure is strongly influenced by the presence of turbulence in the inflow and the wake recovers more rapidly for higher turbulence intensity levels of the incoming flow.

Different methods have been applied to generate a turbulent flow field upstream of the wind turbine. In wind-tunnel experiments, additional roughness elements in front of the wind turbine evoke a turbulent flow, which can be generated by turbulence grids (Medici and Alfredsson, 2006) or obstacles on the floor (Chamorro and Porté-Agel, 2009). Implementing this method in a numerical simulation requires a rather large upstream section, which is computationally expensive, leading to other approaches.

A simple synthetic method avoiding the simulation of atmospheric turbulence was proposed by Mann (1994), e.g. used in Troldborg et al. (2007). The resulting three-dimensional turbulence field is compact and provides turbulence spectra as expected in an ABL. This method, however, is not based on a physical model and only offers a synthetic turbulence field (Naughton et al., 2011). An alternative approach is to couple meteorological data (e.g. wind speed, wind direction, temperature) from a mesoscale simulation on the microscale LES of the wind turbine. However, the two-way coupling as well as the one-way coupling between mesoscale and microscale models, induces different problems (Mirocha et al., 2013; Muñoz-Esparza et al., 2014).

The necessity of synthetic or mesoscale atmospheric parameters can be avoided by the use of a precursor simulation. Wu and Porté-Agel (2012) created a neutral ABL flow forced by a streamwise pressure gradient. The main simulation is initialized with data from the precursor simulation. By applying streamwise periodic boundary conditions, a buffer zone prevented the turbulence in the wake from re-entering the domain and interacting with the wind turbine.

74 Open streamwise boundary conditions do not require a buffer zone. In-
75 stead, the wind-turbine simulation has to be fed continuously with turbulence
76 data from a precursor simulation to generate a fully developed turbulent flow
77 field. Naughton et al. (2011) ensured a turbulent inflow by prescribing instan-
78 taneous velocity components from the precursor simulation at the inflow plane
79 at regular time intervals. Witha et al. (2014) realized a turbulent inflow for
80 an array of wind turbines in a wind park based on a recycling method after
81 Kataoka and Mizuno (2002). The main simulation used the data from the pre-
82 cursor simulation for initialization and persistently extracted turbulence from
83 a region upstream of the wind turbine adding it to the mean inflow profiles.

84 The first goal of our study is to develop and investigate a new method-
85 ology to generate and maintain a realistic background turbulence field in the
86 wind-turbine LES with open horizontal boundary conditions, and by avoiding
87 a continuous turbulent inflow from a precursor simulation. At each timestep
88 of the wind-turbine LES, the flow field shall be perturbed by velocity fluctu-
89 ations extracted from a selected state of the precursor simulation of a neutral
90 ABL. The aim is to maintain the spectral properties of realistic background
91 turbulence and to control the energy of the applied perturbation fields. Here,
92 we describe the new methodology and compare our numerical results with
93 published results from previous simulations and measurements.

94 In addition to a realistic background turbulence field, an LES of wind-
95 turbine wakes require a detailed knowledge and parametrization of the forces
96 exerted by a wind turbine on the atmosphere. In a numerical model the wind-
97 turbine forces can be parametrized as a disc that can either rotate or not.
98 Alternatively, individual rotating lines represent the blades of the wind tur-
99 bine. The respective approaches are termed the actuator disc model (ADM)
100 and the actuator line model (ALM). The impact of wind-turbine parametriza-
101 tions on the wake has been studied focusing on various aspects.

102 Mikkelsen (2003) investigated the parametrization of a wind turbine with
103 the ADM and the ALM, extended for a multiplicity of rotor configurations,
104 e.g. a coned or a yawed rotor. Numerous investigations validating the different
105 wind-turbine parametrizations were performed by e.g. Ivanell et al. (2008),
106 Porté-Agel et al. (2010), Wu and Porté-Agel (2011) and Tossas and Leonardi
107 (2013). All of these studies resulted in a near-wake wind field, sensitive to the
108 wind-turbine parametrization, whereas the far-wake structure depends mainly
109 on the background turbulence. Mirocha et al. (2014) implemented the gener-
110 alized actuator disc wind-turbine parametrization into the Weather Research
111 and Forecasting (WRF-LES) model. This approach enabled the investigation
112 of the interaction of a wind turbine with different ABL stratifications, result-
113 ing in good agreement of the wake characteristics with observations under
114 weakly convective conditions. Numerous studies explored the impact of the
115 distribution of the forces. Ivanell et al. (2008) and Tossas and Leonardi (2013)
116 studied the impact of different smearing parameters of the forces acting on
117 the atmosphere, resulting in numerical instabilities for a tight volume-force
118 distribution at the rotor position. Ivanell et al. (2008), Wu and Porté-Agel
119 (2011) and Gomes et al. (2014) investigated the influence of the number of

120 grid points representing the disc on the wake structure with the result that
121 the wake characteristics are independent of the resolution, if a minimum of ten
122 grid points cover the rotor diameter in the spanwise and the vertical directions.
123 Gomes et al. (2014) also analyzed the effect of the radial dependencies of the
124 applied forces. A strong sensitivity of the near-wake wind field was found in
125 contrast to the far-wake behaviour.

126 Here, we apply a modified version of the classical Rankine-Froude ADM
127 and the blade element momentum (BEM) method for two different airfoils
128 as wind-turbine parametrizations in our numerical simulations. In the second
129 part, systematic investigations of the wake characteristics depending on the
130 two parametrizations, the local blade characteristics, and the rotation of the
131 disc are made.

132 We implement our turbulence preserving method and both wind-turbine
133 parametrizations in the multiscale geophysical flow solver EULAG (Prusa
134 et al., 2008). This LES model resolves all energy containing modes of the turbu-
135 lent transport and scales larger than the spatial resolution of the computational
136 grid. Only the turbulence of the smallest unresolved scales is parametrized
137 using a subgrid-scale (SGS) closure model. The sensitivity of the numerical
138 results towards different SGS closure models (turbulent kinetic energy (TKE)
139 closure, Smagorinsky closure) as well as an implicit LES (Grinstein et al.,
140 2007) constitute the third task investigated.

141 The outline of the paper is as follows: the LES model is presented in Sect. 2,
142 while the turbulence preserving method is formulated in Sect. 3, and the wind-
143 turbine models are described in Sect. 4. The results of the numerical simula-
144 tions studying the influence of the intensity of background turbulence, the
145 wind-turbine parametrizations, the rotation of the wind turbine and the SGS
146 closure models on the wake characteristics follow in Sect. 5. Conclusions are
147 given in Sect. 6.

148 **2 Numerical model framework**

149 An inviscid and incompressible flow through a wind turbine is simulated with
150 the multiscale geophysical flow solver EULAG (Prusa et al., 2008). The geo-
151 physical flow solver EULAG is at least second-order accurate in time and space
152 (Smolarkiewicz and Margolin, 1998) and is well suited for massively-parallel
153 computations (Prusa et al., 2008). It can be run parallel up to a domain de-
154 composition in three dimensions. A comprehensive description and discussion
155 of the geophysical flow solver EULAG can be found in Smolarkiewicz and
156 Margolin (1998) and Prusa et al. (2008).

157 For the numerical simulations conducted herein, the Boussinesq equations
158 for a flow with constant density $\rho_0 = 1.1 \text{ kg m}^{-3}$ are solved for the Cartesian
159 velocity components $\mathbf{v} = (u, v, w)$ and for the potential temperature pertur-

160 bations $\Theta' = \Theta - \Theta_0$ (Smolarkiewicz et al., 2007),

$$\frac{d\mathbf{v}}{dt} = -G\nabla\left(\frac{p'}{\rho_0}\right) + \mathbf{g}\frac{\Theta'}{\Theta_0} + \mathbf{v} + \mathbf{M} + \frac{\mathbf{F}}{\rho_0} \equiv \mathcal{R}^v, \quad (1)$$

$$\frac{d\Theta'}{dt} = \mathcal{H} \equiv \mathcal{R}^\Theta, \quad (2)$$

$$\nabla \cdot (\rho_0 \mathbf{v}) = 0, \quad (3)$$

161 where $\Theta_0 = 301$ K. In Eqs. 1, 2, and 3, d/dt , ∇ , and $\nabla \cdot$ represent the total
 162 derivative, the gradient and the divergence, respectively. The quantity p' rep-
 163 represents the pressure perturbation with respect to the environmental state and
 164 \mathbf{g} is the vector of acceleration due to gravity. The factor G represents geometric
 165 terms that result from the general, time-dependent coordinate transformation
 166 (Wedi and Smolarkiewicz, 2004; Smolarkiewicz and Prusa, 2005; Prusa et al.,
 167 2008; Kühnlein et al., 2012). The SGS terms \mathbf{v} and \mathcal{H} symbolise viscous dis-
 168 sipation of momentum and diffusion of heat, \mathbf{M} denotes the inertial forces
 169 of coordinate-dependent metric accelerations and \mathbf{F} additional external forces
 170 related to the parametrization of the wind turbine in the geophysical flow
 171 solver EULAG. The terms \mathcal{R}^v and \mathcal{R}^Θ summarize symbolically all forces in
 172 the corresponding equations.

173 The acronym EULAG refers to the ability of solving the equations of motions
 174 either in an Eulerian (flux form) (Smolarkiewicz and Margolin, 1993)
 175 or in a semi-Lagrangian (advective form) (Smolarkiewicz and Pudykiewicz,
 176 1992) mode, via

$$\psi^{\xi+1} = \text{LE} \left(\psi^\xi + \frac{1}{2} \Delta t \mathcal{R}^\psi |^\xi \right) + \frac{1}{2} \Delta t \mathcal{R}^\psi |^{\xi+1}, \quad (4)$$

177 where $\psi = (u, v, w, \Theta)$, ξ denotes the timestep and LE is the corresponding
 178 finite-difference operator (semi-Lagrangian/Eulerian). In general, the geophys-
 179 ical flow solver EULAG owes its versatility to a unique design that combines a
 180 rigorous theoretical formulation in generalized curvilinear coordinates (Smol-
 181 arkiewicz and Prusa, 2005) with non-oscillatory forward-in-time (NFT) dif-
 182 ferencing for fluids built on the multi-dimensional positive definite advection
 183 transport algorithm (MPDATA), which is based on the convexity of upwind
 184 advection (Smolarkiewicz and Margolin, 1998; Prusa et al., 2008) and a ro-
 185 bust, exact-projection type, elliptic Krylov solver (Prusa et al., 2008). The flow
 186 solver has been applied to a wide range of scales simulating various problems
 187 like turbulence (Smolarkiewicz and Prusa, 2002), flow past complex or mov-
 188 ing boundaries (Wedi and Smolarkiewicz, 2006; Kühnlein et al., 2012), gravity
 189 waves (Smolarkiewicz and Dörnbrack, 2008; Doyle et al., 2011) or even solar
 190 convection (Smolarkiewicz and Charbonneau, 2013). The turbulence closure
 191 in the geophysical flow solver EULAG can be described by a TKE model, a
 192 Smagorinsky model or an implicit LES, with no turbulence closure model due

193 to not considering the diffusion process. The implicit LES properties of nu-
 194 merical solvers based on MPDATA are documented in e.g. Margolin and Rider
 195 (2002), Margolin et al. (2002) and Margolin et al. (2006) for structured grids.
 196 A detailed description of an implicit LES is given in Grinstein et al. (2007).

197 3 Turbulence preserving method

198 The basic idea of our new methodology that preserves the background tur-
 199 bulence in an LES of a flow through a wind turbine is to extract velocity
 200 perturbations from a precursor simulation of the neutral ABL. The velocity
 201 fields are used to disturb the wind-turbine simulation in a special manner
 202 as described below. For this purpose, a precursor simulation of the turbulent
 203 neutral ABL has to be conducted.

204 3.1 Precursor simulation

205 To drive the neutral ABL flow, an additional forcing $-u_*^2/H$ is applied for the
 206 u -component of Eq. 1, where H is the height of the computational domain.
 207 Sensitivity tests revealed that a value of the friction velocity $u_* = 0.4 \text{ m s}^{-1}$
 208 results in a realistic pressure gradient of the ABL. This forcing is comparable
 209 to the streamwise mean pressure gradient force applied in Wu and Porté-Agel
 210 (2012). The precursor simulation is performed with the same number of grid
 211 points as the wind-turbine simulations, but with periodic boundary conditions
 212 in the horizontal directions. The initial wind speed is set to zero, and the drag
 213 coefficient in the surface parametrization is set to 0.1.

214 Applying only the above forcing, it is a long lasting process until the pre-
 215 cursor simulation is in an equilibrium state. Additional velocity gradients in
 216 the neutral flow can serve as a trigger, breaking the symmetry and acting as
 217 a seed for turbulence to develop. Therefore, the precursor simulation is dis-
 218 turbed by inserting an obstacle in the domain for a few timesteps. The flow
 219 around this obstacle enhances the velocity gradients in the neutral ABL flow,
 220 and the equilibrium state of the precursor simulation is attained more rapidly.

221 3.2 Methodology

222 The perturbation velocities $\mathbf{u}_p^*|_{i,j,k}^\xi$ are extracted from the precursor simulation
 223 according to,

$$223 \mathbf{u}_p^*|_{i,j,k}^\xi = \alpha \cdot \beta \cdot \underbrace{\left(\mathbf{u}_p|_{i^*,j,k} - \underbrace{\frac{1}{n \cdot m} \sum_{i=1}^n \sum_{j=1}^m \mathbf{u}_p|_{i,j,k}}_I \right)}_{II}, \quad (5)$$

224 where $\mathbf{u}_p|_{i^*,j,k}$ is the velocity vector of the precursor simulation in an equilib-
 225 rium state and the term I in Eq. 5 denotes the height-averaged mean value of
 226 the corresponding wind component at each grid point i , j , and k . The indices
 227 of the grid points are denoted by $i = 1 \dots n$, $j = 1 \dots m$, and $k = 1 \dots l$ in the
 228 x , y , and z directions, respectively.

229 The perturbation velocity from Eq. 5 contributes to the velocity field of
 230 the wind-turbine simulation $\mathbf{u}|_{i,j,k}^\xi$ at the initial timestep $\xi = 0$ and at each
 231 following timestep ξ . The values of the precursor simulation $\mathbf{u}_p|_{i^*,j,k}$ are shifted
 232 in the streamwise direction by one grid point every timestep ξ , symbolized by
 233 $i^* = i + \xi^*$, with $i^* \in [1, n]$ and ξ^* representing the number of timesteps since
 234 the start of the simulation. Furthermore, the difference as denoted by II in
 235 Eq. 5 is multiplied with a random number β ranging from -0.5 to 0.5 . Both
 236 the grid point shift and the random number multiplication are necessary to
 237 only apply the spectral energy distribution of the precursor simulation instead
 238 of impressing individual flow patterns onto the wind-turbine simulation. To
 239 account for different magnitudes of the background turbulence, the term II in
 240 Eq. 5 is additionally multiplied by a factor α , representing the amplitude of the
 241 turbulence perturbations (hereafter referred to as perturbation amplitude).

242 Applying this method maintains the spectral properties of the turbulent
 243 fluctuations in the wind-turbine simulation. It offers several possibilities for
 244 the numerical scheme:

- 245 1. Periodic boundary conditions and a buffer zone can be avoided, enabling
 246 open inflow and outflow Neumann boundary conditions and minimising
 247 the domain size of the simulation.
- 248 2. The perturbation data from the precursor simulation are imported only
 249 once and are stored in three 3D fields (u , v , w) during the wind-turbine
 250 simulation.
- 251 3. The method is computationally very efficient, as it allows to reapply the
 252 background turbulence of one precursor simulation to a variety of wind-
 253 turbine simulations.
- 254 4. The response of a wind turbine to different intensities of the background
 255 turbulence can be easily investigated by changing the parameter α in Eq. 5.

256 3.3 Validation of the turbulence preserving method

257 We performed a simulation applying term I from Eq. 5 as wind field. In ad-
 258 dition, the spectral energy distribution of the precursor simulation is applied
 259 with the prescribed methodology. After integrating for the same amount of
 260 time as in the following wind-turbine simulations, this simulation resulted in
 261 the same values of $\langle u \rangle_t$, $\langle v \rangle_t$ and $\langle w \rangle_t$, as well as σ_u , σ_v and σ_w
 262 with $\sigma_i = \sqrt{i'^2}$ as the precursor simulation, validating the mechanism of the
 263 turbulence preserving method.

4 Wind-turbine Parametrization

4.1 Parametrization of the forces

The classical Rankine-Froude theory is the simplest ADM representation of turbine-induced forces in a numerical model where the disc covers the span of the blades. It was introduced by Froude (1889) who continued the work of Rankine (1865) on the momentum theory of propellers. The forces induced by a wind turbine are basically parametrized as a 1D thrust force, which is constant over the disc. Despite its simplicity, this non-rotating ADM has been widely used in LES as it provides reliable results on coarse grids (Calaf et al., 2010; Porté-Agel et al., 2010; Wu and Porté-Agel, 2011; Tossas and Leonardi, 2013; Meyers and Meneveau, 2013). A wind turbine rotates and the incoming profiles of the horizontal wind speed are often vertically sheared ($\partial \mathbf{u} / \partial z \neq 0$). Both processes limit the applicability of the simple ADM parametrization. To circumvent these limitations and to enable an investigation of the impact of the local blade characteristics by comparing to the results of the BEM parametrization (Manwell et al., 2002; Hansen, 2008), we apply a modified version of the Rankine-Froude ADM considering the axial force $F_x(y, z)$ in the streamwise (x) direction and the tangential force $F_\Theta(y, z)$ perpendicular to F_x in the y - z plane,

$$|F_x|_{x_0, y, z} = \frac{1}{2} \rho_0 c'_T A_{x_0, y, z} \langle u_{x_0, y, z}^2 \rangle_t, \quad (6)$$

$$|F_\Theta|_{x_0, y, z} = \frac{1}{2} \rho_0 c'_P A_{x_0, y, z} \langle u_{x_0, y, z}^2 \rangle_t \frac{u_{x_0, y, z}}{\Omega r_{x_0, y, z}}. \quad (7)$$

Both forces F_x and F_Θ result in the total force $\mathbf{F}|_{x_0, y, z}$ (Hansen, 2008), with

$$\mathbf{F}|_{x_0, y, z} = \mathbf{F}_x|_{x_0, y, z} + \mathbf{F}_\Theta|_{x_0, y, z}, \quad (8)$$

where the centre of the rotor is defined by the grid-point coordinates x_0 , y_0 and z_h (hub height). In Eqs. 6 and 7, c_T represents the thrust coefficient ($c'_T = c_T / (1 - a)^2$) and c_P the power coefficient ($c'_P = c_P / (1 - a)^3$). The factor a corresponds to the axial induction factor and can be derived from the one-dimensional momentum theory to a value of $1/3$ for an ideal rotor (Betz, 1926). $A_{x_0, y, z}$ is the area of the rotor at position x_0 covered by grid points in the y - z plane, Ω is the angular velocity of the turbine and $r_{x_0, y, z}$ the radial position inside the rotor ($0 \leq r_{x_0, y, z} \leq R$), with $R = D/2$ and D representing the diameter of the wind-turbine rotor. The time-averaged value of the squared streamwise velocity component at the rotor position x_0 , y , z is denoted by $\langle u_{x_0, y, z}^2 \rangle_t$.

A great improvement of the simple momentum theory was the classical BEM method by Glauert (1963). This method accounts for local blade characteristics, as it enables calculation of the steady loads as well as the thrust and the power for different wind speeds, rotational speeds, and pitch angles of

Table 1 The crucial characteristics of the three different parametrizations A, B, and C of the wind turbine used in this study.

Parametrization	A	B + C
Name	MMT	BEM
Characteristics	Eq. 6 Eq. 7	Eq. 9 Eq. 10

299 the blades. The axial and tangential forces of the BEM method are represented
300 in Eqs. 9 and 10,

$$|F_x|_{x_0,y,z} = \frac{1}{2} \rho_0 \frac{Bc}{2\pi r_{x_0,y,z}} (c_L \cos \Phi + c_D \sin \Phi) \times A_{x_0,y,z} \frac{u_{x_\infty,y,z}^2 (1-a)^2}{\sin^2 \Phi} \quad (9)$$

$$|F_\theta|_{x_0,y,z} = \frac{1}{2} \rho_0 \frac{Bc}{2\pi r_{x_0,y,z}} (c_L \sin \Phi - c_D \cos \Phi) \times A_{x_0,y,z} \frac{u_{x_\infty,y,z} (1-a) \Omega r_{x_0,y,z} (1+a')}{\sin \Phi \cos \Phi}. \quad (10)$$

301 Here, B represents the number of blades, c is the chord length of the blade,
302 c_L is the lift coefficient, c_D is the drag coefficient, Φ is the angle between the
303 plane of rotation and the relative streamwise velocity, and a' is the tangential
304 induction factor. Following Hansen (2008), we calculate a and a' by an iterative
305 procedure from the airfoil data. The upstream velocity $u_{x_\infty,y,z}$ is taken at the
306 first upstream grid point in the x -direction and the corresponding y and z
307 coordinates. With the exception of ρ_0 and B , all other parameters appearing
308 in Eqs. 9 and 10 depend on the radius $r_{x_0,y,z}$ and vary spatially.

309 In this work, the modified version of the Rankine-Froude ADM as well
310 as the BEM parametrization are implemented via Eq. 8 in the geophysical
311 flow solver EULAG. The forces are treated implicitly in the numerical scheme
312 according to Eq. 4. In the geophysical flow solver EULAG, the rotor of a wind
313 turbine is not implemented as a real circular obstacle (e.g. grid-point blocking
314 as in Heimann et al. (2011)) or a permeable rotor (Witha et al., 2014; Tossas
315 and Leonardi, 2013; Gomes et al., 2014). Instead, at every grid point covered by
316 the rotor, the velocity field experiences the turbine-induced force \mathbf{F} according
317 to Eq. 1. This implementation is inspired by the immersed boundary method,
318 successfully applied in the geophysical flow solver EULAG by Smolarkiewicz
319 and Winter (2010). The implicit treatment of the forces in Eq. 4 has a positive
320 effect on the timestep, because there are no large velocity gradients between
321 the rotor area and its surroundings.

322 Altogether, three different parametrizations of wind-turbine induced forces
323 are implemented in the geophysical flow solver EULAG. The respective param-
324 etrizations A, B, and C are listed together with their main characteristics in
325 Table 1. It should be noted that the parametrizations B and C are essentially
326 the same, however, the airfoil data applied in B and C differ. The radial distri-

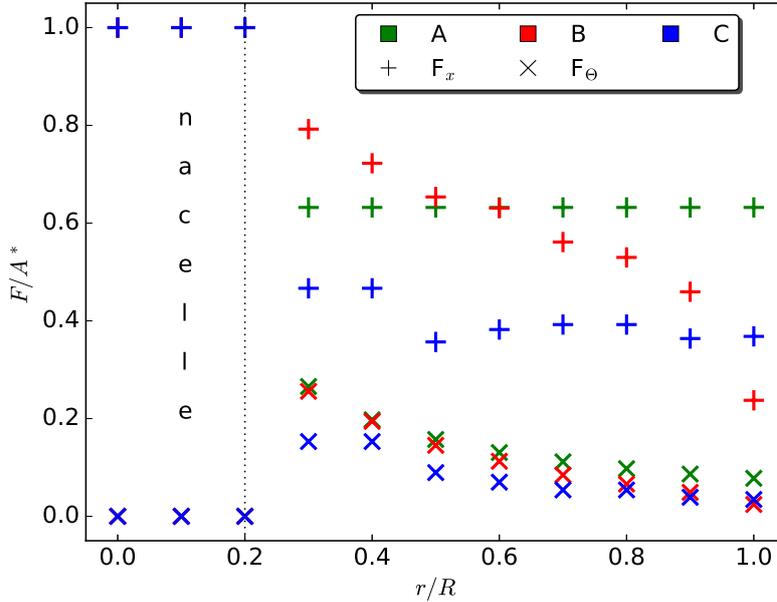


Fig. 1 Radial distributions of the axial and tangential forces F_x and F_Θ normalized by the area A^* for the different wind-turbine parametrizations A, B, and C of Table 1. The values of F_x and F_Θ are normalized by the maximum of the axial force at the nacelle, which is the same in all three parametrizations. The axial forces are represented by (+) and the tangential forces by (\times). They are plotted for each discrete position of the rotor, assuming 21 grid points are covering the rotor with radius R . The nacelle covers 20 % of the blades, denoted by the dotted vertical line. For the calculation of the forces in these schematic illustration, a rotor diameter of 100 m is assumed, together with a rotation frequency $\Omega = 7$ r.p.m. and a constant upstream velocity $u_{x_\infty, y, z} = 8 \text{ m s}^{-1}$.

327 butions of the respective axial and tangential forces are depicted in Fig. 1. In
 328 each parametrization, a nacelle is represented within $r/R \leq 0.2$ by a stronger
 329 drag force in comparison to the blade values and no lift force. The size of the
 330 parametrized nacelle is large compared to a real wind turbine, because the
 331 numerical resolution demands enough grid points representing the nacelle to
 332 avoid instabilities. The tower is not considered in our parametrizations as it
 333 is not the major source of turbulence.

334 Parametrization A represents the modified version of the Rankine-Froude
 335 ADM, hereafter referred to as modified momentum theory (MMT). It can be
 336 applied for a rotating actuator with $F_\Theta \neq 0$ or for a non-rotating actuator
 337 with $F_\Theta = 0$. Parametrization A can be regarded as a simplified version of
 338 parametrization B, as the values of $c'_{T_{blade}} = 1.27$ and $c'_{P_{blade}} = 0.87$ in Eqs.
 339 6 and 7 are deduced from parametrization B. These prescribed values are
 340 comparable to other studies (Meyers and Meneveau, 2013).

341 The BEM method is used to investigate the influence of the blade structure.
 342 The airfoil data are taken from two different wind turbines. The 10 MW refer-
 343 ence wind turbine from DTU (Technical University of Denmark) referred to as
 344 parametrization B (Mark Zagar (Vestas), personal communication, 2015) and
 345 the three-blade GWS/EP-6030x3 rotor (Wu and Porté-Agel, 2011) referred to
 346 as parametrization C. For both wind turbines, the rotor radius as well as the
 347 chord length of the blades are scaled to a rotor diameter of 100 m, to make the
 348 results comparable to each other. The most relevant wind-turbine parameters
 349 used for parametrizations B and C are listed in the Appendix.

350 For the nacelle, $c'_{T_{nacelle}} = 1.48$ and $c'_{P_{nacelle}} = 0$ are chosen in all three
 351 parametrizations. The value of the drag coefficient of the nacelle of 1.0 agrees
 352 with the drag coefficient interval of cylindrically shaped bluff bodies between
 353 0.8 and 1.2 (Schetz and Fuhs, 1996), and has also been used e.g. in El Kasmi
 354 and Masson (2008).

355 4.2 Application of the forces

356 The numerical simulations conducted in this study are performed on an equidis-
 357 tant Cartesian mesh with grid spacings Δx , Δy and Δz , in the stream-
 358 wise, lateral and vertical directions, respectively. It must be noted, that all
 359 parametrizations A, B, and C are coded to perform properly in terrain-following
 360 coordinates with variable vertical grid spacings over hilly terrain.

361 To calculate the forces of the actuator, we use polar coordinates that serve
 362 as a local mesh. The centre coordinate of the polar mesh is the centre of
 363 the rotor. From this position, the polar mesh is described by a very fine grid
 364 with $\Delta r = R/1000$ as radial step size and $\Delta\varphi = 1^\circ$ as azimuthal step size. The
 365 step sizes in the radial and azimuthal directions are fine enough to minimize
 366 the errors that would result from calculating the forces on a Cartesian mesh
 367 (Ivanell et al., 2008). The computational costs arising from such a fine polar
 368 mesh are insignificant, as the disc is always at the same position, making this
 369 calculation of the actuator force in polar coordinates $F_{r,\theta,z}$ only necessary
 370 once.

371 The force acting on each polar grid point $F_{r,\theta,z}$ is transformed to the
 372 corresponding force in Cartesian coordinates $F_{x,y,z}^* = \mathcal{M}_{x,y,z} \cdot F_{r,\theta,z}$ through
 373 the transformation matrix $\mathcal{M}_{x,y,z}$. The force $F_{x,y,z}^*$ contributes to a certain
 374 fraction $\mu \in [0, 1]$ to the actuator force $F_{x,y,z} = \mu \cdot F_{x,y,z}^*$. The fraction μ is
 375 determined by the ratio of the grid-cell volume of the polar coordinate and
 376 the corresponding Cartesian coordinate, i.e. $\mu = 1$ if the Cartesian grid point
 377 is completely covered by the rotor and $\mu = 0$ in case of a rotor-free grid point.
 378 At the edge of the rotor, the fraction $\mu < 1$, because the Cartesian grid cell is
 379 not completely covered by the local polar mesh representing the rotor.

380 A smearing of the turbine-induced forces in the axial as well as in the
 381 radial direction is necessary to avoid numerical instabilities. As a first step,
 382 the forces from Eq. 8 are additionally distributed in the streamwise direction.
 383 This approach is performed for all parametrizations. The forces in Eq. 8 are

384 smeared with a 1D Gaussian function in the x -direction,

$$F_{s_x} = \frac{1}{\sqrt{\pi}\sigma} \exp\left(-\frac{(x-x_0)^2}{\sigma^2}\right). \quad (11)$$

385 Similar to other studies (Meyers and Meneveau, 2013), the value of σ is set to
386 1.5 and is given in absolute values of the radius.

387 In parametrization A, the axial force F_x in the y - z plane only varies with
388 the incoming velocity across the rotor. A moderate velocity gradient results in
389 very similar F_x values and generates large gradients at the edges of the rotor.
390 An additional two-dimensional smearing $F_{s_{y-z}}$ in the y - z plane is introduced
391 to avoid too sharp radial gradients in the turbine-induced forces between the
392 rotor area and the immediate surroundings. The forces of the schematic illustra-
393 tion in Fig. 1 decrease with a step function over the last three grid points
394 $\in [0.8r/R, 1.0r/R]$. The force at each of these outer region grid points is half
395 of the force of the corresponding nearest inner neighbour grid point. $F_{s_{y-z}}$
396 is not applied for the forces in the BEM method, as the parameters in Eqs. 9
397 and 10 already decrease with increasing r .

398 The values of the smearing parameters and of the step function applied
399 on the forces in the y - z plane in parametrization A are chosen in such a way
400 that the integrated force distributed in three dimensions is the same as in the
401 two-dimensional case without smearing. By combining the smearing in the x -
402 direction F_{s_x} and the smearing in the y - z plane $F_{s_{y-z}}$, the difference of the
403 forcings between a 2D and a 3D disc is less than 1% for 21 grid points per
404 disc and decreases for a finer resolution.

405 The parametrization $\mathbf{F}|_{x_0,y,z}$ (Eq. 8) together with the coordinate trans-
406 formation $F_{x,y,z}$ and the applied smearing in the axial F_{s_x} and radial $F_{s_{y-z}}$
407 directions result in a total parametrized force,

$$\mathbf{F}|_{x,y,z} = \mathbf{F}|_{x_0,y,z} \cdot F_{x,y,z} \cdot F_{s_x} \cdot F_{s_{y-z}}, \quad (12)$$

408 where the wind-turbine induced force $\mathbf{F}|_{x,y,z}$ corresponds to the force \mathbf{F} in Eq.
409 1.

410 4.3 Validation of the wind-turbine parametrization

411 We validate our numerical results for the wind-turbine parametrizations A,
412 B, and C at the rotor position (x_0, y, z) and in the wake (x_w, y, z) , whereby
413 $x_w \geq x_0$, with theoretical wind predictions from the one-dimensional momen-
414 tum theory,

$$u_{x_0,y,z} = u_{x_\infty,y,z}(1-a), \quad (13)$$

$$u_{x_w,y,z} = u_{x_\infty,y,z}(1-2a), \quad (14)$$

415 where a is the axial induction factor defined as

$$a := \frac{u_{x_\infty,y,z} - u_{x_0,y,z}}{u_{x_\infty,y,z}}. \quad (15)$$

Table 2 Parameters for the two different wind turbines (wind turbine 1 in laminar flow and wind turbine 2 in the turbulent ABL) with the rotor diameter D , the hub height z_h , the spatial resolution Δ , the rotation frequency of the blades Ω in revolutions per minute (r.p.m.), the location of the centre of the rotor in the simulated domain, as well as the velocity at the hub height u_{x,y,z_h} of the wind turbine and the vertical profile of the incoming velocity $u_{x_\infty,y,z}$. In the prescribed logarithmic wind profile, u_* represents the friction velocity, κ is the von Karman constant ($\kappa=0.4$), and z_0 is the roughness length. All simulations are performed on an equidistant grid with the spacing $\Delta = \Delta x = \Delta y = \Delta z$.

parameters	wind turbine 1	wind turbine 2
grid points	512x128x128	512x64x64
D (m)	4	100
z_h (m)	4	100
Δ (m)	0.1	5
Ω (r.p.m.)	0	7
rotor centre	$x_0 = 120\Delta$ $y_0 = 64\Delta$ $z_h = 40\Delta$	$x_0 = 60\Delta$ $y_0 = 32\Delta$ $z_h = 20\Delta$
u_{x,y,z_h} (m s ⁻¹)	0.08 and 0.10	8.0
$u_{x_\infty,y,z}$	constant wind profile $\frac{du}{dz} = 0$	logarithmic wind profile $u_{x_\infty,y,z} = \frac{u_*}{\kappa} \ln\left(\frac{z}{z_0}\right)$ $u_* = 0.45 \text{ m s}^{-1}$; $z_0 = 0.1 \text{ m}$

Table 3 Theoretically predicted velocities for different axial induction factors a at the rotor position $u_{x_0,y,z}$ and in the wake $u_{x_w,y,z}$ scaled with the upstream velocity $u_{x_\infty,y,z}$ according to Eqs. 13 and 14 and the deviations obtained from the numerical simulations. The deviations are calculated as an average over the disc area.

a	$\frac{u_{x_0,y,z}}{u_{x_\infty,y,z}}$ expected	$\frac{u_{x_0,y,z}}{u_{x_\infty,y,z}}$ deviation	$\frac{u_{x_w,y,z}}{u_{x_\infty,y,z}}$ expected	$\frac{u_{x_w,y,z}}{u_{x_\infty,y,z}}$ deviation
1/3	0.67	2 %	0.33	5 %
1/4	0.75	0 %	0.50	0 %
1/5	0.80	2 %	0.60	4 %

416 Equation 13 follows directly from Eq. 15, and Eq. 14 can be derived from
 417 the Bernoulli equation and Newton's second law of motion (Hansen, 2008).
 418 This comparison is strictly applicable only for laminar and uniform inflow
 419 conditions $u_{x_\infty,y,z}$.

420 Numerical simulations with the set-up as listed in Table 2 for wind turbine
 421 1 are performed with different axial induction factors $a=1/3, 1/4, 1/5$ for
 422 all parametrizations. Exemplary, the results for parametrization A, a non-
 423 rotating disc and $u_{x_\infty,y,z} = 0.08 \text{ m s}^{-1}$ are listed in Table 3. The results for
 424 parametrizations B and C and for $u_{x_\infty,y,z} = 0.10 \text{ m s}^{-1}$ are quantitatively
 425 similar and therefore not shown here.

426 The simulated ratios of $u_{x_0,y,z}/u_{x_\infty,y,z}$ and $u_{x_w,y,z}/u_{x_\infty,y,z}$ for a realistic
 427 value of the axial induction factor of 1/4 are in complete agreement with the
 428 one-dimensional momentum theory. For larger ($a=1/3$) and smaller ($a=1/5$)
 429 a values, the simulation results deviate by less than 5 % from the theoretical
 430 predictions.

Table 4 List of all performed simulations with information of the perturbation amplitude, the type of the wind-turbine parametrization, the tangential force and the SGS closure model used in the LES model.

simulation	perturbation amplitude α	wind turbine parametrization	tangential force F_Θ	SGS closure model
B_1	1	B	$\neq 0$	TKE
B_5	5	B	$\neq 0$	TKE
B_10	10	B	$\neq 0$	TKE
A_1	1	A	$\neq 0$	TKE
C_1	1	C	$\neq 0$	TKE
A_NR	1	A	$= 0$	TKE
B_S	1	B	$\neq 0$	Smagorinsky
B_I	1	B	$\neq 0$	no (implicit LES)

431 Summarizing, we successfully validated our LES model EULAG for the
 432 non-rotating disc of parametrization A and realistic values of the axial induc-
 433 tion factor against the one-dimensional momentum theory.

434 5 Numerical Experiments and Results

435 In this section, a detailed investigation of the reference simulation B_1 (base
 436 case) with $\alpha = 1$ and wind turbine 2 (Table 2) is given to confirm the applica-
 437 tion of the turbulence preserving model in a wind-turbine simulation. Details
 438 of the simulation set-up are listed in Table 2. Further, the dependence of the
 439 wake characteristics of the reference simulation B_1 are investigated regarding
 440 the impact of,

- 441 a , the perturbation amplitude
- 442 b , the wind-turbine parametrization
- 443 c , the rotation of the disc
- 444 d , the SGS closure model.

445 The corresponding parameters of B_1 and of all other simulations are listed in
 446 Table 4.

447 All simulations are performed for 60 min, a period long enough for the
 448 wake to reach an equilibrium state with statistical convergence of the results.
 449 All mean values are averaged over the last 50 min. The temporal average
 450 $\langle \Psi_{x,y,z} \rangle_t$ of a quantity Ψ for a time period t is calculated online in the
 451 numerical model and updated at every timestep according to the method of
 452 Fröhlich (2006, Eq. 9.1). In the following numerical simulations, the rotor
 453 covers 21 grid points. This leads to a high enough resolution according to
 454 investigations of Ivanell et al. (2008), Wu and Porté-Agel (2012) or Gomes
 455 et al. (2014) to avoid any dependence of the wake on the resolution. Generally,
 456 the numerical simulation results are plotted in dimensionless coordinates as a
 457 function of the rotor diameter D . The contour of the actuator in the cross-
 458 sections represents the transition to a force of zero. Furthermore, only a sector
 459 of the complete computational domain is shown in most of the following plots.

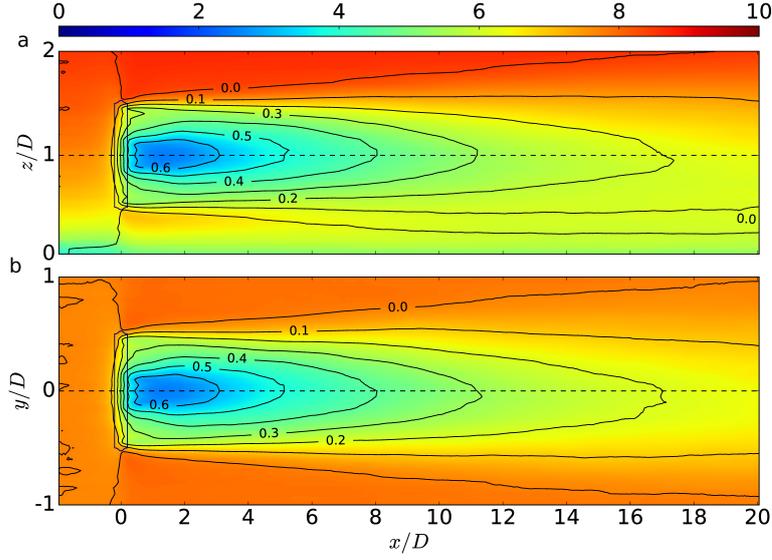


Fig. 2 Streamwise wind field in a vertical x - z cross-section at y_0 in (a) and in a horizontal x - y cross-section at z_h in (b). The contours represent the velocity deficit $(u_{\infty,y_0,k} - u_{i,y_0,k})/u_{\infty,y_0,k}$ in (a) and $(u_{\infty,j,z_h} - u_{i,j,z_h})/u_{\infty,j,z_h}$ in (b). Note, that in these cross-sections, the scale in the z or y -direction is exaggerated compared to the horizontal scale the in x -direction.

460 Now, we investigate the following characteristics of the wake of a wind
461 turbine:

- 462 – The spatial distribution of the velocities u , v and w .
- 463 – The streamwise velocity ratio

$$VR_{x,y,z} = \frac{\langle u_{x,y_0,z_h} \rangle_t}{\langle u_{x_\infty,y_0,z_h} \rangle_t}, \quad (16)$$

464 as it is related to the power loss of a wind turbine.

- 465 – The streamwise turbulent intensity

$$I_{x,y,z} = \frac{\sigma_{u_{x,y,z}}}{\langle u_{x,y,z_h} \rangle_t}, \quad (17)$$

466 with $\sigma_{u_{x,y,z}} = \sqrt{\langle u_{x,y,z}'^2 \rangle_t}$ and $u_{x,y,z}' = u_{x,y,z} - \langle u_{x,y,z} \rangle_t$, as it affects
467 the flow-induced dynamic loads on downwind turbines.

468 5.1 Reference simulation B.1

469 Figure 2 shows the vertical (Fig. 2a) and horizontal (Fig. 2b) cross-sections of
470 the streamwise wind field of simulation B.1. The general wake structure reveals

471 a minimum of the velocity right behind the rotor with a velocity increase in the
 472 radial and streamwise directions. This pattern results from the entrainment of
 473 surrounding air with higher velocity values, it is observed prevalently in field
 474 experiments in the atmosphere (Heimann et al., 2011, Fig. 3) or in wind-tunnel
 475 measurements (Zhang et al., 2012, Fig. 4) as well as simulated numerically
 476 (Porté-Agel et al., 2010, Fig. 5; Wu and Porté-Agel, 2012, Fig. 3; Aitken et al.,
 477 2014, Fig. 5; Mirocha et al., 2014, Fig. 5).

478 The x - y cross-section of u shows a nearly axisymmetric distribution (Fig. 2b),
 479 whereas the x - z cross-section of u displays a non-axisymmetric mean velocity
 480 profile (Fig. 2a) as a consequence of the vertically sheared upstream wind
 481 profile and the effect of the surface. Another feature in the x - z cross-section
 482 (Fig. 2a) represents the region of higher velocity air at the lowest part of
 483 the rotor in comparison to the surroundings. The velocity deficit plotted as
 484 contour lines in Fig. 2 enables a comparison with lidar measurements (Iungo
 485 et al., 2013; Käsler et al., 2010) or with remotely piloted aircraft measurements
 486 (Wildmann et al., 2014). These measurements for similar sized turbines and
 487 wind speeds result in a wind speed deficit of about 50 to 60 % at $x = 4D$, which
 488 is in line with the contours of the reference simulation in Fig. 2.

489 In Fig. 3, the mean values of u , v and w are plotted in a y - z cross-sections
 490 for selected downstream positions at $x = 3D$, $x = 5D$ and $x = 10D$. With in-
 491 creasing streamwise distance from the rotor, the flow field u recovers and starts
 492 to converge towards the upstream wind profile. The general structure of the
 493 position of the velocity minimum as well as the recovery of the wind field
 494 is comparable to published results (e.g., Wu and Porté-Agel, 2012, Fig. 4;
 495 Mirocha et al., 2014, Fig. 4). Depending on the implementation of a nacelle,
 496 the flow field directly behind the centre of the wind turbine changes. Among
 497 others, Wu and Porté-Agel (2011) and Meyers and Meneveau (2013) include
 498 the nacelle, whereas it is neglected in Aitken et al. (2014) and Mirocha et al.
 499 (2014). The slices of the lateral wind component v reveal a maximum at the
 500 upper rotor part and a minimum at the lower part, which corresponds to the
 501 vertical velocity field w with a maximum for $y/D \in [-1, 0]$ and a minimum for
 502 $y/D \in [0, 1]$. The intensity of this rotational effect decreases with increasing
 503 streamwise distance from the rotor. The regions with the maximum swirl of
 504 the flow are veering away from the rotor centre for an increasing downstream
 505 distance. The pattern in v and w is comparable to Mirocha et al. (2014, Fig. 4).
 506 In contrast to our results, the y - z cross-sections in Mirocha et al. (2014) are
 507 asymmetric, which is most likely induced by the weakly convective ABL in
 508 their simulations.

509 In Fig. 4, the temporally averaged velocity component in streamwise di-
 510 rection $\langle u_{x,y,z} \rangle_t$ is plotted as a function of streamwise distance for different
 511 positions (top, bottom, right ($y/D \in [0, 1]$), left ($y/D \in [-1, 0]$)) 60 m away
 512 from the centre of the rotor. These positions, although located outside of the
 513 actuator ($R = 50$ m), are still close enough to represent the effect of the forces
 514 resulting from Eq. 8 on the flow field. In the upstream region, the velocities
 515 at the top and the bottom locations differ due to the incoming logarithmic
 516 wind profile whereas the wind speeds right and left of the rotor are the same.

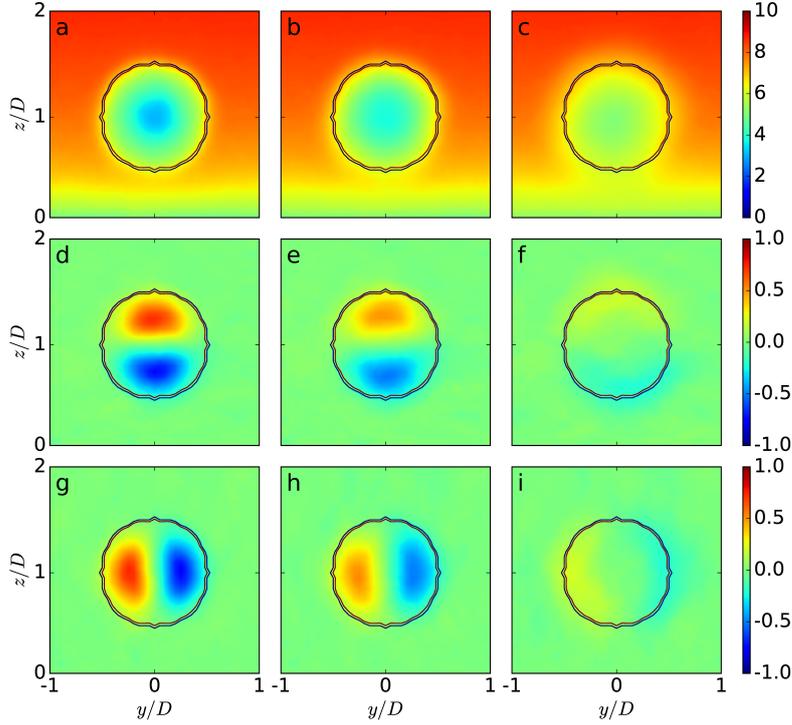


Fig. 3 The averaged values of the base-case simulation (B.1) of $\langle u_{x,y,z} \rangle_t$ in (a)-(c), $\langle v_{x,y,z} \rangle_t$ in (d)-(f) and $\langle w_{x,y,z} \rangle_t$ in (g)-(i) in y - z cross-sections at downstream positions $x = 3D$ ((a), (d), (g)), $x = 5D$ ((b), (e), (h)) and $x = 10D$ ((c), (f), (i)).

517 Approaching the rotor, the flow is decelerated in front of the wind turbine and
 518 accelerated behind it. This behaviour is induced by the flow deceleration due
 519 to the axial force F_x , which causes a pressure increase in front of the rotor
 520 and a decrease behind (Bernoulli equation) (Hansen, 2008). The difference of
 521 the flow in the spanwise direction for $x/D > 2$ results from the rotation of the
 522 actuator, leading to an accelerated (decelerated) flow on the right (left) due to
 523 downward (upward) transport of air with higher (lower) momentum. The flow
 524 recovers with increasing distance and the velocity values start to approach the
 525 values of the incoming wind field for $x \geq 10D$. The effect of the wind turbine on
 526 the wake is not negligible even at a streamwise distance of $x = 20D$ in Fig. 2,
 527 therefore we expect a full recovery in Fig. 4 at positions $x > 20D$.

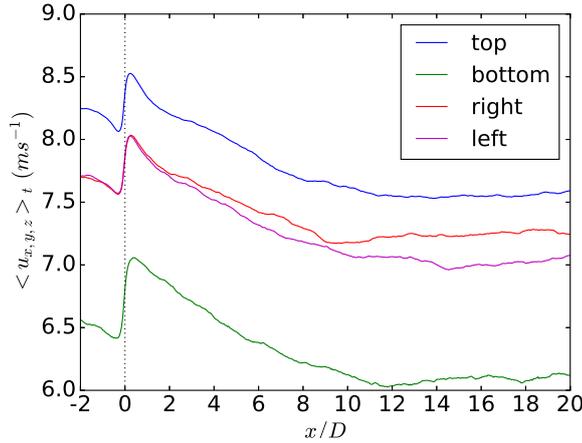


Fig. 4 The velocity component in streamwise direction $\langle u_{x,y,z} \rangle_t$ averaged over the last $t = 50$ min of the base-case simulation B_1 at four positions, which are located 60 m away from the rotor centre ($R = 50$ m), in both spanwise (left and right) and vertical (top and bottom) directions. The spanwise directions correspond to Figs. 2 and 3 with right $\equiv y/D \in [0, 1]$ and left $\equiv y/D \in [-1, 0]$.

528 5.2 Impact of the perturbation amplitude

529 The method of preserving the background turbulence includes the factor α in
 530 Eq. 5, which was introduced as the amplitude of the perturbation. The impact
 531 of α is studied in simulations B_5 ($\alpha = 5$) and B_10 ($\alpha = 10$) and compared to
 532 the reference simulation B_1 ($\alpha = 1$).

533 Figure 5a shows the streamwise profiles of the velocity ratio from Eq. 16 for
 534 different values of the perturbation amplitude α . A larger α value leads to a
 535 progressively shorter streamwise extension of the wake, induced by a stronger
 536 entrainment of ambient air. Further, the minimum of the velocity ratio in the
 537 near wake directly behind the nacelle increases.

538 The markers in Fig. 5a correspond to different wind-turbine studies, as
 539 described in detail in the caption of Fig. 5. The simulation results of B_1 are
 540 comparable to lidar measurements and WRF-LES model results for a stable
 541 ABL (Aitken et al., 2014). By increasing the value of α , the velocity ratio ap-
 542 proaches values found in observations and simulations of cases with enhanced
 543 turbulence. The numerical results of simulation B_5 correspond to a neutral
 544 ABL (Wu and Porté-Agel, 2011; Gomes et al., 2014), whereas the results of
 545 simulation B_10 are almost comparable to measurements and WRF-LES model
 546 results in a convective ABL (Mirocha et al., 2014). This comparison with other
 547 studies leads to the hypothesis that the factor α from Eq. 5 could be related
 548 quantitatively to different levels of atmospheric turbulence.

549 We also tested various precursor simulations (convection or Coriolis force
 550 as trigger to excite turbulence) resulting in different spectral energy densities.
 551 The velocity ratio for a larger amount of the spectral energy density is in

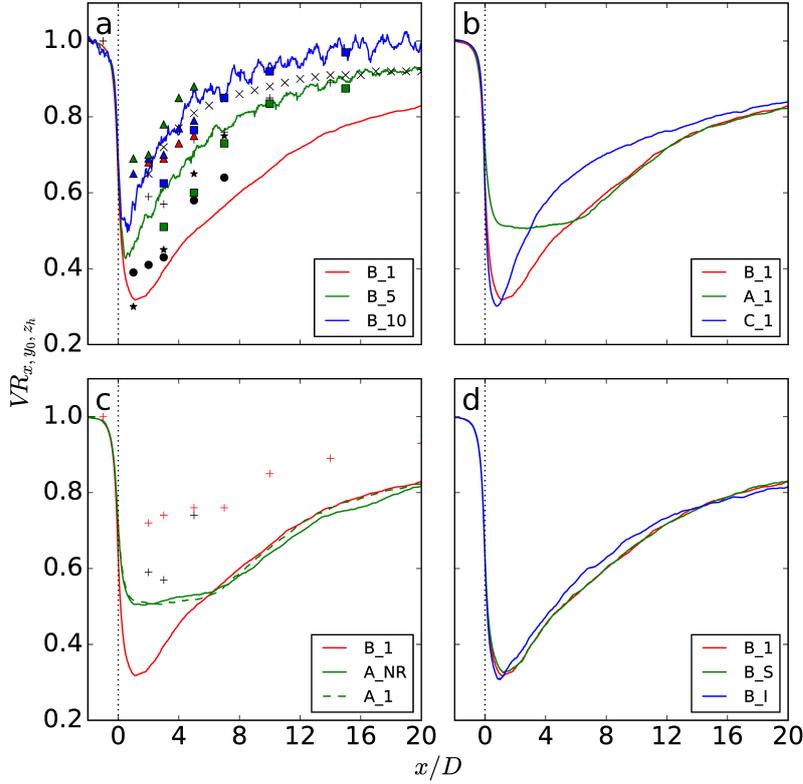


Fig. 5 The streamwise dependence of the velocity ratio from Eq. 16 at y_0 and z_h for all simulations listed in Table 4, grouped together regarding the wake impact of the perturbation amplitude in (a), the wind-turbine parametrization in (b), the rotation of the disc in (c), and the SGS closure model in (d). The markers in (a) and (c) correspond to the results of the velocity ratio from the wake of a wind turbine in various studies: the values marked by a plus sign are extracted out of the LES from Wu and Porté-Agel (2011, Fig. 4) for a neutral ABL. The crosses correspond to the neutral ABL RANS simulation by Gomes et al. (2014, Fig. 1). The circles are extracted from lidar measurements in a stable ABL and the asterisks from the corresponding WRF-LES model simulation of a stable ABL, see Aitken et al. (2014, Fig. 6). The red triangles are extracted from convective ABL measurements, the blue triangles correspond to the WRF-LES model simulation of a convective ABL characterized by a heat flux of 20 W m^{-2} , and the green triangles correspond to the WRF-LES model simulation of a convective ABL characterized by a heat flux of 100 W m^{-2} , investigated in Mirocha et al. (2014, Fig. 8). The green squares correspond to a neutral ABL with a roughness length $z_0 = 1 \times 10^{-5} \text{ m}$, and the blue squares to a value of $z_0 = 1 \times 10^{-1} \text{ m}$ (Wu and Porté-Agel, 2012, Fig. 5). The red plus signs in (c) correspond to the results of the non-rotating disc in Wu and Porté-Agel (2011, Fig. 4) opposed to their rotating results in black.

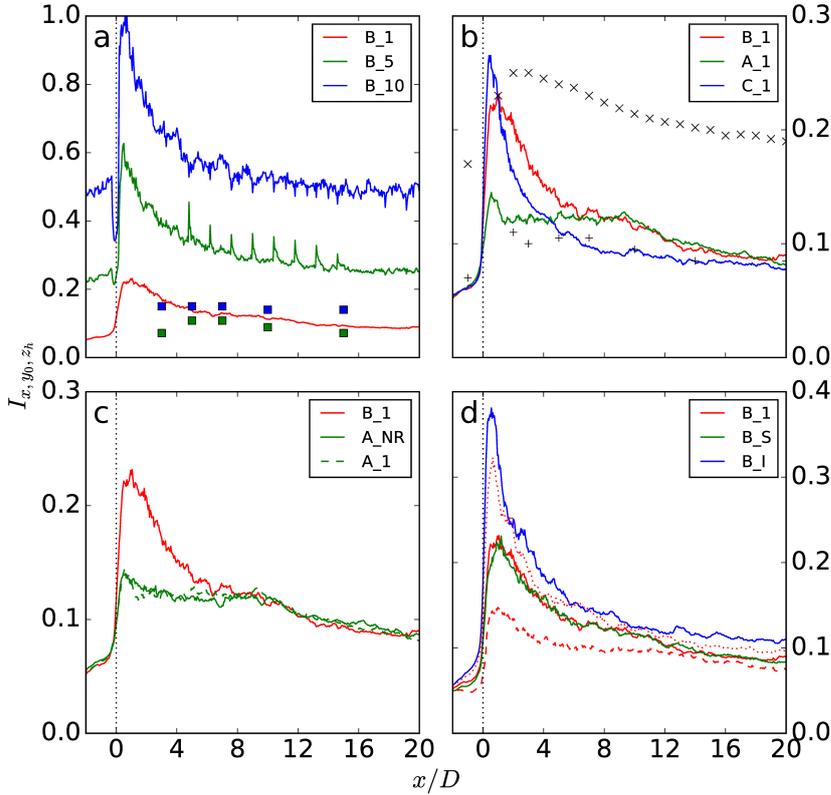


Fig. 6 The streamwise dependence of the turbulence intensity from Eq. 17 at y_0 and z_h for all simulations listed in Table 4, grouped together regarding the wake impact of the perturbation amplitude in (a), the wind-turbine parametrization in (b), the rotation of the disc in (c), and the SGS closure model in (d). The markers in (a) and (b) result from the streamwise turbulent intensity in the wake of a wind turbine in various studies: the green squares in (a) correspond to a neutral ABL with a roughness length $z_0 = 1 \times 10^{-5}$ m, and the blue squares to a value of $z_0 = 1 \times 10^{-1}$ m (Wu and Porté-Agel, 2012, Fig. 8). The values marked by a plus sign in (b) are extracted out of the LES from Wu and Porté-Agel (2011, Fig. 7) for a neutral ABL. The crosses correspond to the neutral ABL RANS simulation by Gomes et al. (2014, Fig. 1). The dotted line in plot (d) represents simulation B.1 with 1/2 times the length scale in the SGS closure model, whereas the dashed line represents simulation B.1 with twice the length scale in the SGS closure model.

552 agreement with a larger value of α (not shown here). The parameter α is
 553 also comparable to the different roughness lengths used in Wu and Porté-Agel
 554 (2012), with a larger roughness length corresponding to a higher perturbation
 555 amplitude.

556 The streamwise profiles of the turbulent intensity in Eq. 17 are presented
 557 in Fig. 6a for different α values. The turbulent intensity I_{x, y_0, z_h} increases

558 with increasing α . In the upstream as well as in the downstream region, the
 559 streamwise distribution of I_{x,y_0,z_h} is proportional to α . Wu and Porté-Agel
 560 (2012) investigate an increase of I_{x,y_0,z_h} for increasing z_0 . We also result in an
 561 increase of I_{x,y_0,z_h} for increasing α , reinforcing our assumption that larger α
 562 values are comparable to a surface with an increased roughness length.

563 We conclude that the entrainment in the wake can be easily modified by
 564 adjusting the value of α in the numerical simulations. In this way, a realistic
 565 level of atmospheric background turbulence intensity corresponding to various
 566 atmospheric stratifications or different roughness lengths can be parametrized
 567 by applying our turbulence preserving model.

568 5.3 Impact of the wind-turbine parametrization

569 The impact of the three wind-turbine parametrizations A, B, and C on the
 570 wake is studied for $\alpha = 1$ in simulations A_1, B_1 and C_1. The different
 571 parametrizations influence the velocity ratio in the wake as documented in
 572 Fig. 5b.

573 A comparison between simulation A_1 and simulation B_1 focuses on the
 574 difference between the MMT and the BEM method. Approaching a down-
 575 stream distance of $x = 5D$, the difference in the wake structure becomes mar-
 576 ginal. Therefore, we define a streamwise distance of $x = 5D$ as the transition
 577 between the near wake and the far wake. Further, the value of the minimum
 578 of the velocity ratio in the near wake is larger for parametrization A in A_1
 579 due to no radial dependence of the thrust and power coefficients in Eqs. 6 and
 580 7.

581 The difference between parametrizations B and C are the local blade char-
 582 acteristics of the two airfoils. In parametrization C the velocity field in the
 583 streamwise direction recovers more rapidly up to approximately $x = 14D$ in
 584 comparison to type B. This is caused by the sharper gradient in the axial
 585 force at the edge of the nacelle between $0.2 r/R$ and $0.3 r/R$ in Fig. 1.

586 The different parametrizations also have an impact on the value of the
 587 maximum of the turbulent intensity in Fig. 6b. The maximum is larger for
 588 parametrization B in comparison to parametrization A. This is caused by the
 589 radial gradient of the axial force in parametrization B, which contrasts a con-
 590 stant force in parametrization A, as shown in Fig. 1. The streamwise turbulent
 591 intensities of parametrizations A and B are very similar in the far wake. The
 592 difference in the maximum between parametrizations B and C correlates with
 593 the gradient of the axial force close to the nacelle in Fig. 1. A larger maximum
 594 corresponds to a sharper gradient. A sharper gradient also results in a more
 595 rapid decline in parametrization C in comparison to parametrization B up to
 596 approximately $x = 14D$.

597 Comparing these results to other studies, the turbulent intensity values
 598 of all three parametrizations are rather small in comparison to the RANS
 599 simulation of Gomes et al. (2014) approaching $x \geq 2D$. A comparison with
 600 the LES of Wu and Porté-Agel (2011) results in a rather good agreement in the

601 near wake for parametrization A and in the far wake for parametrization C.
 602 The agreement of parametrization C is referable to a similar radial distribution
 603 of the forces yielded from the same blade characteristics.

604 We conclude that the MMT is sufficient as simplification of the BEM
 605 parametrization if only the far wake is of interest. In the near wake the ra-
 606 dial dependence of the axial force becomes important. Further, the local blade
 607 characteristics influence the wake up to a downstream distance of $x = 14D$.

608 In the scope of this work, we also implemented an advanced version of the
 609 MMT. It considers the radial distribution of the forces in Eqs. 6 and 7, which is
 610 adopted from the radial chord length dispersion in Micallef et al. (2013). The
 611 forces in Eqs. 6 and 7 are modified similarly to the procedure in Gomes et al.
 612 (2014). Numerical simulations using this approach led to a better agreement
 613 of the near-wake structure with the BEM method in parametrization B in
 614 comparison to the MMT approach (not shown here).

615 5.4 Impact of the rotation of the disc

616 To investigate the impact of the rotation of the actuator on the wake structure,
 617 simulation A_NR with parametrization A, no rotation of the disc ($F_\theta = 0$ in
 618 Eq. 7) and $\alpha = 1$ is performed and compared to simulations A_1 and B_1.

619 The minimum of the velocity ratio in simulation B_1 is smaller in compar-
 620 ison to simulation A_NR. This finding is in agreement with the results of Wu
 621 and Porté-Agel (2011) (markers in Fig. 5c). A comparison between simulation
 622 A_1 and simulation A_NR results in a marginal impact of the tangential force
 623 on the streamwise velocity ratio according to Fig. 5c. Therefore, the difference
 624 between simulation B_1 and simulation A_NR is evoked by the uniform thrust
 625 force distribution over the disc, which has a larger impact on the velocity ratio
 626 than the marginal effect of rotation.

627 Wu and Porté-Agel (2011) show an increase of the turbulence intensity
 628 applying the BEM method instead of the classical Rankine-Froude approach.
 629 The streamwise turbulent intensity at the centre line in Fig. 6c is also larger for
 630 the BEM parametrization in the near wake. The effect of rotation is marginal.
 631 Consequently, not the swirl, but the non-uniform distribution of the axial force
 632 in the BEM method (Fig. 1) is responsible for the near-wake difference in the
 633 streamwise turbulent intensity in Fig. 6c.

634 The rotation of the disc in simulation A_1 leads to a swirl in the wake as
 635 shown in Figs. 7a-c. The rotational effect of the disc is evident at $x = 3D$. Ap-
 636 proaching $x = 10D$, the swirl in the disc region decays while it is transported
 637 outwards. Both effects originate from entrainment processes. At a downstream
 638 position of $x = 20D$, the rotation in the disc region approaches zero, whereas
 639 there is still some swirl in the air around the disc. In contrast to this rotational
 640 behaviour, there is no swirl of the air downstream of the non-rotating disc of
 641 simulation A_NR in Figs. 7d-f. The pattern of the streamwise velocity u in
 642 the rotor region as well as in the surroundings are comparable in both simu-
 643 lations at $x = 3D$ and $10D$. At $x = 20D$, the wake pattern in simulation A_NR

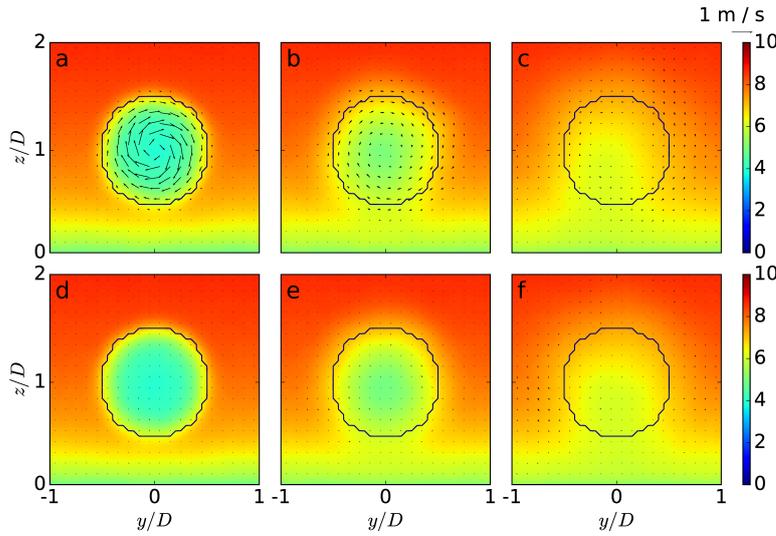


Fig. 7 The averaged value of $\langle u_{x,y,z} \rangle_t$ in a y - z cross-section at downstream positions $x = 3D$ ((a), (d)), $x = 10D$ ((b), (e)) and $x = 20D$ ((c), (f)) for simulation A.1 ((a)-(c)) and simulation A_NR ((d)-(f)). The arrows represent the wind vectors ($\langle v_{x,y,z} \rangle_t$, $\langle w_{x,y,z} \rangle_t$). The magnitude of 1 m s^{-1} is shown at the right edge of the plot.

644 is symmetric, whereas in simulation A.1 it is shifted towards $y/D \in [-1, 0]$.
 645 This asymmetric streamwise velocity field results from the rotation of the disc
 646 and is also prevalent in the study of Wu and Porté-Agel (2012, Fig. 4).

647 This investigation leads to the conclusion that the rotation has a minor
 648 effect on the velocity ratio and on the streamwise turbulent intensity at the
 649 centre line. However, the effect of the tangential force on the v and w wind
 650 components is prevailing even in the far-wake region, with an influence on the
 651 streamwise velocity field in the y - z plane.

652 5.5 Impact of the SGS closure model

653 The impact of the SGS closure models is investigated by comparing the TKE
 654 SGS closure model simulation B.1 with the Smagorinsky SGS closure model
 655 simulation B.S. The geophysical flow solver EULAG provides a reliable numerical
 656 testbed to study the SGS closure model sensitivities. Further, it depends
 657 on the NFT integrations of Eqs. 1 to 3 and therefore offers the possibility to
 658 integrate these equations without an explicit SGS closure model by setting
 659 $\mathcal{V} = 0$ and $\mathcal{H} = 0$ in Eqs. 1 and 2 in the implicit LES B.I.

660 The streamwise dependence of the velocity ratios in Fig. 5d agrees quan-
 661 titatively very well for simulation B.1 and simulation B.S. The contrast to
 662 simulation B.I is insignificant.

663 The turbulent intensities in Fig. 6d are also rather similar for the TKE and
 664 the Smagorinsky SGS closure model. For the implicit LES, the maximum of
 665 I_{x,y_0,z_h} is roughly 1.7 times larger than in the simulations with the SGS closure
 666 model. In the far wake the difference becomes rather small. The dependency of
 667 the difference in the turbulent intensity in the near wake between an implicit
 668 LES and a simulation using an explicit SGS closure model is verified with two
 669 further simulations, modifying the SGS closure model of simulation B_1. In the
 670 first simulation, the length scale of the TKE SGS closure model is multiplied
 671 by a factor of 1/2, resulting in the dotted red line in Fig. 6d, whereas in the
 672 second simulation, the length scale is multiplied by a factor of 2, resulting
 673 in the dashed red line. Decreasing (increasing) the length scale of the closure
 674 model results in a weaker (stronger) damping. A weaker damping induces
 675 larger turbulence, approaching the turbulent intensity behaviour of the implicit
 676 LES, whereas a stronger damping results in a weaker turbulent behaviour.
 677 The streamwise velocity ratios are nearly unaffected by the length scale of the
 678 closure model (not shown here).

679 The agreement between the established SGS schemes (TKE and Smagorin-
 680 sky) is a remarkable result and confirms earlier findings by Smolarkiewicz et al.
 681 (2007). The possibility of an implicit LES of wind-turbine flows enables nu-
 682 merical simulations with stretched or adaptive meshes, where an explicit SGS
 683 parametrization might be difficult and troublesome.

684 The length scale of the closure model offers another tuning parameter in
 685 addition to α , which can explain the difference in the streamwise turbulent in-
 686 tensity in comparison to other simulation results of Wu and Porté-Agel (2011),
 687 Wu and Porté-Agel (2012) and Gomes et al. (2014).

688 6 Conclusion

689 The wake characteristics of a wind turbine in a turbulent and neutral ABL flow
 690 were investigated by means of LES. Besides reliable wind-turbine parametriza-
 691 tions, an effective method to preserve the atmospheric background turbulence
 692 was applied successfully in the numerical solver. The numerical simulations
 693 using these two ingredients result in realistic wake structures, which are quan-
 694 titatively comparable with previous observations and numerical simulation
 695 results.

696 The atmospheric background turbulence field was simulated by a precu-
 697 sor simulation of the neutral ABL using cyclic boundary conditions. Velocity
 698 perturbations were extracted once from the equilibrium state of the precursor
 699 simulation. These perturbation velocities were superimposed on the flow field
 700 of the wind-turbine simulations by a new method suitable for open horizontal
 701 boundaries. This method preserves the atmospheric background turbulence by
 702 applying the spectral energy distribution at every timestep taken from three
 703 3D fields (u , v , w) of the precursor simulation. The newly developed turbu-
 704 lence preserving method uses an empirical factor α , which controls the energy
 705 content of the background turbulence. Larger α values refer to more turbulent

706 flow regimes, e.g. under convective conditions or for flows over a surface with
707 an increased roughness length. An increase of the atmospheric background
708 turbulence, i.e. larger α values, enhance the entrainment of air into the wake,
709 resulting in a shorter streamwise wake extension and an increase of the stream-
710 wise turbulent intensity. The turbulence preserving method as presented here
711 provides a simple and numerically very effective tool for studying the inter-
712 action of ABL flow of different thermal stratifications with a wind turbine
713 by applying the same spectral energy distribution and varying the parameter
714 α . Considering different stratifications of the atmosphere is important, as a
715 near-neutral stratification occurs only with a frequency of roughly 10 % ac-
716 cording to data from a field experiment (SWiFT Facility Representation and
717 Preparedness; 730 days of measurement in the period from 2012 to 2014 (Sue
718 Ellen Haupt (NCAR), personal communication, 2015)).

719 Furthermore, the wake structure was investigated for different wind-turbine
720 parametrizations. We considered the MMT and the BEM method as wind-
721 turbine parametrizations, varied the local blade characteristics in the BEM
722 method and studied the effect of rotation of the actuator. The BEM method
723 yields a more accurate prediction of the near-wake characteristics if the air-
724 foil data of the wind turbine are known. Considering how sparse information
725 on detailed blade geometries is available, the MMT offers an alternative. It
726 was found that the MMT is a reasonable simplification of the BEM model
727 for studies of the far wake, when near-wake characteristics are of secondary
728 importance. The wake structure for the two considered airfoils in the BEM
729 model differs up to a streamwise distance of $14D$. The very far wake is not
730 affected by the blade characteristics. The rotation of the wind turbine leads
731 to a swirl in the wake and impacts on the streamwise velocity field in the y - z
732 plane even in the far wake.

733 The sensitivity of the wake to two SGS closure models (TKE and Smago-
734 rinsky-type models) and numerical simulations without an explicit SGS closure
735 model (implicit LES) was studied. The choice of the SGS closure models has
736 a rather small impact on the wake characteristics. Even the implicit LES re-
737 sults of the streamwise velocity ratio agree surprisingly well with the former
738 simulations reinforcing the suitability of this approach to study a wide class
739 of ABL flows. However, there is a remarkable impact on the streamwise tur-
740 bulent intensity in the near wake, which is strongly affected by the amount of
741 damping in the SGS closure model.

742 In this study, we presented a simple and numerically effective method to
743 perform LES of wind turbines with a realistic background turbulence field.
744 Our turbulence preserving model as well as the wind-turbine models, both
745 implemented in the numerical model EULAG, allow for subsequent future
746 applications for a wide range of scales, for different thermal stratifications, as
747 well as for flows over heterogeneous and hilly terrains.

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756 **Appendix: BEM parameters**

Table 5 List of the BEM method parameters used in parametrization of type B (10 MW reference wind turbine from DTU) (Mark Zagar (Vestas), personal communication, 2015) and type C (three-blade GWS/EP-6030x3 rotor) (Wu and Porté-Agel, 2011). The radius r and the chord length c of the two rotors are scaled to a rotor diameter of 100 m.

parametrization B			parametrization C		
r / m	$c(r) / \text{m}$	$\Theta(r) / ^\circ$	r / m	$c(r) / \text{m}$	$\Theta(r) / ^\circ$
5.0	5.3	13.3	6.7	9.3	20.5
10.0	6.0	13.2	13.3	9.8	20.9
15.0	6.2	10.5	20.0	9.8	19.8
20.0	5.8	9.0	26.6	9.4	16.9
25.0	5.0	7.3	33.3	8.7	13.2
30.0	4.5	5.5	40.0	7.9	10.7
35.0	3.5	3.8	46.7	6.8	9.1
40.0	3.0	2.5	50.0	4.0	6.7
45.0	2.3	1.3			
50.0	1.0	0.2			

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